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A BEM-based actuator disk model for wind turbine wakes considering atmospheric stability

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Abstract: Atmospheric stability affects wind turbine wakes significantly. High-fidelity approaches such as large eddy simulations (LES) with the actuator line (AL) model which predicts detailed wake structures, fail to be applied in wind farm engineering applications due to its expensive cost. In order to make wind farm simulations computationally affordable, this paper proposes a new actuator disk model (AD) based on the blade element method (BEM) and combined with Reynolds-averaged Navier–Stokes equations (RANS) to model turbine wakes under different atmospheric stability conditions. In the proposed model, the upstream reference velocity is firstly estimated from the disk averaged velocity based on the one-dimensional momentum theory, and then is used to evaluate the rotor speed to calculate blade element forces. Flow similarity functions based on field measurement are applied to limit wind shear under strongly stable conditions, and turbulence source terms are added to take the buoyant-driven effects into consideration. Results from the new AD model are compared with field measurements and results from the AD model based on the thrust coefficient, the BEM-AD model with classical similarity functions and a high-fidelity LES approach. The results show that the proposed method is better in simulating wakes under various atmospheric stability conditions than the other AD models and has a similar performance to the high-fidelity LES approach however in much lower computational cost.

Keywords: wind turbine; wake; atmospheric stability; actuator disk; BEM

1. Introduction

As a measure of turbulence exchanges in the atmospheric surface layer (ASL), atmospheric stability can significantly affect the wind turbine wake deficit and its recovery rate. In general, turbulence exchanges between the wake and the atmosphere are depressed under stable conditions. High wake deficits and slow wake recovery thus are usually observed in the stable stratification boundary layer [1,2]. Since wakes play critical roles in wind farm energy production and the fatigue loads of wind turbines, there is an increasing interest in studying effects of atmospheric stability on wakes and developing wake models considering atmospheric stability.

The impact of atmospheric stability on wakes are widely observed in wind tunnel measurements [3–5] and field experiments [1,2,6,7]. According to the wind tunnel measurements in Chamorro et al. [3], the stronger inlet wind shear in the stable case leads to a slightly stronger turbulence intensity and extends the region of enhanced turbulence intensity from a distance of about 4–5.5 rotor diameters to 3 and 6 rotor diameters downwind of the turbine location. In Zhang et al. [4], a 15% smaller velocity deficit at the wake center, a more rapid momentum recovery due to an enhanced radial momentum transport, a 20% higher peak turbulence intensity were observed in the unstable case, as compared to

33 the wake of the same wind turbine in a neutral boundary layer. Hancock et al. [5], Iungo et al. [6] and
34 Han et al. [1] found that the velocity deficit decreases more slowly under stable conditions and more
35 quickly under unstable conditions. According to the field experiments of wakes using lidars [2,7] or
36 masts [1], the double-bell near-wake shape was clearly observed in very stable cases. Measurements
37 showed the strong dependence of the vertical wake propagation on the atmospheric stability [2,7]: the
38 wake tends to follow the terrain down the ridge with a maximum inclination of -28° in very stable cases
39 or be advected upwards by up to 29° above the horizontal plane in unstable cases. Those observations
40 suggest that atmospheric stability should be considered for improved wake models and predictions of
41 wind power harvesting.

42 In the current stage, high-fidelity large eddy simulations with the actuator line model (AL-LES)
43 [8,9] or the actuator disk model (AD-LES) [2,10] are widely used to study the structure and dynamics
44 of wind turbine wakes in varying stability cases. "Precursor" atmospheric simulations of the thermal
45 stratified layer are usually required to provide ambient conditions applied in the simulations of
46 wind turbine wakes [2,11]. Energy equations are modeled while buoyancy and Coriolis force due to
47 planetary rotation are added in the momentum transport equations. There are no special corrections
48 of turbulence terms for wake modeling as compared to that in RANS. As Peña [12] argued, proper
49 characterizations of the shifts in length and velocity scales of the turbulence are required to model a
50 non-neutral atmosphere, which results in the special treatment of the turbulence Prandtl number in the
51 Smagorinsky subgrid-scale model of the Simulator for Wind Farm Applications (SOWFA) [11]. The
52 LES approaches allow to capture the near wake structure, resolve the interactions of tip vortices with
53 large-scale eddies of the ambient flow and wake meandering. Fair agreements between measurements
54 and the LES predictions were presented in literature [2,13]. However, the high-fidelity approaches
55 are computational expensive for wind energy engineering applications. In the AL-LES modeling of
56 rotors, the tip of blade is not allowed to travel more than one mesh cell over a time step [14]. Noting
57 the high resolution mesh in the AL simulations to capture the detail wake structure, this limitation of
58 AL-LES method results in a very small time step (about 0.01 s). Although an AD-LES method with a
59 coarse spatial resolution could bring a relaxed time step and make large-scale wind farm simulations
60 computationally affordable for research purposes, the high-fidelity approaches still have a long way
61 for engineering applications.

