Article

Ultrasonic Transceiver with a Regular/Periodic 1-3 Piezocomposite based on the SAW Resonance Mode on Damping Backing

Alex Mezheritsky

PiezoTech LLC, Indianapolis, Indiana, USA; amezheritsky@gmail.com

Received:

Abstract — A novel effective vibrational mode was disconered in the conventional transducer with an array of orthogonal (square) regular piezoelectric rods in 1-3 piezocomposite, containing the damping backing and front matching layers. The operational resonance in the structure was determined as the Surface Acoustic Wave (SAW) on the backing boundary excited by the adjacent piezo-rods, with its frequency typically near 3 times lower the fundamental half-lambda conventional piezocomposite resonance.

Pulse-echo sensitivity and transmitting sound-pressure-level (*SPL*) in air showed that the signal strength is roughly comparable to the industrial similar air transducers at the frequency range 100...700 kHz, where at these frequencies the lateral and longitudinal piezoelement dimensions in the conventional transducer design are typically close to each other causing interference with unwanted coupling modes.

As was determined theoretically and proved in experiments, the backing *SAW* resonance effect in the transducer performance is inherent just to the regular periodic 1-3 piezocomposite structure, and does occur neither with randomly located/oriented piezo-rods, nor in the homogeneous piezo-plate at least with the same lateral cross-section as the connected to it backing.

The purpose of the article is to investigate a newly discovered operational vibrational mode of a SAW type in 1-3 regular piezocomposite, other than piezoelectric resonance. The investigated phenomena can improve the transceiver sensitivity, bandwidth, providing lower drive voltage, and smaller and lighter weight ultrasonic transducers.

Based on the piezocomposites with thickness' 1...1.5 mm (rod resonance near 2...3 MHz), pillar width 0.2...0.8 mm, kerf width 0.1...0.4 mm, the transceivers with an operating frequency from 140 kHz to 650 kHz were designed and fabricated with a conventional backing of a mixture of high-density tungsten powder and epoxy, and a matching layer of a mixture of low-density glass bubbles and epoxy. Experimental evaluation of their acoustical performance showed expected characteristics suitable for practical applications.

Keywords: piezoelectric materials; piezoelement; 1-3 piezocomposite; resonance and antiresonance frequencies; quality *Q*-factor, surface acoustic wave (*SAW*), interdigital transducer (*IDT*), mass load, *SAW* resonator

1. Introduction

The 1-3 piezocomposite consists of an array of parallel piezoelectric rods embedded in a polymer matrix with electrode layers on the top and bottom surfaces, and incorporates epoxy('ies) for bondlines, backing and matching layers. It improves mechanical and ultrasonic performance compared to a baseline traditional design with a homogeneous piezoelectric plate, for example providing broader bandwidth, shorter pulse, lower *Q*-factor, clear spectra with suppressed planar unwanted modes. In conventional piezocomposite applications, the height of the pillars normally is about one half wavelength at the operating frequency. The piezoelectric layer in the structure is an active resonating transducing element, which converts electrical energy to acoustical energy and vice versa. The most common piezoelectric materials used in ultrasonic transducers are piezoceramics such as lead zirconate titanate (Pb(Zr,Ti)O₃, or PZT), and based on it piezocomposites, which possess relatively strong piezoelectric effect [1-3].

In the traditional operation, a voltage pulse is applied between the front and back surfaces of the piezoceramic rods of particularly a 1-3 piezocomposite to excite a mechanical expansion thickness resonance (rod length) of the piezoelectric plate.

The transducer generally works as a transceiver, generating and receiving the pressure and voltage pulses typically at the transducer half-lambda resonant frequency F_r evaluated [2,4] as:

$$F_r = c/2L \tag{1}$$

where L is the thickness of piezoelectric layer and c is the speed of sound (SOS) for the thickness mode in piezoelectric material. For a thin homogeneous plate the latter parameter $c = \sqrt{C_{33}^E/\rho}$, and for a 1-3 piezocomposite $c = \sqrt{1/\rho S_{33}^E}$, with C_{33}^E and $1/S_{33}^E$ taken as elastic stiffness and Young modulus under short-circuit condition, ρ is the material density. For a typical piezoelectric material, a 2 MHz array needs approximately 1.0 mm thick ceramic layer.

Conventionally, the planar (radial) modes are used for a low frequency range below near 100 kHz, while for a high frequency range above near 700 kHz a thickness vibrational mode of a monolithic, or piezocomposite, plate is exploited [5]. In the present research, a newly discovered operational vibrational mode of a *SAW* type in a 1-3 regular piezocomposite, other than piezoelectric resonance, was proposed and investigated for the 100...700 kHz applications.

Typically a quarter-lambda wavelength thick matching layer is introduced to provide the acoustic matching between the ceramic and loading fluid. The impedance of air is as low as near 100 Rayls, so that a light couplant is used to reduce this impedance mismatch. For a transducer with a single matching layer, its optimal acoustic impedance Z_m value is a geometrical mean [4] $Z_m = \sqrt{Z_o Z_R}$, where Z_0 and Z_R is the acoustical impedance of piezoelectric material and fluid, respectively. The

effect of perfectly matching layer is the energy transmission maximum efficient at the center frequency, providing relatively wide transducer bandwidth. Implementing typically light glass bubbles into epoxies is common practice to form matching material, when the acoustic impedance can be adjusted by the percentage of loaded bubbles to meet the design requirements of transducers [6].

The energy generated in the transducer can radiate in both the forward direction and reverse direction. The purpose of the matching layer is to encourage energy to be propagated in the forward direction with low loss and reflections. On the contrary, the backing layer is designed to maximum attenuate the signal emanating from the back surface of the piezoelectric plate, as well as reduce ringing. If the attenuation of the backing material is sufficiently large, no reflections from the back surface of backing layer can be found [2].

To minimize the ringing, the acoustic impedance of backing can be matched to piezoelectric material. But as a result, half of the energy will be transferred into the backing layer and then lost. A very short pulse could be obtained, but with a relatively low amplitude. Thus, compromise is always taken between sensitivity and bandwidth of the transceiver. Therefore, the acoustic impedance of backing layer is usually slightly lower than the one of piezoelectric layer in order to improve sensitivity at the cost of slightly increased pulse length or ringing – for some particular applications a predetermined waveform, with a definite number of the sine peaks, should be provided. The epoxies loaded with fine dense powders and heavy chips such as tungsten are commonly used as backing materials [2]. The purpose of loading powder into epoxies is to change the acoustic impedance similarly to the technique used for matching layer design, but with opposite effect of increased density, also creating multiple centers of energy dissipation.

The general benefits of the 1-3 piezocomposite structure are higher thickness mode piezoactivity (rods with vibration of k_{33} -type), lower frequency transverse (planar) unwanted modes, lower acoustical impedance and transducer Q-factor. Several unwanted higher frequency lateral resonances caused by the regular piezoceramic-epoxy structure in piezocomposites, are found and described in [7-11]. To damp the unwanted resonances, an irregular piezocomposite structure is recommended at this point as in [12-14].

