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<u>Mariana Pop</u>, <u>Ioana-Monica Sas-Boca</u>\*, Dan Frunza, <u>Florin Popa</u>, <u>Adriana Neag</u>

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Article

# Aspects Regarding the Influence of Hot Deformation on the Mechanical and Structural Properties of 42CrMo4 Steel

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Abstract: The aim of the paper is to analyze the experimental results of the influence of elevated temperatures and strain rate on the mechanical and structural properties of steel 42CrMo4. The experiments were based on uniaxial tension and compression tests at high temperatures between 700 °C and 1000 °C and strain rates range 0.0018 - 0.1 s<sup>-1</sup>. The influence of temperature and strain rate on yield stress, strain to fracture, hardness, structural changes, and fracture characteristics were analyzed. Due to the dynamic recrystallization phenomenon present during the hot tensile tests, an increase in the flow stress is observed at the beginning of the deformation, after which it decreases until the fracture. By increasing the deformation temperature from 700 °C to 1000 °C, the tensile stress decreases significantly for all strain rates. The increase in the strain rate leads to the increase in the tensile stress. In compression tests by increasing the strain rate, the true strain is slightly increasing, but this depends on the temperature. The non-uniformity of deformations obtained at different values of the strain rate and temperature were also analyzed. Analysis by scanning electron microscopy showed the ductile behavior of the material. The degree of damage of the material caused by the presence of cavities increases by increasing the deformation temperature. For all the presented deformation conditions, the presence of the fracture through the ductile fracture mechanism was produced by the localized of necking and the coalescence of microvoids. By increasing the deformation temperature and reducing the strain rate, the fracture behavior of 42CrMo4 steel can be improved.

**Keywords:** steel; tensile test; compression test; elevated temperatures; scanning electronic microscopy

# 1. Introduction

The commercialization of steel is constantly developing, the range of operation of steels is quite wide, in the field of constructions, the field of transports (cars, trucks, aerospace industry, ship building and railways). Currently, to find out the characteristics of the deformation, laboratory equipment and special programs can be used, which are able to simulate physically or mathematically in a simplified way the real operational conditions, the information provided can be used later to determine the formability of the investigated material.

Steel 42CrMo4 is widely used in the machine building industry, being used in the manufacture of high-strength parts of compressors, turbines, working elements of heavy surface and underground equipment, as well as parts of agricultural equipment and other applications. In general, its application is for statically and dynamically stressed components for engines and machines [1–10]. Steel 42CrMo4 is a low alloy steel with chromium, molybdenum and manganese usually used in hardened and tempered state and has wide industrial applicability due to its high strength and hardness, good fatigue behavior and good machinability [9,10]. Despite the efforts made in studying the behavior of 42CrMo4 steel, the effects of hot working process parameters on the stress distribution, strain, and microstructural evolution of hot worked 42CrMo4 steel need to be further investigated to study the workability and to optimize the workability parameters hot [11,12].

Although the properties of 42CrMo4 steel have been studied, there is still a great interest in studying the mechanical properties of this material [13–16]. In the past, many investigations have been carried out on the behaviors of 42CrMo4 steel [15–20]. Lin YC et al. shown that preference for use of 42CrMo4 high-strength steel is mainly due to its good balance of strength, toughness, and water resistance [21,22]. Research was also carried out regarding the influence of temperature, strain rate on the deformation behavior and microstructure of the as extruded 42CrMo4 alloy [23,25]. During hot compression was studied the dynamic recrystallization kinetics of 42CrMo steel [26–30]. Constitutive models and equations were developed to explain the hot behavior of 42CrMo4 steel [31–39]. Nurnberger et al. studied the microstructure behavior of 42CrMo4 steel during continuous cooling from hot deformation temperatures [40–43]. Arun S. has studied the influence of the thermomechanical processing of the 40CrMo4 alloy on the mechanical properties [12].

In recent years, the research of Andreatta F., et al. focused on inclusions in 42CrMo4 QT steel [44], Díaz A., et al. studied the influence of hydrogen on the hydraulic fracture behavior of a 42CrMo4 steel welds [45] and Polášek M., et al., did research on the contact fatigue resistance of gun barrel steels [46].

The fracture behavior can be influenced by certain general factors such as: the temperature of the material during deformation, the speed of stress, the degree of triaxiality of the stress states generated in the material under stress, depending on the complexity of the stress and the presence of stress concentrators in the material. The complexity of the stresses is determined by the way the loads act and the stress concentrators (scratches, holes). The intensity of the stresses produced by a mechanical test in a material with stress concentrators is much higher than in another material where the same mechanical test was produced but which does not have stress concentrators.

The purpose of this work is to study the influence of the deformation process parameters (temperature, strain rate) on the mechanical (yield stress, hardness) and structural properties of the 42CrMo4 alloy. The study of these properties was carried out based on uniaxial tension and compression tests at temperatures between 700 °C and 1000 °C and strain rates range 0.0018 - 0.1 s<sup>-1</sup>. Analysis by scanning electron microscopy showed the ductile behavior of the material.

### 2. Experimental Details

The material used in this study was the commercial steel 42CrMo4, and its chemical compositions are presented in Table 1. The experimental research pursued the study of temperature and strain rate influence on mechanical and structural properties of 42CrMo4 steel. For this purpose, the experimental tests were carried out and included: tensile; compression; hardness; and scanning electronic microscopy (SEM) by using JEOL JMS 5600 LV (Tokyo, Japan) and AZtech software (version 4., Oxford Instruments, High Wycombe, UK).