62 An order-reduced alternative to the LES approach is the AD model based on the thrust coefficient
63 (CT-AD) or based on BEM (BEM-AD) in transient RANS with a large time step (about 1 s) or even
64 in steady RANS. Classical CT-AD approaches combined with turbulence corrections such as the El
65 Kasmi model [15] and the $k - \varepsilon - f_p$ model [16] are now commonly used in wake modeling [15–18]
66 and validated by numerous experiments [3,19,20] under neutral conditions. Recently, some BEM-AD
67 models [10,18] that take into account effects of the turbine blade induced flow rotation are also
68 introduced in RANS. Using BEM to distribute the blade force through the rotor provides an applicable
69 way to model wakes in yaw or pitch conditions and allows to capture the near wake more precisely than
70 the CT-AD. In non-neutral cases, the inlet boundary conditions are imposed with similarity functions
71 of atmospheric stability and special turbulence terms are added to model effects of atmospheric
72 stability on turbulence [1,21]. Model uncertainties might be raised due to the model ability to keep
73 flow homogeneity and the overestimated wind shear by the classical similarity functions. A widely
74 used turbulence model for the thermal stratified boundary layer was developed by Alinot and Masson
75 [22] by calibrating the coefficient of the buoyant terms in the ε transport equation. However, van der
76 Laan et al. [21] showed that this model cannot keep the flow homogeneity in large domain under
77 unstable conditions and thus developed a turbulence model consistent with the classical similarity
78 functions. In general, the height of surface boundary layer decreases with atmospheric stability down
79 to 10 m in very stable cases. The decreased boundary layer height limits the wind shear and results in
80 a fair lower wind shear than the one based on classical similarity functions [1,23]. In Han et al. [1], the
81 unlimited wind shear predicted by the classical similarity functions was observed in overestimating
82 wind speedup for upslope flow and underestimating it for downslope flow. Consequently, a new set

of similarity functions based on field measurements in the full range of the Richardson number is estimated to limit the wind shear in very stable conditions [1]. Han et al. [1] thus proposed the FullyRF turbulence model by extending the Laan model with those observed similarity functions.

In this paper, a BEM-based AD model by using the FullyRF turbulence model [1] is developed to simulate turbine wakes under various atmospheric stability for engineering applications. This paper mainly focuses on how the distributed blade force using BEM helps the predictions of near wake structures and the side effects of applying wind profiles with unlimited wind shear in the classical similarity functions on wake modeling. Before discussing the simulation results and measurements of wakes under varying stability conditions (Section 4), we will start with a description of the BEM-AD method and the associated turbulence model in the next section. A detail introduction of simulation cases including computational domain, meshing and boundary conditions is presented in Section 3. Finally, in Section 5 the results of wake simulations are concluded.

2. Methods

In the modeling of wind turbine wakes under different stability conditions, the following continuity, momentum and energy equations [24] are solved:

$$\frac{\partial}{\partial x_i} (\rho U_i) = 0 \quad (1)$$

$$\frac{\partial}{\partial t} (\rho U_i) + \frac{\partial}{\partial x_j} (\rho U_i U_j) = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[(\mu + \mu_t) \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \right] + S_{u,i} \quad (2)$$

$$\frac{\partial}{\partial t} (\rho \Theta) + \frac{\partial}{\partial x_i} (\rho U_i \Theta) = \frac{\partial}{\partial x_i} \left[\left(\mu + \frac{\mu_t}{\sigma_\theta} \right) \frac{\partial \Theta}{\partial x_i} \right] \quad (3)$$

where ρ is the air density, U_i is the velocity component in the x_i direction, p is the air pressure, μ is the laminar viscosity, μ_t is the turbulent viscosity, $S_{u,i}$ is the momentum term source imposed by the wind turbine rotor in the x_i direction, Θ is the potential temperature and σ_θ is the turbulent Prandtl number.

2.1. Modeling of Wind Turbine

2.1.1. Actuator Disk Model Based on Thrust Coefficient (CT-AD)

In the origin actuator disk model, the momentum source term in the x_i direction due to the thrust is uniformly distributed through the rotor:

$$S_{u,i} = -\frac{T}{V_{\text{disk}}} \frac{U_{\text{ref},i}}{U_{\text{ref}}} = -\frac{\rho C_T U_{\text{ref}} U_{\text{ref},i}}{2\Delta l} \quad (4)$$

where T is the thrust, V_{disk} is the disk volume, C_T is the thrust coefficient, U_{ref} is the upstream reference velocity at hub height, $U_{\text{ref},i}$ is the component velocity U_{ref} of in x_i direction, Δl is the disk depth and the negative sign represents the drag effects of thrust on the flow.

For the upstream flow disturbed by a complex terrain or wind turbine wakes, U_{ref} is unknown and difficult to be evaluated from the local flow directly. According to the one-dimensional momentum theory and ignoring the wind shear, the reference velocity is a function of the rotor-averaged velocity U_{disk} :

$$U_{\text{disk}} = (1 - a_B) U_{\text{ref}} \quad (5)$$

in which the induced factor a_B [25] is related to the thrust coefficient C_T by

$$a_B = \begin{cases} \frac{1}{2}(1 - \sqrt{1 - C_T}), & C_T \leq \frac{8}{9} \\ \frac{C_T - 4a_c^2}{4(1 - 2a_c)}, & C_T > \frac{8}{9} \end{cases} \quad (6)$$

113 where $a_c = 1/3$.

114 In simulations, the disk-averaged velocity U_{disk} is calculated by averaging the local velocity in
115 the disk region, then is applied to estimate the upstream reference velocity U_{ref} based on Equations (5)
116 and (6) and the momentum source $S_{u,i}$ based on Equation (4), respectively.

117 2.1.2. Actuator Disk Model Based on Blade Element Method (BEM-AD)

118 In the BEM-based actuator disk model, the rotor plane consists of N actuator lines and each of the
119 actuator lines is splitted into M element sections (Fig. 1). The element section collects the local velocity
120 and the rotor speed Ω to calculate the element force and applies this force in the neighbor cells of the
121 element section. The reference velocity is firstly assessed from the disk-averaged velocity and then is
122 applied to evaluate the rotor speed.

