

## Article

# Calorimetric Testing of Solar Thermal Absorbers for High Vacuum Flat Panels

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**Abstract:** Among solar thermal collectors, the evacuated flat panel is emerging as a reference technology for operation at higher temperatures of up to 200 °C with an increased annual energy production owing to both direct and diffuse light capture. Accurate measurements of the optical properties of the selective absorbers used in such devices are key for a reliable estimation of the overall performance. These optical properties must be measured under high vacuum at high temperatures, conditions under which the panels are meant to operate. In this study, we accurately measured these properties using a calorimetric technique. The measurement procedure is based on a power balance equation for a flat sample suspended in a high-vacuum chamber with minimal thermal losses and is well adapted for this class of devices. Calorimetric measurements obtained under Sun and LED light revealed excellent reproducibility and good agreement with those obtained using traditional optical analysis at low temperatures in air. When extended up to the absorber stagnation temperature, which often exceeds 300 °C, the calorimetric measurements started to deviate from the optical measurements, indicating the importance of measuring under the operating conditions.

**Keywords:** Selective Solar Absorber; Solar Absorptance; Thermal Emittance; Evacuated Flat Panel; Calorimeter; High Vacuum Flat Panel; Solar Simulator

## 1. Introduction

Solar thermal collectors may play a key role in the decarbonisation process of heat production. Worldwide, there is a huge demand for industrial process heat, and several technologies are available to convert the Sun's light power on a large scale (Farjana et al., 2018; Ismail et al., 2021). The adoption of a proper receiver architecture mainly depends on the operating temperature, and ranges from flat panels to concentrating power plants (Gorjian et al., 2020; Tian and Zhao, 2013). All possible collector designs can be distinguished by the grade of the sun's light concentration, type of tracking system, insulation, and shape of the absorber (Pranesh et al., 2019; Tian et al., 2018; Vengadesan and Senthil, 2020; Verma et al., 2020). Even if the available types of collectors are extremely diverse, they all share two main goals: maximisation of the sun's light conversion into heat and minimisation of thermal losses.

For temperatures of up to 200 °C, a novel architecture is emerging: a high vacuum flat plate (HVFP) collector hosting a planar absorber under high vacuum insulation (Buonomano et al., 2016). The presence of high vacuum in the panel represents the main difference between HVFP and evacuated flat panel (EFP) since EFP can also work at low-medium vacuum pressure (Benz and Beikircher, 1999). For HVFP collectors, convection losses are practically eliminated, and only conduction and radiation losses remain in addition to optical losses. In 2018, Moss et al. tested a small-scale prototype, showing that a reduction in the internal pressure to below  $5 \times 10^{-3}$  mbar minimized the thermal

conduction of residual gases to a negligible level. They also showed that HVFPs can potentially guarantee a larger quantity of annually converted energy by more than 60% with respect to the evacuated tube architecture owing to a wider acceptance angle, a better collection of the diffuse light and a larger fill factor of the gross area. Commercial devices have been introduced to the market (Benvenuti, 2013; TVP Solar SA, 2017), and a recent study reported on the experimental results from a 50 m<sup>2</sup> plant using TVP solar SA panels during a four-month testing period (Gao et al., 2020). An efficiency of 50% was achieved at a fluid temperature of 150 °C above ambient temperature, setting a record for solar thermal collectors.

When conduction losses due to the inevitable point of contact of the absorber with the panel envelope are minimised, the performance of this type of solar panel is mainly defined by the optical properties of the absorber because heat conduction through residual gases is negligible owing to the high vacuum (pressure below 10<sup>-4</sup> mbar). Therefore, establishing the optical properties can lead to a reliable calculation of the overall panel efficiency with a high precision.

In this study, we focus on a procedure for measuring the optical properties of the absorber, namely, the solar light absorptance and thermal emittance. Equations 1 and 2 express the absorptance (or spectrally averaged solar absorptivity)  $\alpha_{Sun}$  and thermal emittance (or spectrally averaged emissivity)  $\varepsilon_{abs}$ , respectively. They are a function of the spectral absorptivity  $\alpha_\lambda$  integrated over the Sun's spectrum  $G_{Sun}$  in the first case and of the spectral emissivity  $\varepsilon_\lambda$  integrated over black body emission  $E_{bb}$  (see Eq. 3) in the second case:

$$\alpha_{Sun} = \frac{\int_{0.3 \mu m}^{2.5 \mu m} G_{Sun}(\lambda) \alpha_\lambda d\lambda}{\int_{0.3 \mu m}^{2.5 \mu m} G_{Sun}(\lambda) d\lambda} \quad (1)$$

$$\varepsilon_{abs}(T_{abs}) = \frac{\int_{2.0 \mu m}^{100 \mu m} E_{bb}(\lambda, T_{abs}) \varepsilon_\lambda d\lambda}{\sigma T_{abs}^4} \quad (2)$$

$$E_{bb}(\lambda, T_{abs}) = \frac{2\pi hc^2}{\lambda^5} \cdot \frac{1}{\exp\left(\frac{hc}{\lambda k_B T_{abs}}\right) - 1} \quad (3)$$

where  $\lambda$  is the electromagnetic wavelength,  $T_{abs}$  is the absorber's temperature,  $c$  is the speed of light,  $h$ ,  $\sigma$  and  $k_B$  are Planck's, Stefan's, and Boltzmann's constants, respectively.

Under the hypothesis of steady-state conditions, the Kirchhoff law of radiation states that  $\alpha_\lambda = \varepsilon_\lambda$ ; in addition, in the presence of an opaque surface, i.e., no transmissivity,  $\alpha_\lambda$  and  $\varepsilon_\lambda$  are complementary to reflectivity:

$$\alpha_\lambda = \varepsilon_\lambda = 1 - \rho_\lambda \quad (4)$$

As a consequence,  $\alpha_\lambda$  and  $\varepsilon_\lambda$  are traditionally determined by optical measurements of the absorber reflectivity  $\rho_\lambda$ , where the subscript  $\lambda$  indicates that they are wavelength dependent quantities.

The measurement of  $\rho_\lambda$  is performed with reflectometers for the visible to the near infra-red (IR) wavelength range and Fourier Transform Infra-Red (FTIR) spectrophotometers for the near IR to the far IR. Both instruments measure the reflection from a sample under illumination by a controlled light beam as a function of the radiation wavelength. Measurements are usually conducted at room temperature, so the dependence of  $\varepsilon_{abs}$  on  $T_{abs}$  in Eq. 2 is only due to the temperature dependence of Planck's law (see Eq. 3), assuming  $\varepsilon_\lambda$  is constant across all temperatures. Considering  $\varepsilon_\lambda$  to be temperature independent inevitably leads to some approximations. Echániz et al. measured  $\varepsilon_\lambda$  for a solar absorber (based on Si, N, Mo over Ag-coated stainless steel) at various  $T_{abs}$ , from 250 °C to 600 °C and results of calculations of the thermal emittance  $\varepsilon_{abs}(T_{abs})$  increased by up to 8% if compared with values extrapolated from  $\varepsilon_\lambda$  measurements at room temperature (Echániz et al., 2015). Similarly large discrepancies were reported also in other works on multi-layered coatings in the same temperature range (Jyothi et al., 2017; Niranjani et al., 2021).