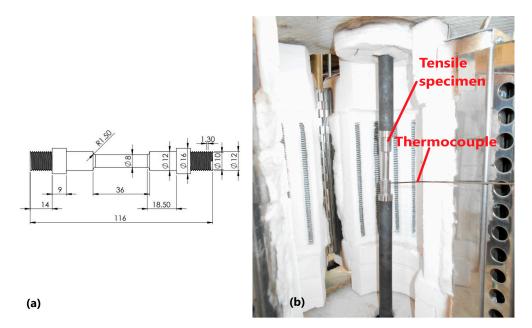
Table 1. The chemical composition of 42CrMo4 steel (wt.%).

| С           | Si      | Mn        | P         | S         | Cr        | Mo         |
|-------------|---------|-----------|-----------|-----------|-----------|------------|
| 0.38 - 0.45 | Max 0.4 | 0.6 - 0.9 | Max 0.025 | Max 0.035 | 0.9 - 1.2 | 0.15 - 0.3 |

The hot tensile Figure 1(a) and compression tests Figure 2(a) were performed on a Heckert type hydraulic press with a maximum force of 200KN (200kN hydraulic Heckert-EDZ-20S testing machine). The hot tensile tests were carried out at temperatures of T=700 °C, T=800 °C, T=900 °C, T=1000 °C, respectively strain rates of 0.0018 s<sup>-1</sup>, 0.012 s<sup>-1</sup>, 0.08 s<sup>-1</sup> in the vertical furnace, with electric heating, in Figure 1(b).

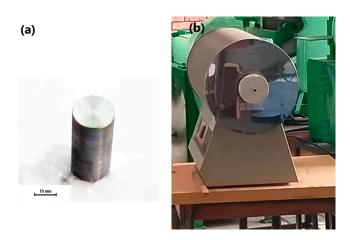
Tensile specimens were prepared from 18 mm diameter extruded round bars. The geometry and dimensions of the specimen are determined by ASTM standards. The dimensions of the initial tensile specimen are shown in Figure 1.





**Figure 1.** The hot tensile tests. (a) Dimension of the tensile specimen; (b) Vertical furnace, with electric heating.

The shape of the initial compression specimen is shown in Figure 2(a).



**Figure 2.** The hot compression tests (**a**) Compression test specimen. (**b**) Electric furnace type Carbolite.

The hot compression tests were carried out on specimens with dimensions of Ø18 x 30 mm heated on electric furnace type Carbolite CTF/12/75/700 at the temperatures T=800 °C, T=900 °C, T=1000 °C, at three strain rates 0.033 s<sup>-1</sup>, 0.066 s<sup>-1</sup>, 0.1 s<sup>-1</sup>.

# 3. Experimental Results

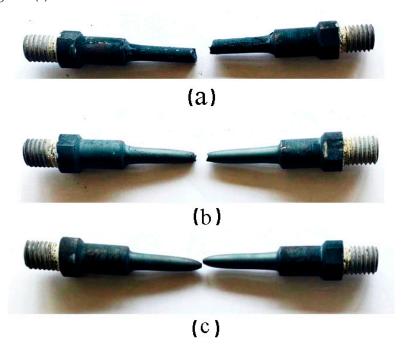
# 3.1. Hot Tensile Behavior

Figures 3–6 show the macro morphology of the specimens obtained after the tensile tests under different temperatures and strain rates conditions.

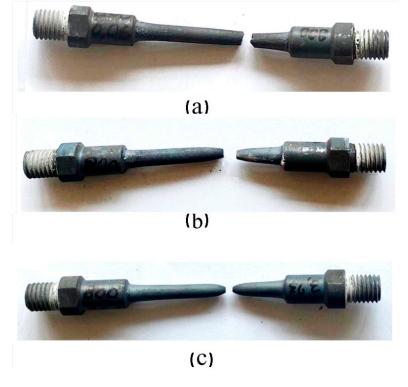
Tensile test specimen tested at 1000 °C from Figure 3 shows ductile fracture. With the increase in strain rates, the necking is more pronounced.

**Figure 3.** Tensile tests at T=1000 °C and strain rates conditions: (**a**)  $\dot{\varepsilon}$ = 0.001833 s<sup>-1</sup>; (**b**)  $\dot{\varepsilon}$ = 0.012833 s<sup>-1</sup>; (**c**)  $\dot{\varepsilon}$ = 0.089722 s<sup>-1</sup>

Tensile test specimen tested at 900 °C, Figure 4 and tensile test specimen tested at 800 °C, Figure 5, shows ductile fracture. In Figure 4(a) and Figure 5(a) the neck is very small, and fracture looks more fragile than Figure 4(b), and Figure 4(c). With the increase in strain rates, the necking is more pronounced, Figure 4(c).

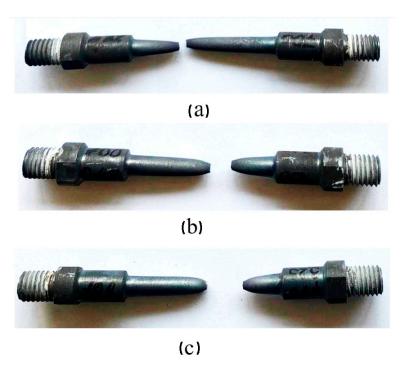


**Figure 4.** Tensile tests at T=900 °C and strain rates conditions: (**a**)  $\dot{\varepsilon}$ = 0.001833 s<sup>-1</sup>; (**b**) $\dot{\varepsilon}$ = 0.012833 s<sup>-1</sup>; (**c**)  $\dot{\varepsilon}$ = 0.089722 s<sup>-1</sup>



**Figure 5.** Tensile tests at T=800 °C and strain rates conditions: (a)  $\dot{\varepsilon}$ = 0.001833 s<sup>-1</sup>; (b)  $\dot{\varepsilon}$ = 0.012833 s<sup>-1</sup>; (c)  $\dot{\varepsilon}$ = 0.089722 s<sup>-1</sup>

Tensile test specimen tested at  $700\,^{\circ}\text{C}$ , Figure 6, shows ductile fracture in all strain rates conditions.



**Figure 6.** Tensile tests at T=700 °C and strain rates conditions: (a)  $\dot{\varepsilon}$ = 0.001833 s<sup>-1</sup>; (b)  $\dot{\varepsilon}$ = 0.012833 s<sup>-1</sup>; (c)  $\dot{\varepsilon}$ = 0.089722 s<sup>-1</sup>

Macro fracture morphology under different temperatures and strain rates shows that the ductile fracture behavior can be observed for all deformation conditions, due to the presence of necking localization before fracture.