For the acoustical performance characterization of the conventional layered transceiver, the transmission line (one-dimensional 1-D approach) theory is typically considered [4]. For this reason, it directly suggests and relates to the averaged (effective), or homogeneous piezoelement and backing bodies of the same planar area. For a transversely structured design more advanced 2-D and 3-D approaches and simulation methods are used in acoustical performance characterization [10].

In case of a 1-3 piezocomposite element (Figure 1a), it is structured with typically periodic piezoelectric rods imbedded into softer and lighter epoxy. In the conventional approach, the effective acoustical impedance of such a structure is proportional to the structural effective (averaged) density, following to the *kerf-to-pitch* piezocomposite parameter. Typically, the effective acoustical impedance of a piezocomposite is near 2 times lower than that in piezoceramics.

To the contrary of the conventional approach for the total piezocomposite structure as a whole, a single piezoelectric rod attached to a transversely wider backing was considered here as a baseline to describe the multiple rods individual and collective behavior on the backing, based on the theory of vibrations and 2-D FEA simulation.

2. Boundary SAW Resonance Performance of the Piezoelectric 1-3 Composite Bonded to Backing

2.1. 1-3 Piezocomposites with Regular Structure for the experiments

In the present research, a novel acoustically effective vibrational mode in the 1-3 piezoelectric transducer was discovered and investigated. Its resonance is based on the SAW-type wave on the backing bonding boundary and has frequency significantly lower the piezocomposite fundamental ($\lambda/2$ mode) thickness resonance. In this case the typical operation transducer frequency (piezocomposite with backing and matching) is near 3 times lower than the conventional piezocomposite thickness resonance (Figures 1b and 1c).

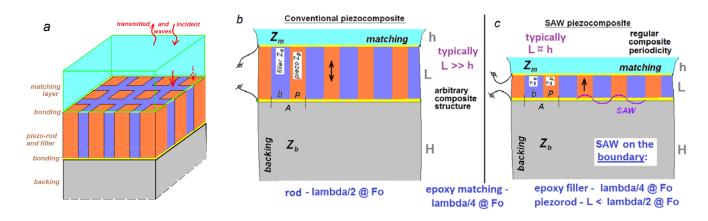


Figure 1. 1-3 epoxy-filled PZT piezocomposite transceiver: a – general view; b –structure for the conventional thickness resonance mode; c –structure for the novel *SAW* resonance mode on backing of the same operation frequency.

As was found, there are two critical factors supporting the transducer *SAW* operation with that new condition: necessary regular periodic transverse structure in the 1-3 piezocomposite, and a solid layer of bonding material (hard epoxy) between 1-3 piezocomposite and backing. The transducer

structure has a conventional damping backing with near 10...20 MRayls acoustical bulk impedance; conventional low impedance matching layer (glass bubbles with epoxy), with its quarter-lambda thickness at the *SAW* resonance frequency.

Practical realization of the frequency range with near 100...500 kHz operation frequency in a conventional transducer design is complicated due to the necessity to use a piezoelement with close lateral and transverse dimensions, causing complex transducer spectra and hence relatively lower acoustical efficiency.

In the present experiments, the 1-3 piezocomposites have near 65% volume of PZT-5A type (CeramTec GmbH, Germany) [3] piezoceramic, with effective acoustical impedance near 17 MRayl. They have typically 12 x 7 mm planar dimensions, with near 2.5 MHz piezoelement thickness resonance. Structural parameters of the piezocomposite samples used in the experiments are shown in Table I. All values and formula further are presented in SI, if other is not specified.

Table I. Structural array piezocomposite parameters, used in experiments.

#	<i>L</i> , mm	p, μm	b , μm	<i>p</i> + <i>b</i> , mm	L/p	L/p+b	$p^2/(p+b)^2$	F , kHz
1	0.76	120	80	0.20	6.3	3.8	0.36	~500 kHz
2	0.63	180	80	0.26	3.5	2.4	0.48	~500 kHz
3	0.88	540	160	0.70	1.6	1.3	0.60	~500 kHz
D1 / D2	2.7/5.0	1000	400	1.40	2.7/5.0	23	0.51	~150 kHz

Reference info: typically L/p > 1; piezomodule d_{33} 440...520 pC/N; capacitance 1900...2800 pF typ.; samples numbering is arranged in ascending order with the pillar (pitch) values; for D #, there are two elements with different piezorod height L.

Relative ceramic volume (effective density) in piezocomposite is determined by the parameter $p^2/(p+b)^2$, and the effective density can be expresses as

$$eff \frac{\rho_{composite}}{\rho_{ceramic}} = \frac{\rho_{ep}}{\rho_{PE}} + \frac{p^2}{(p+b)^2} \left(1 - \frac{\rho_{ep}}{\rho_{PE}}\right). \tag{2}$$

The backing with a height H of a dozen wave-lengths at the operation (SAW) frequency was a typical dense mixture of tungsten powder and chips with epoxy – it provides effective damping backing with acoustical impedance near 15 MRayls and Q-factor near 5 [15]. The quarter-lambda thickness matching layer was a mixture of glass-bubbles with epoxy, providing relatively low acoustical impedance near 1 MRayl. Basic parameters of the materials used in the experiments are shown in Table II.

For the acoustical tests, the distance ~2.5" was used in pitch-catch testing, and with a reflective SS block as a target in pulse-echo tests, both locations are the transducer "far field" area.

Table II. Transceiver Materials Data

	E, GPa	ρ , kg/m ³	<i>c</i> , m/s	Z_{ac} , MRayl	σ	1/Q
Piezoceramic	122	7800	3960 (rod)	30	0.35	1/100
Epoxy in composite	8.1	1200	2600 (rod)	3	~0.40	1/10
Matching (glass bubbles&epoxy)	2.4	390	2500 (plate)	0.9	~0.40	1/5
Backing (W powder&epoxy)	22	13000	1300 (600 SAW)	17	~0.35	1/6
Air	0.00015	1.25	350	0.00045	-	-

where E – Young's/bulk module; ρ – density; c – SOS; Z_{ac} – acoustical impedance; σ – Poisson coefficient; 1/Q – attenuation coefficient, with the quality factor Q.

An excitation spike of 300 V amplitude was applied to the transducer structure, so that with an output signal typically 100 mV, it provides near -70 dB pulse-echo sensitivity. In the basic characterization, together with the fundamental thickness resonance, typical higher frequency spurious resonances were noticed, related to a lateral Lamb wave in just a piezocomposite plate, with rods filled with epoxy [7-11].

2.2. SAW on the Backing Boundary – Acoustical Effect Prediction

A regular piezo-rod structure in the conventional 1-3 piezocomposite, under applied voltage or pressure, causes localized deformation of the backing boundary pre-surface with the same structural periodicity. When the excitation frequency coincides with that *SAW* resonance, the total transceiver structure and synchronized collective piezorods motion exhibit the resonance characteristics suitable for its ultrasonic operation and performance.

The effect is some similar to how the piezoelectric *SAW* devices with interdigital electrodes (*IDT*) work [16]. One of the differences is that an in-phase voltage is applied to all piezo-rods in piezocomposite, while in a *SAW* device, the counter-phase voltage is applied to every other electrode strips in the electrode comb array structure. It is also known that in the *SAW* resonator with *IDT* on a piezo-substrate, the phase velocity of *SAW* is considerably reduced since the metal *IDT* implies the mass load on the substrate. As a *FEA* simulation showed, there is a similar effect for the propagating *SAW*, when an isolated loading mass shifts down the working frequency as well, from the "ideal" resonance determined by the *SAW* speed of sound.