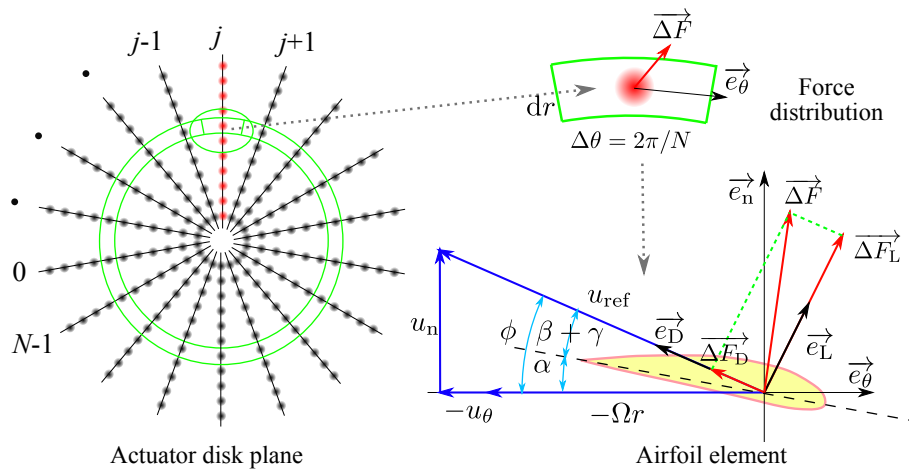


Figure 1. Schematics of the BEM-based acutor disk model.

123 By transforming the local velocity at the blade element into polar velocity components (u_r , u_θ ,
124 u_n), the force of the blade element is:

$$\vec{\Delta F} = \frac{B\Delta\theta}{4\pi} \rho u_{\text{rel}}^2 c (C_L \vec{e}_L + C_D \vec{e}_D) \Delta r \quad (7)$$

125 where B is the number of the blades and c is the chord length, Δr is the length of the element section.
126 The drag coefficient of the element section, C_D and its lift coefficient C_L which are functions of the attack
127 anlg α , are estimated from XFOIL [26] and then corrected by three-dimensional rotational effects of the
128 blades based on Du et al. [27]. According to Fig. 1, $\alpha = \phi - (\beta + \gamma)$ where $\phi = \arctan [u_n / (\Omega r + u_\theta)]$
129 is the flow angle, β is the blade installation anlg and γ is the pitch angle.

130 The element force is distributed across neighbor cells. The force added to a cell is calculated by:

$$\vec{\Delta F}_{\text{cell}} = \sum_i^{N \cdot M} \frac{1}{s^3 \pi^{3/2}} \exp\left(-\frac{s_i^2}{s^2}\right) \vec{\Delta F}_i F_{\text{tip}} F_{\text{hub}} \Delta V_{\text{cell}} \quad (8)$$

131 where s_i the distance of the i -th element to the cell and s is the cut-off length scale that takes a value
132 between 2 and 3 cell sizes [28]. F_{tip} and F_{hub} are the Prandtl tip loss and hub loss functions [29]:

$$F_{\text{tip}} = \frac{2}{\pi} \arccos \left[\exp \left(-\frac{B(R-r)}{2r \sin \phi} \right) \right] \quad (9)$$

$$F_{\text{hub}} = \frac{2}{\pi} \arccos \left[\exp \left(-\frac{B(r-R_{\text{hub}})}{2r \sin \phi} \right) \right] \quad (10)$$

133 where R is the rotor radius, R_{hub} is the hub radius, r is the radial distance of the element to the rotor
134 center.

135 2.2. Turbulence modeling

136 To model effects of atmospheric stability on wind turbine wakes, the extended $k - \varepsilon$ model
137 proposed by Han [1] is applied, in which the eddy viscosity is calibrated with atmospheric stability:

$$\mu_t = \frac{\phi_\varepsilon}{\phi_m \phi_k^2} \cdot C_\mu \frac{k^2}{\varepsilon} \quad (11)$$

138 where $C_\mu = 0.033$, k and ε are the turbulence kinetic energy (TKE) and its dissipation (TKD),
139 respectively. $\phi_m \equiv \frac{\kappa z}{u_*} \frac{\partial U}{\partial z}$ is the dimensionless gradient of wind speed, $\phi_k \equiv \sqrt{C_\mu} k / u_*^2$ is the
140 dimensionless TKE and $\phi_\varepsilon \equiv \kappa z \varepsilon / u_*^3$ is the dimensionless TKD where $\kappa = 0.4$, the friction speed
141 $u_* = \left(\overline{u'w'^2} + \overline{v'w'^2} \right)^{1/4}$, u' , v' and w' are the fluctuations of the longitudinal, lateral and vertical
142 velocity components. The dimensionless gradient of potential temperature, ϕ_h also can be defined as
143 $\frac{\kappa z}{\theta_*} \frac{\partial \Theta}{\partial z}$ where the scaling temperature $\theta_* = -\overline{w'\theta'}/u_*$ and θ' is the fluctuation of potential temperature.

144 In the extended approach, the turbulence kinetic energy k and its dissipation ε are modeled by

$$\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_i}(\rho U_i k) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + \mathcal{P} + \mathcal{B} - \varepsilon - S_{k,SBL} \quad (12)$$

$$\frac{\partial}{\partial t}(\rho \varepsilon) + \frac{\partial}{\partial x_i}(\rho U_i \varepsilon) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right] + (C_{\varepsilon 1} \mathcal{P} - C_{\varepsilon 2} \varepsilon) \frac{\varepsilon}{k} + S_{\varepsilon,ABL} + S_{\varepsilon,wake} \quad (13)$$

where $\sigma_k = 1.0$, $\sigma_\varepsilon = 1.3$, $C_{\varepsilon 2} = 1.92$, and $C_{\varepsilon 1} = 1.24$. \mathcal{P} and \mathcal{B} are the TKE source production due to
shear and buoyancy:

$$\mathcal{P} = \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_j}{\partial x_i} \quad (14)$$

$$\mathcal{B} = - \frac{g_i}{\Theta_0} \frac{\mu_t}{\sigma_\theta} \frac{\partial \Theta}{\partial x_i} \quad (15)$$

145 where Θ_0 is the potential temperature on the ground.