The results obtained from the experimental tests are presented in Table 2.

**Table 2.** Experimental results.

| Crt No. | Material | 10   | <b>d</b> o | 11   | <b>d</b> 1 | <b>d</b> r | <b>E</b> 1 | ε <sub>d</sub> | V     | έ          | T    |
|---------|----------|------|------------|------|------------|------------|------------|----------------|-------|------------|------|
|         |          | [mm] | [mm]       | [mm] | [mm]       | [mm]       | [%]        | [%]            | [m/s] | $[s^{-1}]$ | [°C] |
| 1       | 42CrMo4  | 36   | 8          | 59.6 | 5.1        | 6.3        | 65.56      | 36.3           | 0.066 | 0.001833   | 1000 |
| 2       | 42CrMo4  | 36   | 8          | 57.0 | 5.0        | 6.5        | 58.33      | 37.5           | 0.066 | 0.001833   | 950  |
| 3       | 42CrMo4  | 36   | 8          | 56.7 | 4.3        | 6.5        | 57.50      | 46.3           | 0.066 | 0.001833   | 900  |
| 4       | 42CrMo4  | 36   | 8          | 56.3 | 3.9        | 6.3        | 56.39      | 51.3           | 0.066 | 0.001833   | 850  |
| 5       | 42CrMo4  | 36   | 8          | 56.0 | 2.9        | 6.8        | 55.56      | 63.8           | 0.066 | 0.001833   | 800  |
| 6       | 42CrMo4  | 36   | 8          | 49.9 | 3.4        | 5.5        | 38.61      | 57.5           | 0.066 | 0.001833   | 750  |
| 7       | 42CrMo4  | 36   | 8          | 42.2 | 2.3        | 7.5        | 17.22      | 71.3           | 0.066 | 0.001833   | 700  |
| 8       | 42CrMo4  | 36   | 8          | 61.2 | 3.9        | 6.7        | 70.00      | 51.3           | 0.462 | 0.012833   | 1000 |
| 9       | 42CrMo4  | 36   | 8          | 53.7 | 3.0        | 6.1        | 49.17      | 62.5           | 0.462 | 0.012833   | 900  |
| 10      | 42CrMo4  | 36   | 8          | 50.0 | 2.8        | 5.8        | 38.89      | 65.0           | 0.462 | 0.012833   | 800  |
| 11      | 42CrMo4  | 36   | 8          | 48.0 | 2.6        | 5.6        | 33.33      | 67.5           | 0.462 | 0.012833   | 700  |
| 12      | 42CrMo4  | 36   | 8          | 63.5 | 5.5        | 6.4        | 76.39      | 31.3           | 3.230 | 0.089722   | 1000 |
| 13      | 42CrMo4  | 36   | 8          | 63.1 | 3.0        | 6.0        | 75.28      | 62.5           | 3.230 | 0.089722   | 900  |
| 14      | 42CrMo4  | 36   | 8          | 59.7 | 2.6        | 6.1        | 65.83      | 67.5           | 3.230 | 0.089722   | 800  |
| 15      | 42CrMo4  | 36   | 8          | 43.5 | 2.2        | 7.5        | 20.83      | 72.5           | 3.230 | 0.089722   | 700  |

As can be seen by increasing the temperature at the same strain rate, the longitudinal deformation increases, determining the increase in material formability.

Where:

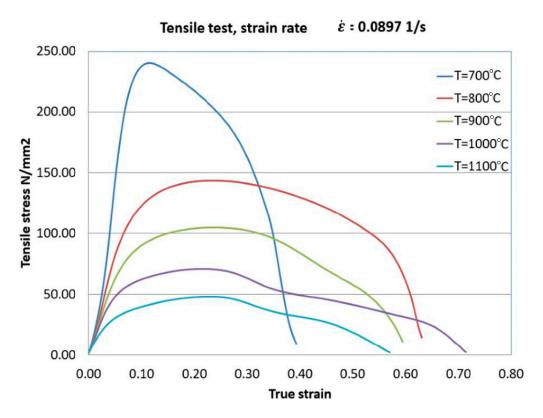
$$\varepsilon_l = \frac{l_1 - l_0}{l_0} \cdot 100 \tag{1}$$

$$\varepsilon_d = \frac{d_0 - d_1}{d_0} \cdot 100 \tag{2}$$

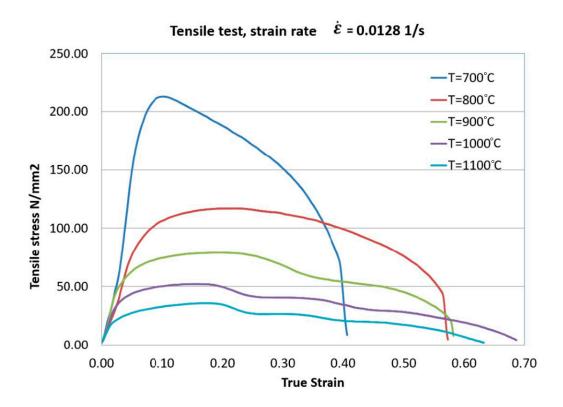
Figures 7-9 show the tensile stress evolution according to the true strain of the specimen for different experimental conditions. The flow curves obtained at different temperatures: 700 °C, 800 °C, 900 °C, 1000 °C and 1100 °C, at strain rate 0.0897 s<sup>-1</sup> (Figure 7); at strain rate 0.0128 s<sup>-1</sup> (Figure 8); strain rate 0.0018 s<sup>-1</sup> (Figure 9). Increasing strain rate was observed also an increased uniformity of curves. At the strain rate of 0.0897 s<sup>-1</sup> (Figure 7) and 700 °C, a maximum tensile stress of 240 N/mm<sup>2</sup> more than to the tensile stress 213 N/mm<sup>2</sup> at strain rate 0.0128 s<sup>-1</sup> (Figure 8), 158 N/mm<sup>2</sup>, on 0.0018 s<sup>-1</sup> strain rate (Figure 9) at the same condition of temperature. At 800 °C were obtained a maximum tensile stress of 144 N/mm<sup>2</sup> (Figure 7) comparative with a maximum tensile stress of 117 N/mm<sup>2</sup> (Figure 8) and a maximum tensile stress of 91 N/mm<sup>2</sup> (Figure 9). At 900 °C were obtained a maximum tensile stress of 105 N/mm<sup>2</sup> (Figure 7) comparative with a maximum tensile stress of 79 N/mm<sup>2</sup> (Figure 8) and a maximum tensile stress of 60.4 N/mm<sup>2</sup> (Figure 9). At 1000 °C were obtained a maximum tensile stress of 71 N/mm<sup>2</sup> (Figure 7) comparative with a maximum tensile stress of 52 N/mm<sup>2</sup> (Figure 8) and a maximum tensile stress of 37.5 N/mm<sup>2</sup> (Figure 9). Similarly, at 1100 °C were obtained a maximum tensile stress of 48 N/mm<sup>2</sup> (Figure 7) comparative with a maximum tensile stress of 35.6 N/mm<sup>2</sup> (Figure 8) and a maximum tensile stress of 24.6 N/mm<sup>2</sup> (Figure 9). The work hardening and dynamic softening stages can be observed here.

