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## Article

# Dimensional Accuracy and Mechanical Characterization of Inconel 625 components in Atomic Diffusion Additive Manufacturing

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**Abstract:** Metal material additive manufacturing (MEAM) has risen interest in the last five years as an alternative to powder bed processes. MEAM is promising for generating shelled components with defined infill structures, making it very interesting for lightweight engineering. Atomic Diffusion Additive Manufacturing (ADAM) is a filament based MEAM process patented by Markforged Inc. providing closed process chain from preprocessing to the final sintering of printed green parts. This study focuses on the Inconel 625, which is of high interest in the aerospace industry and assesses its dimensional accuracy and tensile properties regarding different print orientations and solid, triangular, respectively gyroid infill structures. The results showed that neither the dimensional accuracy nor the sintering shrinkage was significantly influenced by the printing orientation or the infill structure. In the context of lightweight engineering, the infill structures proved beneficial especially within the elastic region. Generally, triangular infill patterns resulted in higher stiffness, while gyroids led to more ductile specimens. A mass-related evaluation of tensile testing elucidates that with aid of the infill structures weight savings of 40 % resulted in mechanical performance decreased by only 20 % on average, proofing its high potential for lightweight design.

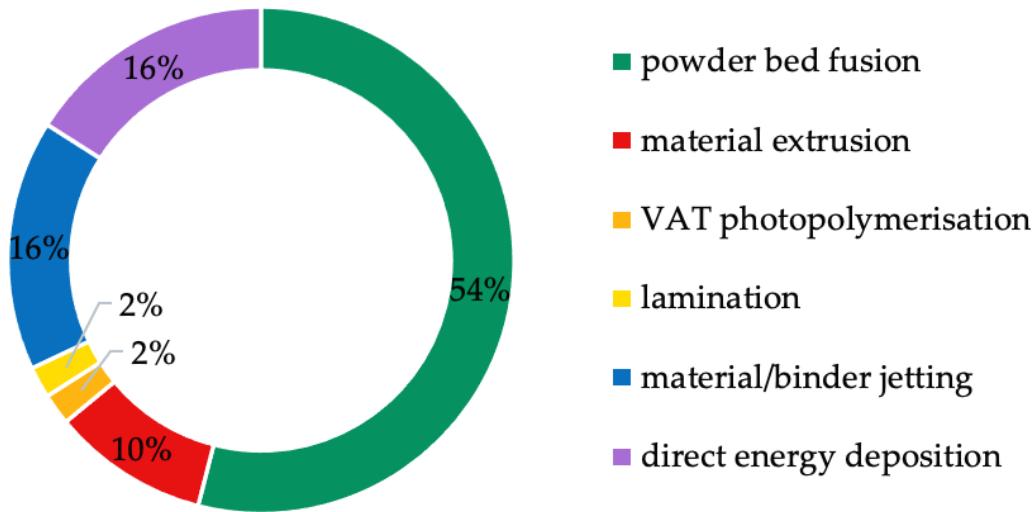
**Keywords:** ADAM; metal material extrusion; metal additive manufacturing; infill structures

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## 1. Introduction

A wide variety of processes for the additive manufacturing (AM) of polymer, metal or ceramic components have been developed over the past few decades [1]. Powder bed fusion, including selective laser sintering, selective laser melting, electron beam melting, and direct metal laser sintering, direct energy deposition, and material/binder jetting are the most widely utilized metal AM techniques, as depicted in Figure 1.

Currently, powder-bed-based processes are primarily used for metallic components. These have several disadvantages. On the one hand, the price of the powder and the equipment is very high due to the use of powerful lasers, or the vacuum technology required for electron beam applications, on the other hand, the manufactured parts have a welded microstructure, which usually requires additional heat treatment [1]. Metal filament printing via extrusion in combination with the posttreatment steps of debinding and sintering is an economical and straightforward to operate alternative. Thermoplastics, ceramics, metals, hybrid materials, and reinforced composites like metal - polymer composites can all be manufactured using extrusion additive manufacturing (EAM). In metal EAM (MEAM), the production of a feedstock made of metallic powder and binder is the initial stage. After the printing process the binder is removed during two posttreatment steps, debinding and sintering, thus creating the final metal component [2].



**Figure 1.** The metal additive manufacturing market in 2020 according to [2].

Especially in times where product development needs to be operated in a more sustainable manner regarding the availability of resources and the life cycle of products, MEAM shows great potential [3]. Further, due to the extrusion process, shelled parts with defined infill structures are favorably in MEAM, which is accompanied by reduced material use, printing times and the possibility to tailor the components effective properties. These shelled structures with infill patterns enhance significantly the lightweight design potential [4] and are also known from nature, e.g. the human bone, to adapt the effective mechanical properties for their purpose [5].