The surface wave physically occurs in an effective pre-layer of backing, including a thin layer of hard bonding epoxy, used to connect the backing to the piezocomposite rods and filler. The condition for a relatively low-frequency *SAW* resonance (caused by its low *SOS*), in the first approximation, requires for the two adjacent rods distance to be an integer multiple to the surface acoustical wavelength:

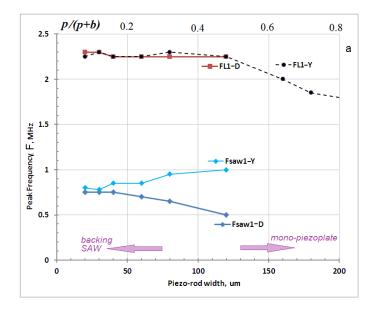
$$A \equiv b + p \cong \lambda \cdot N \quad , \tag{3}$$

where N=1,2,3... is a positive integer; A is the pitch of piezocomposite, as a sum of the kerf b and pillar width p. As an estimation for the SAW wavelength $\lambda = c_{saw}/F$, the backing bulk SOS is roughly 1500 m/s, and the shear (SAW) SOS is estimated as $c_{saw} = \sim 0.45$ $c_b \sim 600$ m/s, then the backing SAW wavelength $\lambda \approx 1.5$ mm is expected for 500 kHz frequency.

2.3. FEA Simulation of SAW Resonances on Backing with 1-3 Piezocomposite Rod Structure

COMSOL (Comsol Inc., Burlington, MA) software was used to simulate the *SAW* propagation on the backing pre-surface, loaded with the periodic piezoceramic rod structure bonded on the backing boundary. In a simplified linear array of the axial and transverse 2D model, an elementary cell of a regular piezocomposite was considered, under continuous wave excitation (*CW*, applied voltage 100 VAC). It was supposed that a 3D model provides some shifted parameters of secondary significance. In the model design, soft type piezoceramic PZT-5A [3], with damping coefficient Q = 20, a piezoceramic rod with basic height (length) L in a range from 50 μ m to 1.5 mm, and total width from 20 μ m to 240 μ m (minimum 10 μ m kerf) in the fixed pitch 250 μ m for certainty, were used in the simulation.

The backing parameters are presented in Table II, so that for a total backing height H 4 mm, with periodicity (pitch) 250 µm, it provides a longitudinal backing resonance much lower 100 kHz. Both the fundamental (lowest) backing *SAW* and longitudinal piezo-rod resonance frequencies are relatively stable versus piezo-rod width, as shown in Figure 2a. For the backing *SAW* resonance, it has a broader peak bandwidth for wider rods.



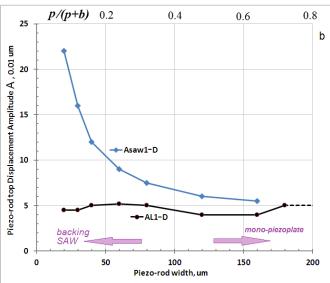


Figure 2. Simulation resonance frequencies (a) and top piezo-rod displacement (b) dependences on the rod width (pillar) under constant pitch $(p+b=250 \mu m)$; the piezo-rod length 0.76 mm.

In Figure 2a, the data are presented for the saw1-Y and saw1-D fundamental (lowest) SAW resonance taken respectively from the piezocomposite electrical admittance (Y = 1/Z) and from the piezo-rod top displacement (D); and for L1-Y and L1-D similarly for the fundamental (lowest) longitudinal resonance of the piezo-rod in piezocomposite; where L1 denotes the fundamental rod length vibrational mode. The respected displacement resonance amplitudes are presented in Figure 2b.

There are a number of SAW harmonics excited in the structure, and data for four lowest SAW modes were taken for analysis. For the baseline case of negligible piezo-rod's mass, the resonance frequencies are just directly proportional to their consecutive orders (N = 1, 2, 3...). For some longer piezo-rods (Figure 3a), when their own rod's longitudinal resonance coincides with one of the SAW resonances, a coupled modes vibration takes place. Under such conditions, the largest top piezo-rod displacement is provided at the lowest SAW resonance, as shown in Figure 3b.

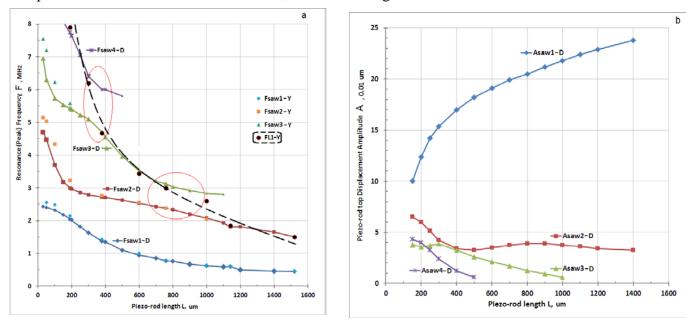


Figure 3. Simulation results for the resonance frequencies (a) and top piezo-rod displacement (b) vs. rod length under constant pitch ($p+b=250 \mu m$) and pillar width p (25 μm). The SAW and rod-length (piezocomposite thickness) resonance branches are shown inside the red circles where they are most acoustically coupled.

As follows from the Figure 2b data, for the piezocomposite structure with relatively narrow kerf, the top piezo-rod displacement is relatively low and close to that of a conventional homogeneous piezo-plate. For a larger kerf, the fundamental *SAW* mode shows large and rapid increase in the top piezo-rod displacement at the backing *SAW* resonance - the effect directly determining the transducer higher vibrational sensitivity and efficiency.

The resonance characteristics variation under just the kerf width varying, with all other structural array parameters taken constant, is presented in Figure 4a. For a relatively wider kerf, or the same larger pitch, the spectra effectively goes from a mono-frequency resonance to the multi-resonance peaks of *SAW* harmonics.

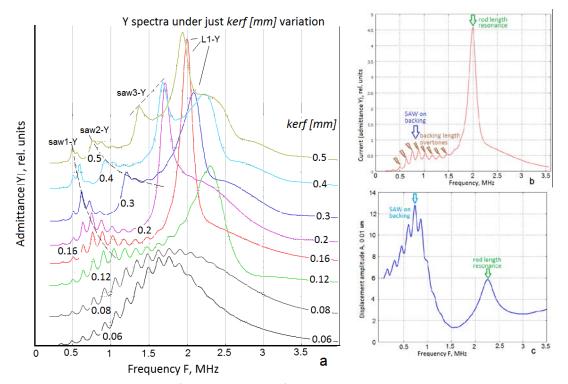


Figure 4. Simulated spectra (admittance IYI, CW) transformation under composite piezo-rod structural variation: a - just kerf variation (pillar width 50µm); b and c – comparative peak amplitudes admittance and top rod displacement (pitch 250µm and pillar 50µm), respectively; rod length 0.76 mm in the structure of just piezo-rods on backing.