The influence of temperature and strain rate on the maximum ultimate tensile stress is shown in Figures 7–9. In all studied deformation conditions by increasing temperature the maximum stress decrease, we have observed, Figure 10.

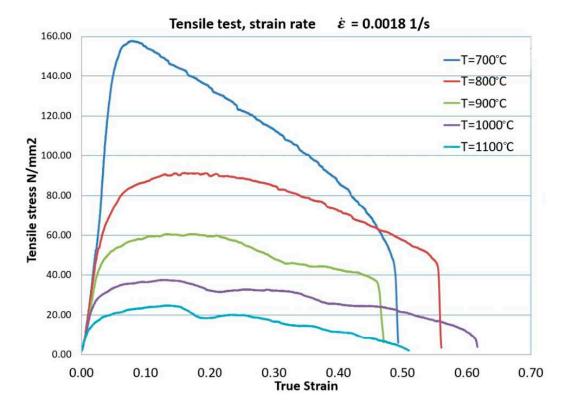
Along with temperature increase there is a major decrease of stress curves. Along with strain rate increase there is a major increase of stress curves, Figure 10.



**Figure 7.** Variation of tensile stress as a function of true strain, at strain rate of  $\dot{\varepsilon} = 0.0897 \text{ s}^{-1}$ .



**Figure 8.** Variation of tensile stress as a function of true strain, at strain rate of  $\dot{\varepsilon} = 0.0128 \text{ s}^{-1}$ .



**Figure 9.** Variation of tensile stress as a function of true strain, at strain rate of  $\dot{\varepsilon} = 0.0018 \text{ s}^{-1}$ .

The flow stress decreases with the decrease of strain rate and the increase of temperature as can be seen in Figure 9. The explanation for this phenomenon is that the low strain rate leads to a longer time for the accumulation of energy, at the same time the high temperature favours nucleation, the growth of dynamically recrystallized grains and by eliminating the barriers given by the dislocations the flow stress decreases [41,42]. Due to the combined effects of work hardening and softening due to high temperature, the yield stress curves show different hot deformation behaviors of the material. As can be seen in the Figure, at low deformation temperatures (700 °C), the yield stress increases to a maximum value and then monotonously decreases until fracture [23].

From Figures 7–9, it can be seen that in the first stage of the deformation work hardening is present. The yield stress then increases in the second stage to a maximum value for all the strain rates and in the third stage of the deformation the tension decreases until the fracture occurs.

The influence of strain rate and temperature on the maximum tensile stress is shown in Figure 10.

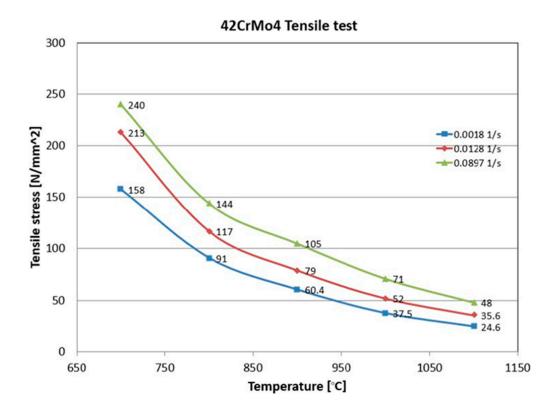


Figure 10. Variation of tensile stress with temperature for different strain rates.

The increase in the strain rate leads to the increase in the tensile stress, as can be seen.

At low strain rates Dynamic restoration (DRV) and dynamic recrystallization (DRX) have enough time to occur, so that the effect of work hardening is removed, which leads to the reduction of the stress necessary for deformation.

### 3.2. Compression Tests Behavior

The compression test is the most general test used to study the formability of materials. Carrying out the test at high temperature raises problems related to the presence of friction on the contact surfaces and the control of the process parameters (temperature, strain rate). The values obtained by the compression test on the press at the temperature of 800 °C, 900 °C, 1000 °C are presented in Tables 3–5.

| Table 3. Experimenta | l values of the compre | ession test at the te | emperature of 800 °C. |
|----------------------|------------------------|-----------------------|-----------------------|
|----------------------|------------------------|-----------------------|-----------------------|

| Crt.<br>No. | Material | T<br>[°C] | d <sub>0</sub><br>[mm] | h <sub>0</sub><br>[mm] | V<br>[m/s] | d <sub>min</sub><br>[mm] | d <sub>max</sub><br>[mm] | hı<br>[mm] | εh<br>[%] | Ed<br>[%] | $rac{d_{min}}{d_{max}} \ egin{array}{c} - egin{array}{c} & & & \ & & \ & \ & \ & \ & \ & \ & \ $ | έ<br>[s-1] |
|-------------|----------|-----------|------------------------|------------------------|------------|--------------------------|--------------------------|------------|-----------|-----------|---|------------|
| 1.          | 42CrMo4  | 800       | 18                     | 30                     | 0.003      | 18.35                    | 27                       | 17.1       | 43        | 5         | 1.47139   | 0.1        |
| 2.          | 42CrMo4  | 800       | 18                     | 30                     | 0.002      | 18.2                     | 25.4                     | 19.2       | 36        | 41        | 1.395604  | 0.066      |
| 3.          | 42CrMo4  | 800       | 18                     | 30                     | 0.001      | 18.25                    | 23.1                     | 23.2       | 22        | 28        | 1.265753  | 0.033      |