Moreover, when measuring thin-film structures at high temperatures in air, an oxidation process may occur, which can affect the optical properties of the film. For this reason, when the absorber inside a solar collector works under high vacuum at high temperatures, measurements of the optical properties should be performed under the same conditions. Some research groups have developed instruments capable of measuring samples at high temperatures under vacuum (Le Baron et al., 2019; Mercatelli et al., 2015). However, these methods are quite complicated and expensive. Simpler

and less expensive methods would be very useful in assessing the optical properties of selective absorbers developed for working at high operating temperatures in HVFPs (De Maio et al., 2021a, 2021b).

When measuring under high vacuum, a calorimetric approach using a power balance equation can be used effectively for measuring  $\alpha_{\text{Sun}}$  and  $\varepsilon_{\text{abs}}$  if controlled boundary conditions exist. The idea of a calorimetric measurement was already used in 1978 in testing selective coatings of metallic substrates (Willrath and Gammon, 1978). More recently, in 2015, Kreamer et al. proposed a calorimetric device for measuring  $\alpha_{\text{Sun}}$  and the thermal emittance of both sides of a selective solar absorber,  $\varepsilon_{\text{abs}}$  and  $\varepsilon_{\text{sub}}$ , respectively (Kraemer et al., 2015). What neither of the two previous calorimetric testing methods can provide is tests of the absorber under sunlight.

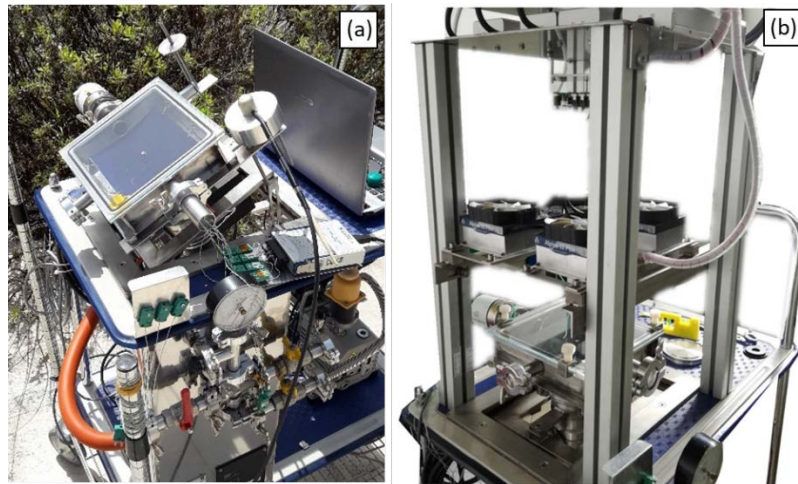
In this work, we describe a novel device for calorimetric measurements of solar absorptance and the effective thermal emittance of highly selective flat absorbers under high vacuum. The test equipment was a panel-like high vacuum envelope designed for outdoor testing, direct exposure to sunlight, and indoors testing using an LED-based solar simulator (D'Alessandro et al., 2021). This allows for a determination of the thermal emittance and solar absorptance of the selective solar absorber up to the stagnation temperature with high accuracy.

## 2. Experimental set-up

The test equipment was aimed at reproducing a HVFP as closely as possible. In Fig. 1(a), the device configuration for outdoor testing is depicted. It consisted of a rectangular stainless steel vacuum chamber closed by a dismountable cover glass. A vacuum seal was obtained using a Viton O-Ring hosted in a groove on the chamber flange, which was directly compressed by the glass plate. Low pressure in the chamber was guaranteed by a pumping system composed of primary and turbo-molecular pumps connected in series. Flat absorber samples (dark blue surface in the picture), 140 mm  $\times$  150 mm, were placed inside the vacuum chamber and suspended by four stainless steel springs placed at the two opposite long edges of the sample. The dimensions of the sample and chamber were such that they reproduced the unit cell of an actual TVP Solar SA panel. In this panel, an array of holes was present in the full-scale absorber to allow metal pins to support the glass against atmospheric pressure. The vertical and horizontal pitches of the array were 140 mm and 150 mm, respectively. Therefore, the aim of this equipment was to test unit cell absorber samples in an environment as close as possible to that of the entire panel.

The pressure was measured during each test using a vacuum gauge (PKR 251, Pfeiffer). The temperatures of the absorber, cover glass, and chamber walls were measured using K-type thermocouples and read out using a USB module (National Instruments 9211). The glass and chamber temperature sensors were placed on the vacuum side so that the external air convection, natural or forced, did not affect the measurements. An additional thermocouple was crimped to a lug and screwed to the central hole of the sample using an M3 stainless steel bolt.

The entire system (vacuum chamber, pumping system, measurement probes, and electronics) was mounted on a trolley for outdoor testing, as shown in Fig. 1(a). This trolley was equipped with a tiltable support consisting of two slotted rods manually fixed by two screws to set the vacuum chamber at an angle with respect to the vertical. A meridian and pyranometer were attached parallel to the chamber to check the angle of the latter and to measure the intensity of the solar irradiation incident over time, respectively. The pyranometer used was a thermopile-type from Apogee, with a response time of 0.5 s, a measurement repeatability of less than 1%, and a calibration uncertainty of 5%. In addition, a second pyranometer (CMP11 secondary standard pyranometer from Knipp&Zonen) with a lower uncertainty of 1% and a longer response time of 5 s was used to normalise the average value of the first pyranometer. The measurements from the sensors (thermocouples, pressure gauge, and pyranometer) were recorded in a 1 s time interval.



**Figure 1.** Experimental set-up for calorimetric testing in outdoor (a) and indoor (b) configuration. For outdoor testing the support positions of the chamber are tilted perpendicular to the Sun's beam. For indoor tests an LED light source (D'Alessandro et al., 2021) illuminates the chamber.

For indoor testing, the chamber was set horizontally and illuminated using an LED light source, as described in a previous paper (D'Alessandro et al., 2021). Figure 1(b) shows the LED system with its supporting structure when placed on the same trolley. The system was composed of four LED lamps arranged along the vertices of a square parallel to the plane of the sample, with each LED lamp attached to a modified CPU cooler. It is possible to separately adjust the position of each lamp in the horizontal plane, whereas the vertical distance can only be set for the four lamps together. The intensity is controlled by a proportional voltage signal driving the LEDs power supplies, and the incident light power density reaching the chamber can be fine-tuned between  $400 \text{ Wm}^{-2}$  and  $1.3 \text{ kWm}^{-2}$ . This allowed for an estimation of the incident power on the sample surface with a 1% accuracy (D'Alessandro et al., 2021). As LEDs emit light only in the range  $0.3 - 0.7 \text{ } \mu\text{m}$  (CREE, 2014), indoor testing of absorptance is limited to this spectral range, whereas measurements with the complete Sun spectrum are performed outdoors.

### 3. The Calorimetric Equation

The typical calorimetric equation for a system under study can be written as follows:

$$m_{abs}c_p(T)\frac{dT_{abs}}{dt} = \dot{Q}_{heating} - \varepsilon_e(T)\sigma A_{abs}(T_{abs}^4 - T_{amb}^4) - K_{all}(T_{abs} - T_{amb}) \quad (5)$$

where  $m_{abs}$ ,  $c_p(T)$ ,  $T_{abs}$  and  $A_{abs}$  are the mass, temperature-dependent heat capacity, temperature, and absorber surface, respectively.  $\dot{Q}_{heating}$  is a positive contribution to the absorber's temperature derivative  $dT_{abs}/dt$  owing to the absorbed light.  $\varepsilon_e(T)$  is the effective emittance that regulates the radiative exchange with the surrounding surfaces at ambient temperature  $T_{amb}$  and depends on the thermal emittance, shape, and relative position of all surfaces of the cavity (Howell et al., 2021). In our system  $\varepsilon_e$  can be calculated using a model of parallel plates with thermal emittances  $\varepsilon_1$  and  $\varepsilon_2$ ,

$$\varepsilon_e = \frac{1}{\frac{1}{\varepsilon_1} + \frac{1}{\varepsilon_2} - 1} \quad (6)$$

Finally,  $K_{all}$  is the parameter that includes all other heat losses, and in the following sections, its contribution to the calorimetric equation is calculated and demonstrated to be negligible.