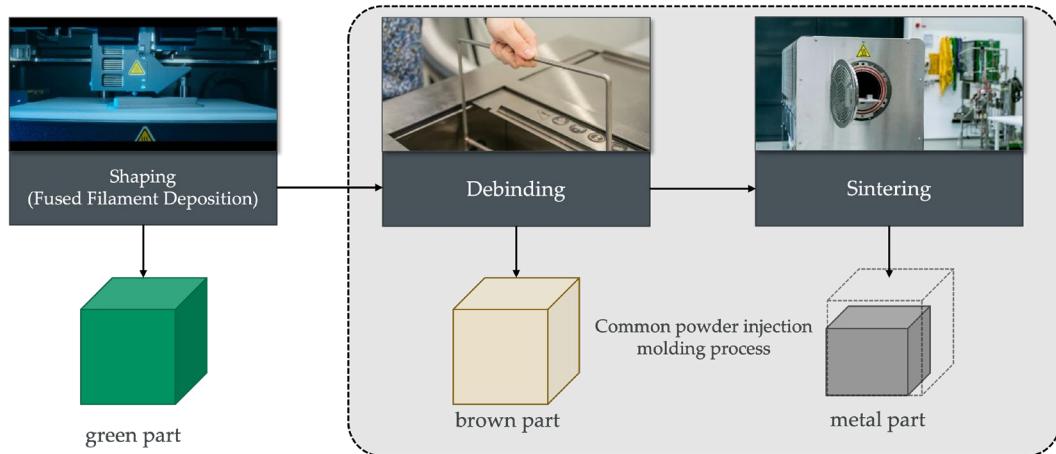
In the current research, studies investigating Atomic Diffusion Additive Manufacturing (ADAM), a MEAM process patented by Markforged Inc. (Boston, MA, USA), are predominately on the stainless steel 17-4 PH [6–17], mostly with a focus mechanical properties. Besides 17-4 PH, few studies investigate copper filaments [18–20], while research on other materials such as the coldwork tool steel D2 [21] and Inconel 625 [22] is very scarce. Thus, our work focuses on Inconel 625 which, due its prominence in aerospace industry, is of high interest for lightweight engineering applications. The influence of infill patterns on the printability, accuracy and mechanical properties have been studied extensively for polymers, for instance in [23–25]. Nevertheless, due to additional steps of debinding and sintering those results cannot be simply transferred to ADAM parts. Kiswanto et al. investigated the influence of a triangular infill structure on surface roughness and impact toughness of ADAM produced Inconel 625 [22], but no studies evaluating the influence of infill structures on shrinking behavior or tensile properties of ADAM fabricated Inconel 625 have been conducted yet. Further, most studies focus on solid filled specimens which is a shortcoming as shelled and structure-filled parts are very prominent in EAM in general and can hardly be manufactured with powder bed AM processes.

Thus, it is this study's objective to assess the commercial ADAM process chain for Inconel 625 components with different infill structures regarding the process robustness, dimensional accuracy, surface roughness, and effective tensile properties. Further, a mass-related evaluation of the tensile test allows to assess the current state of the art of lightweight engineering performance of Inconel 625 components in MEAM.

### 1.1 Influence of design and printing parameters on the mechanical properties in MEAM

ADAM is a MEAM process which uses a feedstock composed of metallic powder and paraffin-based binder. As the flowchart in Figure 2 shows, the filament is deposited layer-wise, and referred to as green part. In the next step the main part of the binder is removed in the debinding process, leaving behind a carbon-backbone that ensures the integrity of the now obtained brown part. These residues are eliminated in the early stages of the sinter process after the first sinter necks have formed.

After the sintering process has been finished a full metal part with a linear shrinkage of about 13-20 % is obtained [8,26-28].



**Figure 2.** Schematic overview of the MEAM process after [3].

The shrinkage in MEAM is effect of the sintering process, whereas its anisotropy is affected by the printing direction and its value is controlled by the binding agent content [29]. Accordingly, the remaining anisotropy in the microstructure is caused by pores and not by the grains shape or size. In this regard, the parts differ heavily from powder bed additive manufacturing metal parts whose anisotropic properties arise from the welded microstructure.

Studies for MEAM in general have shown that different printing parameters e.g. build orientation, type of infill, infill degree and layer thickness influence the mechanical properties of the final part [30]. As already known from polymer EAM, thinner layer thickness leads to better mechanical properties due to the decrease of voids between deposited paths, hence a lower surface roughness. These effects were investigated by Kurose et al., which compared the mechanical properties of 316L stainless steel parts with layer thicknesses of 0.1 mm and 0.3 mm [31]. For both build orientations, flatwise and upright, their results show the specimens with a layer thickness of 0.1 mm withstanding a higher tensile strength. Also, the strain break was higher for the samples with 0.1 mm layer thickness. In addition, when printing with 0.3 mm layer thickness, voids around the outside lines of the internal structures were observed, which were not existent in the samples with the layer thickness of 0.1 mm [31]. Thus, the better mechanical performance of smaller layer thicknesses can be attributed to a better linking between shell and infill structure.

In general, the type and degree of infill influence the occurrence of printing defects, affecting the mechanical properties of the parts, have been studied extensively for thermoplastic polymers. Khan et al. investigated the effect of four different infill patterns in FDM printed PLA parts: rectilinear, concentric, honeycomb, and Hilbert Curve [23]. Their data shows that rectilinear and honeycomb infills provide the maximum strain. The rectilinear infill pattern also achieved the highest Young's modulus and yield strength. Dezaki et al. compared the tensile strength of FDM manufactured PLA parts with concentric, triangular, and zig-zag infill [24]. They achieved the highest load bearing specimens with a concentric, in tensile load infill due to those patterns providing lower void fractions than other infill types. Also, the results of Dezaki et al. show an increase in tensile strength with increasing density. Parab et al. also investigated the optimal infill pattern for FDM PLA parts [25]. They tested triangular, gyroid and rectilinear infill patterns at different infill degrees. Their results showed that a triangular infill can withstand greater yield loads than the gyroid and rectilinear infill patterns. They also state gyroid infill showing mostly isotropic properties, hence being useful for parts that are subjected to multiple loading directions.