Table 4. Experimental values of the compression test at the temperature of 900 °C.

| Crt.<br>No. | Material | T<br>[°C] | d <sub>0</sub><br>[mm] | h <sub>0</sub><br>[mm] | V<br>[m/s] | d <sub>min</sub><br>[mm] | d <sub>max</sub><br>[mm] | hı<br>[mm] | εh<br>[%] | εd<br>[%] | $rac{d_{min}}{d_{max}} \ egin{array}{c} & & & \ & & \ & \ & \ & \ & \ & \ & \ $ | έ<br>[s <sup>-1</sup> ] |
|-------------|----------|-----------|------------------------|------------------------|------------|--------------------------|--------------------------|------------|-----------|-----------|--|-------------------------|
| 1.          | 42CrMo4  | 900       | 18                     | 30                     | 0.003      | 19.5                     | 27.7                     | 15.7       | 47        | 53        | 1.420  | 0.1                     |
| 2.          | 42CrMo4  | 900       | 18                     | 30                     | 0.002      | 20.6                     | 27.8                     | 18.1       | 39        | 54        | 1.349  | 0.066                   |
| 3.          | 42CrMo4  | 900       | 18                     | 30                     | 0.001      | 23.3                     | 30.2                     | 21.9       | 27        | 67        | 1.296  | 0.033                   |

**Table 5.** Experimental values of the compression test at the temperature of 1000 °C.

| Crt.<br>No. | Material | T<br>[°C] | d <sub>0</sub><br>[mm] | h <sub>0</sub><br>[mm] | V<br>[m/s] | d <sub>min</sub><br>[mm] | d <sub>max</sub><br>[mm] | h <sub>1</sub><br>[mm] | Eh<br>[%] | &d<br>[%] | $rac{d_{min}}{d_{max}} \ egin{array}{c} - \end{bmatrix}$ | έ<br>[s <sup>-1</sup> ] |
|-------------|----------|-----------|------------------------|------------------------|------------|--------------------------|--------------------------|------------------------|-----------|-----------|---|-------------------------|
| 1.          | 42CrMo4  | 1000      | 18                     | 30                     | 0.003      | 20.8                     | 30                       | 13.1                   | 56        | 66        | 1.43269   | 0.1                     |
| 2.          | 42CrMo4  | 1000      | 18                     | 30                     | 0.002      | 18.4                     | 26.4                     | 18.3                   | 39        | 46        | 1.43478   | 0.066                   |
| 3.          | 42CrMo4  | 1000      | 18                     | 30                     | 0.001      | 18.45                    | 26.1                     | 21.1                   | 29        | 45        | 1.41463   | 0.033                   |

$$\delta = ln \frac{l_1}{l_0} \tag{2}$$

The shapes of the specimens resulting from the compression tests for different strain rates and different temperatures are presented in Figures 11–13.



**Figure 11.** Compression tests at T= 800 °C (a)  $\dot{\varepsilon}$  =0.033 s<sup>-1</sup>, (b)  $\dot{\varepsilon}$  = 0.066 s<sup>-1</sup>, (c)  $\dot{\varepsilon}$  =0.1 s<sup>-1</sup>.



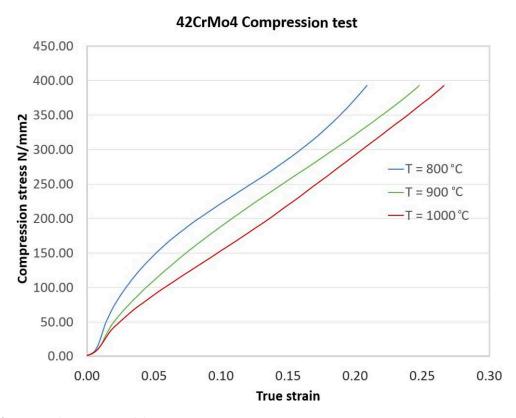
**Figure 12.** Compression tests at T= 900 °C (a)  $\dot{\varepsilon}$  =0.033 s<sup>-1</sup>, (b)  $\dot{\varepsilon}$  =0.066 s<sup>-1</sup>, (c)  $\dot{\varepsilon}$  =0.1 s<sup>-1</sup>.



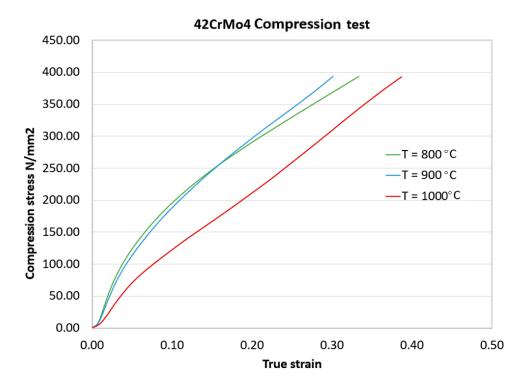
**Figure 13.** Compression tests at T= 1000 °C (a)  $\dot{\varepsilon}$  =0.033 s<sup>-1</sup>, (b)  $\dot{\varepsilon}$  =0.066 s<sup>-1</sup>, (c)  $\dot{\varepsilon}$  =0.1 s<sup>-1</sup>.

