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Article

Study on the Influence Law of Ce on Microstructure and High Temperature Oxidation Resistance of Ti-Al-Si Composite Coating

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Abstract: High temperature titanium alloys are widely used in aerospace hot parts, however, the thermal barrier temperature of 600 °C limits its service temperature. In this paper, a two-step hot-dip plating method is proposed to prepare the composite coating containing Ce on the surface of titanium alloy, which can make the oxidation resistance temperature of titanium alloy reach 800 °C The microstructure, phase composition and element distribution of Ce-containing Ti-Al-Si hot-dip coating, Ce-containing Ti-Al-Si pre-oxidation coating and Ce-containing Ti-Al-Si high-temperature oxidation coating were studied by X-ray diffraction (XRD), scanning electron microscopy (SEM) and energy dispersive spectroscopy (EDS), respectively. The results show that the Ti-Al-Si hot-dip coating containing Ce is mainly composed of (Ti,Ce)(Al,Si)3 alloy phase layer, and the Ti-Al-Si pre-oxidation coating containing Ce is mainly composed of Ti-Al binary system phase layer and dense Al₂O₃ layer rich in CeO₂. Ce in the high temperature oxidation coating of Ti-Al-Si containing Ce is mainly distributed in Ti₃Al, TiAl, TiAl₃+Ti₅Si₃ mixed phase layer and Al₂O₃. The addition of Ce can promote the formation of continuous Al₂O₃ layer, improve the uniformity of composite coating, prevent the spread of cracks, and improve the high temperature oxidation resistance of Ti-Al-Si coating.

Keywords: Ti-Al-Si composite coating; titanium alloy; Ce; resistance to high temperature oxidation

1. Introduction

Titanium alloys have excellent properties such as high strength, good corrosion resistance, non-magnetic properties and good welding performance, and are widely used in some cutting-edge fields such as aerospace, military industry, Marine development and petrochemical industry [1,2]. Aviation materials need to withstand high temperature, high pressure, high stress, strong corrosion and other extremely harsh environmental erosion in the service process. High temperature titanium alloy has excellent mechanical comprehensive properties, which well meets the conditions of aviation materials in high altitude service. High temperature titanium alloy is widely used in aircraft engines [3-5]. It can greatly reduce their weight and improve their flight speed for its density is 60% lower than steel and other superalloys, or its weight is 45% lighter than standard low carbon steel [2].

The initial service temperature of high temperature titanium alloy is about 350 °C, and it has been continuously developed. At present, the service temperature of high temperature titanium alloy with relatively perfect technology is about 600 °C [6]. Because titanium alloy has a thermal barrier temperature of 600 °C [7], which means titanium alloy will undergo secondary oxidation at 600 °C to generate oxides that are no longer dense, so the use temperature of titanium alloy cannot be higher than 600 °C. In order to further improve the service temperature of high-temperature titanium alloys, two solutions are

proposed. One is to alloying high-temperature titanium alloys [6,8], which is to add other elements to titanium alloys to modify titanium alloys. This method can improve the service temperature of high temperature titanium alloy, but there are great uncertainties. Adding different kinds of alloying elements to high temperature titanium alloy and conducting different heat treatment will generate different structures, which has a great impact on its mechanical properties. If properly controlled, the performance of titanium alloy will be greatly improved, and conversely, it is also likely to reduce the performance of titanium alloy. Moreover, due to the high dissolved oxygen content of titanium at high temperatures, its surface forms a non-dense porous mixed oxide layer rather than a dense anti-oxidation protective film, which can not effectively prevent further oxidation of the substrate. Therefore, the effect of alloying high temperature titanium alloys to improve their service temperature is not ideal [5,9,10].