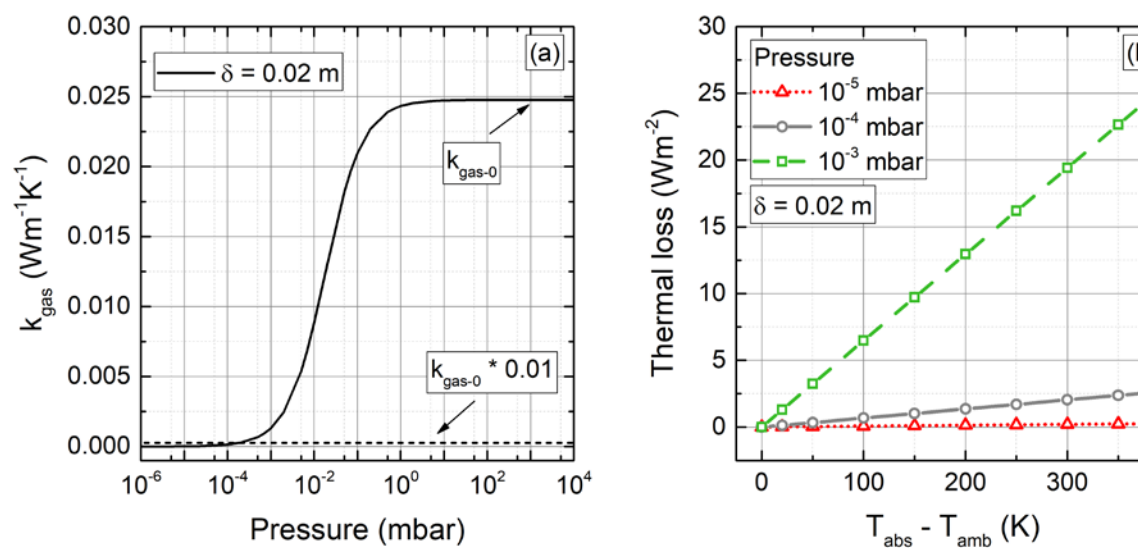
### 3.1 Thermal loss due to residual gas

The first term that contributes to  $K_{all}$  is the thermal conductivity  $k_{gas}$ . The test equipment included a pumping system capable of maintaining the chamber under high vacuum. Under such conditions, thermal losses via residual gases are generally negligible, but some quantitative considerations must be made (Arya et al., 2018a, 2018b). Equation (7) describes the thermal conductivity  $k_{gas}$  of the residual gases as a function of the Knudsen number  $Kn$ .

$$k_{gas} = \frac{k_{gas-0}}{1+2\beta Kn} \quad (7)$$

$Kn$  is a dimensionless parameter defined as the ratio between the mean free path of the gas particles and the characteristic size  $\delta$  of the chamber. When the gas temperature and  $\delta$  are constant,  $Kn$  is inversely proportional to the pressure.  $k_{gas-0}$  is the thermal conductivity of air at ambient pressure, and  $\beta$  is a parameter which depends on the nature of the residual gases and the accommodation fraction of the surfaces (Dushman et al., 1949). In Fig. 2(a), the thermal conductivity of residual gases is reported as a function of pressure for  $\delta = 0.02$  m, which is representative of the distance between the sample and the walls of the chamber facing it. Between a pressure of  $10^3$  mbar and 10 mbar, there is no change in  $k_{gas}$ . In this range, pressure alteration causes variations in particle collisions per unit volume, which is counterbalanced by variations in the mean free path (continuous matter domain).

In the range of  $10^{-4}$  – 10 mbar,  $k_{gas}$  decreases with pressure, and for a pressure of  $10^{-4}$  mbar and below,  $k_{gas}$  remains below 1%. Such a strong reduction in pressure leads to a condition in which the mean free path of particles becomes smaller than  $\delta$ ; thus, the number of collisions between particles is lower than that between particles and walls (molecular domain) (Beikircher et al., 1995). It is possible to calculate thermal losses in terms of  $Wm^{-2}$  as  $k_{gas}$  times the temperature difference between the absorber and the environment,  $T_{abs} - T_{amb}$ , divided by  $\delta$ . Figure 2(b) shows the resulting loss for three internal pressures of  $10^{-5}$ ,  $10^{-4}$  and  $10^{-3}$  mbar with  $\delta = 0.02$  m. When the pressure is equal to or below  $10^{-4}$  mbar the losses do not exceed  $5 Wm^{-2}$  even for values of  $T_{abs} - T_{amb}$  as high as 400 K. In contrast, at a pressure of  $10^{-3}$  mbar the losses are significantly higher, as the dependence of  $k_{gas}$  on pressure is not linear. The pressure of  $10^{-4}$  mbar is therefore a threshold that guarantees only a few  $Wm^{-2}$  of losses, which is negligible when compared to the typical  $1 kWm^{-2}$  irradiance in solar applications. During all experimental runs, the internal pressure of the chamber was always below such threshold, and  $T_{abs} - T_{amb}$  never exceeded 300 K, ensuring that any contribution to  $K_{all}$  due to residual gases could be neglected.



**Figure 2.** Effect of the pressure of the residual gas inside the vacuum chamber on the thermal conductivity  $k_{gas}$ . (a)  $k_{gas}$  is expressed in  $Wm^{-1}K^{-1}$  and is reported as a function of pressure for the characteristic length  $\delta = 0.02$  m. (b) Thermal losses due to residual gases are reported in  $Wm^{-2}$  relative to  $\delta = 0.02$  m and for the three pressure levels of  $10^{-5}$ ,  $10^{-4}$  and  $10^{-3}$  mbar as a function of  $T_{abs} - T_{amb}$ .