The variation of the compression stress according to the true strain for different temperatures and strain rates is presented in Figures 14–16. At lower strain rate of  $0.33~\rm s^{-1}$  (Figure 14) were obtained

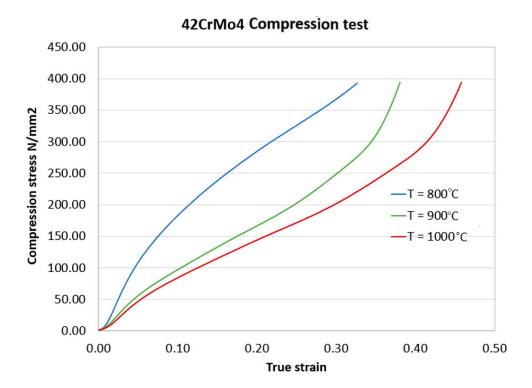
an expected increase of compression stresses, by increasing the true strain. By increasing the temperature from  $800\,^{\circ}\text{C}$  to  $1000\,^{\circ}\text{C}$  were obtained lower values of compressing stresses for all values of strain rates. From Figure 15 (0.33 s<sup>-1</sup>) we observe a slight deviation from this rule, in the range 0-0.15 true strain.



**Figure 14.** The variation of the compression stress vs. true strain,  $\dot{\varepsilon} = 0.033 \text{ s}^{-1}$ .



**Figure 15.** The variation of the compression stress vs. true strain,  $\dot{\varepsilon} = 0.066 \text{ s}^{-1}$ .



**Figure 16.** The variation of the compression stress vs. true strain,  $\dot{\varepsilon} = 0.1 \text{ s}^{-1}$ .

Analyzing the figures, is observed that once the strain rate increases, the true strain is slightly increasing, but this depends on the temperature. Thus, at a temperature of  $1000\,^{\circ}$ C, the higher strain of 0.46 is reached at a strain rate of  $0.1\,\mathrm{s}^{-1}$ . By increasing the strain rate, the compression stress decreases for the same true strain.

Figures 17–19 show the variation curves of the non-uniformity of deformations obtained at different values of the strain rate and temperature.

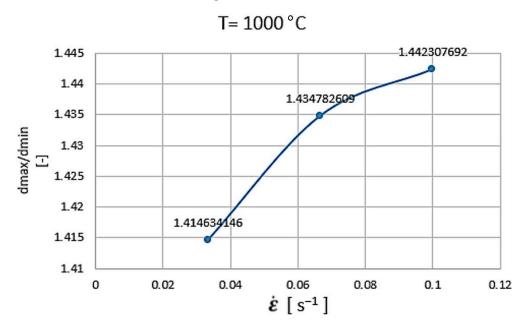


Figure 17. The variation of deformation non-uniformity vs. strain rate at 1000 °C.

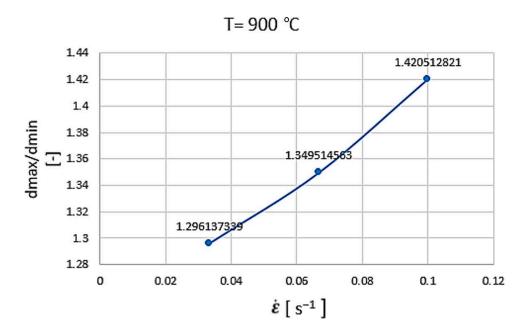


Figure 18. The variation of deformation non-uniformity vs. strain rate at 900 °C.

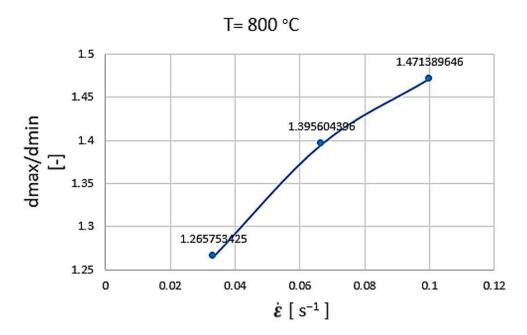


Figure 19. The variation of deformation non-uniformity vs. strain rate at 800 °C.

### Form 1

It can be observed, in all cases, an increase in the non-uniformity of the deformation with the increase in the strain rate.

### 3.3. Structural Analysis

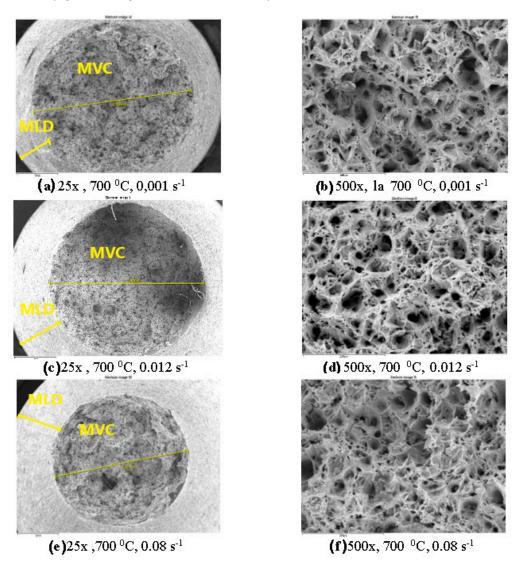
In order to analyze the influence of temperature and strain rate on structural evolution SEM microscopy images were examined.

Below are presented the images from the scanning electron microscopy made on the tensile tested specimens at different temperatures and strain rates. It was found that with the increase of the temperature at which the test takes place, the formability increases too. At the same time, at a higher magnification, the ductile fracture pattern of the samples can be observed in Figures 20–22.