In contrast, the research on MEAM in general is very scarce in parts contrary to the results obtained for the thermoplastics. Damon et al. for instance state that the yield strength of their SS 316L samples is not affected significantly by the build orientation (flatwise:  $155 \pm 6$  MPa, upright:  $165 \pm 12$

MPa), due to the pore channels of both their build orientations being 45 ° to the load direction [32]. Tosto et al. analyzed the influence of the building orientation on the tensile properties of SS 316L [33]. Their results show that flatwise-printed specimens display higher values for tensile strength. The difference in tensile properties between flatwise and upright printed samples was explained by voids which run perpendicular to the testing direction.

Regarding different infill structures, Gonzalez-Gutierrez et al. focused on the effects of different infill patterns in copper parts, they analyzed that rectilinear and diagonal patterns do not influence the flexural properties of the specimens significantly [34]. While the influence of the infill structures density was shown to be significant for the effective mechanical properties [3], studies comparing different infill structures for MEAM or ADAM in general are yet missing.

### 1.2 Sinter shrinkage of MEAM parts

During sintering, pores in the part are removed leading to a full dense metal part due to growth and adhesion amongst the metal particles, thus causing the part to shrink. The sintering is the intrinsic deformation mechanism of MEAM and consists of at least six different diffusion controlled [35] mass transport mechanisms, contributing to the bonding between the metal particles by forming so-called sinter necks. This shrinkage does not occur equally in all directions but depends on the orientation of the part [35–37]. Kurose et al. studied the shrinkage of parts printed with 316L stainless steel [31]. Their specimens all exhibited anisotropic shrinkage behavior. This shrinkage can be influenced by the orientation of the filler particles [27]. Also, the orientation of the polymeric binder in the part can lead to gaps between the deposited stands which effects the shrinkage and density [31]. Singh et al. investigated the influence of different sintering temperatures on the shrinkage and mechanical properties of FDM stainless steel (SS 17-4 PH MIM) [38]. With higher sintering temperatures the shrinkage and the relative density increased. A maximum density of 96.5 % was achieved at a sintering temperature of 1360 °C for 3 h with a shrinkage of 14.1 %. Tosto et al. also measured the sintered SS 316L parts for the two build orientations and compared them to the green part. While the flatwise samples reveal a linear shrinkage in XY-direction of approximately 20 %, the upright ones show more of an anisotropic behavior, which is due to gravity while sintering [33].

The shrinkage thus depends on the binding agent fraction and the voids in the green part after the extrusion process. Hence, by knowing the characteristic porosity and binding agent content, the shrinkage in the different directions can be calculated [39,40]. By using the finite element method, this can also be expanded to analysis hotspots for potential sinter failure [29]. For mitigating the sinter shrinkage, scaling factors based on a either theoretical or empirical obtained linear shrinkage are used in order to scale the CAD geometry before printing [41]. This approach is also used in the preprocessing software Eiger by Markforged Inc. (Boston, MA, USA), which scales green parts automatically based on empirical scaling factors for the specific material.

Recent approaches also deploy machine learning algorithms for prediction either the void percentage after sintering [42] or the dimensional changes directly [43].

## 2. Materials and Methods

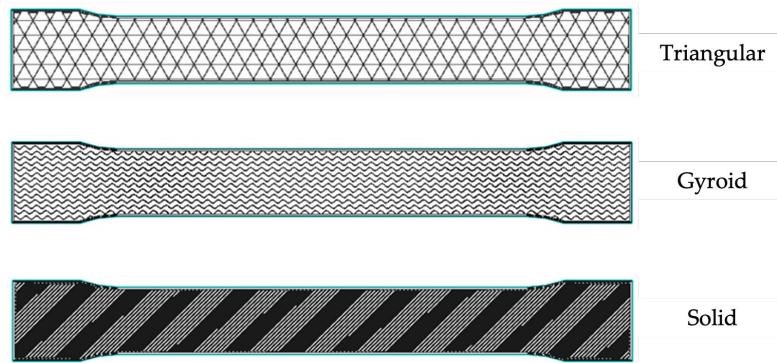
This paper assesses the commercial ADAM process by Markforged Inc. (Boston, MA, USA) for the production of Inconel 625 (Material number 2.4856) [44] components with different infill structures. The specimens are analyzed regarding the influence of the infill structure on sinter shrinkage – and its associated geometric deviations –, surface roughness, and effective tensile properties to evaluate their potential for lightweight engineering.

Thereby, ADAM process is a closed process with little parameters editable for the operator following an easy to print and use methodology. Accordingly, the possibilities to adjust printer settings, sinter conditions or similar are very limited.

## 2.1 Preparation process of tensile specimens

The Inconel 625 filament from Markforged (Markforged Inc., Boston, MA, USA) [44] with an average diameter of 1.75 mm was used as received. The specimens were preprocessed using Eiger as slicing software, the Markforged Metal X for printing. After printing the green parts were debindered and sintered using the Markforged Wash-1 and Sinter-1, respectively.

For the preprocessing the number of wall layers, respectively roof and floor layers was chosen to have a constant shell thickness of 0.5 mm. Also, the scaling factors were calculated automatically by Eiger and times for debinding and drying were recommended. The scaling factors were 1.204 for the flatwise specimens, respectively 1.191 for the upright printed ones. The orientation was varied between flatwise and upright and the solid, triangular honeycombs and gyroids were used as infill structures, resulting in six different configurations, with  $n = 5$  specimens for each configuration. Figure 3 shows the cross section of the specimens by the example of the flatwise printed specimens.