The second solution is to prepare a protective coating on the surface of the titanium alloy [11] without changing the titanium alloy substrate, so as to prevent the oxidation of the titanium alloy. Because Ti-Al coating has the advantages of high specific strength, high specific elastic modulus, low density, good oxidation resistance, high corrosion resistance, high creep resistance and high fatigue resistance, it has great development potential [12]. That's why a layer of Al is initially plated on the surface of titanium alloy to form a TiAl3rich layer. TiAl3 is oxidized into a dense Al2O3 film in contact with air, which can well protect the titanium alloy matrix at a temperature lower than 800°C [13]. However, the aircraft has severe friction with the air flow during high-speed flight, making the actual working temperature of high temperature titanium alloy is higher than 600 °C, or even more than 800 °C. Then the Al₂O₃ film can not effectively protect the titanium alloy substrate, because a plurality of TiAl phase layers are generated, and the thermal expansion coefficient of the phase layer is not matched with the substrate, resulting in penetration cracks. The addition of Si element can reduce the number of transverse cracks [3], and compared with Ti-Al layer, Ti-Al-Si coating has lower brittleness and better plasticity [14]. When Si is added as an alloying element, the formation of Si-rich phase can extend the life of Ti-Al protective layer [15], form an oxygen diffusion barrier to prevent the internal oxidation of Al and Ti, inhibit the formation of harmful nitrides such as TiN and Ti2AlN, and enhance the adhesion between scale and substrate during the oxidation process [16]. However, due to the high affinity between Ti and Si atoms [17,18], Ti tends to combine with Si to form Ti-Si intermediate phase first, and cannot form TiAl3 phase with good oxidation resistance, which cannot protect the substrate.

Therefore, a two-step hot-dip plating method was proposed in this paper, which firstly hot-dip plating Al on the surface of titanium alloy, preferentially form TiAl3 phase, and then hot-dip plating alloy liquid containing Si. Referring to the ternary phase diagram of Ti-Al-Si, it can be seen that Si can be solidly dissolved in TiAl3, replacing some of Al atoms in TiAl3 to form Ti(Al,Si)3 secondary solid solution [19]. Compared with other methods for preparing Ti-Al-Si coating, such as the self-generated gradient hot-dipping infiltration (SGHDI) [3,5] and pulsed laser cladding [20], the two-step hot-dip plating method can not only better control the content of added Si, inhibit the formation of τ_2 phase, but also make the shape of the sample free from the restriction of quartz glass tube, and the equipment is simple and easy to operate.

Adding rare earth elements to superalloys can effectively improve the oxidation resistance of the alloys at high temperatures [21]. The rare earth elements have purification, modification, and alloying effects [22]. Among them, Ce element can reduce the oxygen concentration in the alloy [23] to avoid tissue oxidation. At the same time, the formed cerium oxide nanoparticles can refine the grain of titanium alloy and play a role in fine crystal strengthening [24]. Ce element can also inhibit the formation of τ_2 phase, which may have an adverse effect on the overall oxidation resistance of the coating [3], and improve the high temperature oxidation resistance of the coating. In summary, the Ce element can improve the performance of the coating from many aspects, so choose to add Ce to the Ti-Al-Si coating.

2. Experimental Method and Process

2.1. Coating Preparation

The wire cut Ti65 samples with length, width and height of 15mm, 10mm and 3mm respectively were polished with 400#, 600# and 800# SiC sandpaper until the surfaces were bright and silver-white. Then the polished Ti65 samples surfaces were cleaned with metal degreasing agent, and then washed with clean water and put into the ultrasonic cleaning machine (Yu Meng brand, Shenzhen Fangao Microelectronics Co., LTD., Shenzhen, China) in alcohol to clean and remove water, and finally put into the drying oven (Xiangtan Mita Electric Furnace Co., LTD., Xiangtan, China) to dry for use. With purity of 99.999wt.%Al, purity of 99.999wt.%Si, high purity Ce, prepared a certain quality of pure Al liquid and Al-10wt.%Si-1wt.%Ce alloy liquid into two different crucible, and then put them into the pit furnace (Dongtai Aozhing Mechanical and Electrical Equipment Factory, Jiangsu, China), dissolved by heating at 750 °C. After the solution in the crucible was completely dissolved, argon gas was poured into the pit furnace. Then skimmed off the alumina film on the pure Al liquid and immersed the dried Ti65 sample in the pure Al liquid holding for a certain time for the first hot-dip plating. Skimmed off the oxide film on the Al-10wt.%Si-1wt.%Ce alloy liquid and extracted the sample from the pure Al liquid, then quickly transferred it to the Al-10wt.%Si-1wt.%Ce alloy solution for the secondary hotdip plating. After holding for a certain period of time, the sample was extracted from the alloy solution and the excess hot-dip solution on the surface was removed. Finally, the sample was quenched and dried.