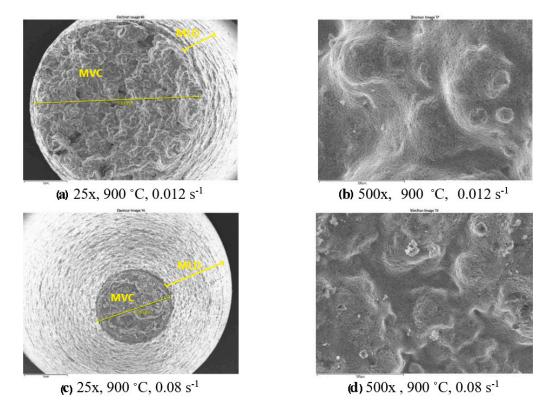
SEM images show classic cup-and-cone fracture surfaces at low magnification (25x) and high magnification (500x) at 700 °C (Figure 20); 900 °C (Figure 21), 1000 °C (Figure 22); (at different strain rates: strain rate (a), (b)  $0.001 \, \text{s}^{-1}$ ; (c), (d) strain rate  $0.012 \, \text{s}^{-1}$ ; (e), (f) strain rate  $0.08 \, \text{s}^{-1}$ . is presented in Figures 20 and 22 and at 900 °C, Figure 21 at strain rate  $0.001 \, \text{s}^{-1}$  (a), (b) and (c), (d) strain rate  $0.08 \, \text{s}^{-1}$ .

Ductile fracture characterized by microvoid coalescence (MVC) in the breaking section is described by all the tensile tests performed. The broken cone-head sample and the images at 500x magnification highlight the agglomeration of dimples in the breaking surface. The intergranular fracture occurs. All fracture surfaces are covered with distinct elongated dimples which indicate the ductile nature of the material at these deformation conditions.

A detached embrittlement mechanism is a transgranular fracture in the peripheral region [43] extending from the front of the notch, with detached developing along the propagation path of a crack that likely goes through the matrix lattice only (MLD).



**Figure 20.** SEM microstructures after the tensile tests for different strain rates at  $700 \,^{\circ}\text{C}$ : .(a) 25x,  $0.001 \,^{\circ}\text{s}^{-1}$ ; (c) 25x,  $0.012 \,^{\circ}\text{s}^{-1}$ ; (e) 25x,  $0.08 \,^{\circ}\text{s}^{-1}$ ; (b) 500x,  $0.001 \,^{\circ}\text{s}^{-1}$ ; (d) 500x,  $0.012 \,^{\circ}\text{s}^{-1}$ ; (f) 500x,  $0.001 \,^{\circ}\text{s}^{-1}$ .



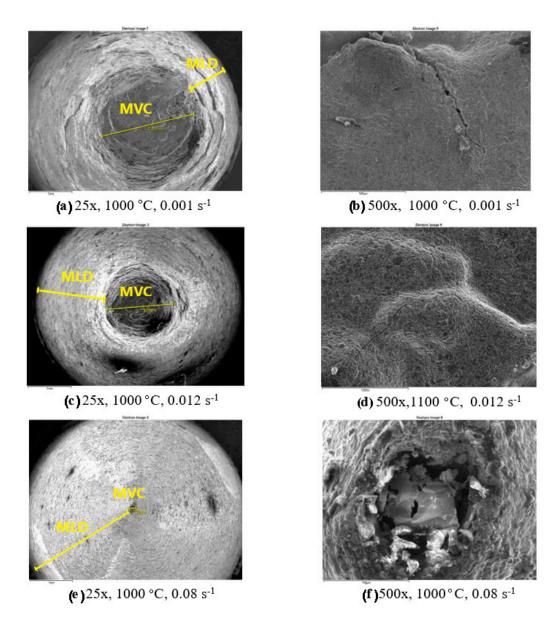
**Figure 21.** SEM microstructures after the tensile tests for different deformation conditions (900 °C): (a) 25x,  $0.012 \text{ s}^{-1}$ ; (c) 25x,  $0.08 \text{ s}^{-1}$ ; (b) 500x,  $0.012 \text{ s}^{-1}$ ; (d) 500x,  $0.08 \text{ s}^{-1}$ .

At low values,  $0.001~\rm s^{-1}$  of strain rate, the cracks are visibly larger Figure 22a,b, compared with  $0.012~\rm s^{-1}$  Figure 22c,d and with  $0.08~\rm s^{-1}$  Figure 22e,f.

The shape of the fracture surfaces indicates the fracture mechanism, which is a ductile one. The fractures occurred perpendicular to the stress direction. As can be seen, the fractures occurred intergranular, dimples can be identified on the break surfaces. Ductile fracture involves the growth and coalescence of voids that lead to the formation of specific formations called dimples.

An advanced detached embrittlement is present in the peripheral region at a temperature of 1000  $^{\circ}$ C under conditions of low strain rate, 0.001  $\rm s^{-1}$ .

During deformation at high temperatures, microscopic cavities appear at the grain boundaries, the phenomenon being called cavitation. The presence of this phenomenon, in some situations, can determine the occurrence of premature fracture in the material, at much lower strain than would occur in the case of a controlled flow-localization. The strain rate and deformation temperature influence the degree of cavitation. The presence of tensile stress also favors the appearance of the cavity phenomenon.



**Figure 22.** SEM microstructures after the tensile tests for different deformation conditions (1000 °C): (a) 25x, 0.001 s<sup>-1</sup>; (c) 25x, 0.012 s<sup>-1</sup>; (e) 25x, 0.08 s<sup>-1</sup>; (b) 500x, 0.001 s<sup>-1</sup>; (d) 500x, 0.012 s<sup>-1</sup>; (f) 500x, 0.001 s<sup>-1</sup>

The microstructures after the compression tests for different deformation conditions were presented in Figure 23. The SEM images show the microstructure in the longitudinal direction of the specimens, Figures 11–13, subjected to compression.

Following the microscopic analysis, it was observed that after hot plastic deformation, by compression, the structural constituents resulting from cooling in the air from the deformation temperature, are of the ferrite-pearlitic type (very fine pearlite, Figure 23b,d,f-h) as well as of the bainitic type. Compared to specimen deformed at room temperature, Figure 23 (a), the structure is ferrite-pearlitic type, with a more pronounced elongation perpendicular to the direction of compression, in the case of ferrite, which appears light in color.

In the case of a compression test, the cavitation phenomenon is not observed. The microscopic cavities produced during the tensile stress can be removed by a subsequent compression stress. In the case of hot processing, the cavity can lead to the appearance of premature failures during deformation, but it can also provide inferior properties to the final part.