On the admittance piezocomposite characteristic, the *SAW* resonance is some hardly identified (Figure 4b), however the piezo-rod top displacement amplitude is significantly higher than that at the piezocomposite plate fundamental mode (Figure 4c), making it obviously beneficial at least for more efficient energy transmission into load medium. Note that the fundamental *SAW* resonance on the backing boundary works also like a high-stop filter for the lower frequency longitudinal backing resonances, represented as a comb of peaks in Figure 4.

The waveform displacement and stress (von Mises) pattern in the resonating transducer structure at the fundamental *SAW* vibrational mode are shown in Figure 5. For a low (negligible) mass of tiny piezo-rods, the space *SAW* waveform is close to the ideal total sine period (Figure 5a). The effect of piezo-rod width variation on the stress distribution is demonstrated in Figures 5b-d. As seen from the simulation, the effective *SAW* in-depth into the backing equals to near same *SAW* wavelength, estimated as a motion intensity transition from the *SAW* to the bulk flat waves in the backing. For a

narrower kerf (Figure 5d), the vibrational displacement is getting lower, so that in a limit of zero kerf (homogeneous piezoplate) the SAW effect on backing disappears.

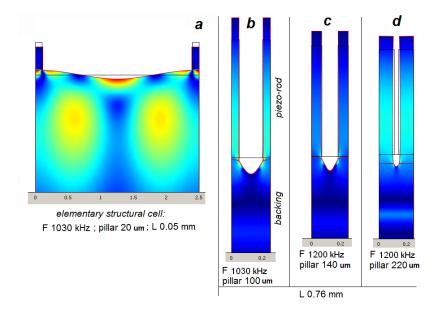


Figure 5. Space waveform displacement and stress (CW) distribution at the SAW backing resonance under composite structure (a), and their transformation under variation of piezo-rod height and width (b-c). Backing length 4 mm; pitch 250 µm.

2.4. Basic estimation for the 1-3 Piezocomposite Rod Structure with SAW Vibrational Mode on Backing

Various types of *SAW* devices such as filters, resonators, and sensors have been reported by researchers and are used in many industries and electronic equipment [16]. They basically consist of and are employing an electro-acoustic interdigital transducer (*IDT*) fabricated on a piezoelectric substrate [16,18], for transmitting and receiving acoustical waves. An *IDT* consists of metallic comb-shaped electrodes of sub-micron thickness normally placed over the surface. That *IDT* mass loading effect on the *SAW* resonance has close similarities to the *SAW* phenomenon on the backing under current consideration. In the literature, a reduction of *SAW* velocity was investigated caused by the mass loading of metal *IDT*, and also as an additional structured mass' along the wave propagation on the surface.

Typically metal *IDT*s fabricated over the substrate introduce *secondary* effects such as *re-emission* (reflection) the waves. Due to the *IDT* mass load effect the *SAW* phase velocity reduces from the *ideal* (unloaded) velocity. The *IDT* fabricated over the substrate affects mostly the surface wave velocity and the bulk wave velocity is not much affected since the bulk waves propagate interior to the device

substrate. The velocity (c_b) of BAW is normally two times as higher the SAW velocity (c_{saw}) [3,17] in the same material (in the backing under current consideration).

The decrease in *SAW* phase velocity caused by mass load of *IDT*-like rod can be estimated using an elementary structural unit. For the lowest first fundamental *SAW* resonance, using a correction factor based on the spring-mass model, it can be wrote

$$F_{saw} \approx \frac{c_{saw}}{A} \frac{1}{\sqrt{1 + (\sim L/A)}}$$
, (4)

where A = p + b is equal to the SAW "natural" wavelength. As an estimate, typically in practice $L \sim 2A$, then $F_L \approx \frac{c_L}{2L}$, where F_L is the piezo-rod $\lambda/2$ resonance frequency, and c_L is the longitudinal speed of sound in piezoceramic, so that typically in general $F_{saw} \approx 0.4 F_L$, and lies close to the $\lambda/4$ -type ineffective piezocomposite resonance (see Table II). To the contrary, it is a SAW resonance on the backing boundary, sophisticated with an "appendant" mass of the piezo-rods.

For the rod extensional fundamental frequency higher than the lowest SAW resonance on backing $(F_L > F_{sawl})$, for simplicity considering the non-loaded condition as a reference, the rod influence on the actual SAW resonance is determined by the loading rod mass. Just note that in the opposite case, with the rod fundamental frequency lower the SAW resonance $(F_L < F_{sawl})$, the rod influence on the SAW resonance is determined by the loading rod elasticity, not its mass [19]. The resonance modes in the transitional frequency area have a dispersion character typical for modes' coupling (Figure 3). As a reference point, the "ideal" (free surface) SAW harmonic resonances, tied to the composite basic structural configuration, are determined as $F_{saw,N} \approx c_{saw}/\lambda$, with N = A = p + b as the periodic piezocomposite pitch integer multiple (harmonic order N = 1,2,3...) to the SAW wavelength.

SAW fundamental resonance frequency variation for small loading masses (3D-configuration) can be described also as in [18,19]

$$\frac{\Delta F}{F_{saw}} \approx -2F_{saw} \frac{m}{Z_b} \approx -2 \frac{c_{saw}}{\lambda} \frac{L \rho_{pe}}{\rho_b c_b} \approx -(\sim 0.7) \frac{\rho_{pe}}{\rho_b} \frac{L}{p+b} , \qquad (5)$$

where m is the rod mass per base square, Z_b and c_b are the cell acoustic impedance and bulk SOS of the backing. It particularly follows that the SAW resonance frequency vs. relative rod's width is near *constant*, as supported by the simulation results presented in Figure 2.

The latter effect partly refers to the $\operatorname{spring}(k)$ -mass(M) model, where the resonance is determined as $F_{res} \propto \sqrt{k/M}$. Then the frequency effect of mass variation (loading), with a parameter of loading mass (M) relative to the effective mass of moving unit cell in the SAW, can be estimated as

$$\frac{\Delta F}{F_{saw}} \approx \frac{1}{\sqrt{1 + \frac{M}{\rho_b \,\lambda^3}}} - 1 \approx \frac{1}{\sqrt{1 + \frac{\rho_{pe} \, p^2 \, L}{\rho_b \, (p+b)^3}}} - 1 \approx -0.5 \, \frac{\rho_{pe}}{\rho_b} \, \frac{L}{p+b} \, \frac{p^2}{(p+b)^2} \quad . \tag{6}$$

Extending expression (5), including larger connected mass', it can be expressed as:

$$F_{saw,N} \approx N \frac{c_{saw}}{p+b} \frac{1}{\sqrt{1 + \frac{\rho_{pe}}{\rho_b} \frac{L}{(p+b)} N}}$$
 (7)

There are several particular conclusions to be noticed. The rod length derivative of the resonance frequency $(\Delta F/F)/\Delta L \sim N$ is proportional to the SAW harmonic order, that agrees well with the simulation data of Figure 3. If the rod resonance is taken as $F_L \approx \frac{c_L}{2L}$, and the backing acoustical impedance Z_b is some lower than that in piezoceramic, then the condition of maximum modes coupling with equal frequencies $F_{saw,N} = F_L$ is as follows

$$\left(2N\frac{c_{Saw}}{c_L}\frac{L}{(p+b)}\right)^2 = 1 + \frac{\rho_{pe}}{\rho_b}\frac{L}{(p+b)}N.$$
 (8)

If $a=2\frac{c_{saw}}{c_L}$; $t=\frac{\rho_{pe}}{\rho_b}$, then $\frac{L}{(p+b)}N=\frac{t+\sqrt{t^2+4a^2}}{2a^2}$, so that for high-order SAW harmonics with

$$N >> 1$$
 the condition $\frac{L}{p+b}\Big|_{\substack{max. modes \\ coupling}} \sim 1/N$ is required for the high-order modes coupling.

