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Article

Chemical Modeling of Constant Volume Combustion of Mixture of Methane and Hydrogen Used in Spark Ignition Otto Cycles

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Abstract: This paper develops a chemical model for a closed constant volume combustion of gaseous mixtures of methane and hydrogen. Since the combustion is strongly depending on temperature, pressure and fuel composition, they were chosen the actual corresponding thermodynamic systems using this kind of combustion, respectively the spark ignition reciprocating engines in order to evaluate the combustion parameters and exhaust flue gases composition. The actual cycles impose extra restrictive operational conditions through the engine volumetric compression ratio, the geometry of combustion volume, the mode to prepare the mixture of methane and hydrogen, the cooling system and the delivered power. The chemical model avoided the unknown influences in order to explain accurately the influence of hydrogen upon the constant volume combustion and flue gases composition. The model adopted simplifying hypotheses, respectively isentropic compression and expansion processes, closed constant volume combustion developed by two successive steps obeying to the energy and mass conservation laws and, flue gases exhaust described also by two steps, i.e. an isentropic expansion through the flow section of exhaust valves followed by a constant pressure stagnation (this succession is corresponding in fact to a throttling direct process). The chemical model supposed the homogeneous mixtures of gases with variable heat capacities function of temperatures, the Mendeleev - Clapeyron ideal gas state equation and, the variable chemical equilibrium constants for chosen chemical reactions. It was assumed that the flue gases chemistry is prevailing during the isentropic expansion and flue gases exhaust and has mainly a thermal ground. The chemical model allowed evaluation of flue gasses composition and of noxious emissions.

Keywords: closed constant volume combustion; Otto cycle; operational constraints; flue gases composition; harmful noxious

1. Introduction

The combustion of mixtures of fossil fuel with hydrogen showed firstly that harmful emissions, respectively CO and NO_x, are directly proportional to the hydrogen mole fraction in the mixed fuel, and are larger as larger the flame temperature, i.e. low excess oxygen and/or high preheating of the air before combustion. Therefore, the actual theoretical/experimental researches are developing means to reduce these emissions through new design of combustion zone (e.g. MILD combustion, combustion inside a porous material), new methods to mix different fossil fuels with hydrogen (e.g. dual fueling in diesel engines), new ideas regarding the pre and post combustion processing of working fluids (e.g. mixing blended fuel with hydrogen, post combustion catalytic treatment). The combustion of the fossil fuel and hydrogen mixtures in spark ignition engines had activated basic



researches regarding adaptation of actual spark ignition engines. More of them are tackling specific research targets.

[1] used a predictive model for lean hydrogen combustion in an Otto engine; the modeling put together a quasi-dimensional multi-zone combustion zone with a combustion model; knowing the unburned fuel it was evaluated the laminar flame speed and additionally an improved flame geometry was adopted.

[2] developed a methanol-syngas prototype engine which used gasoline and dissociated methanol gas as dual fuel, and they was found that the presence of hydrogen obtained by dissociation improved the efficiency, increased the maximum pressure on the cycle and the released heat in the recycling exhaust heat.

[3] simulated by AVL Boost the combustion of pure H₂ and compared with combustion of H₂O micro-emulsion fuel and pure gasoline fuel organized inside a single-cylinder stationary spark ignition engine; this simulation comparatively revealed the characteristics regarding the pollutants, energy efficiency for all three chosen fuels.

[4] experimentally explored the combustion of hydrous ethanol/hydrogen in spark ignition engine; the classical injection of the hydrous ethanol during the intake process was combined with the hydrogen injection into the cylinder during the compression stroke; hydrogen addition increases the maximum pressure and temperature on the cycle, reduce CO and HC emissions, but rises NO_x emissions.

[5] based on AVL Boost program and on some selected experiments concluded that changing the liquid fuel with the gaseous one it is assuring complete combustion, reductions of HC and CO emissions, modification of the brake power, improved efficiency with lean mixtures, but increase of NO_x.

[6] searched the effect of ethanol-gasoline-hydrogen in a lean-burn SI engine for an optimized blend ethanol-gasoline combined with hydrogen; it was found that hydrogen presence shows a significant improvement of brake power and thermal efficiency and improves the combustion and reduces hydrocarbon emissions.

[7] studied the impact of dual fuel mixtures; they examined the effects of combustion of mixed secondary fuels (hydrogen, methane, butane, propane) with primary fuels (gasoline, iso-octane, benzene, toluene, hexane, ethanol, methanol) on the theoretical performance of a SI engine; the results showed that the ratios of hydrogen, methane, butane, propane considerably modifies the energy and exergy performances of the engine.

[8] researched optimal fuel compositions and the control parameters of the spark ignition engine fueled with a mixture of syngas-biogas-hydrogen and it was setting up a flexible electronic control unit for the engine included in a solar-biomass hybrid renewable energy system, controlling the relation engine load - fuel flow – air/fuel mixture flow.

[9] investigated the effects of hydrogen addition on engine performance using spark ignition gasoline with a compression ratio of 15; the best hydrogen fraction was found to be a balance between improvement in burning efficiency and increase in heat loss.

Because all actual researches following the use of hydrogen in reciprocating engines were organized for limited experiments with imposed operational restrictive conditions such as constant revolution per minute, constant power, specific cooling, accurate instrumentation, the research results are at this time somewhat poor and sometimes contradictory. This paper develops a chemical model for a closed constant volume combustion of gaseous mixtures of methane and hydrogen used in Otto cycles for adopted simplifying hypotheses, respectively actual volumetric compression ratio, isentropic compression and expansion processes, closed constant volume combustion developed by two successive steps obeying to the energy and mass conservation laws and, flue gases exhaust described also by two steps, i.e. an isentropic expansion through the flow section of exhaust valves followed by a constant pressure stagnation (this succession is corresponding in fact to a throttling direct process). These simulation restrictive conditions allowed the generalized evaluation of all state parameters along the cycle, i.e. temperatures, pressures, and working fluid composition/pollutants.

2. The approach of the closed constant volume combustion in Otto engine cycles

2.1. The assumptions for the Otto cycle, see Figure 1

Let's suppose the general basic Otto cycle interrelating with the environment by heat, mass and power transfers, see Figure 1. The engine cycle used to describe truthfully the constant volume combustion of a homogeneous mixture of CH₄ and H₂ and to evaluate the flue gases composition and harmful emissions, is consisting of below processes:

- 1 – 2 isentropic compression of a mixture CH₄/H₂/air prepared by a carburetor technique;
- 2 – 3 closed constant volume combustion;
- 3 – 4 isentropic expansion of flue gases and,
- 4 – 5 exhaust of flue gases, assimilated to a throttling process (succession of 4 – 4t isentropic expansion through flow section of exhaust valves and 4t – 5 constant pressure stagnation);
- 5 – 1 final constant pressure exhaust of flue gases.

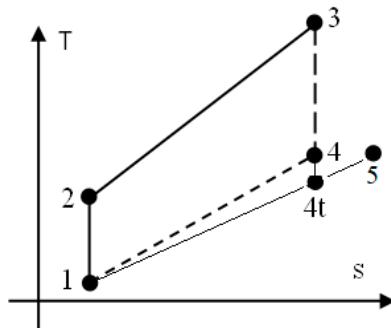


Figure 1. The basic Otto cycle.

2.2. The isentropic compression of a mixture of CH₄/H₂/air, prepared by a carburetor, see Figure 2

The assumptions were related to 1 kmole fuel with different mole composition and mixed with air having different values of excess oxygen:

- they were considered 1 kmole fuel with x kmole H₂ and (1 – x) kmole CH₄;
- the needed air for combustion was evaluated by the imposed excess oxygen, exo, respectively (1 + exo)(2 – 1.5x) kmole O₂ and 3.7619(1 + exo)(2 – 1.5x) kmole N₂;
- the imposed x and exo, x = (0, 0.25, 0.5, 0.75); exo = (0, 0.25, 0.5, 0.75, 1, 1.5, 2);
- the volumetric compression ratio, $\pi_{12} = V_2/V_1 = 10$.