## 3.2 Thermal loss due to supporting springs

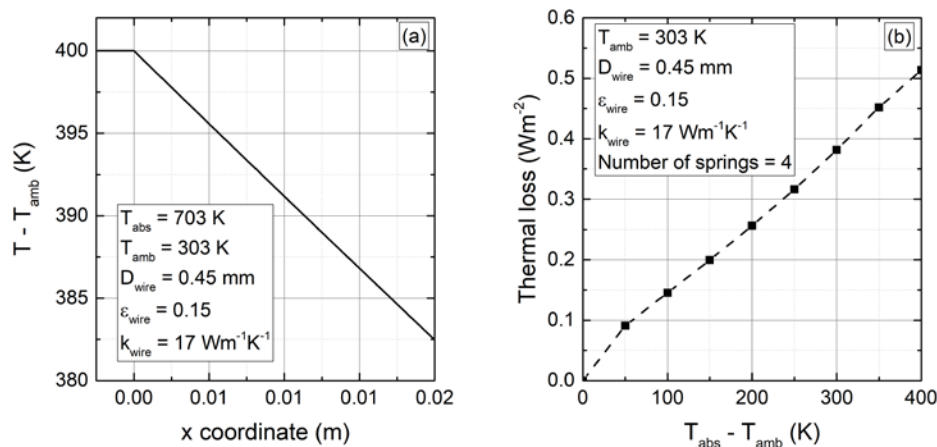
An additional contribution to  $K_{all}$  to be considered is the contact between the sample and the stainless steel supporting springs. In the worst-case scenario, this contact was perfect, and the ends of the two springs were at  $T_{abs}$  and  $T_{amb}$ , respectively, dissipating heat via conduction and radiative emission toward the chamber walls. The following differential Eq. (8) governs the temperature distribution over the length of a straight wire under steady-state conditions and circular cross-sections (Masuda and Higano, 1985):

$$\frac{d^2T}{dx^2} = \frac{4\varepsilon_w\sigma}{k_w D_w} (T^4 - T_{amb}^4) \quad (8)$$

where  $T$  is the temperature of the wire at the spatial coordinate  $x$ ,  $k_w$ ,  $D_w$  and  $\varepsilon_w$  are the wire's thermal conductivity, diameter, and thermal emittance, respectively. Assuming contact with the sample at  $x = 0$ , the corresponding heat loss through the wire is:

$$\dot{Q}_s = \frac{\pi}{4} D_w^2 k_w \left. \frac{dT}{dx} \right|_{x=0} \quad (9)$$

Using the finite element method, we numerically solved the differential Eq. 8 and the resulting temperature distribution is shown in Fig. 3(a) as  $T - T_{amb}$  (with  $D_w = 0.45$  mm,  $\varepsilon_{wire} = 0.15$ ,  $k_w = 17$  Wm<sup>-1</sup>K<sup>-1</sup>,  $T_{amb} = 303$  K, and  $T_{abs} - T_{amb} = 400$  K). The distribution is roughly linear, meaning that the conduction mechanism is dominant over the radiative one. In Fig. 3(b) the thermal losses calculated by Eq. 9 are reported for the case of a 140 mm x 150 mm sample area and four supporting springs. The power losses are extremely low (below 1 Wm<sup>-2</sup>), so the impact of the supporting springs is surely negligible and will not be accounted for in  $K_{all}$ .



**Figure 3.** (a) Temperature distribution over the wire for  $T_{abs} - T_{amb} = 400$  K as a result of the differential Eq. 8 solved by a finite element method. (b) Thermal loss calculated by Eq. 9 is expressed in Wm<sup>-2</sup> for four springs in perfect contact with the sample and chamber as a function of  $T_{abs} - T_{amb}$ . The surface referred to is an area of 140 x 150 mm<sup>2</sup> of the sample.

## 3.3 Thermal losses due to the presence of a thermocouple

The presence of a stainless steel bolt and nut attached to the absorber for fastening the thermocouple is an additional source of radiation loss to be considered. This could affect the tests because the thermal emittance and light absorptance are likely to differ with respect to the absorber under test. However, the total exposed surface was more

than two orders of magnitude lower than that of the investigated sample, and even if its radiative properties differed from those of the absorber, its contribution to  $K_{all}$  could be neglected. Therefore, the role of the bolt and nut is limited to its thermal capacitance, which is added to the thermal capacitance of the sample.

### 3.4 Formulation of the sample's calorimetric equation

Under the previous assumptions, the term  $K_{all}$  can be neglected, and the power-balance equation of the absorber can be rewritten as follows:

$$\frac{(m_{abs}c_{p_{abs}} + m_{th}c_{p_{th}})}{A_{abs}} \frac{dT_{abs}}{dt} = P_l \alpha_l \tau_g - \varepsilon_{eabs}(T) \sigma (T_{abs}^4 - T_g^4) - \varepsilon_{es}(T) \sigma (T_{abs}^4 - T_v^4) \quad (10)$$

where  $m_{abs}$ ,  $c_{p_{abs}}$ ,  $m_{th}$  and  $c_{p_{th}}$  are the mass and specific heat of the absorber and the thermocouple fastening bolt and nut, respectively.  $A_{abs}$  is the sample surface area of the side exposed to the light source.  $P_l$  and  $\alpha_l$  are the light power density incident on the glass closing the vacuum chamber and the spectrally averaged absorptivity of the sample, respectively. Depending on the adopted light source,  $P_l$  and  $\alpha_l$  are specified as  $P_{Sun}$  and  $\alpha_{Sun}$  in the case of outdoor testing and as  $P_{LED}$  and  $\alpha_{LED}$  for indoor testing.  $\tau_g$  and  $T_g$  are the light transmittance and the measured temperature of the glass, respectively. The product of  $P_l$ ,  $\alpha_l$  and  $\tau_g$  defines the heating of the sample within the chamber.

Commercial flat absorbers are thin metal foils, typically aluminium or copper, coated on one side with a selectively absorbing coating. As the area of the coated side is equal to that of the uncoated side,  $A_{abs}$  represents both sides, and herein it is 140 mm × 150 mm. The thermal emission mechanism is modelled according to the Stefan-Boltzmann formulation, which expresses the radiative exchange by the difference between the fourth power of the two facing surface temperatures and an emittance parameter that considers such surfaces as non-ideal blackbodies.  $\varepsilon_{eabs}$  and  $\varepsilon_{es}$  are then the effective thermal emittance of the two sides of the sample. The first refers to the absorbing side facing the glass at  $T_g$  and the second to the substrate side facing the stainless steel chamber at temperature  $T_v$ .

Assuming  $T_g = T_v$ , one can rewrite the radiative terms in Eq. 10, which is proportional to the total effective thermal emittance,  $\varepsilon_{et}$ , as shown in Eq. 11. It obviously depends on the thermal emittance of each side of the sample and the walls of the chamber, being the sum of  $\varepsilon_{eabs}$  and  $\varepsilon_{es}$ .

$$\frac{(m_{abs}c_{p_{abs}} + m_{th}c_{p_{th}})}{A_{abs}} \frac{dT_{abs}}{dt} = P_l \alpha_l \tau_g - \varepsilon_{et}(T) \sigma (T_{abs}^4 - T_g^4) \quad (11)$$

The previous simplification is acceptable because, during all runs, the maximum difference between  $T_g$  and  $T_v$  never exceeds 10 K, having an impact of less than 1% on the radiated power, and hence, also on the calculated  $\varepsilon_{et}$ .