The source term $S_{\varepsilon,wake}$ is applied to correct the fast wake recovery of the standard $k - \varepsilon$ model
[15]:

$$S_{\varepsilon,wake} = C_{4\varepsilon} \frac{\mathcal{P}^2}{\rho k} \quad (16)$$

146 where $C_{4\varepsilon} = 0.37$.

147 The TKE source term $S_{\varepsilon,ABL}$ is applied to maintain flow velocity profiles under non-neutral
148 conditions over a flat ground:

$$S_{k,ABL} = \left\{ \phi_m - \phi_\varepsilon - \zeta + \frac{C_\mu^{-1/2} \kappa^2}{\sigma_k} \frac{\zeta^2}{\phi_m} \left[\phi_k'' - \frac{\phi_k' \phi_m'}{\phi_m} + \frac{\phi_k'}{\zeta} \right] \right\} \frac{u_*^3}{\kappa z} \quad (17)$$

where $\phi' = \partial \phi / \partial \zeta$, $\phi'' = \partial^2 \phi / \partial \zeta^2$ for $\phi \in \{\phi_k, \phi_m, \phi_\varepsilon\}$. The TKD source term due to buoyancy, $S_{\varepsilon,ABL}$
is modelled by:

$$S_{\varepsilon,ABL} = C_{\varepsilon 3} \mathcal{B} \frac{\varepsilon}{k} \quad (18)$$

149 in which

$$C_{\varepsilon 3} = \frac{C_{\varepsilon 1} \phi_m - C_{\varepsilon 2} \phi_\varepsilon}{\zeta} + \frac{C_\mu^{-1/2} \kappa^2}{\sigma_\varepsilon} \frac{\phi_k}{\phi_m} \left[\frac{\zeta \phi_\varepsilon''}{\phi_\varepsilon} - \frac{\zeta \phi_\varepsilon' \phi_m'}{\phi_\varepsilon \phi_m} - \frac{\phi_\varepsilon'}{\phi_\varepsilon} + \frac{\phi_m'}{\phi_m} + \frac{1}{\zeta} \right] \quad (19)$$

150 According to the Monin-Obukhov similarity theory [30], ϕ_m , ϕ_h , ϕ_ε and ϕ_k are functions of the
151 stability parameter $\zeta = z/L$ where L is the Obukhov length:

$$L \equiv -\frac{u_*^3}{\kappa \frac{g}{\Theta} w' \theta'} \quad (20)$$

152 Based on measurements of flows over flat terrain in the ASL, the classical similarity functions commonly
153 used in literature are given as [31,32]:

$$\phi_{m,cls}(\zeta) = \begin{cases} (1 - \gamma_m \zeta)^{-1/4} & -2 < \zeta < 0 \\ 1 + \beta_m \zeta & 0 < \zeta < 1 \end{cases} \quad (21)$$

$$\phi_{h,cls}(\zeta) = \begin{cases} \chi(1 - \gamma_h \zeta)^{-1/2} & -2 < \zeta < 0 \\ \chi + \beta_h \zeta & 0 < \zeta < 1 \end{cases} \quad (22)$$

$$\phi_{\epsilon,cls}(\zeta) = \begin{cases} 1 - \zeta & \zeta < 0 \\ \phi_{m,cls} - \zeta & \zeta > 0 \end{cases} \quad (23)$$

$$\phi_{k,cls}(\zeta) = \sqrt{\frac{\phi_{\epsilon,cls}(\zeta)}{\phi_{m,cls}(\zeta)}} \quad (24)$$

154 where $\gamma_m = \gamma_h = 16$, $\beta_m = \beta_h = 5$ [32] and $\chi = 0.9$.

155 According to Grachev et al. [33], the applicability of the Monin-Obukhov similarity theory is only
156 valid for the flux Richardson number R_f below 0.25. In the full range of R_f , the similarity functions
157 based on a field measurement reads [1]:

$$\phi_{m,exp}(\zeta) = \begin{cases} \frac{2+100\zeta^2}{2+200\zeta^2} & \zeta < 0 \\ (1 + 6\zeta)^{\frac{1+16\zeta^2}{1+40\zeta^2}} & \zeta > 0 \end{cases} \quad (25)$$

$$\sigma_{\theta,exp}(\zeta) = \begin{cases} 0.9(1 - 2\zeta)^{-1/2} & \zeta < 0 \\ 0.9(1 + 6\zeta)^{-1/8} & \zeta > 0 \end{cases} \quad (26)$$

158 where $\sigma_{\theta,exp}(\zeta) = \phi_{h,exp}(\zeta) / \phi_{m,exp}(\zeta)$.

159 Eqs. (17) and (19) can be adapted with the classical similarity functions (the Laan model [21]) or
160 the similarity functions in the full range of R_f (the FullyRF model [1]).

Table 1. Turbulence model for non-neutral conditions.