2.2. High Temperature Oxidation Test

By observing the morphology of the coating with different hot-dipping time, the Ti65 sample with dense coating and moderate thickness of 15 min Al+15 min Al-10 wt.%Si-1wt.%Ce was selected for high temperature oxidation experiment. The dried samples were put into small corundum crucibles that were washed, dried and marked, and put into a box furnace (Xiangtan Mita Electric Furnace Co., LTD., Xiangtan, China) for preoxidation treatment at 500°C /1h+850°C /2h, and then took out the crucibles. After they were completely cooled, they were weighed with an electronic balance (Shanghai Huachao Industrial Co., LTD., Shanghai, China) (The samples were heated and weighed in the crucibles to prevent the peeling of the oxide film from affecting the test results). A small crucible was repeatedly weighed 3 times, and the final weight was taken as the average of the 3 times of weight, which is recorded as the original weight. Then put the small crucibles into the box furnace for high temperature oxidation treatment at 800 °C and recorded the time. Every day took out 3 small crucibles at the same time (the sample in each small crucible had the same process parameters), and then weighed them after they were completely cooled. Each crucible was weighed 3 times and taken the average, and there were 3 weight values. The increased weight of the sample after high temperature oxidation was obtained by subtracting the obtained 3 weight values from the weight of the small crucible without high temperature oxidation, and then take the average value.

2.3. Microstructure Detection

The thickness of the alloy layer was measured by ZEISS optical microscope software. The final coating thickness was obtained from 3 samples with the same process parameters. 10 coating thickness values were measured at different parts of each sample, totaling 30 thickness values, and the average thickness of the coating was taken. Scanning electron microscope (SEM, ZEISS EVO MA10, ZEISS, Jena, Germany) and energy dispersive spectrometer (EDS, OXFORD X-MAXN, ZEISS, Jena Germany) were used to observe the microstructure of the coating and the surface of the sample, the distribution of each element and they could also determine the content of each element. X-ray diffractometer (XRD, Ultima IV, Rigaku Co., Tokyo, Japan) was used to detect the phase on the surface of high temperature oxidation resistant samples.

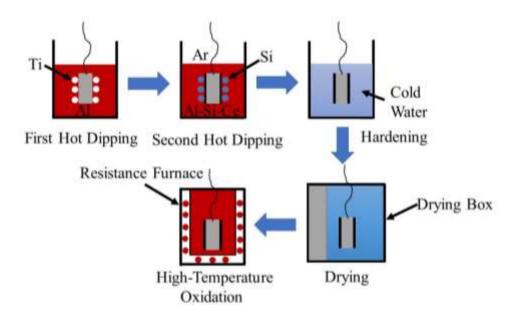


Figure 1. Schematic diagram of experimental process

3. Results and Discussion

3.1. Microstructure Characterization of Coatings

3.1.1. Hot-dip Coating Tissue

As can be seen from Figure 2, the generated hot-dip coating has obvious delamination. Figure2(a-b) are divided into three layers outwards from the substrate: dense alloy layer, loose alloy layer and liquid layer; Figure 2(c-d) are divided into two layers outwards from the substrate: dense alloy layer and liquid layer. EDS results (Table 1) show that the content of Al in region 1# is 99.73%, the content of Si is 0.27%, and the atomic percentage of elements in region 2# is Al:Si:Ti= 63.75:9.55: 26.57, the Ce content is 0.13%, so the region of 1# is the Al-Si layer and the region of 2# is the (Ti,Ce)(Al,Si) 3-phase layer. In other words, the (Ti,Ce)(Al,Si)3 layer and Al-Si layer are successively formed from the substrate after the second hot-dip plating of Al+Al-10wt.%Si-1wt.%Ce.