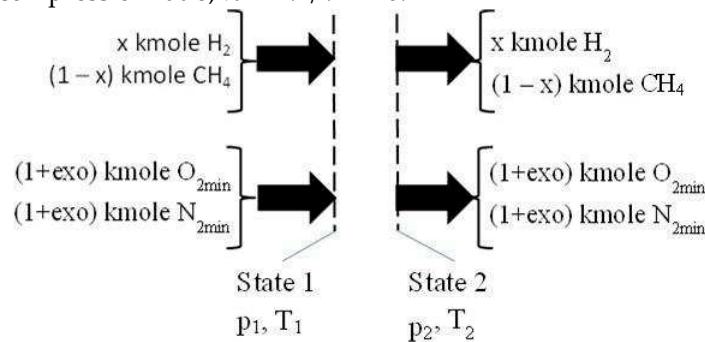


Figure 2. The isentropic compression 1 – 2. O_{2min} and N_{2min} correspond to stoichiometric combustion.

The parameters for the state 2 were evaluated through equations 1, 2 and 3

isentropic exponent

$$k_{12} = \frac{r_{CH4} \cdot (h_{2,CH4} - h_{1,CH4}) + r_{H2} \cdot (h_{2,H2} - h_{1,H2}) + r_{O2} \cdot (h_{2,O2} - h_{1,O2}) + r_{N2} \cdot (h_{2,N2} - h_{1,N2})}{r_{CH4} \cdot (u_{2,CH4} - u_{1,CH4}) + r_{H2} \cdot (u_{2,H2} - u_{1,H2}) + r_{O2} \cdot (u_{2,O2} - u_{1,O2}) + r_{N2} \cdot (u_{2,N2} - u_{1,N2})} \quad (1)$$

isentropic compression ending temperatures, T_2 and T_1 :

$$T_2 = T_1 \cdot \pi_{12}^{k_{12}-1}, \quad T_1 = 298 \text{ K} \quad (2)$$

isentropic compression ending pressures, p_2 and p_1 :

$$p_2 = p_1 \cdot \pi_{12}^{k_{12}}, \quad p_1 = 1 \text{ bar} \quad (3)$$

where $r, \hbar, \mathbf{u} = \hbar - R \cdot (T - T_1)$ are the mole composition of wholly mixture $\text{CH}_4/\text{H}_2/\text{air}$ [kmole/kmole], and the relative physical enthalpies and internal energies [kJ/kmole] that were interpolated by 5th order polynomials function of temperatures, see Table 1, $R = 8.3145 \text{ kJ/kmole.K}$ is the universal constant of ideal gas;

Table 1. Polynomials of physical enthalpy of chemical species participating in the constant volume combustion.

	$\hbar = h - h_{298}$, h is the relative physical enthalpy of gaseous chemical species, $h_{298} = h(298\text{K})$, [kJ/kmole]					
	$\hbar = c_0 + c_1 \cdot T + c_2 \cdot T^2 + c_3 \cdot T^3 + c_4 \cdot T^4 + c_5 \cdot T^5$ [kJ/kmole], for $298 \text{ K} \leq T \leq 5200 \text{ K}$					
	\hbar_{CH_4} for $298 \text{ K} \leq T \leq 1500 \text{ K}$					
	c0	c1	c2	c3	c4	c5
\hbar_{CO_2}	-10295.73122	+29.18222805	+0.01976308361	-0.000006164587807	+9.718495272E-10	-5.984216612E-14
$\hbar_{\text{H}_2\text{O}}$	-7820.523838	+24.56140839	+0.006095193576	-0.000001576697822	+2.155196747E-10	-1.189998347E-14
\hbar_{O_2}	-8310.810038	+26.11440421	+0.006486554878	-0.000001879231979	+3.125634850E-10	-2.048055699E-14
\hbar_{N_2}	-7820.523838	+24.56140839	+0.006095193576	-0.000001576607822	+2.155196747E-10	-1.189998347E-14
\hbar_{CO}	-77895.91761	24.23615845	+0.006957194766	-0.000002000452191	+3.006868933E-10	-1.79961500E-14
\hbar_{H_2}	-8523.602676	+28.58874847	-0.0002520099380	+0.000001072813673	-2.405013187E-10	+1.755178738E-14
\hbar_{OH}	-8479.625232	+28.19400310	+0.0006776079283	+7.234592056E-7	-1.956721062E-10	+1.522710549E-14
\hbar_{O}	-6430.497036	+21.74597694	-0.0006119709133	+1.784585133E-7	-2.341992360E-11	+1.452421376E-15
\hbar_{H}	-6191.086828	+20.77251773	+0.00001104663432	-4.160102635E-9	+7.310338752E-13	+4.846517103E-17
\hbar_{N}	-6167.889202	+20.64973571	+0.0001887590700	-9.970607532E-8	+1.832411454E-11	-2.236164634E-16
\hbar_{NO}	-8047.826762	+25.12271666	+0.006900181303	-0.000002036851415	+3.093429512E-10	-1.856273997E-14
\hbar_{NO_2}	10209.25264	+29.28985710	+0.01839187998	-0.000006044547788	+9.772891559E-10	-6.122580161E-14

h_{CH_4}	-8143.263322	+21.07695175	+0.01570851794	+0.00002221853471	-1.604634302E-8	+3.367601937E-12
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2.3. The closed constant volume combustion, see Figure 3

The closed constant volume combustion was designed in two successive steps. The first step gives the energy input for the second one, respectively the higher heating value, HHV, and the physical enthalpy of intaking chemical species, see Figure 3. This first step develops a fictitious combustion without dissociation, and performed at the constant pressure p_2 and the constant temperature T_2 . The chemical reactions pertaining to this first step are those of direct oxidation of the fuel:

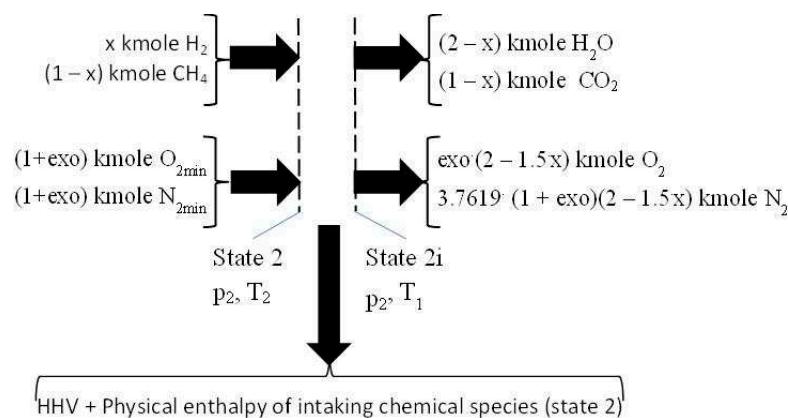
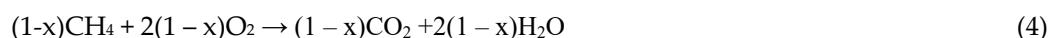


Figure 3. The first step of combustion giving the heat input for the second one. HHV is higher heating value, $O_{2\min}$ and $N_{2\min}$ values for stoichiometric combustion.

The HHV was evaluated by the enthalpies of formation of involved chemical species, h_{f0} , see Table 2.

$$HHV = (1-x) \cdot h_{f0,CH_4} - (1-x) \cdot h_{f0,CO_2} - (2-x) \cdot h_{f0,H_2O} \quad [\text{kJ}/\text{kmole fuel}] \quad (6)$$

The physical enthalpy of intaking chemical species were given by a fictitious constant pressure cooling of the intaking gases.

$$ph = xh_{2,H_2} + (1-x)h_{2,CH_4} + (1+exo)(2-1.5x)h_{2,O_2} + 3.6719(1+exo)(2-1.5x)h_{2,N_2} \quad [\text{kJ}/\text{kmole fuel}] \quad (7)$$

where the physical enthalpies of intaking gases correspond to T_2 .

Table 2. Enthalpy of formation of chemical species involved in the constant volume combustion.

	CH ₄	CO ₂	H ₂ O	H ₂	O ₂	N ₂	CO	H	O	N	OH	NO	NO ₂
h_{f0} [kJ/kmole]	-74873	-393522	-285830	0	0	0	-110527	+217999	+249170+472680	+38897	+90921	+33100	

The second step of constant volume combustion, 2i – 3, is conceived as a heating with dissociation of flue gases and consuming the heat released in the first step, i.e. HHV and physical enthalpy of intaking chemical species (state 2), see Figure 4. The inlet chemical species have the temperature T_2 and pressure p_2 . The outlet temperature and pressure, T_3 and p_3 , were computed using the mass and energy conservation law and the Mendeleev – Clapeyron state equation.