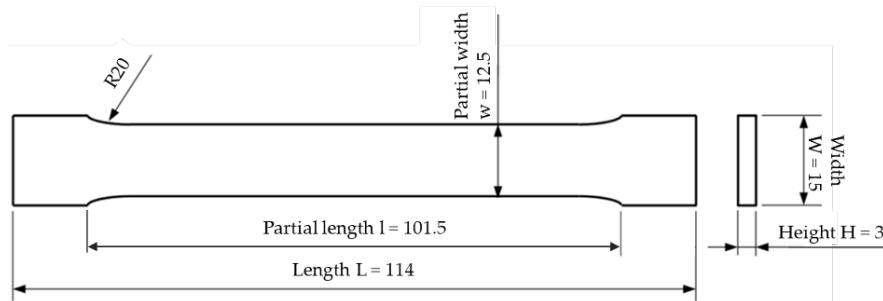


**Figure 3.** Sketch of the cross-sections of the used specimens made in Eiger.

The printing parameters for all configurations are summarized below in Table 1 and the geometry of the specimens is shown in Figure 4.

**Table 1.** Process parameters for all configurations in Eiger.

	Solid flatwise	Solid upright	Gyroid flatwise	Gyroid upright	Triangular flatwise	Triangular upright
As printed (mm)	137.2 x 18.1 x 3.6	135.8 x 18.1 x 3.6	137.2 x 18.1 x 3.6	135.8 x 18.1 x 3.6	137.2 x 18.1 x 3.6	135.8 x 18.1 x 3.6
As sintered (mm)	114.0 x 15.0 x 3.0	114.0 x 15.0 x 3.0				
Layer height (mm)	0.125	0.125	0.125	0.125	0.125	0.125
Roof and floor layer	-	-	4	4	4	4
Wall layers	2	2	4	4	4	4
Nozzle temperature	220 °C	220 °C				
Bed temperature	112 °C	112 °C				
Chamber temperature	48 °	48 °	48 °	48 °	48 °	48 °
Print time	2h 37 min	5h 30 min	2h 22min	5h 24 min	2h 5min	5h 5min
Wash time	12 h	12 h	4 h	4 h	4 h	4 h
Dry time	4 h	4 h	4 h	4 h	1 h	1 h
Printed part mass (g)	42.47	50.28	24.61	29.74	24.82	32.74
Final part mass (g)	37.83	38.57	21.91	20.27	22.1	22.94
Metal volume (cm <sup>3</sup> )	7.74	9.15	4.48	5.41	4.52	5.96



**Figure 4.** Tensile specimen with a total length of 114 mm. Specimens were printed upright and flatwise.

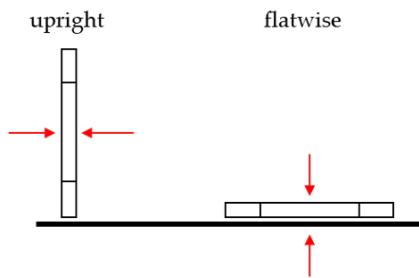
After printing, the specimens were put into the Wash-1 for solvent debinding, where 3M Novec 72DA (3M, Saint Paul, MN, USA) served as debinding fluid. For the solvent debinding and adjacent drying the times as recommended by Eiger in Table 1 were used. The thermal debinding and final sintering was conducted using the Sinter-1. The furnace works up to a peak temperature of 1300 °C depending on the materials' needs and operates automatically from start to finish, according to the material selected by the operator. Since it is a closed system, with materials and debinding and sinter settings pre-defined for the Markforged materials, no information about the temperature, respectively holding times, heating or cooling rates is provided. The whole thermal debinding and sintering process took about 24 h to complete. All specimens were sintered flatwise.

## 2.2 Dimensional deviations and surface roughness

The main influence of the dimensional deviation is attributed by the sinter shrinkage, whereby the comparison is conducted to investigate, if the infill structure influences the shrinkage behavior or affects the surface, for instance by small sink marks. Accordingly,

the length, width, and height of the specimens were measured before and after sintering in triplicate. The sintered parts were also weighed to be compared to the final metal part mass predicted by Eiger.

To make a statement about the surface roughness of the sintered parts, the mean roughness value  $R_a$  and the mean roughness depth  $R_z$  were measured on the front and back of each part in triplicate, see Figure 5. For the tactile measurement of the surface, the MahrSurf PS 10 (Mahr GmbH, Goettingen, Germany) was used.



**Figure 5.** Schematic of the surfaces where the roughness was measured.

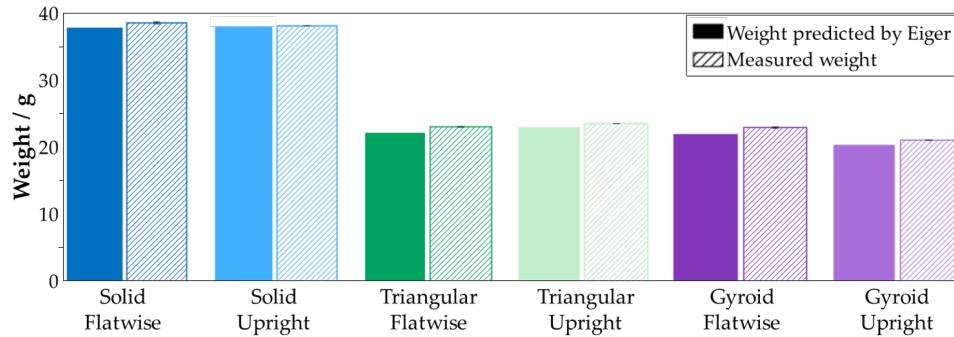
## 2.4 Tensile testing

Tensile testing was performed according to DIN EN ISO 6892-1 procedure A1 and A2 [45]. Five specimens per infill type and orientation were tested on a universal testing machine with a 50 kN Xforce load cell (ZwickRoell GmbH & Co. KG, Ulm, Germany). Strain was measured with a clip-on extensometer 5025-1. A preload of 5 MPa was set, and the test speed ( $0.0067 \text{ s}^{-1}$ ) was controlled via a constant strain rate (measurement with extensometer). The Young's modulus was determined through linear regression between 20 and 40 MPa.