Turbulence Model	Similarity functions
Laan	classical
FullyRF	$\phi_{m,exp}, \phi_{h,exp}, \phi_{k,cls}, \phi_{\epsilon,cls}$

161 3. Simulation Details

162 3.1. The H93-2.0MW Turbine Test Case

163 Since the FullyRF similarity functions are derived from field measurements in the Jingbian wind
164 farm where measurements of wind turbine wakes were also carried out [34], the wake models are firstly
165 assessed in this area. In the experimental campaign, two masts numbered M1 and M3 are installed
166 near a H93-2.0MW wind turbine (#14) to capture the wake profiles. On the masts, cup anemometers
167 were installed at 30 m, 50 m and 70 m. Measurements of the sonic anemometer installed on M1 at 30 m
168 were applied to estimate the Obukhov length L . Three cases (1) M1 \rightarrow #14 \rightarrow M3, (2) M3 \rightarrow #14 \rightarrow
169 M1 and (3) #14 \rightarrow M1 are studied with longitudinal distances to the upstream wind turbine rotors
170 of 1.25D, 2.15D and 5D where D is the rotor diameter. As M3 is near #14 in case (1) and the terrain

171 slope in the wake of cases (2) and (3) is below 2° , the wake is expected to follow the terrain. This is
 172 especially for some stable cases as observed in [7]. The average wind direction misalignment at M3
 173 and #14 due to the complex terrain that is about 2.05° in unstable cases, 3.75° in neutral cases and 3.94°
 174 in stable cases, is also applied to correct the inflow wind direction at M3. Since the flow tends to follow
 175 the terrain under stable conditions, this misalignment of wind direction is enhanced by atmospheric
 176 stability. Taking these terrain effects into considerations, simulations of wakes of the H93-2.0MW wind
 177 turbine in flat terrain are expected to be able compared with measurements.

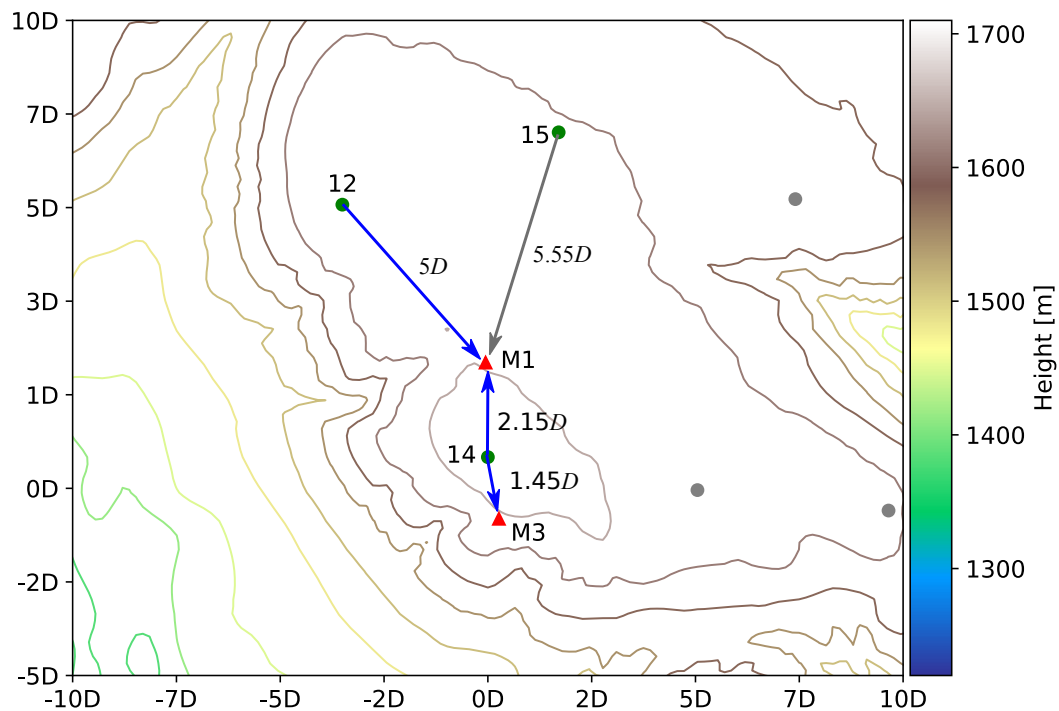


Figure 2. Complex terrain around the H93-2.0MW wind turbine (#14).

178 The power curves and thrust coefficient curves of the H93-2.0MW turbine are shown in Fig. 3.
 179 Results of BEM are compared with manufacture curves, measurements and other CFD results. The
 180 observed power curve and the thrust coefficient curve measured from the strain gauges installed at
 181 the bottom of the tower [34] show good agreement with the performance curves from BEM. The large
 182 discrepancy of the thrust coefficient from the manufacture and from the BEM calculations are probably
 183 due to different implementations of BEM.