With the increase of hot-dip plating time of Al-10wt.%Si-1wt.%Ce, the (Ti,Ce)(Al,Si)³ phase layer gradually becomes dense which is because Ti atoms fully diffuse outward to make up for the vacancy of the loose alloy layer, so the (Ti,Ce)(Al,Si)³ phase layer becomes denser. The quality of the alloy layer obtained is the best when the hot-dip plating is 15minAl-10wt.%Si-1wt.%Ce. If the hot-dip plating time is increased, although the alloy layer is dense, it is easy to crack.

This method of preparing the coating achieves the purpose of making Al react with the substrate to form TiAl₃ phase layer first, and then let Si atoms and Ce atoms diffuse into it. At the same time, the formation of dense (Ti,Ce)(Al,Si)₃ phase layer is promoted by controlling the hot-dip plating time.

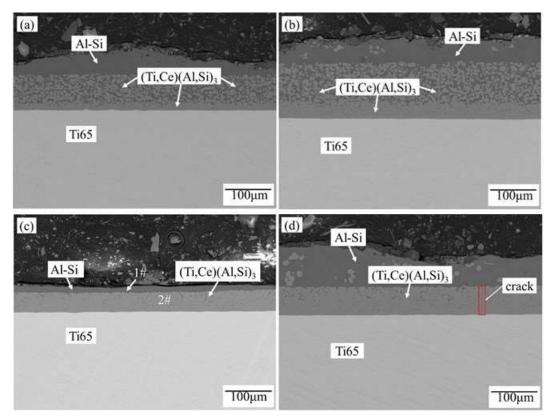


Figure 2. Microstructure of Ti-Al-Si composite coating at 750°C under different hot-dip plating time,(a) 15min Al+5min (Al-10wt.%Si-1wt.%Ce), (b)15min Al+10min (Al-10wt.%Si-1wt.%Ce), (c)15mim Al+15min (Al-10wt.%Si-1wt.%Ce) (d)15min Al+20min (Al-10wt.%Si-1wt.%Ce)

Table 1. EDS data (at.%) and phase components of each phase layer in Figure 2(c)

Analysis region	Al	Si	Ti	Ce	Phase
1#	99.73	0.27	0	0	Al-Si
2#	63.75	9.55	26.57	0.13	(Ti,Ce)(Al,Si) ₃

3.1.2. Growth Kinetics of Coatings

Analysis of the growth kinetics of (Ti,Ce)(Al,Si)₃ alloy layer is helpful to understand the formation mechanism of (Ti,Ce)(Al,Si)₃ alloy layer. In addition, the (Ti,Ce)(Al,Si)₃ alloy layer is the key coating to resist high temperature oxidation of titanium alloy, so it is necessary to analyze the growth kinetics of the (Ti,Ce)(Al,Si)₃ alloy layer. There is an empirical formula[25]:

$$\delta = kt^n \tag{1}$$

where δ is the alloy layer thickness (μ m), t is the hot immersion time (min), k is the rate constant, n is the kinetic index, and the kinetic index $n \le 0.5$ is the diffusion-controlled growth, n > 0.5 is interfacial controlled growth (reaction controlled growth).

The kinetic index n > 0.5 in the fitting formula of Figure 3(a) indicates that hot-dip plating Al controls the growth of (Ti,Ce)(Al,Si)₃ alloy layer by reaction. In the fitting formula of Figure 3(b) the kinetic index $n \le 0.5$ indicates that hot-dip plating of Al-Si-Ce controls the growth of (Ti,Ce)(Al,Si)₃ alloy layer through diffusion.