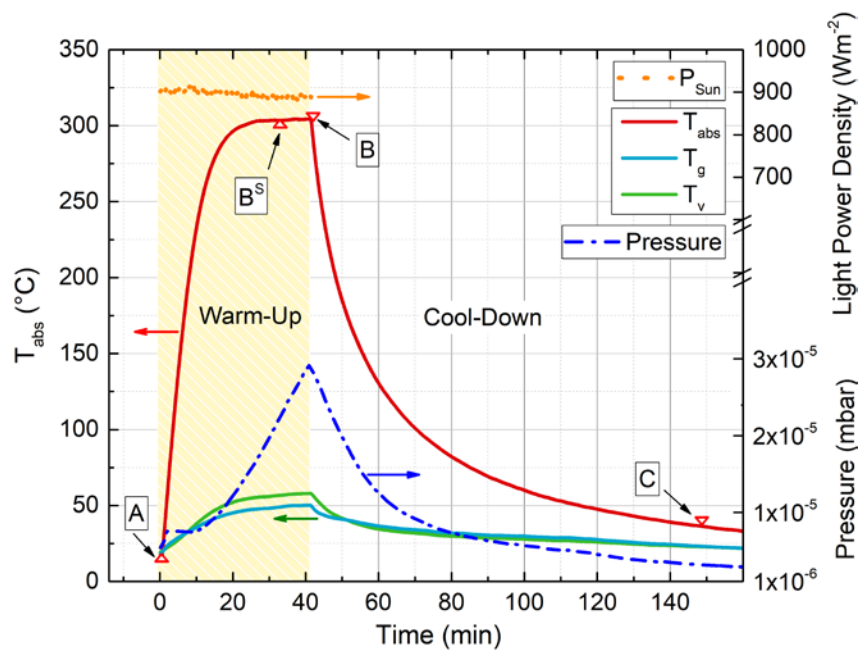
Equation 11 is, therefore, the simplified power balance equation, which regulates the temperature of the sample  $T_{abs}$  over time  $t$ . This equation is used to calculate  $\alpha_l$  and  $\varepsilon_{et}$  for the absorber sample because all other quantities can be readily measured.

## 4. Experimental calorimetric procedure

The calorimetric measurement procedure was divided into two steps: warm-up and cool-down. The first step involves exposing the sample to light through the cover glass and allowing its temperature to increase up to the stagnation temperature  $T_s$ . This temperature is the maximum achievable by the sample and occurs when the radiative emission equals the absorbed light in power. For all experimental runs, the sample was maintained at a stagnation temperature for at least 10 min. The second step, cool-down, consists of letting the sample cool by means of radiative emission, while it is shielded from any incident light by means of an aluminium shield placed on top of the cover glass.

The data analysis started with the cool-down. During this step, Eq. 11 has zero  $P_l$ , so  $\varepsilon_{et}$  can be calculated as the only unknown of the equation. Then, the warm-up is used to calculate  $\alpha_l$  from Eq. 11 using  $\varepsilon_{et}$  from the cool-down.

In Fig. 4, all measured quantities during a single outdoor test are reported as a function of the relative time. The start of the warm-up phase was set at zero relative time. Before this moment, the sample was within the vacuum chamber with the system already oriented perpendicular to the Sun and the glass covered with an aluminium shield. Under this condition, the sample was at a temperature close to that of the vacuum chamber. When the shield was removed, the sample was exposed to solar irradiation, and the warm-up phase began (point A in Fig. 4). During this phase,  $T_{\text{abs}}$ ,  $T_g$  and  $T_v$  increased with the internal pressure, never exceeding the threshold of  $5 \times 10^{-5}$  mbar during any run. Outdoor tests were conducted around solar noon on clear-sky days to minimise variations in solar irradiation. The maximum recorded variation in  $P_{\text{Sun}}$  over the entire warm-up phase was 5% of the initial value. The absorber temperature increases up to the stagnation temperature  $T_s$ , point B<sup>s</sup> in Fig. 4. This condition was maintained for 10 min, and then, the glass was again covered with the aluminium shield, see point B, allowing the sample to cool to a temperature close to that of the vacuum chamber, point C.

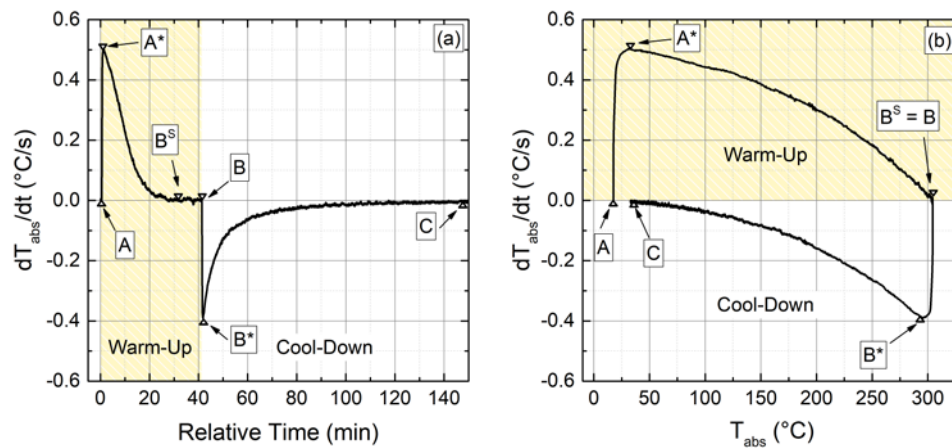


**Figure 4.** Time evolution of the sun light power density, chamber pressure, and temperature of different chamber elements during the outdoor experiment.

In Fig. 5, the time derivative of  $T_{\text{abs}}$  is reported as a function of relative time (a) and as a function of  $T_{\text{abs}}$  (b). According to Eq. 11, the maximum temperature derivative of the sample is expected at the start of the warm-up phase, as thermal losses are at a minimum. Experimentally, it is observed that the maximum temperature derivative occurs with a slight delay with respect to the sample illumination (point A\* in Figs. 5(a) and 5(b)). This unavoidable effect, due to the presence of a thermocouple that slightly increases the thermal capacity of the sample, was taken into account when analysing the data. After reaching the maximum  $T_{\text{abs}}$ , the derivative had a progressively lower value, reaching zero at the stagnation temperature, point B<sup>s</sup>. Therefore, the time interval with the sample at the stagnation temperature is characterised by a zero derivative of  $T_{\text{abs}}$  over time. Point B coincides with the illumination stopped by covering the glass with the aluminium shield; it defines the start of the cool-down phase, which has negative values for the  $T_{\text{abs}}$  derivative. Analogous to the warm-up phase, there is a delay for this derivative, and its absolute value reaches the maximum at point B\* before decreasing very close to zero at point C in Fig. 5. When the absorber approaches the ambient temperature, the emitted power becomes very small, and therefore cool-down lasts much longer than the warm-up phase. The measurements were stopped when the absorber reached below 50 °C.

The experimental procedure for indoor testing was similar. The vacuum chamber was set horizontally, and the LED system was mounted on top of it with lamps facing the cover glass. Because of the transient time necessary for the stabilisation of the emitted light (D'Alessandro et al., 2021), the LEDs were switched on 30 min before the warm-up phase started, with the glass covered by the aluminium shield. Like the outdoor tests, the warm-up phase was started by removing the aluminium shield and was stopped by repositioning it.





**Figure 5.**  $T_{\text{abs}}$  time derivative for an outdoor experiment is reported as function of relative time (a) and  $T_{\text{abs}}$  itself (b). The yellow background defines the warm-up, while the white indicates the cool-down. A 10-point moving average is used to reduce the noise.

## 5. Experimental Results

The experimental analysis was carried out on a sample cut from the Mirotherm® absorber from Alanod, the same absorber mounted in the TVP solar SA panels (Buonomano et al., 2016; Gao et al., 2020). It consisted of a 0.6 mm thick aluminium roll coated with a selective absorption layer on one side.

### 5.1 Optical Measurement Results

Optical measurements of the absorber reflectivity  $\rho_\lambda$  were performed for both the coated and uncoated sides of the absorber and the results were used to calculate the spectral absorptivity  $\alpha_\lambda$  and emissivity  $\varepsilon_\lambda$  (Eq. 4). Calibrated references were used to normalize the reflectivity to 1.