Based on the estimation and simulation results, for the structure of piezo-rods connected to backing, several features of vibrational performance can be noticed related to the characteristic structural parameters:

- The *SAW* resonance frequency(s) is inversely proportional to the composite pitch. Caused particularly by the kerf increase, with the fixed rod width and length, all *SAW* resonance branches are going down, in respect to the piezo-rod thickness mode frequency in piezocomposite. The strongest *SAW* modes are when located below the thickness (rod length) resonance, and with descending *SAW* resonance intensities when above it.
- When the rod width and kerf are *varying simultaneously* and in-opposite directions, like a wider rod width under fixed pitch, then both (e.g. lowest) *SAW* and length rod resonance frequencies remain relatively constant.
- When the kerf is getting much less than the rod width (or under the kerf-to-pitch ratio reaching zero), the *SAW* resonances disappear, mostly due to the mass loading effect the total resonance spectra is getting closer to that of the elementary monolithic rod, or plate, connected to the backing of the same transverse cross-section.
- When the rod height varies, with all other structural parameters fixed, the *SAW* resonances disappear with increased rods, mostly due to the mass loading effect.

2.5. Consecutive FEA Simulation of Air-coupled 1-3 Piezocomposite Transducer with SAW Resonance on Backing

COMSOL *FEA* simulation was conducted to analyze the expected effects with *SAW* resonance on the transducer backing, with step-by-step adding structural elements into the total rods-epoxy-backing-matching transducer structure – a respective elementary cell in the 2D simulation model is shown in Figure 5.

The consecutive transformation of the electrical admittance spectra in the regular periodic piezocomposite assembly is shown in Figure 6. The matching-to-air boundary condition was set just through the air acoustical loading impedance.

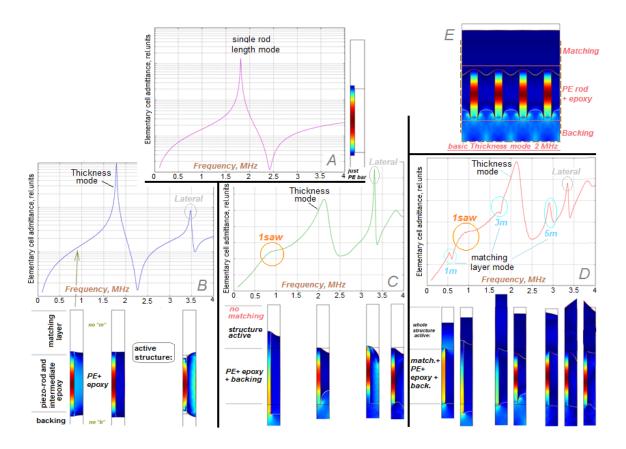


Figure 6. FEA simulation of the electrical admittance and stress, with consecutive structural complexity of the piezocomposite transducer in 2D model with a periodic elementary cell containing: a single free rod (*A*); just 1-3 piezocomposite plate (*B*); piezocomposite connected to the backing (*C*); full assembly with matching layer (*D*). The spectra includes the fundamental thickness mode (E), piezocomposite lateral, backing boundary SAW and matching layer resonances, at the different assembly stages. Backing 4 mm; rod L 0.76 mm; pitch 250 μm; pillar 70 μm.

A detailed description of the effects is based on the vibrational analysis of a rod connected (bonded) to the infinite boundary (backing). It determines specific features of the rod vibration, different from the case when the transverse cross-areas of the piezoelement and connected to it backing are finite and equal.

The basic stages and corresponding conclusions in the analysis of the data presented in Figure 6 are as follows. A free piezo-rod has its own length fundamental resonance (A). Then, in a piezoelement structured into a piezocomposite, additionally to that basic resonance, a higher frequency lateral mode appears (B) [10]. And then most importantly, the piezocomposite plate just connected to the backing provides an extra low-frequency vibrational mode identified as a SAW resonance on the backing boundary with piezo-rods array (C). Finally, connecting the matching layer with its $\lambda/4$ -resonance provides maximum transceiver efficiency - when it coincides with particularly the *SAW* resonance (D). In that totally assembled piezocomposite transducer, the thickness resonance mode per se has a conventional character of motion of the parts (E).

3. Experimental Performance of a Transceiver with piezoelectric 1-3 Composite Bonded to the Backing Layer – SAW Mode

3.1. Impedance Characteristics

For the SAW effect demonstration and performance comparison, two fully assembled transducers were built - one with a traditional monolithic piezo-plate, and the second with a regular periodic 1-3 piezocomposite (see Table I, sample #2), both with close overall dimensions and thickness resonances. The backing in both cases is a plate made of tungsten powder/chips mixed with epoxy; and the matching layer is a mixture of glass-bubbles with epoxy, with its thickness corresponding to the $\lambda/4$ resonance at 500 kHz, close to the expected SAW resonance on the backing boundary. As shown in Figure 7, the transducer with monolithic piezo-plate demonstrates a relatively smooth resistance characteristic, without any specific resonances at low frequencies up to the fundamental thickness mode. In the case of 1-3 piezocomposite there is a characteristic peak at near 600...700 kHz identified as the backing SAW resonance, gained by the fundamental $\lambda/4$ matching layer resonance tuned to the same frequency, with its higher-order multiple overtones seen as well.

To preliminary evaluate the transducer acoustical sensitivity, a simple express method was used with a rubber put on the front matching layer, and then the recorded resonance peak resistance variation was used as a measure of acoustical efficiency, as demonstrated in Figure 7 with multiple curves in the yellow area, reflecting different level of rubber damping. In the latter case, the most sensitive resonance is the $\lambda/4$ matching layer resonance located/coinciding with interpreted as a *SAW* resonance on the backing. The $\lambda/4$ matching layer resonance coupled to the backing *SAW* of the 1-3 piezocomposite plate bonded to the backing is deeply damped with quality factor near several units Q

~ 3...5, and demonstrates relatively strong sensitivity. Further experiments supported that conclusion with more details.

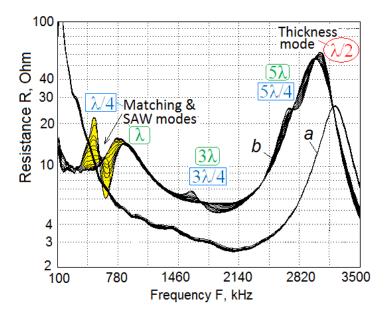


Figure 7. Comparative characteristics of the resonance vibrational modes for a conventional homogenous piezoelement (a) and regular piezocomposite (b) with close overall dimensions, all are in the fully assembled transducers with backing and matching (at 500 kHz). The resonator OD 30 x H 0.72 mm, (b) piezocomposite sample 2 (Table 1), (a) same overall dimensions conventional piezoplate. Yellow area – most effective front surface sensitivity.