### 3. Results and Discussion

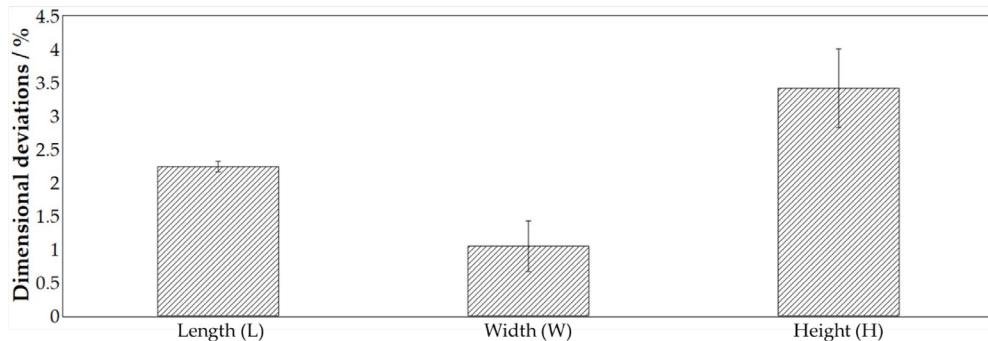
#### 3.1. Dimensional deviations

In Figure 6, the weight of the sintered parts is compared to the predicted final weight by Markforged 3D printing software Eiger. Eiger's predicted final metal part mass deviates from the measured one by an average of 3.09 %, with Eiger always being the lighter value. Thus, the mass prediction in the preprocessor is quite reliable for the different infill structures and print orientations.



**Figure 6.** Comparison of the measured weight of metal specimens with different infills and printing orientations with predicted weight by the printing software Eiger.

Further, the post-sintered dimensions of the showed that neither the infill type nor the printing orientation influenced the sinter-induced dimensional change significantly. This was to be expected due to all parts in this study being printed with a shell of four wall, four bottom and four top layers, both solid. The remaining infill was printed using the triangular and gyroid patterns. Thus, also the sintering shrinkage was not affected by the infill structure. This is also reflected in the deviations between the sintered specimens and the initial CAD model in Figure 7.

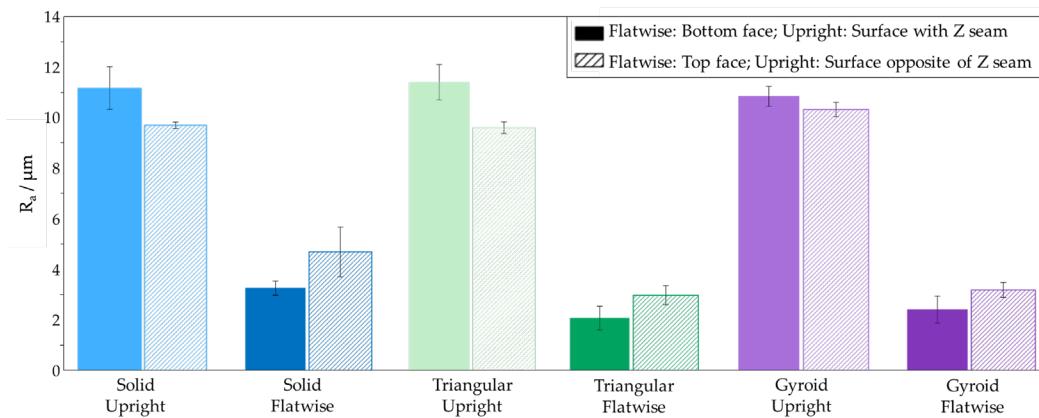


**Figure 7.** Dimensional deviations between the sintered specimens and the CAD model.

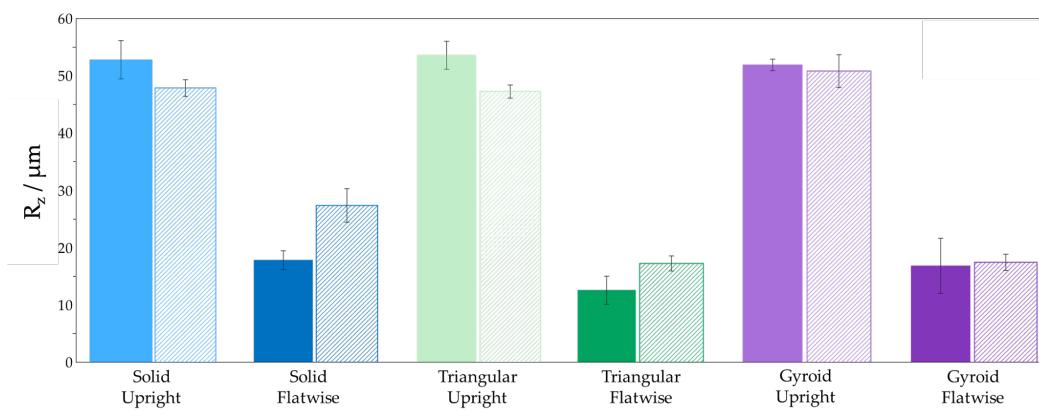
The deviations vary between 1.04 % (width) and 3.41 % (height). The overall average deviation amounts to 2.23 %. This non-uniform sinter shrinkage being one of the key concerns which complicates the design process of ADAM or MEAM parts in general [31,36] and must be considered by tolerance compensation when using ADAM parts in product development. The average shrinkage in z-direction (H) was the largest with 15.50 % for flatwise printed parts and 14.77 % for upright ones, due to the force of gravity during sintering. The smallest shrinkage was observed in x-direction (L) at 14.25 %. Thus, differing by 1.25 %. When compared to other MEAM results, those of Kurose et al [31] show an average linear shrinkage varying by 4 % between 13 % and 17 % or Abe et al. [39] with a linear shrinkage between 15 % and 17 %. Thus, we conclude that for no configurations air/gas entrainment or similar occurred during debinding or sintering, which would have been a major concern for geometrical deviations. Further, the results show that the sinter shrinkage can be controlled within about 4 % deviation regardless of the part's orientation or infill structure used.