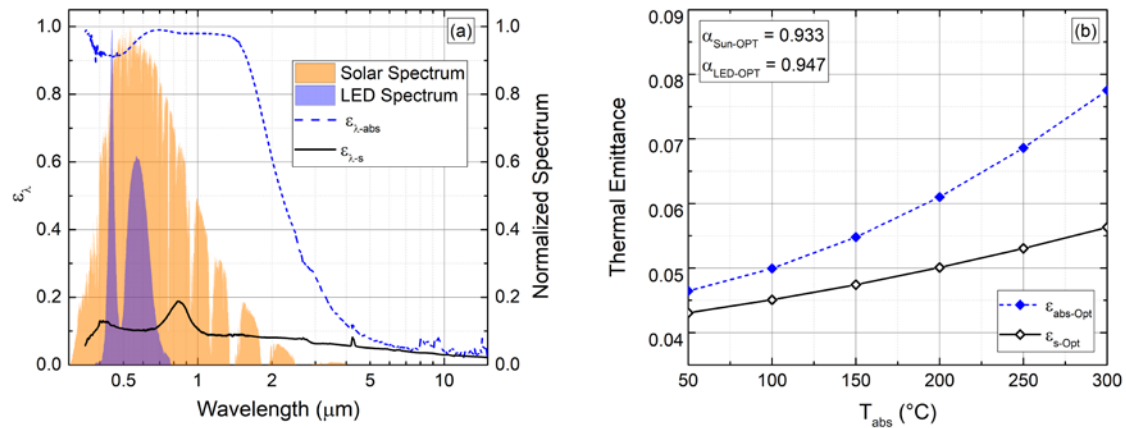
Through Eqs. 1 and 2, we calculated the light absorptance and thermal emittance values. The latter can then be optically estimated for higher temperatures using Eq. 2 and the hypothesis that  $\varepsilon_\lambda$  does not change with temperature. Hereinafter, the subscript Opt will be adopted for light absorptance and thermal emittance obtained by optical measurements.

In Fig. 6(a), the spectral emissivity derived from the optical reflectivity measurements at room temperature is shown,  $\varepsilon_{\lambda-\text{abs}}$  for the coated and  $\varepsilon_{\lambda-s}$  uncoated side of the absorber. The analysis between 0.35  $\mu\text{m}$  and 1.70  $\mu\text{m}$  was performed using an integrating sphere from Ocean Optics equipped with a halogen lamp and connected to an Optical Spectrum Analyser through a fibre optic cable. Then, between 1.20  $\mu\text{m}$  and 20.0  $\mu\text{m}$ , a FTIR from Jasco equipped with a diffuse reflectivity accessory was used.

In the overlapping region (between 1.20  $\mu\text{m}$  and 1.75  $\mu\text{m}$ ), the emissivity is the mean of the results of the two instruments. The coated side showed a good selectivity with high emissivity in the solar range (blue dashed line in Fig. 6(a)) which rapidly decreased at longer wavelengths. In contrast, the aluminium substrate has an emissivity  $\varepsilon_{\lambda-s}$  (black solid line) lower than  $\varepsilon_{\lambda-\text{abs}}$ , particularly in the solar range, as there is no coating applied on the back of the absorber. In the same figure, the normalized solar spectrum incident on the terrestrial surface, standard ASTM G173-0 (orange area), and the normalized spectrum of the adopted LED lamps from CREE (violet area) are also reported.

The spectrally averaged absorptivity of the coated side was calculated using Eq. 1, using the corresponding light spectra with  $\alpha_{\text{Sun-Opt}}$  and  $\alpha_{\text{LED-Opt}}$  values of 0.933 and 0.947, respectively. The optical measurements were highly accurate and reproducible, and the uncertainty in  $\alpha$  was estimated to be less than 0.003. The spectrally averaged emissivity  $\varepsilon_{\text{abs-Opt}}$  and  $\varepsilon_{s-\text{Opt}}$ , relative to the spectral emissivity  $\varepsilon_{\lambda-\text{abs}}$  and  $\varepsilon_{\lambda-s}$ , were calculated using Eq. 2 for several  $T_{\text{abs}}$  values (see Fig.

6(b)).  $\epsilon_{\text{abs-Opt}}$  is always higher than  $\epsilon_{\text{s-Opt}}$  over the entire investigated temperature range of 50 °C - 300 °C, with a difference that ranges between 10% at  $T_{\text{abs}} = 50$  °C and 40% at 300 °C.



**Figure 6.** Results of the optical analysis are reported for both sides of the commercial absorber Mirotherm® from Alanod. (a) Spectral emissivity curve of coated,  $\epsilon_{\lambda\text{-abs}}$ , and uncoated,  $\epsilon_{\lambda\text{-s}}$ , side as measured by reflectometry and FTIR measurements at room temperature and shown as a function of wavelength. Also, normalized solar spectrum, ASTM G173-0 standard (orange area), and adopted LED spectrum (violet area) are reported. (b) Thermal emittance of coated,  $\epsilon_{\text{abs-Opt}}$ , and uncoated,  $\epsilon_{\text{s-Opt}}$ , substrate side are estimated using Eq. 2 and reported as a function of  $T_{\text{abs}}$ . Absorptance is temperature independent with its values also reported in the figure.

## 5.2 Calorimetric measurements: total effective thermal emittance $\epsilon_{\text{et}}$

Samples with an area of 210 cm<sup>2</sup> (140 × 150 mm) were weighed using a balance with a sensitivity of 0.001 g to precisely determine their mass ( $m_{\text{abs}}$ ).

Estimations of  $\alpha$  and  $\epsilon_{\text{et}}$ , resulting from 12 calorimetric runs performed under different experimental conditions (6 performed outdoors and the other 6 indoors), are compared. Regarding the outdoor testing, the vacuum chamber walls were cooled by natural convection in all runs, from #1 to #6. For indoor testing, runs #1 and #2 refer to a  $P_{\text{LED}}$  close to 1 kWm<sup>-2</sup> with the chamber cooled by forced convection (electric fan); run #3 had a similar  $P_{\text{LED}}$  but without cooling; runs #4, #5 and #6 were performed without cooling too, but at reduced  $P_{\text{LED}}$ .

In Fig. 7(a), the  $\epsilon_{\text{et}}$  temperature dependence, as a result from an analysis of the cooling-down phase (grey line) of a single run, is reported as function of  $T_{\text{abs}}$  along with the points B, B\* and C. At the start of the cool-down phase, point B, the calculated  $\epsilon_{\text{et}}$  is zero. At lower  $T_{\text{abs}}$ , as the cooling progresses, there is a rapid increase in the calculated  $\epsilon_{\text{et}}$  up to point B\*. This anomaly is due to the thermocouple fastening and is determined by the shape of  $dT_{\text{abs}}/dt$  in the B – B\* interval, as seen in Fig. 5 and it was excluded from the analysis. In the following interval B\* – C, the calculated  $\epsilon_{\text{et}}$  is reduced together with  $T_{\text{abs}}$ , as was expected from the optical measurements, see Fig. 6(b). As the sample temperature approaches point C, the calculated  $\epsilon_{\text{et}}$  fluctuates. This can be explained by the reduced derivative values of  $T_{\text{abs}}$  when the sample has a temperature close to the ambient one. The intrinsic noise of the thermocouple signal affects the estimation in this temperature region. To obtain an  $\epsilon_e(T)$  function that can be used to compare the data and calculate thermal radiative losses, a fitting procedure using a second order polynomial function  $\epsilon_{\text{etF}}$  in the range B\* – C was performed, see Fig. 7(a).