For the same sample #2 as in Figure 7, the full impedance loops of the basic thickness and SAW resonances are shown in Figure 8, demonstrating a deeply damped character of the $\lambda/4$ matching layer resonance coupled to the backing SAW. A specific of the SAW resonance is that its active resistance is close to the minimum transducer resistance (real part of the impedance), including its thickness resonance.

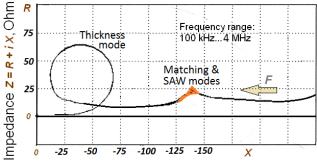


Figure 8. Resonance characteristics in the frequency range 0.1...4 MHz including the low-frequency SAW and basic thickness vibrational modes for a regular periodic piezocomposite transducer with backing and matching layer (500 kHz). The resonator OD 30 x H 0.72 mm, sample #2.

3.2. Backing SAW Resonance Mode Strength in 1-3 Piezocomposite Transceiver

Three piezocomposites with different structural configurations were used in the experiments, as presented in Table I. The impedance Z characteristics just of the free original piezocomposite plates are relatively smooth below the fundamental thickness resonance, which is proportional to the plate's inverse thickness (1), and with some planar mode low-frequency weak "ripple", as shown in Figure 9 A. Being attached to the backing, the 1-3 regular periodic piezocomposites demonstrate complex spectra, with additional SAW fundamental mode at near 600 kHz (Figure 9 B). The data are in a good agreement with the simulation results and theoretical interpretation given to the SAW resonance, including the mass loading effect.

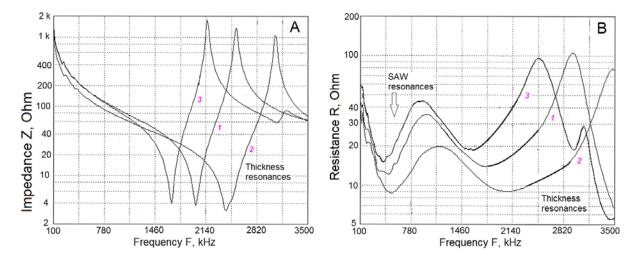


Figure 9. Effects of the backing in piezocomposites with three different thickness': experimental electrical impedance Z for initial free piezocomposite plates (A) and resistance R for those piezocomposites just attached to the 4mm backing (B); all samples as in Table 1.

The resistance R of the complex impedance Z characteristic was chosen as a parameter better reflecting the SAW resonance behavior with deep damping. In the experiments, the backing with quality factor $Q \sim 5$ (a mixture of tungsten powder and chips with epoxy), was connected to the piezocomposite plate through a thin layer of hard epoxy – on this stage just before connecting a front matching layer.

As the acoustical strength of the SAW effect is highly sensitive to the mechanical bonding between 1-3 piezocomposite plate and backing, to demonstrate the transient resonance characteristics caused by the bonding conditions, a piezocomposite plate was put on the backing with freshly prepared glue (LOCTITE hard epoxy). And then the resistance R characteristics transformation in the glue curing process was further under monitoring. As shown in Figure 10, initially the piezocomposite just exhibits some damping on its basic $\lambda/2$ thickness resonance (as it's loaded with a lower-impedance

epoxy liquid), then upon the bonding epoxy getting harder and solid, its intensity is further going down, while to the contrary the new SAW resonance aroused and strengthened. A novel resonance with near 2 times lower frequency is getting stronger, reaching its maximum strength when the bonding epoxy is completely cured. This resonance is identified as a SAW vibrational type occurred on the backing boundary in the regular 1-3 piezocomposite, being out of the piezoelectric rods resonance in the structure at that frequency.

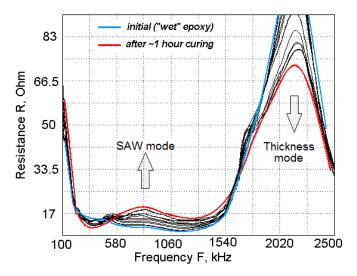


Figure 10. Comparative transient transformation of the low-frequency SAW vibrational mode resonance under curing of epoxy (increased bonding strength) between piezocomposite element and backing. The resonator OD 30 x H 0.72 mm. Rising SAW mode is identified as an additional resonance in the structure.

Further extended experiments showed that the *SAW* resonance, and related effects, is provided only by a regular structure - the rods must be well arranged in the "dice & fill" 1-3 piezocomposite.

The SAW mode on the backing boundary neither occurs with irregularly oriented and distributed rods, when piezoelectric fibers (a bunch) are imbedded randomly in a polymer matrix in the form of a 1-3 composite (for ex. [13,14]), nor with a homogeneous conventional piezoelectric monolithic plate/rod, specifically when the cross area for both piezoelement and backing are equal. The letter is demonstrated in Figure 7, where both comparative characteristics for 1-3 piezocomposite and conventional piezoelectric plates are presented. Note that as was found, the strongest SAW resonance is provided with a relatively hard bonding epoxy between piezoelectric structure and backing supporting a collective character of motion in the multiple regular elementary cells.

3.3. Pulse-Echo Characteristics of 1-3 Piezocomposite Transceiver in Air operating at the SAW Mode on the Backing Boundary

Total transducer assembly was built with regular periodic 1-3 piezocomposites, backing and a single front $\lambda/4$ matching layer. For generating and analyzing the acoustical waves, pulse/receiver P/R

Panametrics 5077 (GE Panametrics, Lago Vista, TX) was used as a basic device, with a general purpose oscilloscope, such as Agilent 54622 (Agilent Technologies, Santa Clara, CA) to provide pulse visualization. Impedance/Gain-Phase HP 4194A Analyzer (Hewlett-Packard, Palo Alto, CA) was used for electrical immittance measurements, with 0.5 V max output. All measurements were performed in air under atmospheric pressure (air-coupled), at room temperature.

Acoustical signal strength measurements and Q-factor estimation were done with a transducer excited with a spike single cycle and 2% duty cycle tone burst. The SAW operational mode on the backing boundary as excited in the experiments, showed expected promising results, specifically for the applications where a relatively low Q-factor $\sim 3...10$ is required for higher resolution.

The transducer has been tested in both pulse-echo and pitch-catch modes. For a typical pulse recorded (Figure 11), with a spike input 400 V_{0-p} , the signal output was ~70 m V_{p-p} , with a transfer level -75 dB for a full 4" pitch-catch transmitter-receiver path in atmospheric air. The pulse has relatively short rise- and decay-times, with near 2 to 3 periods each.

In the SPL measurements with a calibrated microphone, a transmitter with SAW resonance on backing at near 150 kHz (sample D, Table I) with similar input conditions demonstrated ~50 Pa effective acoustical pressure (pulse maximum), or SPL 128dB re 20 μ Pa @4"@150kHz@400V_{0-p}. Then an estimated transceiver peak pressure sensitivity is roughly near -83dB re 1V @1 μ bar.