### 3.2. Surface roughness

The results of the surface roughness analysis are shown in Figure 8 and Figure 9. In the context of this study, we refer to the general surface, using the primary profile without any filter. Independent of the infill type, the upright printed parts show a higher roughness compared to the flatwise printed parts at the surface with the Z seam. This seam shows the over extrusion at layer transitions leading to printing artifacts and therefore a higher surface roughness. For the flatwise printed parts, the top face shows the highest roughness. This finding is consistent with the results in [46] and is due to remaining filament from previous paths and unbalanced pressure at the printing system [47]. Therefore, a slight over-extrusion at the top face can be the reason for the higher surface roughness. Also, the bottom layer roughness is restricted to the surface structure of the printing bed. The overall trend of flatwise printed parts having a lower surface roughness is also visible. This can be explained that when the layers are stacked during printing, they do not fuse together perfectly, as the filament of the lower layer has already cooled down by the time the layer above is printed. The result of this effect is the characteristic unevenness that is considered as the “typical FDM surface”. Further, these results confirm the values of  $R_a$  measured by Kiswanto et al. [22], which proofs a good reproducibility of the ADAM fabricated Inconel 625 parts.



**Figure 8.** Surface roughness mean value analysis of sintered specimens with different printing orientations and infill patterns.

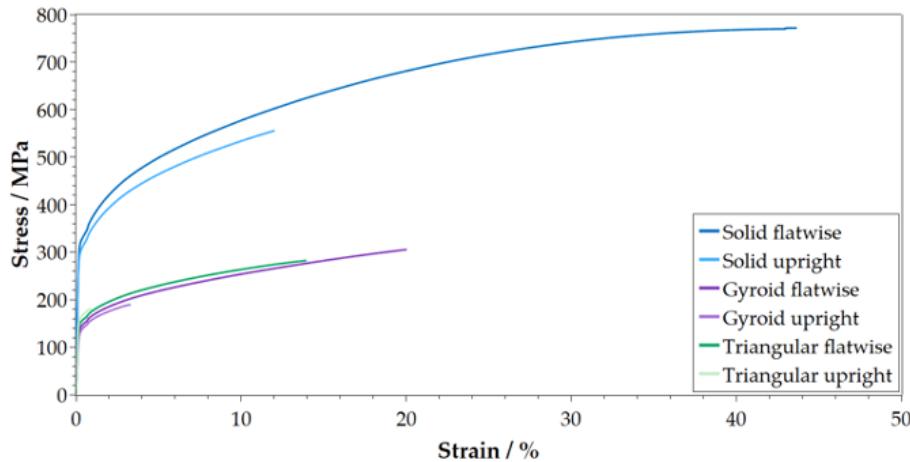


**Figure 9.** Surface Roughness mean depth analysis of the sintered specimens with different printing orientations and infill patterns.

### 3.3 Tensile testing

The results of the tensile testing in Figure 10 show that flatwise printed specimens with solid infill exhibit the greatest normal force and ultimate tensile strength (UTS) ( $770 \pm 2$  MPa) followed by upright printed ones with solid infill (UTS:  $572.8 \pm 10.7$  MPa). Specimens with triangular and gyroid

infill performed inferior with ultimate tensile strength of  $308 \pm 2$  MPa (flatwise),  $203 \pm 21$  MPa (upright) and  $284 \pm 4$  MPa (flatwise),  $194 \pm 5$  MPa (upright), respectively. While five specimens per configuration were tested, with the exception of triangular flatwise, we observed at least one outlier, which we define as failure outside of the valid testing range defined by the testing standard [45]. We have noted the number of outliers (*out*) in the summarized results shown in Table 2 as well as in the caption of Figure 10. For the evaluation all five specimens were used since the deviations between the measurements – including the outliers – were small. We rather assume that we can use the number of outliers as an indicator of the configuration's reliability.



**Figure 10.** Overview of the tensile test results  $n = 5$ . All curves represent the average for each condition (Solid: *out* = 1; Gyroid: *out* = 3; Triangular flatwise: *out* = 0; Triangular upright: *out* = 2).

**Table 2.** Tensile testing data compared to the Inconel 625 datasheet [44], *out* represents the number of outliers per configuration.