The results show that the growth of (Ti,Ce)(Al,Si)₃ alloy layer is affected by two hot-dipping. It can be seen from the fitting curve that with the increase of hot-dipping plating time, the alloy layer thickness shows an increasing trend as a whole. Compared with the

steepness of the curves in Figure 3(a-b), it can be seen that the hot-dipping Al plating time has a greater impact on the coating thickness which means the coating thickness increases more with the increase of the hot-dipping Al plating time.

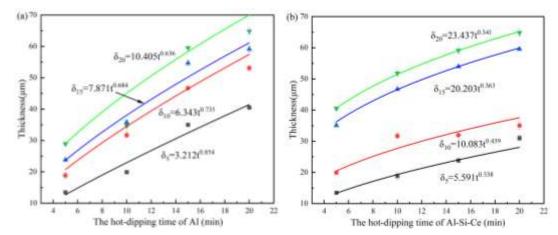


Figure 3. Growth kinetics curve of (Ti,Ce)(Al,Si)₃ alloy layer,(a) the variation curve of coating thickness with hot-dip plating Al time, (b) the variation curve of coating thickness with hot-dip plating Al-Si-Ce time

3.2. Microstructure Characterization of High Temperature Pre-oxidation Coating

It can be seen from Figure 4 that Ti65 alloy after pre-oxidation, the previous single (Ti,Ce)(Al,Si)₃ alloy layer changes into a multiphase layer structure. Under high temperature, Al atom further diffuses towards the substrate, and Ti atom also diffuses outwards, and the two atoms combine to form more dense TiAl alloy layer. The O atom reacts with the Al atom in the outermost (Ti,Ce)(Al,Si)₃ alloy layer to form dense Al₂O₃, and reacts with the Ce atom to form CeO₂. According to EDS data (Table 2), Ti₃Al layer, TiAl layer, TiAl₂ layer and CeO₂-rich Al₂O₃+(Ti,Ce)(Al,Si)₃ mixed phase layer are formed outward from the substrate successively. Ce atoms have diffused inward from the outermost (Ti,Ce)(Al,Si)₃ alloy layer into the newly formed Ti-Al phase layers.

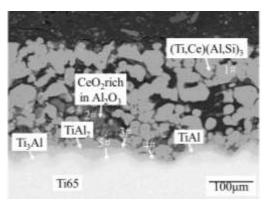


Figure 4. Microstructure of the coating after hot-dipping Al+15min(Al-10wt.%Si-1wt.%Ce) and preoxidation at 500° C/1h+850°C /2h

Table 2. EDS data (at.%) and phase components of each phase layer in Figure 4

Analysis region	О	A1	Si	Ti	Ce	Phase
1#	0	67.51	4.77	27.63	0.09	(Ti,Ce)(Al,Si)3
2#	55.55	41.76	0	2.57	0.12	Al ₂ O ₃ +CeO ₂
3#	0	64.24	2.13	33.57	0.06	$TiAl_2$
4#	0	41.43	2.28	55.90	0.39	TiAl
5#	0	20.90	0	78.88	0.22	Ti ₃ Al

3.3. High Temperature Oxidation Properties of Coatings

3.3.1. High Temperature Oxidation Coating Morphology

It can be seen from Figure 5(a) that the coating oxidized 23d at 800°C is still relatively uniform and dense. Combined with the XRD pattern in Figure 6 and the EDS data analysis in Table 3, it can be seen that after high temperature oxidation, not only TiAl phase but also Ti₅Si₃ phase is formed. That is, with the increase of high temperature oxidation time, Si atoms in (Ti,Ce)(Al,Si)₃ phase also diffuse to the substrate. However, due to the obstruction of Ti₃Al layer and TiAl layer, Si atoms can not be directly combined with the substrate. EDS data shows that after high temperature oxidation, Ti₃Al layer, TiAl layer, TiAl₃+Ti₅Si₃ mixed phase layer and CeO₂-rich Al₂O₃+(Ti,Ce)(Al,Si)₃ mixed phase layer are formed outwards from the substrate.