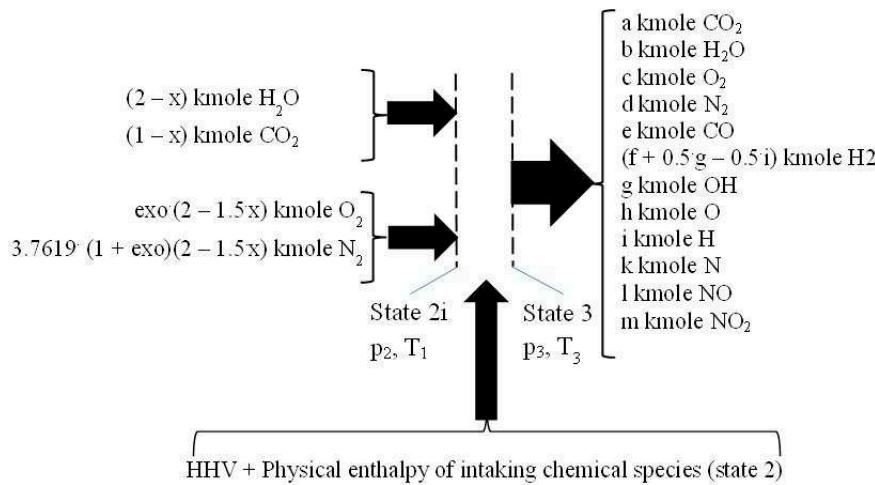


Figure 4. The second step of combustion consuming the heat released in the first one.

The second step of combustion has counted the following chemical reactions, of dissociation and of recombination:



The mass balance equations are stated by:

- four equations related to the balance of kmoles number of major chemical species, CO₂ and H₂O and O₂ and N₂,
- eight equations interrelating all chemical species through chemical equilibrium relations and,
- one mass balance for the whole process 2i – 3.

The balance of kmoles number of major chemical species:

$$a = (1 - x) - e \quad [\text{kmole}] \quad (16)$$

$$b = (2 - x) - g - f \quad [\text{kmole}] \quad (17)$$

$$c = \text{exo}(2 - 1.5x) + 0.5e + 0.5f - 0.5h - 0.5l - m \quad [\text{kmole}] \quad (18)$$

$$d = 3.7619(1 + \text{exo})(2 - 1.5x) - 0.5k - 0.5l - 0.5m \quad [\text{kmole}] \quad (19)$$

For a chemical reaction $v_A \text{ A} + v_B \text{ B} \rightarrow v_C \text{ C} + v_D \text{ D}$ the equilibrium constant is $K = \frac{y_C^{v_C} \cdot y_D^{v_D}}{y_A^{v_A} \cdot y_B^{v_B}} \cdot P^{v_C + v_D - v_A - v_B}$ where $P = \frac{\text{reaction pressure}}{\text{normal pressure (0.1 MPa)}}$

The natural logarithms of chemical equilibrium constants, for P = 1, were interpolated by polynomials of 7th order, see Table 3, using values from [10] (p.773).

Table 3. Polynomials of natural logarithms of chemical equilibrium constants.

$\ln(K) = a_0 + a_1 \cdot T + a_2 \cdot T^2 + a_3 \cdot T^3 + a_4 \cdot T^4 + a_5 \cdot T^5 + a_6 \cdot T^6 + a_7 \cdot T^7$ for $500 \text{ K} \leq T \leq 3400 \text{ K}, P = 1$								
	ln(K1)	ln(K2)	ln(K3)	ln(K4)	ln(K5)	ln(K6)	ln(K7)	ln(K8)
a	+9.232631128	+1.136595616	+7.643759089	+8.166909930	+1.543529198E-	+7.070194312	+2.948541681	+1.070424351
7	9E-22	E-21	E-22	E-22	21	E-22	E-22	E-22
a	-	-	-	-	-	-	-	-
6	1.469466515E-	1.77695367E-	1.219664280E-	1.298258556E-	-2.454651317E-17	1.125159755E-	4.688934093E-	1.700672496E-
17	17	17	17	17		17	18	18
a	+9.908436246	+1.173485363	+8.256023802	+8.743314794	+1.653766691E-	+7.586041432	+3.159040657	+1.144079969
5	E-14	E-13	E-14	E-14	13	E-14	E-14	E-14
a	-	-	-	-	-	-	-	-
4	3.683648801E-	4.258599458	3.081726001E-	3.246702476E-	-6.143247539E-10	2.2820123437	1.173468673E-	4.240167126E-
10	E-10		10	10		E-10	10	11
a	+8.221477783	+9.246236875	+6.907189777	+7.238919496	+0.000001370114	+6.294886934	+2.616956936	+9.424074844
3	E-7	E-7	E-7	E-7	593	E-7	E-7	E-8
a	-	-	-	-	-	-	-	-
2	0.0011215185	0.0012233842	0.00094671770	0.00098688815	-0.001868142287	0.00085923084	0.00035673514	0.00012786331
45	39	73	09		02	18	83	
a	+0.905689424	+0.957598285		+0.7698066320	+0.7977691487	+1.509735517	+0.695832294	+0.2880804502
1	6	5						+0.1027180385
a	-370.3187386	-385.6441663	-323.0833734	-330.5159532	-638.9027541	-289.1945752	-121.5994613	-59.57996704
0								

The equations interrelating all chemical species through chemical equilibrium relations:

$$\exp(\ln(K1)) = \frac{e^2 \cdot c}{a^2} \cdot P \quad \text{for eq. 8} \quad (20)$$

$$\exp(\ln(K2)) = \frac{(f + 0.5g - 0.5i) \cdot g^2}{b^2} \cdot P \quad \text{for eq. 9} \quad (21)$$

$$\exp(\ln(K3)) = \frac{(f + 0.5g - 0.5i)^2 \cdot c}{b^2} \cdot P \quad \text{for eq. 10} \quad (22)$$

$$\exp(\ln(K4)) = \frac{h^2}{c} \cdot P \quad \text{for eq. 11} \quad (23)$$

$$\exp(\ln(K5)) = \frac{k^2}{d} \cdot P \quad \text{for eq. 12} \quad (24)$$

$$\exp(\ln(K6)) = \frac{i^2}{(f + 0.5g - 0.5i)} \cdot P \quad \text{for eq. 13} \quad (25)$$

$$\exp(\ln(K7)) = \frac{i^2}{d \cdot c} \quad \text{for eq. 14} \quad (26)$$

$$\exp(\ln K8)) = \frac{m^2}{d \cdot c^2} \cdot \frac{1}{P} \quad \text{for eq. 15} \quad (27)$$

The mass balance for the whole process 2i - 3

$$2 \cdot x + 16 \cdot (1 - x) + 32 \cdot (1 + \text{exo}) \cdot (2 - 1.5 \cdot x) + 120.3808 \cdot (1 + \text{exo}) \cdot (2 - 1.5 \cdot x) = 44 \cdot a + 16 \cdot b + 32 \cdot c + 28 \cdot d + 28 \cdot e + 2 \cdot (f + 0.5 \cdot g - 0.5 \cdot i) + 17 \cdot h + 14 \cdot k + 30 \cdot l + 46 \cdot m \quad (28)$$

The energy balance equation:

$$HHV + ph = [a \cdot (u_{f0,CO2} + u_{3,CO2}) - (1 - x) \cdot u_{f0,CO2}] + [b \cdot (u_{f0,H2O} + u_{3,H2O}) - (2 - x) \cdot u_{f0,H2O}] + c \cdot u_{3,O2} + d \cdot u_{3,N2} + e \cdot (u_{f0,CO} + u_{3,CO}) + (f + 0.5 \cdot g - 0.5 \cdot i) \cdot u_{3,H2} + g \cdot (u_{f0,OH} + u_{3,OH}) + h \cdot (u_{f0,O} + u_{3,O}) + i \cdot (u_{f0,H} + u_{3,H}) + k \cdot (u_{f0,N} + u_{3,N}) + l \cdot (u_{f0,NO} + u_{3,NO}) + m \cdot (u_{f0,NO2} + u_{3,NO2}) \quad (29)$$