Configuration	<i>out</i>	Young's modulus in GPa	Yield strength in MPa	Ultimate tensile strength in MPa	Strain at break in %
Solid flatwise	1	$194.75 \pm 7.01$	$327.5 \pm 1.80$	$769.75 \pm 1.48$	$46.31 \pm 0.82$
Solid upright	1	$202.25 \pm 26.6$	$308.25 \pm 1.64$	$572.75 \pm 10.69$	$14.06 \pm 1.32$
Gyroid flatwise	3	$87.50 \pm 2.50$	$145.5 \pm 0.5$	$308.00 \pm 2.00$	$20.62 \pm 0.50$
Gyroid upright	3	$77.00 \pm 4.00$	$138.00 \pm 1.00$	$194.00 \pm 5.00$	$4.33 \pm 0.70$
Triangular flatwise	0	$98.6 \pm 6.92$	$155.4 \pm 1.96$	$284.2 \pm 4.17$	$15.11 \pm 0.52$
Triangular upright	2	$101.33 \pm 6.65$	$166.67 \pm 0.94$	$203.00 \pm 21.12$	$2.83 \pm 0.90$
Datasheet [44]	n.a.	n.a.	334	765	42

Accordingly, the results indicate that flatwise printed specimens with triangular infill are most robust from a process point of view. Regarding the solid specimens, it is reported also in other studies, for instance [3], that solid filled parts in MEAM tend be over-extruded leading to slightly higher porosities in particular between the contour and inner layer path. These stochastically artefacts can lead to an undesired failure. We further refer also for the other configurations to porosities between contour and infill structure as predominant cause for the outliers.

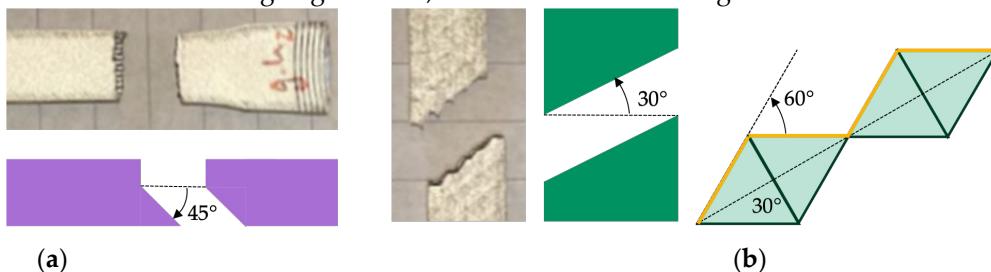
By analyzing the fracture surfaces of the flatwise printed specimens, we observed that the gyroid infill structure collapsed in all five specimens either during debinding or sintering. As we show in Figure 11, the collapsed infill accumulated at the bottom of the specimen resulting in a “half-filled” configuration.



**Figure 11.** Collapsed infill structure of all 5 gyroid flatwise specimens (left) in comparison to the successful sintered infill structure of the gyroid upright specimens (right).

Accordingly, this states that gyroid flatwise specimens cannot be manufactured robustly with ADAM at the current state.

Nevertheless, a concern also known from polymer EAM and early MEAM studies is that only the contour is load bearing and the sparse infill does not contribute to the strength of the specimens. While the solid configuration fractured in a straight line, for both infill configurations different fracture surfaces were visible. The results clearly show that for the flatwise printed specimens the crack propagated through whole structure, as for the collapsed gyroid the crack runs straight through the shell before being deflected at approximated  $45^\circ$  degree by the accumulated material as illustrated in Figure 12 a. As for the triangular infill, equilateral triangles are used with an inner angle of  $60^\circ$ , we can observe from the specimens that the crack path follows the shape of the infill, which results in an effective breaking angle of  $30^\circ$ , as we illustrated it in Figure 12 b.



**Figure 12.** Illustration of the fractures for: (a) collapsed gyroid flatwise; (b) triangular flatwise, where the crack-path (yellow) follows the contour of the equilateral triangle infill structure.

The upright printed specimens all fractured between two layers, resulting in horizontal fracture lines. This result is similar to the results found by Jasmin et al. [48], which concluded that there are no significant differences in the fracture mechanisms for upright printed solid and filled polymer specimens. However, it can be indicated clearly that the intact gyroids enable greater movement than the stiff triangles, which results in a nearly doubled fracture elongation at circa identical effective density. This is a further proof that the infill structure is load bearing and has a significant influence on the effective mechanical performance of the part.

It is nevertheless interesting to observe that the collapsed gyroid structure in the flatwise printed specimens is equivalent to the intact triangular ones regarding UTS and yield strength while showing a significantly higher strain at break. Only the Young's modulus is significantly higher for the flatwise printed triangular specimen, indicating that the infill pattern predominantly influences the Young's modulus.

The result of flatwise printed specimens exhibiting the greatest tensile stress at maximum load is consistent across all infill patterns, due to the tensile strength becoming optimal when the parts are being oriented along the direction of loading stress. Hence these results could be transferred from what was already known for polymer parts, see for instance [49]. Generally, intralayer fractures mainly depend on the strength of the material extruded itself, while the interlayer strength depends on strength of bonds between layers. During tensile testing the bond between two stacked layers is weak, due to unideal fusion of the layers. Therefore, interlayer fractures occur more easily than intralayer fractures and thus flatwise printed specimens exhibit greater tensile stresses [50].

Accordingly, we observed intralayer fracture for flatwise printed specimens and interlayer fracture for upright printed ones.

Analogously to the UTS results, flatwise printed specimens also exhibit the greatest ultimate elongation, up to  $46.3 \pm 0.2$  % for specimens with solid infill. This even exceeds the ultimate elongation stated in the data sheet (42 %). The percentage elongation shows how upright printed specimens behave more brittle in comparison to flatwise printed ones. These results are comparable with the findings by Corapi et al. [51].