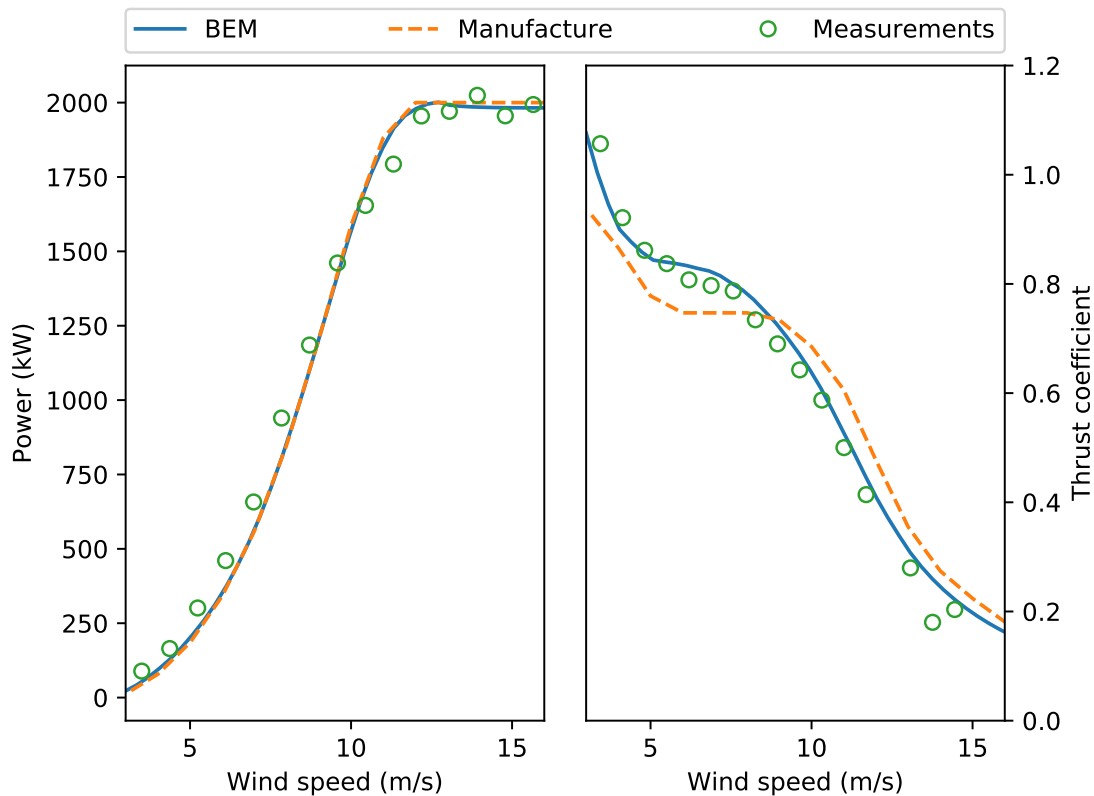


Figure 3. Performance curves of the H93-2.0MW turbine.

184 3.2. The NTK500/41 Turbine Test Case

185 Another test case is based on lidar measurements of wakes of a NTK500/41 turbine at the Risø
 186 Campus test site of DTU in Denmark [2] and is used to check whether the FullyRF similarity functions
 187 are site-dependent in wake simulations. The NTK500/41 turbine is a stall-regulated 500 kW wind
 188 turbine equipped with LM 19.1 m blades and its rotor speed is fixed at 27.1 rpm. A pulsed lidar
 189 mounted on a platform at the rear of the nacelle, pointed its laser downstream the turbine where a
 190 constant downhill slope of about 0.3% was observed. Inlet meteorological properties such as wind
 191 speed, wind direction, air temperature and atmospheric pressure were measured from a 57 m tall
 192 meteorological mast located upstream the turbine. Details of test cases are listed in Tab. 2. LES
 193 simulations of wakes of the NTK500/41 turbine [2] will be compared with the models proposed in this
 194 paper.

Table 2. Details of test cases.

Wind turbine	D (m)	H (m)	Measurements	Wake range
H93-2.0MW	93	67	Mast measurement	$1.25 D$, $2.15 D$ and $5 D$
NTK500/41	41	36	Lidar scanning and LES simulations	$1 D$ to $8 D$

195 For the NTK500/41 turbine, the power curve from BEM calculations has good agreements
 196 with the RANS computations in fully turbulence model [35] below 15 m/s, with LES computations
 197 [2] and measurements [35] below 10 m/s (Fig. 4). The measured thrust coefficient from strain
 198 gauges measurements [36] is shown to have good agreements with the BEM computations and the
 199 LES computations above 6 m/s. The thrust coefficient applied in all simulations is from the BEM
 200 computations.