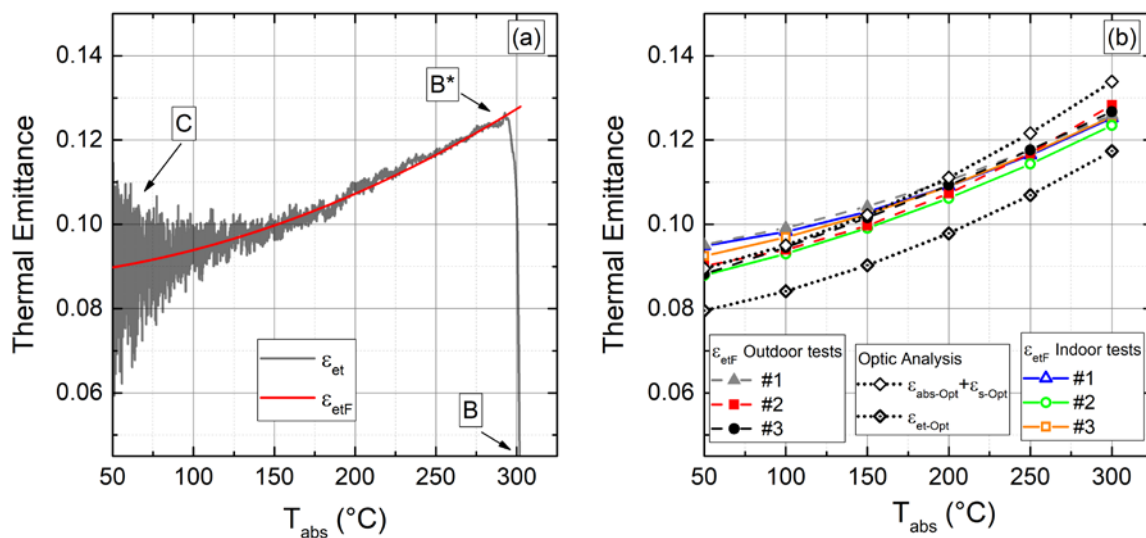
Figure 7(b) shows different  $\epsilon_{\text{etF}}(T)$  values resulting from runs #1 to #3 of the outdoor and indoor measurement campaigns (coloured lines). They all are in good agreement, and no trend can be extrapolated by comparing the indoor and outdoor results, implying that even if the test facility has different illumination, the resulting  $\epsilon_{\text{etF}}(T)$  is not affected. More experimental details regarding these external conditions are reported in the next section in Tables 1 and 2.

The six  $\epsilon_{\text{etF}}(T)$  results in Fig. 7(b) show a maximum difference of  $\pm 2.5\%$  and  $\pm 4\%$  with respect to the mean values in the absorber temperature range above and below 150 °C, respectively. Figure 7(b) also shows a comparison of the

optically determined total emittances ( $\varepsilon_{\text{abs-Opt}} + \varepsilon_{\text{s-Opt}}$  empty black diamonds) and the effective total thermal emittance  $\varepsilon_{\text{et-Opt}}$  (black diamond with dots) calculated using Eq. 6, assuming a glass thermal emittance of 0.85 and a stainless steel thermal emittance of 0.13. At temperatures below 150 °C there is a substantial agreement between  $\varepsilon_{\text{etF}}$  of the six experiments and the sum  $\varepsilon_{\text{abs-Opt}} + \varepsilon_{\text{s-Opt}}$ . With an increase in  $T_{\text{abs}}$ , the calorimetric results tend to be lower than the sum  $\varepsilon_{\text{abs-Opt}} + \varepsilon_{\text{s-Opt}}$ . At  $T_{\text{abs}} = 200$  °C, the sum  $\varepsilon_{\text{abs-Opt}} + \varepsilon_{\text{s-Opt}}$  starts to depart from the calorimetric values, and at  $T_{\text{abs}} = 300$  °C it overestimates the total emittance by 0.01, i.e., by 9% of the calorimetric result, see Fig. 7(b). In contrast, the total effective emittance  $\varepsilon_{\text{et-Opt}}$  calculated using Eq. 6 underestimates the calorimetric outcome over the entire investigated temperature range. This discrepancy reduces with temperature, ranging from 11% at  $T_{\text{abs}} = 50$  °C to 8% at  $T_{\text{abs}} = 300$  °C.

Neither the sum of the spectrally averaged emissivity of either side nor further correction by Eq. 6 can be used to obtain an accurate estimation of the actual effective total thermal emittance of the absorber when operating at a high temperature. The reason for this discrepancy may be a change in the spectral emissivity of the sample at high temperatures (Echániz et al., 2015).

The presented test equipment and procedure allow for the estimation of the actual thermal emittance of the absorber during operation in a HVFP without extending the optical measurements performed at room temperature to a high temperature.



**Figure 7.** (a) An example of calorimetrically measured effective thermal emittance  $\varepsilon_{\text{et}}$  is reported together with the polynomial fitting curve  $\varepsilon_{\text{etF}}$  as a function of  $T_{\text{abs}}$ . (b) Polynomial fitting curves are reported as a function of  $T_{\text{abs}}$  for three indoor and three outdoor experiments together with the total thermal emittance derived by optical reflectivity measurements as the sum of  $\varepsilon_{\text{abs-Opt}}$  and  $\varepsilon_{\text{s-Opt}}$  (empty black diamonds), and as the effective emittance  $\varepsilon_{\text{et-Opt}}$  calculated by Eq. 6 (black diamonds with central dot).

### 5.3 Calorimetric measurements: light absorptance

After the cool-down phase of each run, the warm-up phase was analysed to calculate  $\alpha_l$  as the only unknown quantity of the calorimetric Eq. 11, adopting the  $\varepsilon_{\text{etF}}(T)$  function resulting from the fitting of  $\varepsilon_{\text{et}}$  of the same test as the total effective thermal emittance. Using this approach, every run is a stand-alone experiment and does not correlate with the other results. The absorption mechanism of light, impinging on the glass cover, is modelled by the product of  $P_l$ ,  $\alpha_l$  and  $\tau_g$ . The glass transmittance is measured as the ratio between the reading of the secondary pyranometer when exposed perpendicularly to the incident light underneath and above the glass. For both, the Sun and the LED light, the measured  $\tau_g$  had the same value of 0.915.

Similar to  $\varepsilon_{\text{et}}$ , the calculated  $\alpha_l$  from the warm-up phase also shows an anomaly due to the delayed response of the  $T_{\text{abs}}$  derivative. The delayed response impacts the measurement for no more than 20 °C and  $\alpha_l$  is calculated only in the interval A\* – B\*. For each indoor and outdoor run, the measured  $\alpha_l$  in A\* – B\* can be assumed to be constant with a

variation of  $\pm 1.5\%$  around the mean value.  $\alpha_{\text{Sun}}$  and  $\alpha_{\text{LED}}$  resulting from the outdoor and indoor experimental campaigns are reported in the following Tables 1 and 2, respectively.

**Table 1.** Summary of six outdoor calorimetric experiments is reported. The table shows  $P_{\text{Sun}}$  measured at the start and end of the warm-up phase at points A\* and B<sup>s</sup>, the temperatures at the stagnation of the absorber, glass, and vacuum chamber,  $T_s$ ,  $T_g$  and  $T_v$ , respectively.  $\alpha_{\text{Sun-Opt}}$  and  $\alpha_{\text{Sun}}$  for each test are reported.  $\alpha_{\text{Sun}}$  is the averaged value calculated for A\* – B<sup>s</sup>.