The Mendeleev - Clapeyron state equation for gaseous mixtures:

$$p \cdot V = \left(\sum n_i \right) \cdot R \cdot T \xrightarrow{V=const.} \quad (30)$$

$$\frac{p_3}{p_2} = \frac{T_3}{T_2} \cdot \frac{(\sum n_i)_3}{(\sum n_i)_2} = \frac{T_3}{T_2} \cdot \frac{a + b + c + d + e + (f + 0.5 \cdot g - 0.5 \cdot i) + g + h + k + l + m}{x + (1 - x) + (1 + \text{exo}) \cdot (2 - 1.5 \cdot x) + 3.8619 \cdot (1 + \text{exo}) \cdot (2 - 1.5 \cdot x)}$$

2.4. The isentropic expansion 3 - 4, see Figure 5

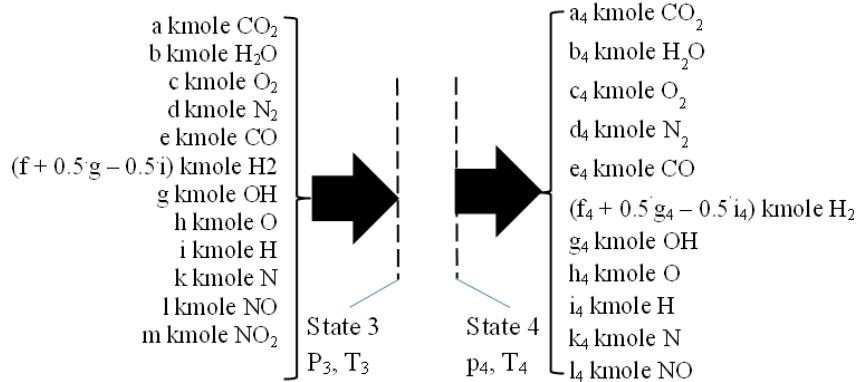


Figure 5. The isentropic reactive expansion 3-4.

During the isentropic expansion, the chemistry of flue gases is also prevailing due to high temperatures and pressures, and therefore the composition of flue gases is changing by variable temperature and pressure. This process was solved defining a proper isentropic exponent quantifying the variable chemical composition of flue gases. The equations defining a reactive isentropic expansion allowed the evaluation of parameters of state 4. The parameters for the state 4 were evaluated through equations 30, 31 and 32, evaluating the flue gases composition through two computational loops regarding relationship composition – temperature and pressure, similar as in the case of constant volume combustion.

$$\text{isentropic exponent} \quad k_{34} = \frac{\Delta H_{34}}{\Delta U_{34}} \quad \text{where} \quad (31)$$

$$\Delta H34 = [a4 \cdot (h_{f0,CO2} + h_{4,CO2}) - a \cdot (h_{f0,CO2} + h_{3,CO2})] + [b4 \cdot (h_{f0,H2O} + h_{4,H2O}) - b \cdot (h_{f0,H2O} + h_{3,H2O})] + [c4 \cdot h_{4,O2} - c \cdot h_{3,O2}] + [d4 \cdot h_{4,N2} - d \cdot h_{3,N2}] + [e4 \cdot (h_{f0,CO} + h_{4,CO}) - e \cdot (h_{f0,CO} + h_{3,CO})]$$

$$+ [(f4 + 0.5 \cdot g4 - 0.5 \cdot i4) \cdot h_{4,H2} - (f + 0.5 \cdot g - 0.5 \cdot i) \cdot h_{3,H2}] + [g4 \cdot (h_{f0,OH} + h_{4,OH}) - g \cdot (h_{f0,OH} + h_{3,OH})] + [h4 \cdot (h_{f0,CO2} + h_{4,O}) - h \cdot (h_{f0,O} + h_{3,O})] + [i4 \cdot (h_{f0,H} + h_{4,H}) - i \cdot (h_{f0,H} + h_{3,H})] + [k4 \cdot (h_{f0,N} + h_{4,N}) - k \cdot (h_{f0,N} + h_{3,N})] + [l4 \cdot (h_{f0,NO} + h_{4,NO}) - l \cdot (h_{f0,NO} + h_{3,NO})] + [m4 \cdot (h_{f0,NO2} + h_{4,NO2}) - m \cdot (h_{f0,NO2} + h_{3,NO2})]$$

$$\Delta U34 = [a4 \cdot (u_{f0,CO2} + u_{4,CO2}) - a \cdot (u_{f0,CO2} + u_{3,CO2})] + [b4 \cdot (u_{f0,H2O} + u_{4,H2O}) - b \cdot (u_{f0,H2O} + u_{3,H2O})] + [c4 \cdot u_{4,O2} - c \cdot u_{3,O2}] + [d4 \cdot u_{4,N2} - d \cdot u_{3,N2}] + [e4 \cdot (u_{f0,CO} + u_{4,CO}) - e \cdot (u_{f0,CO} + u_{3,CO})] + [(f4 + 0.5 \cdot g4 - 0.5 \cdot i4) \cdot u_{4,H2} - (f + 0.5 \cdot g - 0.5 \cdot i) \cdot u_{3,H2}] + [g4 \cdot (u_{f0,OH} + u_{4,OH}) - g \cdot (u_{f0,OH} + u_{3,OH})] + [h4 \cdot (u_{f0,CO2} + u_{4,O}) - h \cdot (u_{f0,O} + u_{3,O})] + [i4 \cdot (u_{f0,H} + u_{4,H}) - i \cdot (u_{f0,H} + u_{3,H})] + [k4 \cdot (u_{f0,N} + u_{4,N}) - k \cdot (u_{f0,N} + u_{3,N})] + [l4 \cdot (u_{f0,NO} + u_{4,NO}) - l \cdot (u_{f0,NO} + u_{3,NO})] + [m4 \cdot (u_{f0,NO2} + u_{4,NO2}) - m \cdot (u_{f0,NO2} + u_{3,NO2})]$$

isentropic compression ending temperatures, T_3 and T_4 : $T_4 = T_3 \cdot \left(\frac{1}{\pi_{12}}\right)^{k34-1}$, T_3 previously computed (32)

isentropic compression ending pressures, p_3 and p_4 : $p_4 = p_3 \cdot \left(\frac{1}{\pi_{12}}\right)^{k34}$, p_3 previously computed (33)

2.5. The flue exhaust process 4 - 5, see Figure 6

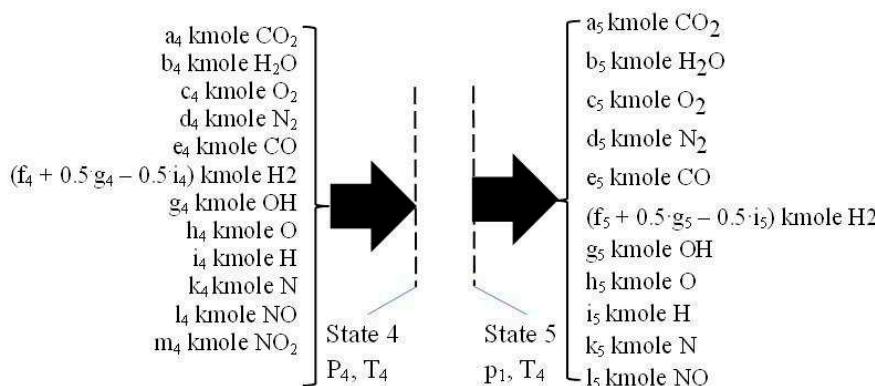


Figure 6. The flue gases exhaust process 4 - 5.

This process is considered as a throttling one, described by constant enthalpy. They were used the ideal gas model and thus they were supposed that the temperature is almost constant because the composition variation of flue gases is not significant.

This simplified approach allowed the evaluation of final composition of the state 5 only on the basis of mass conservation law given by known values of chemical equilibrium constants for $T_5 = T_4$ and $p_5 = 1$ bar.

3. The numerical solving and results

The isentropic compression and expansion can be solved through a proper mathematical procedure, the most simple is an iterative trial and error method. The compression is very easily to be solved because the cylinder intaking fluid has a constant composition. During expansion, the composition is changing but, we know the state 3 composition previously computed and the state 4 composition evaluated through eqs. 31 to 33.