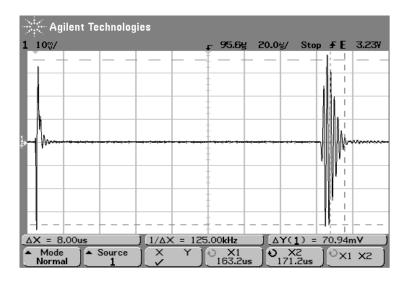


Figure 11. Typical pulse waveform of the novel piezocomposite transducer with the SAW operational vibrational mode on the backing in 1-3 piezocomposite resonator at a relatively low-frequency ~600 kHz. Sample #2 (Table I), pitch-catch with 4" distance.

The performance of the proposed novel transducer is compared with that of a conventional air-coupled transducers operating in a similar frequency range. Compared to one of the industrial prototypes such as Sonda 007CX; QMI AS400C [20] at 400 kHz, the piezocomposite transducer with

SAW vibrational mode demonstrates similar acoustical performance suitable for practical applications, with usable sensitivity, efficiency and short pulse duration, being well-damped.

When the transducer was prepared in its full structure, effectiveness of its sensitivity performance was first evaluated with a simple express method - applying some soft material (like a rubber) on the front transducer surface and recording its electrical impedance variations under that influence. It's supposed that largest impedance variation is at the resonance frequency(-ies) of maximum transducer sensitivity of the acoustically active front surface. As shown in Figure 12, and supported by the data of Figure 7, the optimal transceiver operational frequency is near 500 kHz, with the sensitivity intensity dynamic range near $\sim 2.4...4$ times between the maximum-to-minimum peaks, with effective Q-factor $\sim 3...5$ [15].

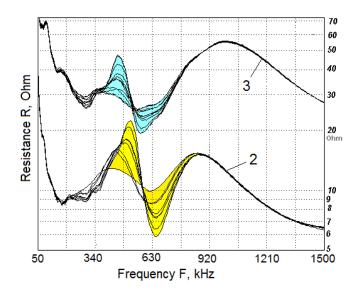


Figure 12. Effectiveness of the front surface sensitivity characteristic of the low-frequency SAW vibrational mode resonance on backing for two regular periodic 1-3 piezocomposite elements (#2 and 3, Table 1), with the backing close to optimal and guarter-lambda matching layer. The area of highest sensitivity is shown in color.

The effect of matching layer thickness on the piezocomposite transducer performance during its operation based on the backing SAW has a typical resonance character - as was shown in an experiment, there is the optimal matching layer thickness providing maximum signal amplitude (Figure 13). It corresponds to the $\lambda/4$ matching layer thickness at the 500 kHz operation frequency of the backing SAW resonance.

The experiment was made in the pulse-echo regime, with a 2.5" SS target and the transducer using regular periodic 1-3 piezocomposite as presented in Table I, sample 2.

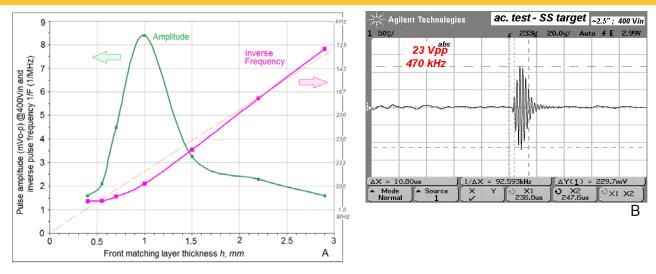


Figure 13. Experimental data of the pulse strength under variation of the matching layer thickness in a piezocomposite transceiver with SAW operational mode on backing: A –signal amplitude and effective frequency of the pulse; B – pulse waveform with maximum output.

4. Discussion

The Rayleigh *SAW* resonance discovered in the piezocomposite backing boundary is a powerful and promising tool for the air (gas) transducer operation. It requires first of all rigid piezorod-to-backing bonding and regular periodic structure array, at least with several adjacent piezorods.

The condition for the relatively low-frequency *SAW* resonance, in the first approximation, requires for the rods periodicity to be an integer multiple to the surface acoustical wavelength on the backing boundary, which is predominantly of a Rayleigh (Stoneley) type. The surface wave physically occurs in a layer of backing, involving bonding epoxy between the rods and backing, with an effective depth in the backing comparable to the SAW wavelength. As was found, just a hard epoxy for bonding provides maximum acoustical signal strength. Moreover, for a piezocomposite element, it also provides better bonding quality on the boundary between epoxy filler and backing, with its extra acoustical coupling.

For the described *SAW* resonance effect, a regular periodic 1-3 (or 2-2) structure is needed. Its acoustical performance is a collective effect between at least several adjacent piezo-rods with a regular space translation. In the latter case, a wider transducer bandwidth, or even a phase array, can be achieved under special structural apodization if needed.

As was predicted and experimentally verified, no effective (i.e. collective) backing *SAW* resonance effect exists with varying planar rod dimensions, and/or irregularly distributed rods, when the piezoelectric rods (or a fibers bunch) are imbedded randomly in a polymer matrix in the form of a 1-3 composite (Figure 14). Note that the *SAW* effect does not exist also with a homogeneous

conventional piezoelectric monolithic plate/rod, specifically when the overall cross area for both the piezoelement and backing are equal.

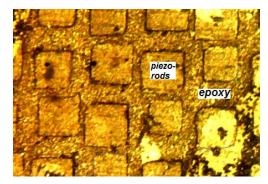


Figure 14. An example of irregular 1-3 piezocomposite structure that causes ineffective SAW behavior on the bonding boundary: piezo-rods of different cross-sections, and their irregular (random) relative planar disposition.

As a possible variant, a monolithic piezoplate, with the bonding side grated, can be used instead of conventional rods, or alternatively such regular grating can be put on the backing boundary surface, as shown in Figure 15. The regular voids can be filled with a relatively soft material. The vibrational operation of that structure is very similar to the separated piezo-rods considered, just with larger effective mass for the "rods".

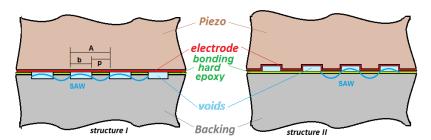


Figure 15. Alternative structures with a boundary SAW resonance effect using monolithic bulk piezoplate - with shallow regular grooves at least on one of the boundary surfaces, piezoceramic and/or backing, as shown in structure I and II.

5. Conclusions

Surface acoustical wave on the damping backing boundary of a piezocomposite transceiver is an effective resonating mechanism which was discovered and then specifically investigated for the air (gas) transducer application. As was shown, the novel *SAW* vibrational mode with a significantly lower natural resonance in a regular periodic 1-3 piezocomposite transducer is an effective mechanism for the air-coupled ultrasonic applications. Collective *SAW* synchronized motion of the backing surface on the piezocomposite bonding boundary requires a regular (translational periodicity) rods' structure to provide strong resonance efficiency of the transceiver, to the contrary of the conventional thickness mode approach with resonating piezoelectric rods.

Along with less transducer weight (a thinner piezocomposite required), the described *SAW* resonance effect on the backing effectively "bridges the gap" in the transitional intermediate frequency range 100...700 kHz, problematic for the conventional airborne transceiver designs typically with the unwanted strongly coupled thickness and planar resonances in a monolithic piezoelement with an aspect ratio close to 1. For practical evaluation, a typical piezocomposite structure (2...3 MHz thickness resonance) was used, and it provided -64 dB pulse-echo signal strength at the backing *SAW* 500 kHz resonance in air with a wide bandwidth and Q-factor near 5, being equipped with the backing and corresponding quarter-lambda conventional matching layer at that frequency.