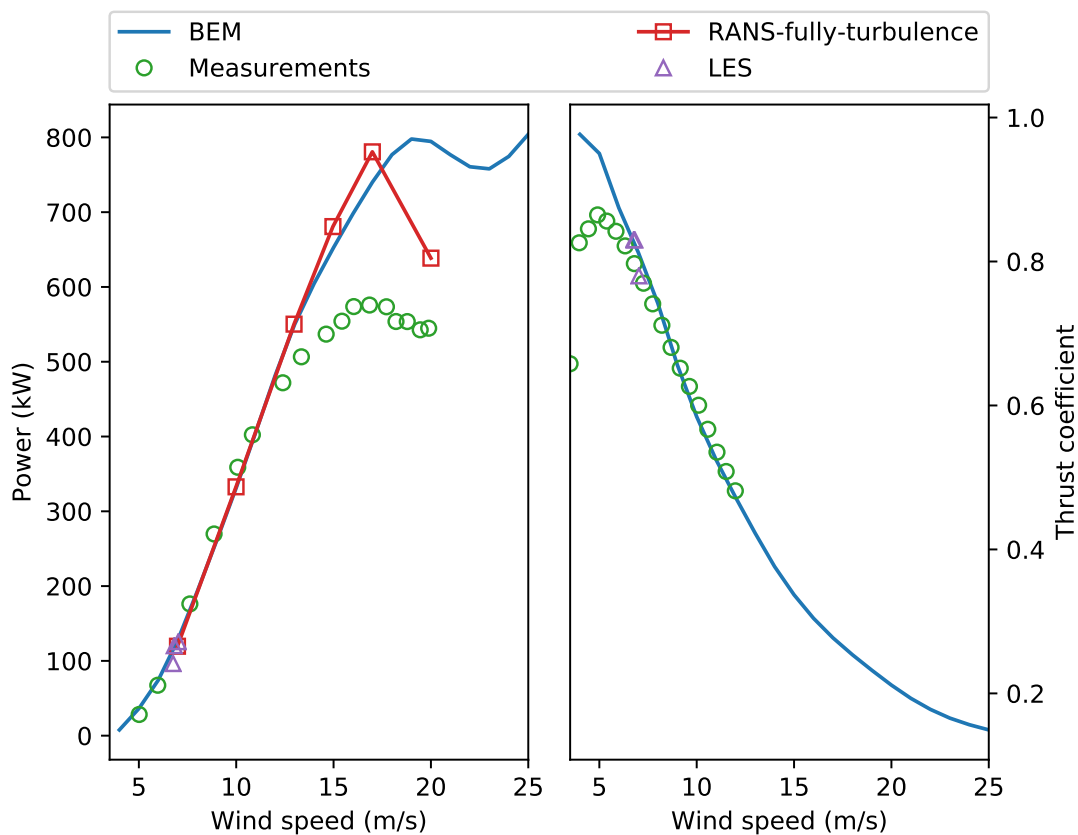


Figure 4. Performance curves of the NTK500/41 turbine.

201 3.3. Computational Domain and Meshing

202 The computational domain has a length of $20D$, a width of $10D$ and a height of $10D$ (Fig. 5). All the
 203 AD models are implemented in the open source CFD software OpenFOAM [37]. The background mesh
 204 whose refinement level is 0, consists of $200 \times 60 \times 60$ for the H93-2.0MW turbine and $100 \times 60 \times 60$ for
 205 the NTK500/41 turbine in length, width and height. The vertical grids are clustered near the ground
 206 and the first cell height above the ground is set to be $5.5 z_0$ [1]. The mesh is refined around the disk
 207 region to ensure 80 to 120 cells through the rotor diameter.

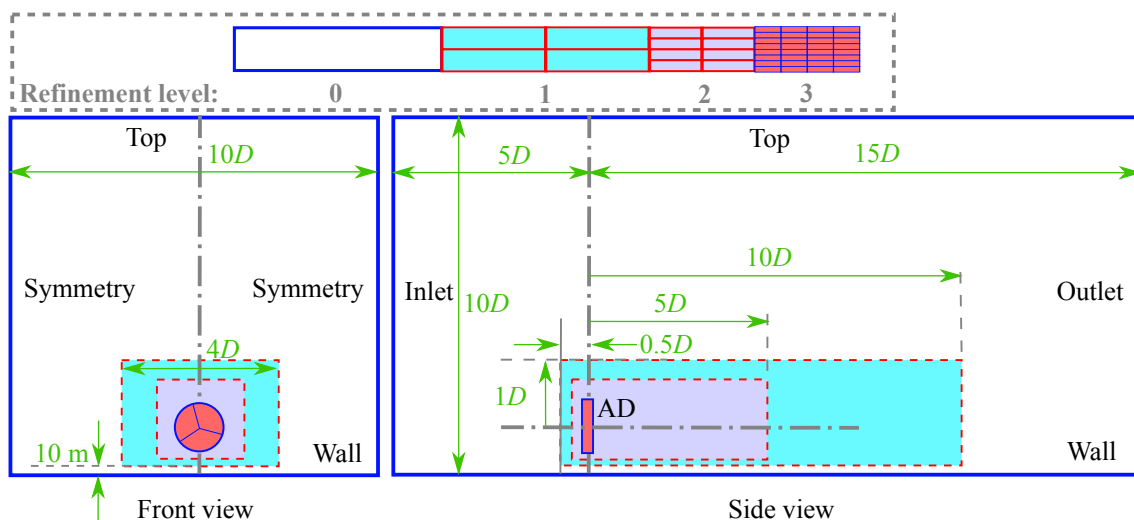


Figure 5. Computational domain and meshing settings.

208 3.4. Boundary Conditions

209 The boundary conditions consistent with similarity functions are applied to modeling the
 210 atmospheric boundary stratification. According to Han et al. [1], we apply the following inlet
 211 conditions in wake simulations:

$$U(z) = \int_{z_0}^z \frac{u_*}{\kappa z} \phi_m \left(\frac{z}{L} \right) dz \quad (27)$$

$$\Theta(z) = \Theta_0 + \int_{z_0}^z \frac{\theta_*}{\kappa z} \phi_h \left(\frac{z}{L} \right) dz \quad (28)$$

$$\varepsilon = \frac{u_*^3}{\kappa z} \phi_\varepsilon \left(\frac{z}{L} \right) \quad (29)$$

$$k = \frac{u_*^2}{\sqrt{C_\mu}} \phi_k \left(\frac{z}{L} \right) \quad (30)$$

212 where the roughness length z_0 approximates 0.05 m for the H93-2.0MW turbine [1] and 0.095 m for the
 213 NTK500/41 turbine [2]. The vertical profiles of wind speed and potential temperature are estimated in
 214 numerical integration. Zero gradients of $U, \Theta, \varepsilon, k$ applied at the outlet. ε, k and vertical gradients of
 215 U, Θ are set to be fixed values on the top boundary [1]. And the turbulent law of wall presented in
 216 [1] is applied to the first layer of cells above the wall. The left and right sides of the computational
 217 domain are set to be symmetry.