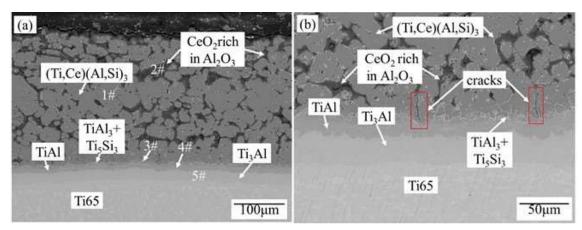


Figure 5. Microstructure of Ti65 alloy oxidized coating at high temperature after 15min Al+15min (al-10wt. %Si-1wt.%Ce), (a)23d, (b)14d

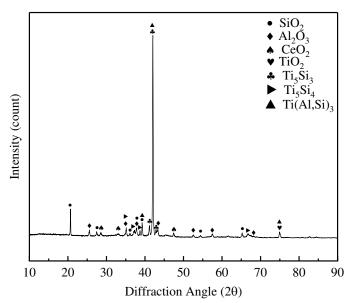


Figure 6. XRD pattern of Ti-Al-Si composite coating at 800°C for 23d oxidation

Table 3. EDS data (at.%) and phase components of each phase layer in Figure 5(a)

		1	1			0 ()
Analysis region	О	Al	Si	Ti	Ce	Phase
1#	0	66.37	3.78	29.74	0.11	(Ti,Ce)(Al,Si)3
2#	59.96	37.59	0	2.30	0.15	Al ₂ O ₃ +CeO ₂
3#	0	36.22	20.36	43.35	0.07	TiAl3+Ti5Si3
4#	0	37.55	3.93	58.28	0.24	TiAl
5#	0	21.18	0	78.61	0.21	Ti ₃ Al

The fitting formula of high temperature oxidation weight gain curve of the substrate at 800°C is as follows:

$$\Delta m' = 0.544 t^{0.932} \tag{2}$$

After hot-dip plating 15min Al+15min Al-10wt.%Si-1wt.%Ce on Ti65 alloy, the fitting formula of high temperature oxidation weight gain curve at 800°C is as follows:

$$\Delta m = 0.963 t^{0.328} \tag{3}$$

After hot plating Al on Ti-6Al-4V alloy, the fitting formula of high temperature oxidation weight gain curve at 800° C is as follows:

$$\Delta m = 0.582 t^{0.5} [26] \tag{4}$$

After hot impregnation of Ti-Al-Si on Ti-6Al-4V alloy, the fitting formula of high temperature oxidation weight gain curve at 800° C is as follows

$$\Delta m = 0.624 t^{0.5} [3,5] \tag{5}$$

where Δm is the oxidation weight gain of the coated sample (mg/ cm²), Δm ' is the oxidation weight gain of the substrate (mg/ cm²), and t is the oxidation time (h).

It can be seen from the high temperature oxidation weight gain curve (Figure 7) that the weight of uncoated Ti65 alloy and coated Ti65 alloy is increasing with the increase of high temperature oxidation time under the condition of high temperature oxidation at 800°C. This is because during the high temperature oxidation process, the O atom diffuses into the coating and generates CeO₂, Al₂O₃ and other oxides. Therefore, the weight of the alloy is increased. Comparing the two weight gain curves and formula (2), (3), (4) and (5), it is obvious that the weight gain rate of Ti65 substrate is the fastest and the weight gain is the most, while the sample of hot-dip plating Al+Al-Si-Ce has the slowest oxidation weight gain rate and the least weight increase. (2)/(3)=0.565t^{0.604}, which indicates the excellent property of the coating becomes more obvious with the increase of high temperature oxidation time. Therefore, the Ti-Al-Si coating with Ce element can greatly improve the high temperature oxidation resistance of Ti65 alloy.