The closed constant volume combustion must use two iterative loops, one for flue gases composition, and the next one for final temperature and pressure. The first step consider a constant

volume heating of flue gases without dissociation in order to preliminary determine the temperature and pressure of state 3 through energy conservation law and Mendeleev – Clapeyron state relation. These parameters permitted the evaluation of a first computational composition of flue gases, in state 3, by mass conservation law. This new flue gases composition, in state 3, allowed the re-evaluation of temperature and pressure of state 3 by energy conservation law and Mendeleev-Clapeyron state equation. The iterative loops end when the difference between two successive steps submit to an imposed error of computed parameters.

The numerical results emphasized the behavior of variations of pollutants emissions (CO and NOx) in a good qualitative agreement with the experiments performed elsewhere.

The values of CO mole fractions in flue gases are included in Figure 7 and Table 4. The maximum mass ppm, found in state 5, is around 170 ppm [mg CO/kg flue gases], for $\text{exo} = 0$ and $x = 0.75$. The throttling, decreasing the pressure, gives CO emissions almost twice larger for a stoichiometric combustion ($\text{exo} = 0$). All numerical results showed that the presence of hydrogen increases very slightly the CO emissions.

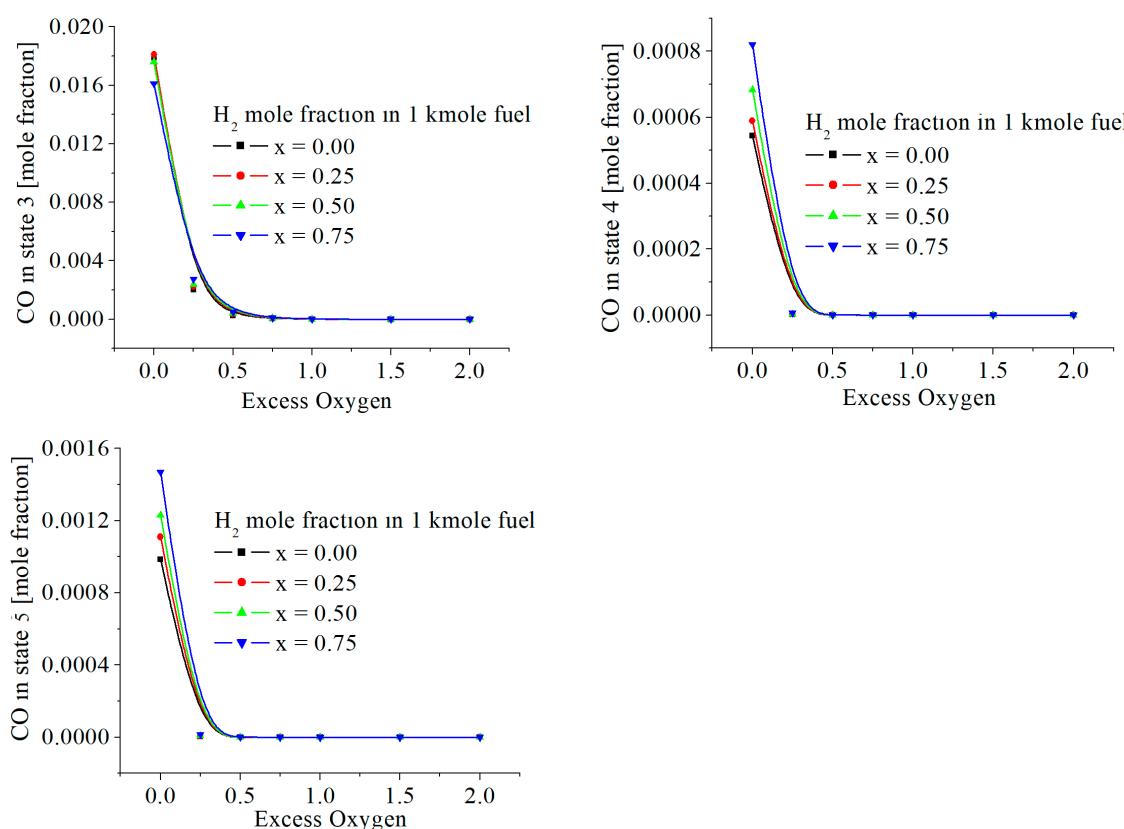


Figure 7. The CO mole fractions in states 3, 4 and 5, function of excess of oxygen and x .

Table 4. CO mole fractions in states 3, 4 and 5.

	$x = 0.00$	$x = 0.25$	$x = 0.50$	$x = 0.75$	$x = 0.00$	$x = 0.25$	$x = 0.50$	$x = 0.75$	$x = 0.00$	$x = 0.25$	$x = 0.50$	$x = 0.75$
excess oxygen	State 3				State 4				State 5			
0	0.0179	0.0181	0.0176	0.0161	5.44E-4	5.9E-4	6.84E-4	8.21E-4	9.86E-4	0.00111	0.00123	0.00147
0.25	0.00201	0.00224	0.0024	0.00273	2.06E-6	2.66E-6	3.52E-6	6.4E-6	4.96E-6	6.66E-6	8.29E-6	1.52E-5
0.50	2.73E-4	3.6E-4	4.34E-4	4.92E-4	5.7E-8	8.01E-8	1.24E-7	2.82E-7	1.32E-7	1.78E-7	2.78E-7	6.43E-7
0.75	5.83E-5	6.91E-5	8.93E-5	1.05E-4	1.85E-9	2.56E-9	4.19E-9	1.06E-8	3.91E-9	5.46E-9	8.94E-9	2.28E-8
1.00	1.09E-5	1.34E-5	1.73E-5	2.26E-5	5.49E-11	7.94E-11	1.34E-10	3.98E-10	1.12E-10	1.63E-10	2.77E-10	8.27E-10
1.50	4.53E-7	6.48E-7	7.34E-7	1.06E-6	4.94E-14	8.53E-14	1.35E-13	4.6E-13	9.5E-14	1.65E-13	1.01E-13	9.04E-13
2.00	1.83E-8	2.58E-8	3.45E-8	5.35E-8	2.73E-17	5.97E-17	1.05E-16	6.99E-15	5.33E-17	1.1E-16	1.94E-16	8.39E-16

The values of NO are in Figure 8 and Table 5. The numerical results showed two influences of hydrogen. The first influence gives NO mole fractions proportional to the mole ratio of hydrogen in fuel and larger than in the case of combustion without hydrogen. The second influence highlighted a maximum of these emissions for $0 < \text{exo} < 0.5$.

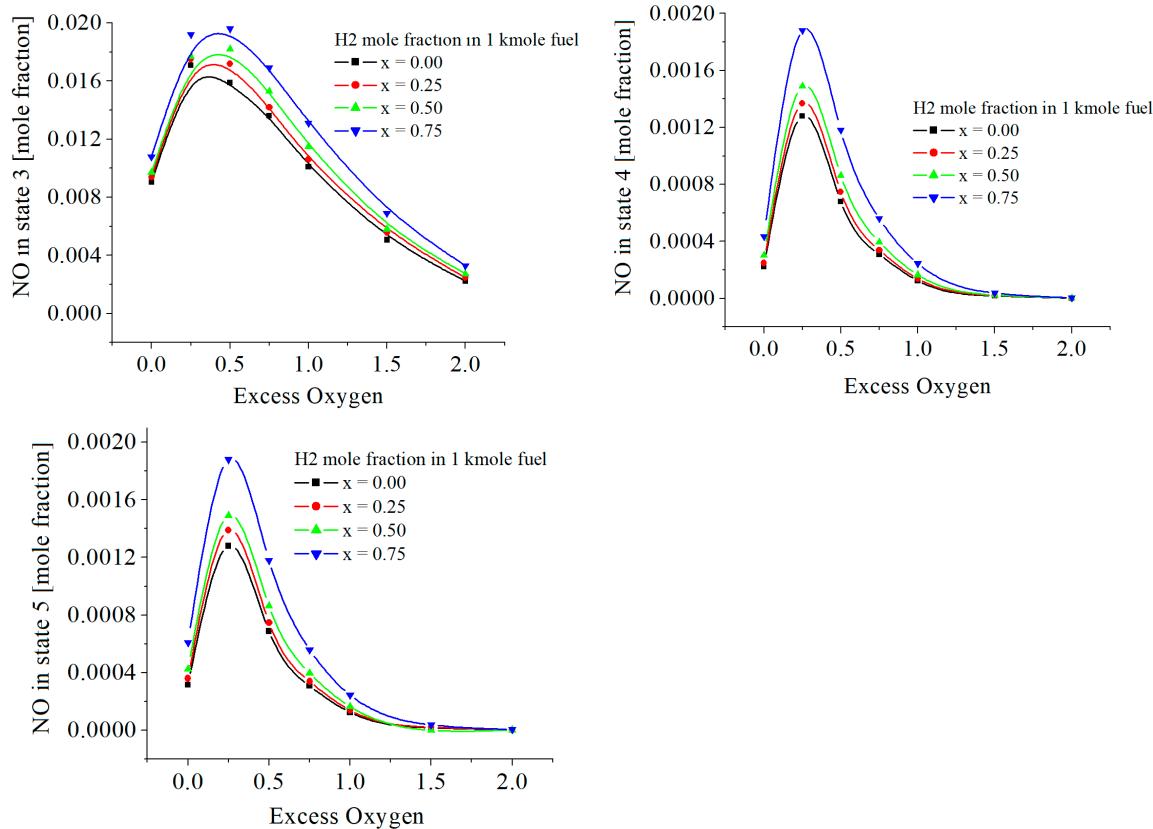


Figure 8. The NO mole fractions in states 3, 4 and 5, function of excess of oxygen and x.