Outdoor test	$P_{\text{Sun}}$ in A*/(Wm <sup>-2</sup> )	$P_{\text{Sun}}$ in B <sup>s</sup> /(Wm <sup>-2</sup> )	$T_s$ /(°C)	$T_g$ in B <sup>s</sup> /(°C)	$T_v$ in B <sup>s</sup> /(°C)	$\alpha_{\text{Sun}}$
#1	910	921	308	56	59	0.930
#2	910	885	302	54	59	0.915
#3	906	897	305	50	58	0.910
#4	923	942	306	56	60	0.915
#5	915	926	311	50	59	0.932
#6	923	943	313	51	60	0.909

$$\bar{\alpha}_{\text{Sun-Opt}} = 0.933 \pm 0.003$$

Table 1 summarises the experimental results of the six outdoor tests. It has been reported  $P_{\text{Sun}}$  at A\* and B<sup>s</sup>, that is, at the start and end of the interval were used to measure  $\alpha_{\text{Sun}}$ . The absorber, glass, and vacuum chamber temperatures ( $T_s$ ,  $T_g$  and  $T_v$ ) measured at stagnation are also reported. The last column shows the mean value of the calculated  $\alpha_{\text{Sun}}$  of the interval A\* – B<sup>s</sup>. In the last line of the table  $\alpha_{\text{Sun-Opt}}$  is also shown. The six runs were conducted on clear-sky days in Naples (40° 51' N – 14° 18' E) in October 2020 around noon. The chamber was then exposed to sunlight and ambient air. There is a good agreement between the six calorimetric measurements and the optical measurements, with a maximum discrepancy of 0.024, corresponding to 2.5% of  $\alpha_{\text{Sun-Opt}}$ . Considering all calorimetric measurements as equivalent, we can calculate the average value of  $\alpha_{\text{Sun}}$  and its standard deviation as  $\bar{\alpha}_{\text{Sun}} = 0.92 \pm 0.01$ , which agrees with  $\alpha_{\text{Sun-Opt}} = 0.933 \pm 0.003$ .

**Table 2.** Summary of six indoor calorimetric experiments. The table shows the control voltage of the LED system and the corresponding  $P_{\text{LED}}$ , the temperatures at the stagnation of the absorber, glass, and vacuum chamber,  $T_s$ ,  $T_g$  and  $T_v$ , respectively.  $\alpha_{\text{LED}}$  for each test is reported.  $\alpha_{\text{LED}}$  is the averaged value calculated for A\* – B<sup>s</sup>. The last line reports  $\alpha_{\text{LED-Opt}}$  and the  $\alpha_{\text{LED}}$  averaged over all six measurements.

Indoor test	Control Voltage /V	$P_{\text{LED}}$ /(Wm <sup>-2</sup> )	$T_s$ /(°C)	$T_g$ in B <sup>s</sup> /(°C)	$T_v$ in B <sup>s</sup> /(°C)	$\alpha_{\text{LED}}$
#1	6.50	1060	330	49	48	0.944
#2	5.80	981	318	46	38	0.928
#3	5.57	951	320	70	55	0.929
#4	4.00	715	289	66	58	0.949
#5	3.00	552	263	60	53	0.946
#6	2.00	392	226	49	44	0.940

$$\alpha_{\text{LED-Opt}} = 0.947 \pm 0.003$$

$$\bar{\alpha}_{\text{LED}} = 0.939 \pm 0.009$$

Table 2 is analogous to Table 1 but refers to the six indoor runs carried out with the LED system. For this reason,  $P_{\text{LED}}$  is identified by a single value, which is constant throughout the warm-up phase, whereas the Sun's irradiance intrinsically fluctuates. Each  $P_{\text{LED}}$  is coupled to a control voltage, which is the electric potential driving the control electronics of the LED system and ranges between 2 and 10 V (D'Alessandro et al., 2021). Indoor runs #1 and #2 have glass and vacuum chamber temperatures that are sensibly lower than for run #3 because of the presence of an external fan for cooling the chamber walls. Although the difference was approximately 20 °C, the resulting  $\alpha_{\text{LED}}$  exhibited no relevant change. Indoor runs #4, #5, and #6 were performed at a lower  $P_{\text{LED}}$  to obtain lower  $T_s$ . In this case, no significant variations in  $\alpha_{\text{LED}}$  could be detected. A comparison of the  $\alpha_{\text{LED}}$  estimations obtained under different conditions reveals a good agreement with the optically evaluated absorptance  $\alpha_{\text{LED-Opt}}$  reported in Table 2. The difference between the

average  $\alpha_{LED}$  ( $\bar{\alpha}_{LED} = 0.939 \pm 0.009$ ) and  $\alpha_{LED-Opt}$  ( $0.947 \pm 0.003$ ) was slightly smaller than in the direct sun illumination case, which could be due to the high temporal stability of the LED system.

The good agreement between the calorimetric and optical absorptance measurements globally validated our model and procedures. In particular, it confirms the appropriate modelling of the thermal losses in Eq. 11, as  $\varepsilon_{etF}$  can be used to correctly estimate the losses during the warm-up phase and it also validates the assumption of a temperature and power independent absorptance  $\alpha_i$ . The proposed model and equipment can measure the thermal emittance and solar absorptance of selective solar absorbers with a precision higher than 0.004 and 0.01, respectively. The larger error in the solar absorptance is essentially due to the 1% calibration error of the chosen pyranometer, which influences the incident power estimation.

## 6. Conclusion

We presented a novel method and experimental setup based on a power balance equation for estimating the optical properties of a flat selective absorber under high vacuum up to the stagnation temperature. These are appropriate conditions for assessing the performance of HVFPs. Optical measurements of the surface reflectivity were performed at room temperature of both sides of a commercial absorber sample. Outdoor and indoor (using a solar simulator) calorimetric measurements were conducted using the same commercial absorber (Mirotherm from Alanod) and the results were compared.

The effective thermal emittance of the sample, when measured calorimetrically, was not dependent on the test conditions. Outdoor runs returned the same results as those obtained indoors under controlled ambient conditions (with and without external fan cooling of the chamber, and incident light power on the cover glass ranging between 1 kWm<sup>-2</sup> and 0.4 kWm<sup>-2</sup>). All measured effective thermal emittances are contained within a variation of  $\pm 0.0035$  for an absorber temperature below 150 °C and within  $\pm 0.0025$  for above. These values correspond to 4% and 2.5% of the measured emittance, respectively.

The sum of the optically evaluated thermal emittances of both sides of the absorber was in good agreement with the effective thermal emittance calorimetrically estimated for temperatures below 150 °C. Above this threshold, the mathematical correction of the optical properties of the absorber based on the hypothesis of infinite plates fails. It results in a discrepancy of -11% at 50 °C which is reduced to -8% at 300 °C, underestimating the calorimetric results over the entire investigated temperature range. Such differences correspond to an error in the emitted power calculation of about 60 Wm<sup>-2</sup> at 300 °C. The measured light absorptance agreed with that obtained from the optical measurements in both outdoor and indoor tests with a maximum variation of 2.5%.

The emissivity discrepancies measured for the Mirotherm samples highlight the importance of measuring the radiative properties of an absorber up to high temperatures and under high vacuum to correctly assess the performance of selective solar absorbers, especially those developed for HVFPs.

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