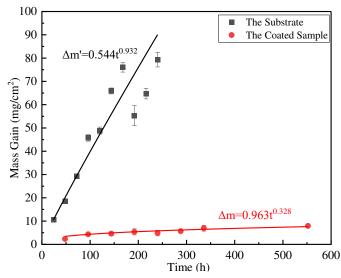


Figure 7. High temperature oxidation weight gain curves of substrate and sample

3.3.3. Crack

Figure 5 (b) shows the coating structure of Ti65 alloy after hot-dip plating 15minAl+15minal-10wt. %Si-1wt.%Ce for 14 days. Obvious longitudinal cracks appeare in the coating, but the cracks do not extend to the substrate, but stop at the TiAl3+Ti5Si3 mixed phase layer, indicating that the dense TiAl3+Ti5Si3 mixed phase layer, TiAl layer and Ti3Al layer hinder the crack propagation, thus blocking the path of O2 entering the substrate.

3.4.1. The Mechanism of Coating Formation

The Ti-Al binary system phase formed during hot-dip plating may be TiAl₃, TiAl₂, TiAl, and Ti₃Al and their Gibbs free energies are as follows [27]:

$$\Delta G_f (TiAl_3) = -40349.6 + 10.36525T(K)$$
(6)

$$\Delta G_f (TiAl_2) = -43858.4 + 11.02077T(K)$$
 (7)

$$\Delta G_f (TiAl) = -37445.1 + 16.79376T(K)$$
 (8)

$$\Delta G_f (Ti_3AI) = -29633.6 + 6.70801T(K)$$
(9)

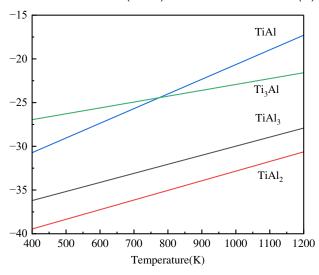


Figure 8. Gibbs free energy curves of some Ti-Al binary system phases with temperature

Although TiAl₂ has the lowest Gibbs free energy of formation, it can only nucleate at the TiAl interface [13]. The activation energies of intermetallic compounds in the Ti-Al system are Q_{Ti3Al} > Q_{TiAl2} > Q_{TiAl3} [5], so Ti65 alloy reacts with pure Al to preferentially form TiAl₃ layer, and then reacts with Al-10wt.%Si-1wt.%Ce alloy liquid. Because Si atoms tend to bond with Ti65 alloy substrate, Si atoms will enter the coating to replace Al atoms in TiAl₃ layer. Due to the high activity of Ce element, Ce atoms can replace Ti atoms [28,29], and finally form (Ti,Ce)(Al,Si)₃ composite phase layer, and the excess alloy liquid forms Al-Si layer. The reaction formulas are as follows:

$$Ti + 3Al \rightarrow TiAl_3$$
 (10)