Table 5. NO mole fractions in states 3, 4 and 5.

x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75
exces											
s											
oxyg											
en											
0	9.04E-03	9.37E-03	9.71E-03	1.08E-02	1.16E-07	2.48E-04	3.03E-04	4.35E-04	3.15E-04	3.61E-04	4.25E-04
0.25	1.71E-02	1.75E-02	1.77E-02	1.92E-02	1.62E-05	1.37E-03	1.49E-03	1.88E-03	1.28E-03	1.39E-03	1.49E-03
											6.09E-04

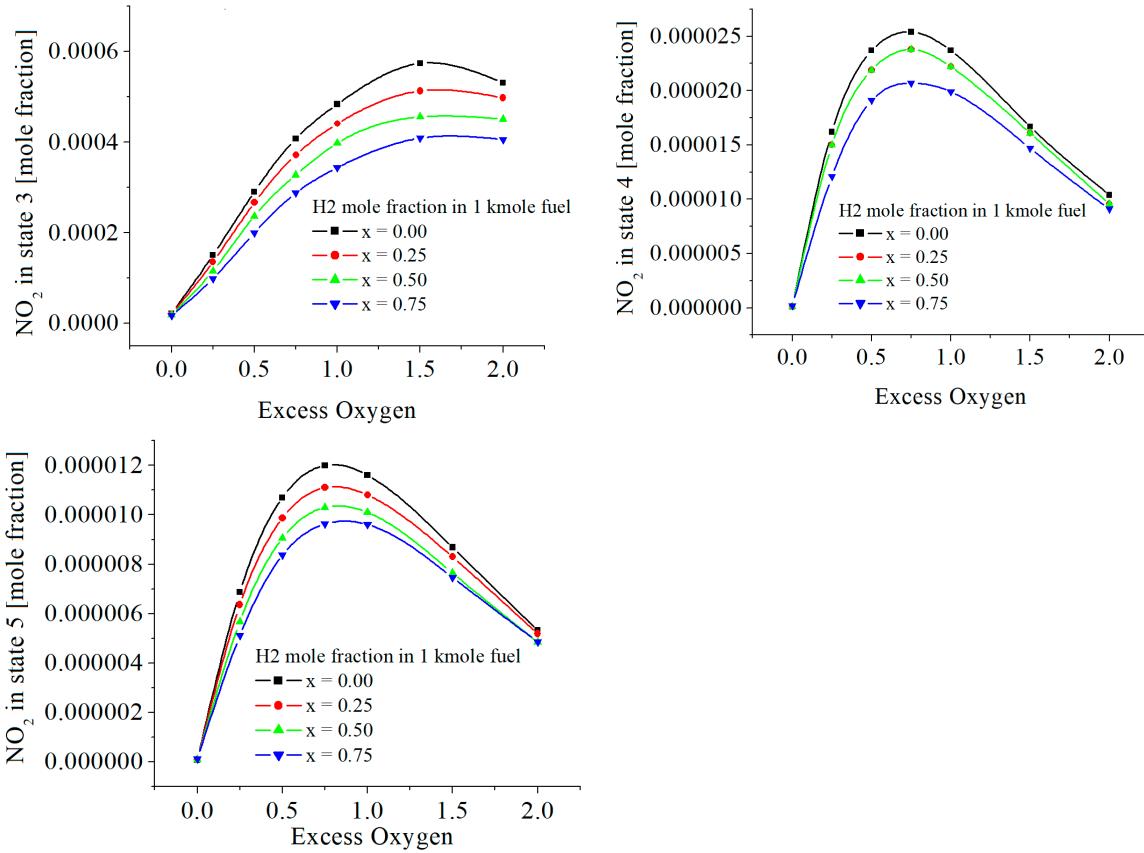
Table 5. NO mole fractions in states 3, 4 and 5 (continuation).

x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75
exces											
s											
oxyg											
en											
0.50	1.59E-02	1.72E-02	1.82E-02	1.96E-02	2.37E-05	7.48E-04	8.63E-04	1.18E-03	6.88E-04	7.48E-04	8.63E-04
0.75	1.36E-02	1.42E-02	1.53E-02	1.69E-02	2.54E-05	3.39E-04	3.97E-04	5.60E-04	3.08E-04	3.39E-04	3.98E-04
1.00	1.01E-02	1.06E-02	1.15E-02	1.31E-02	2.37E-05	1.39E-04	1.65E-04	2.45E-04	1.25E-04	1.39E-04	1.65E-04
1.50	5.07E-03	5.57E-03	5.81E-03	6.88E-03	1.67E-05	2.14E-05	2.48E-05	3.85E-05	1.81E-05	2.14E-05	1.43E-15
2.00	2.23E-03	2.47E-03	2.73E-03	3.29E-03	1.04E-05	2.61E-06	3.13E-06	5.23E-06	2.09E-06	2.61E-06	3.13E-06
											5.23E-06

The values of NO_2 are given in Table 6 and Figure 9. NO_2 emissions are smaller as larger the hydrogen mole fraction in the fuel. This is contrasting to the emissions of NO. Similarly, NO_2 emissions have a maximum depending on the oxygen excess.

Table 6. NO₂ mole fractions in states 3, 4 and 5.