Overall, specimens with solid infill exhibit greater tensile strength than ones with triangular and gyroid infill due to their higher density. Comparing triangular and gyroid filled specimens, the triangular infill performs slightly better, not including flatwise printed gyroid parts with collapsed infill. This agrees with the results of studies with FDM printed polymers as for instance reported by Parab et al., which showed for PLA parts that the triangular pattern's mechanical properties exceeds the gyroid pattern [25]. When comparing build orientations, flatwise printed specimens show a more ductile behavior, with elongations up to  $46.3 \pm 0.2$  % and lower Young's modulus than upright printed ones. The layers of upright printed specimens are built vertically up to each other. Therefore, they are only interlocked by the adhesion among those layers, but not by the printed inner lines. Hence, those parts are fractured more quickly when exposed to the applied force than flatwise printed ones [52].

Flatwise printed samples obtain the highest measured tensile strength with a difference of approx. 27 % against upright orientated specimens. Similar results were achieved by Zaldivar et al. They tested the influence of the build orientation on the mechanical behavior of ULTEM 9085 and found the tensile strength of flatwise printed specimens being 21 % higher than the one of upright printed parts [53].

It is to consider that the triangular specimens have a relative density of 58 % (flatwise) and 60 % (upright), respectively 58 % (flatwise) and 54 % (upright) for the gyroid specimens. Thus, in the context of lightweight engineering, it is interesting how the mass related properties correlate to the relative density. To analyze the results in this context, Table 3 shows the mass-related data of the tensile testing, while the deviation to the flatwise solid reference is given in brackets.

**Table 3.** Mass-related (MR) tensile testing data, *out* represents the number of outliers per configuration. The relative values compared to the reference are given in brackets.

Configuration	out	MR Young's modulus in GPa/kg	MR Yield strength in MPa/kg	MR Ultimate tensile strength in MPa/kg	MR Strain at break in %/kg
Solid flatwise	4	5148 (100 %)	8657 (100 %)	20348 (100 %)	1224 (100 %)
Solid upright	4	5244 (102 %)	7992 (92 %)	14850 (73 %)	365 (30 %)
Gyroid flatwise	2	3994 (77 %)	6640 (77 %)	14058 (69 %)	941 (77 %)
Gyroid upright	2	3799 (74 %)	6808 (79 %)	9571 (47 %)	214 (18 %)
Triangular flatwise	5	4462 (87 %)	7032 (81 %)	12860 (63 %)	684 (56 %)
Triangular upright	3	4417 (86 %)	7266 (84 %)	8849 (44 %)	123 (10 %)

The data elucidates the reinforcing effect of the infill structure, while reducing the mass at about 40 % the mass-related Young's modulus and Yield strength only decrease between 13 and 26 %. Nevertheless, this effect significantly smaller with relative values about 10 % higher than the relative density for gyroid flatwise, respectively 5 % for triangular flatwise. Due to the explained disadvantages of upright printed specimens in the plastic regime, the upright printed specimens show relative values about 10 % smaller than the relative density. This correlates to the strain at break, where the upright printed samples could not achieve 20 % of the mass-related reference. Regarding the collapsed infill for the flatwise printed gyroids, these specimens show a remarkable strain at break. This can be associated to thicker areas inside the specimens due to the collapse, which can mitigate crack growth better than the small facets of the intact triangular infill.

Thus, when comparing the weight savings with the percentual decrease of the mass-related properties in Table 4, it becomes clear that within the elastic regime the infill structures show clear

advantages in the context of lightweight engineering since, about 40 % of weight can be saved by only loosing 20 % of mechanical performance. Based on this, the triangular infill pattern proofed superior in regards of stiffness and strength, while gyroids led to more ductile specimens.

Table 4. Comparison of percentual weight savings and percentual decrease in mass-related tensile testing data. MR = mass related, YM = Young's modulus, YS = Yield strength, UTS = Ultimate tensile strength.

Configuration	Saved Weight	MR-YM	MR-YS	MR-UTS	MR-Strain
Gyroid flatwise	42 %	22 %	23 %	31 %	23 %
Gyroid upright	46 %	26 %	21 %	53 %	82.5 %
Triangular flatwise	42 %	13 %	19 %	37 %	44 %
Triangular upright	40 %	14 %	16 %	56 %	90 %

#### 4. Conclusion

This study aimed to investigate the influence of the printing orientation and type of infill pattern on the mechanical properties and shape deviations of specimens manufactured with Inconel 625 using the ADAM process. The shrinkage of the parts was not influenced by print orientation or the infill pattern. Further, the predicted final weight by the slicing software showed only minor deviations. The surface roughness was influenced strongly by the build orientation and printing bed surface structure but showed no dependence on the infill structure.

As expected, the results showed parts with solid infill exhibiting the greatest UTS (up to  $770 \pm 2$  MPa). Triangular and gyroid infills performed inferior mainly due to their lower infill degree of 59 % on average. Regarding the build orientations, flatwise printed specimens were more ductile, showing a higher elongations and lower Young's Modulus. The dimensions (length, width, and height) of the specimens, however, were neither significantly influenced by the infill pattern nor the printing orientation, due to all parts being printed with a shell of four outlines, four bottom and four top layers. The shrinkage because of the sintering process was observed to be largest in z-direction (15.50 %) on account of the force of gravity during sintering. Based on the evaluation of the fracture surface is it important to note that the gyroid pattern could not withstand the washing and/or sintering process when printed in a flatwise orientation. The infill pattern collapsed with all the infill accumulating at the bottom of the sample. Therefore, the degree of infill using a gyroid pattern needs to be increased for successful printing results. The evaluation in the context of lightweight engineering shows, that infill structures are beneficial within the elastic regime where 40 % of weight savings resulted in only 20 % decrease of the specimens' mechanical performance. The infill patterns also showed slight advantages for the ultimate tensile strength, while having a harmful influence on the strain at break.

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