Time

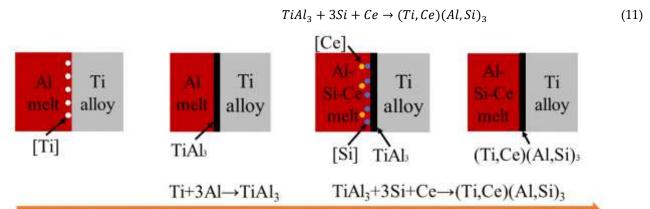


Figure 9. Schematic diagram of formation mechanism of alloy layer

3.4.2. High-Temperature Resistance Mechanism

After high temperature oxidation, the coating changes from a single (Ti,Ce)(Al,Si)₃ alloy layer into Ti3Al layer, TiAl layer, TiAl3+Ti5Si3 mixed phase layer, and CeO2-rich Al₂O₃+(Ti,Ce)(Al,Si)₃ mixed phase layer. This indicates that with the increase of high temperature oxidation time, all the excess Al-Si layers are consumed, forming (Ti,Ce)(Al,Si)3 phases, and Ti, Al, Si atoms are diffused with each other, forming Ti-Si binary system phases and multiple Ti-Al binary system phases. In the process of high temperature oxidation, O₂ reacts with Al atoms in (Ti,Ce)(Al,Si)₃ to form dense Al₂O₃. As rare earth elements have a strong affinity with O and S [30], Ce tends to combine with Al₂O₃ in the coating to form a CeO₂-rich Al₂O₃ phase. Ce element can not only promote the formation of continuous Al₂O₃ layer [31], but also the doping of CeO₂ helps to improve the uniformity and crystallinity of the composite coating, thus improving the high-temperature oxidation resistance of the Al₂O₃ coating [32]. The formed dense TiAl₃+Ti₅Si₃ mixed phase layer, TiAl layer and Ti3Al layer can prevent crack extension, because with the increase of Al content in TiAl compound and the increase in thermal expansion coefficient, making the newly formed TiAl and TiAl3 reduce the mismatch of thermal expansion coefficient between the coating and the base alloy [13]. At the same time, Ce element diffuse into TiAl3+Ti5Si3 mixed phase layer, TiAl layer and Ti3Al layer. The steel containing rare earth has low high Angle boundary and mean crystal orientation error. It can effectively prevent crack propagation [33], so cracks in the coating cannot spread to the TiAl3+TisSi3 mixed phase layer, TiAl layer and Ti3Al layer.

In the process of high temperature oxidation, a dense continuous CeO₂-rich Al₂O₃ oxide film is formed to prevent the diffusion of O atoms into the substrate. A TiAl₃+Ti₅Si₃ mixed phase layer, a TiAl layer and a Ti₃Al layer containing Ce are formed to prevent the propagation of cracks, thus preventing the diffusion of O atoms. The reaction equations are as follows:

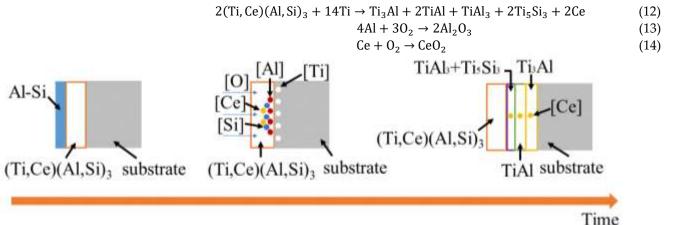


Figure 10. Schematic diagram of high-temperature oxidation mechanism

4. Conclusions

In this study, Ti-Al-Si composite coating containing Ce was prepared on Ti65 alloy by two-step hot-dip plating method. The addition of Ce element improved the high temperature oxidation resistance of Ti-Al-Si composite coating. The main conclusions are as follows:

- (1) The two-step hot-dip coating is mainly composed of (Ti,Ce)(Al,Si)₃ alloy layer. With the increase of the second step hot-dip plating time of Al-10wt.%Si-1wt.%Ce, the alloy layer gradually becomes dense.
- (2) In the process of pre-oxidation and high-temperature oxidation, the substrate reacts with (Ti,Ce)(Al,Si)₃ alloy layers to form multiple Ti-Al binary system phase layers, among which the dense TiAl₃+Ti₅Si₃ mixed phase layer, dense TiAl layer and dense Ti₃Al layer containing Ce element can prevent the crack propagation to the substrate.

(3) During the high-temperature oxidation process, the Ce element combines with Al₂O₃ in the form of CeO₂, which can promote the formation of continuous Al₂O₃ layer, and the high-temperature oxidation resistance of the Al₂O₃ layer containing CeO₂ is improved.

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Conflicts of Interest: The authors declare no conflict of interest.

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