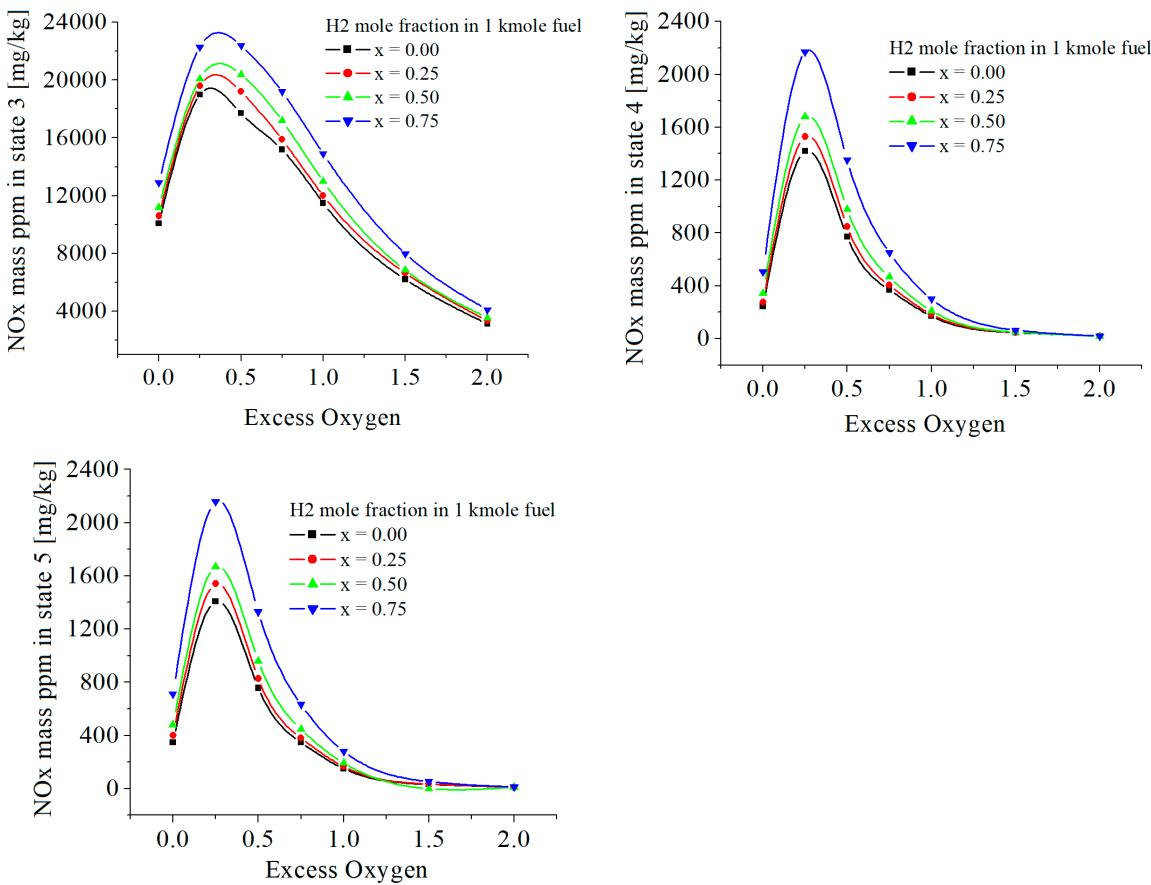
	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75
exc												
ess												
oxy												
gen												
					State 3				State 4			
0	2.21E-05	2.09E-05	1.93E-05	1.78E-05	1.16E-07	1.21E-07	1.21E-07	1.79E-07	9.01E-08	9.73E-08	1.06E-07	1.38E-07
0.2	1.51E-04	1.36E-04	1.16E-04	9.89E-05	1.62E-05	1.50E-05	1.50E-05	1.21E-05	6.87E-06	6.36E-06	5.68E-06	5.11E-06
0.5	2.91E-04	2.67E-04	2.37E-04	2.00E-04	2.37E-05	2.19E-05	2.19E-05	1.91E-05	1.07E-05	9.87E-06	9.06E-06	8.37E-06
0.7	4.08E-04	3.72E-04	3.28E-04	2.88E-04	2.54E-05	2.38E-05	2.38E-05	2.07E-05	1.20E-05	1.11E-05	1.03E-05	9.63E-06
1.0	4.84E-04	4.41E-04	3.98E-04	3.44E-04	2.37E-05	2.22E-05	2.22E-05	1.99E-05	1.16E-05	1.08E-05	1.01E-05	9.61E-06
1.5	5.74E-04	5.13E-04	4.56E-04	4.09E-04	1.67E-05	1.61E-05	1.61E-05	1.47E-05	8.69E-06	8.30E-06	7.65E-06	7.47E-06
2.0	5.31E-04	4.98E-04	4.51E-04	4.06E-04	1.04E-05	9.57E-06	9.57E-06	9.09E-06	5.33E-06	5.19E-06	4.85E-06	4.86E-06

**Figure 9.** The NO₂ mole fractions in states 3, 4 and 5, function of excess of oxygen and x.

The cumulative values of NO_x are given in Table 7 and Figure 10. The numerical results showed similar influences of hydrogen as in the case of NO emissions because the values of NO are more larger than those of NO₂.

Table 7. NO_x mass ppm.

	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75	x = 0.00	x = 0.25	x = 0.50	x = 0.75
exc												
ess												
oxy												
gen												
					State 3				State 4			
0	1.01E+04	1.06E+04	1.12E+04	1.29E+04	2.46E+02	2.76E+02	3.44E+02	5.08E+02	3.47E+02	4.02E+02	4.82E+02	7.12E+02
0.2	1.90E+04	1.96E+04	2.01E+04	2.23E+04	1.42E+03	1.53E+03	1.68E+03	2.17E+03	1.41E+03	1.54E+03	1.67E+03	2.16E+03
0.5	1.77E+04	1.92E+04	2.04E+04	2.24E+04	7.70E+02	8.47E+02	9.80E+02	1.35E+03	7.58E+02	8.27E+02	9.61E+02	1.33E+03
0.7	1.52E+04	1.59E+04	1.72E+04	1.92E+04	3.70E+02	4.03E+02	4.67E+02	6.51E+02	3.48E+02	3.82E+02	4.48E+02	6.33E+02
1.0	1.15E+04	1.20E+04	1.30E+04	1.49E+04	1.70E+02	1.84E+02	2.11E+02	3.00E+02	1.51E+02	1.66E+02	1.93E+02	2.82E+02
1.5	6.22E+03	6.66E+03	6.86E+03	8.00E+03	4.56E+01	4.82E+01	5.01E+01	6.50E+01	3.28E+01	3.57E+01	3.53E-03	5.33E+01
2.0	3.14E+03	3.35E+03	3.56E+03	4.11E+03	1.86E+01	1.79E+01	1.76E+01	2.01E+01	1.06E+01	1.09E+01	1.10E+01	1.33E+01

**Figure 10.** The NO_x mass ppm [mg/kg] in states 3, 4 and 5, function of excess of oxygen and x.

The chemical modeling revealed that all interrelated parameters have variable values depending on the hydrogen mole fraction in the fuel and on the excess oxygen imposed for a certain combustion process, e.g. the temperatures and pressures in states 2, 3, 4, 5 and extra, the mole flue gases composition regarding the so called minor chemical species as H₂, OH, H, O, N. Tables 8 and 9 display selected numerical results for x = 0 and x = 0.75. The presence of hydrogen very slightly modified the temperatures and pressures of states 2, 3 and 4. However the values of x and exo modified notably the flue gases composition, see Figures 11–15.

Table 8. Selected numerical results for $x = 0$.

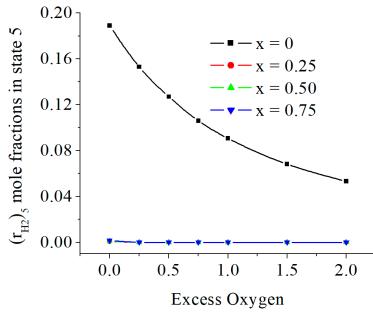
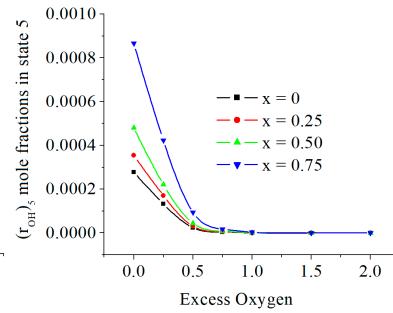
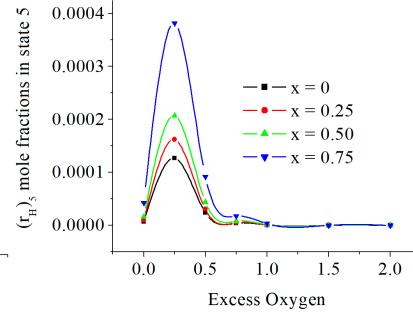
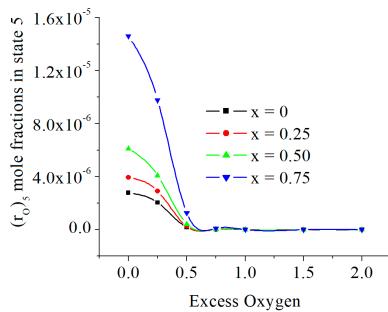
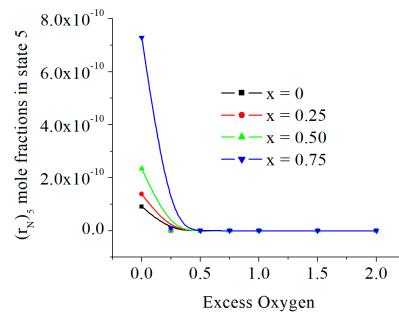
exo	0.00	0.25	0.50	0.75	1.00	1.50	2.00
T_2/p_2 [K/bar]	683.96/22.95	690.13/23.16	694.55/23.31	697.87/23.42	700.45/23.51	704.22/23.63	706.83/23.72
T_3/p_3 [K/bar]	3163/106	2859/96	2587/88	2350/83	2152/79	1843/73	1603/68
$T_4 = T_5$ [K]	1948.00	1670.00	1455.00	1288.32	1148.67	943.44	794.49
p4	6.50	5.60	4.98	4.50	4.19	3.70	3.80