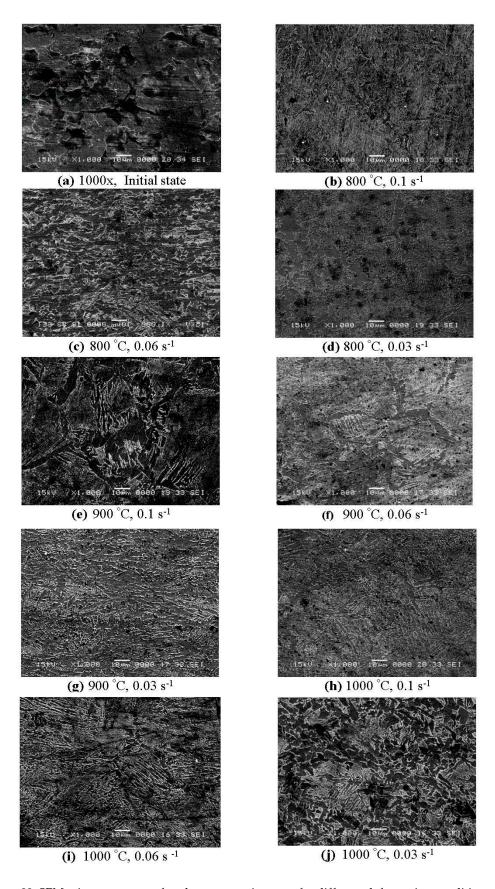
By scanning electron microscope analysis of the surface many lamellas on the fracture surface were found. These lamellas indicated bainitic structures Figure 23c,e,f–j. Figure 23 (j), the structure is predominantly bainitic-pearlitic type, grey and dark colored.

The nucleation, growth and coalescence of the cavity are the main stages of the cavitation phenomenon, which most often overlap during tensile stress. The possible nucleation mechanisms are the intersection of intergranular boundaries with non-deformable secondary phase particles or grain boundaries; the sliding of grains along their boundaries; the agglomeration of holidays at the borders between grants [47,48].

Diffusive growth and plasticity controlled growth are the mechanisms of cavity growth. When the size of the cavity is very small, diffusive growth predominates. With the increase in the size of the cavity, the increase in diffusion decreases very quickly and the growth mechanism through the plastic flow of the surrounding matrix becomes predominant [47,49].

The interconnection of the neighboring cavities determines the coalescence of the cavity. This is greatly influenced by the sensitivity of the material to the speed of deformation. It can be produced longitudinal or transverse directions in the requested material. Transverse direction coalescence is more important and can lead to fracture.

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**Figure 23.** SEM microstructures after the compression tests for different deformation conditions (800 °C, 900 °C and 1000 °C) at magnification 1000x, (a) Initial state; 0.001  $\rm s^{-1}$ ; (b) 800 °C, 0.1  $\rm s^{-1}$ ; (c) 800 °C, 0.06  $\rm s^{-1}$ ; (d) 800 °C, 0.03  $\rm s^{-1}$ ; (e) 900 °C, 0.1  $\rm s^{-1}$ ; (f) 900 °C, 0.06  $\rm s^{-1}$ ; (g) 900 °C, 0.03  $\rm s^{-1}$ ; (h) 1000 °C, 0.1  $\rm s^{-1}$ ; (i) 1000 °C, 0.06  $\rm s^{-1}$ ; (j) 1000 °C, 0.03  $\rm s^{-1}$ .

By analysing the microstructures of samples deformed at different temperatures, with different strain rates, in addition to the presence of pearlite and ferrite, arranged in a network in the case of samples deformed at  $800\,^{\circ}\text{C}$  and  $900\,^{\circ}\text{C}$ , the presence of formations that have the appearance of upper bainite can be observed.

### 4. Conclusions

Following the tests, the influence of different parameters of the deformation process (temperature, strain rate, state of stress) on the formability of the 42CrMo4 alloy was studied. The deformation conditions were temperatures (700 °C, 800 °C, 900 °C, 1000 °C) and different strain rates at the two stress states: tensile (0.001833 s<sup>-1</sup>, 0.012833 s<sup>-1</sup>, 0.089722 s<sup>-1</sup>) and compression (0.033 s<sup>-1</sup>, 0.066 s<sup>-1</sup>, 0.1 s<sup>-1</sup>).

As a result of these experimental tests, the following were observed:

- 1. Tensile stress significantly depends on temperature and strain rate. Yield stress increases by increasing strain rate and decreases by increasing temperature.
- 2. The experimental curves show typical characteristics that demonstrate the presence of dynamic recrystallization at high temperatures and low strain rates. At the deformation temperature of 800 °C, 900 °C, 1000 °C, when the balance between hardening and dynamic recovery takes place, the flow curves show a steady phase, in which the flow stress is approximately constant.
- 3. In case of compression tests is observed that once the strain rate increases, the true strain is slightly increasing, depending on temperature. The greatest non-uniformity of the deformations is observed in the case of the temperature of 1000 °C. By increasing the deformation speed, the non-uniformity of the deformations increases.
- 4. Scanning electron microscopy investigations demonstrate the influence of temperature and strain rate on the formability and structural change of the material.
- For all the presented deformation conditions. the presence of the fracture through the ductile fracture mechanism was produced by the localized of necking and the coalescence of microvoids.
- 6. By increasing the deformation temperature and reducing the strain rate. the fracture behavior of 42CrMo4 steel can be improved.

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# Nomenclature

lo. do the initial dimensions (length and diameter) of the sample [mm];

l<sub>1</sub>. d<sub>1</sub> the final dimensions of the sample [mm];

dr diameter of the sample in the fracture area [mm];

 $\varepsilon_1$ ,  $\varepsilon_d$  longitudinal and transversal strains [%];

V the deformation speed [mm/s];

 $\dot{\varepsilon}$  the strain rate [s<sup>-1</sup>];

T the deformation temperature [°C];

h<sub>0</sub>, h<sub>1</sub> initial and final height of the sample [mm];

dmin, dmax minimum and maximum diameter of the deformed sample [mm].

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