The benefits of using relatively slow *SAW* waves are smaller volume and weight of the total transducer, clear well-damped spectra, with lower excitation voltage required because of a thinner piezoelement needed. It provides acceptable acoustical efficiency and sensitivity, along with low transducer *Q*-factor and consequently relatively wide *BW*. This is the case where a strong regular periodicity of the 1-3 piezocomposite greatly matters for successful transceiver operation.

References

- 1. Lee, H.J.;.Zhang, S. Design of low-loss 1-3 piezoelectric composites for high-power applications. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **2012**, vol. 59, no. 9, pp. 1969-1975.
- 2. Smith, W.A.; Auld, B.A. Modeling 1-3 composite piezoelectrics: thickness mode oscillations. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **1991**, vol. 38, no.1, pp.40-47.
- 3. CeramTec GmbH, *Piezo Ceramic Components. Materials*, Ebersbach, Germany. **2019**: https://www.ceramtec.com/ceramic-materials/soft-pzt/
- 4. Hayward, G.; Gachagan, A. An evaluation of 1-3 connectivity composite transducers for air-coupled ultrasonic applications. J. Acoust. Soc. Am., **1996**, vol. 99, no. 4, pp. 2148-2157.
- 5. Mezheritsky, A.V. Invariants of coupling coefficients in piezoceramics. *IEEE Trans. Ultrason., Ferroel.* and Freq. Control, **2003**, vol. 50, no. 12, pp. 1742-1751.
- 6. Toda, M. New type of matching layer for air-coupled ultrasonic transducer. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **2002**, vol. 49, no. 7, pp. 972-976.
- 7. Gerton, D.; Casula, O., etc. Theoretical and experimental investigations of lateral modes in 1-3 Piezocomposites. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **1997**, vol. 44, no. 3, pp. 643-650.
- 8. Gururaja, T.R.; Schulze, W.A., etc. Piezoelectric composite materials for ultrasonic transducer Applications. Part I: resonant modes of vibration of PZT rod-polymer composites. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **1985**, vol. SU-32, no. 4, pp. 481-498.
- 9. Robertson, D.; Hayward, G., etc. Comparison of the frequency and physical nature of the lowest order parasitic mode in single crystal and ceramic 2-2 and 1-3 piezoelectric composite transducer. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **2006**, vol. 53, no.8, pp.1503-1512.
- 10. Certon, D.; Patat, F., etc. Two dimensional modeling of lateral modes in 1-3 piezocomposites. in *IEEE Ultrasonics Symp. Proc.*, **1994**, pp. 991-994.
- 11. Rouffaud, R.; Hladky-Hennion, A.-C.; Pham-Thi, M.; Bantignies, C.; Levassort, F. Influence of 1-3 piezocomposite fabrications on lateral modes. in *IEEE Ultrasonics Symp. Proc.*, **2012**, pp. 1-4.
- 12. Smart Materials. 1-3 Random Fiber Piezocomposites. Sarasota, FL. **2019**: https://www.smart-material.com/13Crand-product-main.html.

- 13. Qi, W.; Cao, W. Finite element analysis of periodic and random 2-2 piezocomposite transducers with finite dimensions. *IEEE Trans. Ultrason., Ferroel. and Freg. Control*, **1997**, vol. 44, no.5, pp.1168-1171.
- 14. Ruddle, J.; Cass, R.; Mohammadi, F. Advanced cerametrics harvests clean energy. *American Ceramic Society Bulletin*, **2007**, vol. 86, no. 10, pp. 24-27.
- 15. Mezheritsky, A.V. A method of "weak resonance" for piezoelectrics quality factor and coupling coefficient measurements. *IEEE Trans. Ultrason., Ferroel. and Freq. Control*, **2005**, vol. 52, no. 11, pp. 2120-2130.
- Ramakrisnan, N.; Nemade, H.B.; Palathinkal, R.P. Resonant frequency characteristics of a SAW device attached to resonating micropillars. Sensors, 2012, vol. 12, pp. 3789-3797.
- 17. Alhazmi, H.; Guldiken, R. Quantification of Bolt Tension by Surface Acoustic Waves: An Experimentally Verified Simulation Study. *Acoustics* **2019**, *1*, 794-807.
- 18. Pomorska, A.; Schukin, D.; Hammond, R., etc. Positive frequency shift observed upon absorbing icro-sized solid objects to a quartz crystal microbalance from the liquid phase. *Anal. Chem.*, **2010**, vol. 82, pp. 2237-2242.
- 19. Plessky, V.; Koskela, J. Coupling-of-modes analysis of SAW devices. *Int. J. High Speed Electronics and Systems*, **2000**, vol. 10, no. 4, pp. 867–947.
- 20. Quality Material Inspection (QMI), Inc, Huntington Beach, CA, **2019**: http://www.gmi-inc.com/AIRSCAN.htm .

Author(s) Contributions: The only author fully contributed to formal analysis, investigation, data curation writing and editing.

Funding: This research received no external funding.

Conflicts of Interest: The author declare no conflict of interest.

Nomenclature

pillar – piezoceramic rod (with length L and width p) in piezocomposite structure, with its length-to-width aspect ratio (L/p);

kerf – cut/dicing width (b), particularly for a orthogonal (square) array filled with epoxy;

pitch (kerf plus pillar width $A \equiv b + p$) – a shift between piezocomposite cuts, or structural periodicity;

CW – continuous wave;

SAW – surface acoustic wave;

BAW – bulk acoustic wave;

BW – frequency bandwidth;

PE – piezoelement;

SOS – speed of sound;

SPL – sound pressure level;

SS – metal stainless steel reflecting target in acoustic test;

IDT – interdigital transducer;

F – current frequency;

R – electrical resistance R = ReZ component of impedance Z;

saw1-Y and saw1-D — fundamental (lowest) SAW resonance determined from the piezocomposite electrical admittance $(Y = 1/\mathbb{Z})$ and from the piezo-rod top displacement (D);

N - SAW harmonic order 1,2,3...;

L1-Y and L1-D – fundamental (lowest) longitudinal resonance of the piezo-rod in piezocomposite;

 F_{sawN-Y} and F_{sawN-D} – SAW resonance harmonics (N = 1,2,3...) determined from the piezocomposite electrical admittance (Y = 1/Z) and from the piezo-rod top displacement (D);

 λ – wave-length, used for SAW, matching layer and piezorod longitudinal resonances;

 F_L – fundamental $\lambda/2$ rod resonance frequency;

 $n_{\rm m}$ – matching layer vibrational mode of the $n_{\rm m}$ = 1,3,5... order;

 Z_{ac} , Z_0 , Z_R , Z_m , Z_b – acoustical impedance, particularly of piezoelectric material, loading fluid, front matching and backing layers, respectively;

c, c_b , c_{saw} , c_L - speed of sound (SOS), particularly bulk and SAW on backing, and longitudinal SOS in piezorod;

 ρ , ρ_b , ρ_{PE} – density, particularly backing and piezoceramic density;

E, σ , Q – Young module, Poisson coefficient and material quality factor;

M, m – mass in the vibrational models.