Table 8. Selected numerical results for $x = 0$ (continuation).

exo	0.00	0.25	0.50	0.75	1.00	1.50	2.00
$(r_{H_2})_5$ [kmole/kmole]	1.89E-01	1.53E-01	1.27E-01	1.06E-01	9.07E-02	6.82E-02	5.32E-02
$(r_{OH})_5$ [kmole/kmole]	2.78E-04	1.33E-04	2.41E-05	4.12E-06	6.42E-07	1.34E-08	2.04E-10
$(r_H)_5$ [kmole/kmole]	6.91E-06	1.27E-04	2.39E-05	4.11E-06	6.41E-07	1.34E-08	2.04E-10
$(r_O)_5$ [kmole/kmole]	2.78E-06	2.05E-06	1.75E-07	1.22E-08	7.35E-10	2.31E-12	4.46E-15
$(r_N)_5$ [kmole/kmole]	9.12E-11	7.84E-13	4.29E-15	2.11E-17	8.79E-20	1.33E-24	8.81E-30

Table 9. Selected numerical results for $x = 0.75$.

x	0.75	0.75	0.75	0.75	0.75	0.75	0.75
exo	0.00	0.25	0.50	0.75	1.00	1.50	2.00
T_2/p_2 [K/bar]	703.97/23.62	706.75/23.72	708.75/23.78	710.27/23.83	711.46/23.87	713.21/23.93	714.42/23.973
T_3/p_3 [K/bar]	3278/102	2993/94	2714/88	2485/83	2283/80	1954/74	1704/70
$T_4 = T_5$ [K]	2075.00	1790.00	1575.00	1390.00	1240.00	1010.00	850.00
p4	6.50	5.65	5.20	4.60	4.30	3.85	3.50
$(r_{H_2})_5$ [kmole/kmole]	1.60E-03	2.02E-05	1.11E-06	5.36E-08	2.72E-09	6.25E-12	1.31E-14
$(r_{OH})_5$ [kmole/kmole]	8.66E-04	4.23E-04	9.42E-05	1.76E-05	3.17E-06	7.89E-08	1.60E-09
$(r_H)_5$ [kmole/kmole]	4.27E-05	3.82E-04	9.20E-05	1.75E-05	3.16E-06	7.89E-08	1.60E-09
$(r_O)_5$ [kmole/kmole]	1.46E-05	9.77E-06	1.26E-06	1.03E-07	7.62E-09	2.97E-11	8.91E-14
$(r_N)_5$ [kmole/kmole]	7.29E-10	1.06E-11	1.26E-13	8.30E-16	5.08E-18	0.00E+00	0.00E+00

**Figure 11.** The H_2 mole fractions in state 5 [kmole/kmole]**Figure 12.** The OH mole fractions in state 5 [kmole/kmole]**Figure 13.** The H mole fractions in state 5 [kmole/kmole]**Figure 14.** The O mole fractions in state 5 [kmole/kmole]**Figure 15.** The N mole fractions in state 5 [kmole/kmole]

The mole fractions of H_2 in state 5 are significant for $x = 0$ and more diminished when the fuel is containing hydrogen and for larger oxygen excess, $x > 0$, and $exo < 0$, see Figure 11.

The mole fractions of OH in state 5 are relatively small and they are increasing when x is increasing, and they are decreasing when exo is rising, see Figure 12.

The mole fractions of H is directly proportional to x and inversely proportional to exo , and have a maximum for $0 \leq exo \leq 0.5$, see Figure 13.

The mole fractions of O and N are directly proportional to x and inversely proportional to exo . The O mole fraction are at the level of mass ppm, but those of N are below mass ppb (parts per billion), see Figures 14 and 15.

3. Discussions

Chemical modeling of constant volume combustion of mixture of methane and hydrogen used in spark ignition Otto cycles revealed all features regarding flue gases parameters, temperatures, pressures, composition and the magnitude of pollutants function of the excess oxygen quantifying the engine load and of the hydrogen contents in the fuel.

The chemical model used the mass and energy balance equations applied to some cycles delivering different cyclic power, specified indirectly by oxygen excess, in order to find trustworthy numerical results. The gaseous chemical species were considered ideal gases with variable heat capacities depending on the temperature and with enthalpies depending also on the gaseous mixture composition function of temperature and pressure.

The different power was simulated through the oxygen excess, as it is actually proceeding when it is changing the cyclic fuel consume. The temperature ratio, T_3/T_2 , on the closed constant volume combustion was in the range 4.65 to 2.26 corresponding to an exo from 0 to 2.

The inlet temperature of reactants, T_2 around $700\text{ K} \pm 14\text{ K}$, was approximately constant imposed by adopted volumetric compression ratio and methane/hydrogen/air mixture composition, see Tables 8 and 9. This temperature was assimilated to a pre-heating and was quantified in the energy balance equation of combustion.

The numerical results have qualitatively and quantitatively similarities with experimental researches regarding pollutants, i.e. CO and NO_x. Therefore, the reduction of these pollutants asks very probably post combustion devices/processes in order to diminish them. Besides, they were found new features related to a large domains of x and exo , see below comments.

The numerical results emphasized the behavior of variations of pollutants emissions (CO and NO_x) in a good qualitative agreement with the experiments performed elsewhere. The values of CO mole fractions in flue gases are included in Table 4 and Figure 7. The maximum mass ppm found in state 5 is around 170 ppm [mg CO/kg flue gases], for $exo = 0$ and $x = 0.75$.

The throttling 4 – 5, decreasing the pressure, gives CO emissions almost twice larger for a stoichiometric combustion ($exo = 0$). All numerical results showed that the presence of hydrogen increases imperceptibly the CO emissions.

The numerical results showed two influences of hydrogen on the NO mole fractions, the first showed values directly proportional to the mole ratio of hydrogen in the fuel and larger than in the case of combustion without hydrogen. The second influence highlighted a maximum of these emissions for $0 < exo < 0.5$.

NO₂ emissions are smaller as larger the hydrogen mole fraction in the fuel. This is unlike the emissions of NO. But similarly, NO₂ emissions have a maximum depending on oxygen excess.

The cumulative NO_x showed similar influences of hydrogen as in the case of NO emissions because the values of NO are more larger than those of NO₂.

The presence of hydrogen very slightly modified the temperatures and pressures of states 2, 3 and 4.

The values of x and exo modified impressively the flue gases mole fractions of minor chemical species, the mole fractions of H₂ in state 5 are significant for $x = 0$ and more diminished when the fuel is containing hydrogen, $x > 0$, and for increased exo .

The mole fractions of OH in state 5 are relatively small and they are increasing when x is increasing, and they are decreasing when exo is rising.

The mole fractions of H are directly proportional to x and inversely proportional to exo , and have a maximum for $0 \leq exo \leq 0.5$.

The mole fractions of O and N are directly proportional to x and inversely proportional to exo , the O mole fractions are at the level of mass ppm, but those of N are below mass ppb (parts per billion).

The numerical modeling showed that the chemical reaction 10 is redundant when $x > 0$, respectively it was giving values either complex or below zero for chemical species H_2 resulting from this dissociation reaction. This indicates that the main chemical reaction producing H_2 by dissociation is given by eq. 9. Instead of eq. 10 it was involved the mass balance eq. 28 to verify our presumption.

3. Conclusions

The paper presents a chemical model for a closed constant volume combustion of gaseous mixtures of methane and hydrogen considering the spark ignition reciprocating engines in order to evaluate the combustion parameters and exhaust flue gases composition. The chemical model avoided the unknown influences in order to explain accurately the influence of hydrogen upon the constant volume combustion and flue gases composition. The model adopted simplifying hypotheses, respectively isentropic compression and expansion processes, closed constant volume combustion developed by two successive steps obeying to the energy and mass conservation laws and, flue gases exhaust described also by two steps, i.e. an isentropic expansion through the flow section of exhaust valves followed by a constant pressure stagnation (this succession is corresponding in fact to a throttling direct process). The chemical model supposed the homogeneous mixtures of gases with variable heat capacities function of temperatures, the Mendeleev - Clapeyron ideal gas state equation and, the variable chemical equilibrium constants for chosen chemical reactions. It was assumed that the flue gases chemistry is prevailing during the isentropic expansion and flue gases exhaust. The chemical model allowed evaluation of flue gasses composition and noxious emissions. for all states after combustion, 3, 4, 5.

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