218 The detail information of inlet conditions and resulting inlet velocity profiles are shown in Tab.
 219 3 and in Fig. 6. The inlet velocity at hub is set to be 6 m/s for the H93-2.0MW turbine and about 7
 220 m/s for the NTK500/41 turbine [2]. In the LES cases of the NTK500/41 turbine, the time-varying
 221 vertical structure of the whole atmosphere boundary layer was simulated based on the second GABLS
 222 experiment [38] to assess the mean and velocity profiles imposed at the inlet of the computational
 223 domain. In general, the surface-layer height decreases with atmospheric stability and thus limits the
 224 wind shear [39] under stable conditions. For very stable cases, the classical similarity functions strongly
 225 over-predicts the wind speed from 40 m upwards [12], which results in large discrepancy of velocity
 226 profile from measurements and based on the classical similarity functions and a larger shear compared
 227 with the measurements. The FullyRF similarity functions limit the wind shear under stable conditions
 228 and thus show good agreements with measurements. For the unstable test case, good agreements
 229 between the wind speed profile measured at the meteorological mast and the modeled inflow in LES,
 230 FullyRF model and Laan are observed, however the neutral wind profile is further off in LES.

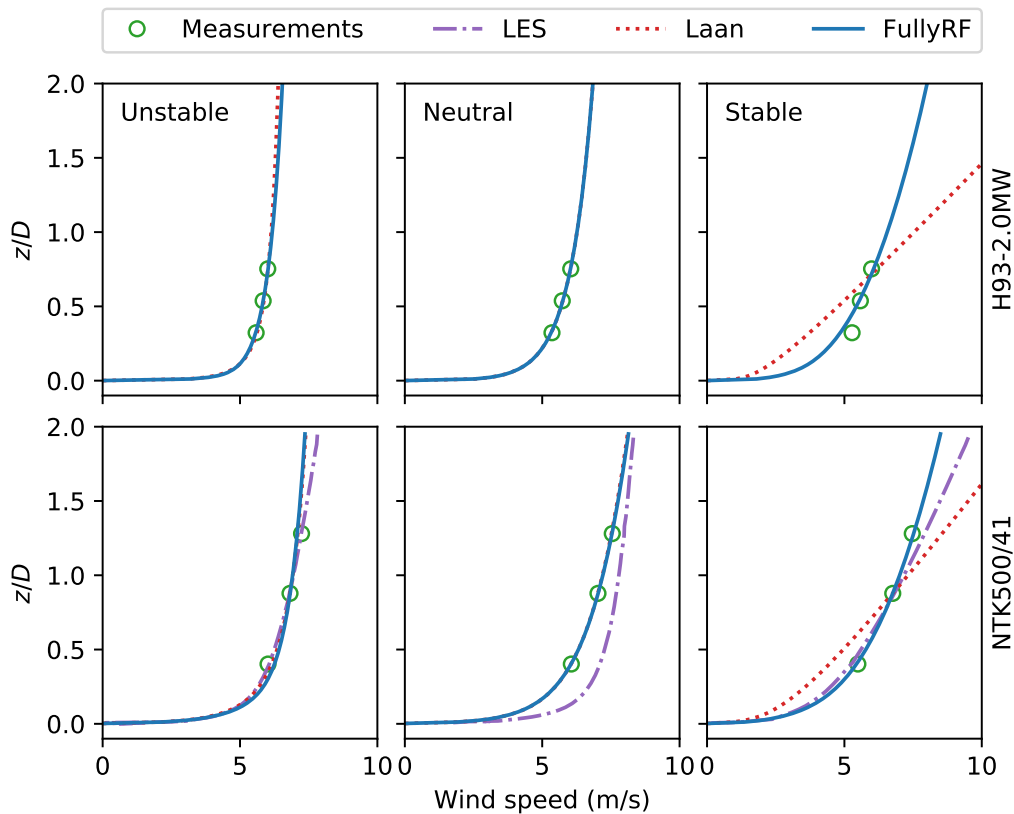


Figure 6. Inlet speed profiles.

Table 3. Details of inlet conditions.

Stability	Model	H93-2.0MW		NTK500/41		
		L (m)	u_* (m/s)	L (m)	u_* (m/s)	U_{hub} (m/s)
Unstable	All	-30	0.425	-84.8	0.525	6.82
Neutral	All	∞	0.333	1677.7	0.464	7.03
Stable	Laan	30	0.131	29.0	0.223	6.76
	FullyRF		0.222		0.316	

231 4. Results and Discussion

232 4.1. Wakes of the H93-2.0MW turbines

233 Fig. 7 to Fig. 9 show the wake profile at $1.45D$, $2.15D$ and $5D$ under different stability conditions.
 234 The significant effects of atmospheric stability that can be observed from measurements are: (1)
 235 Wake deficits increase with atmospheric stability. Higher wake deficit are observed under stable
 236 or neutral conditions than unstable conditions: the maximum wake deficit at 70 m height in $1.45D$
 237 downstream the turbine in the unstable case rises from 0.5 to 0.6 in the neutral or stable case. (2)
 238 Wake recovers much slower under neutral and stable conditions than under unstable conditions. The
 239 normalized wake velocity at 70 m height under unstable, neutral and stable conditions rises to 0.73,
 240 0.64 and 0.62, respectively. This is mainly because turbulence is depressed under stable conditions and
 241 the momentum and energy exchanges between the wake and the atmosphere are thus weakened. (3)
 242 Double-bell near-wake shape at 70 m height, probably due to a lower energy extraction around the
 243 blade root [2], are observed for neutral and stable cases at $1.45D$. A local minimum deficit occurs in the
 244 wake center and the difference between this minimum value and the local maximum deficit around
 245 the wake center ranges from 0.1 (neutral) to 0.2 (stable). (4) The double-bell near-wake shape is slightly

246 asymmetric and this asymmetric can be enhanced by atmospheric stability. More deficit is observed on
 247 the left side of the turbine when looking at the turbine from downwind [40]. The asymmetric wake is
 248 also observed by Menke et al. [7].

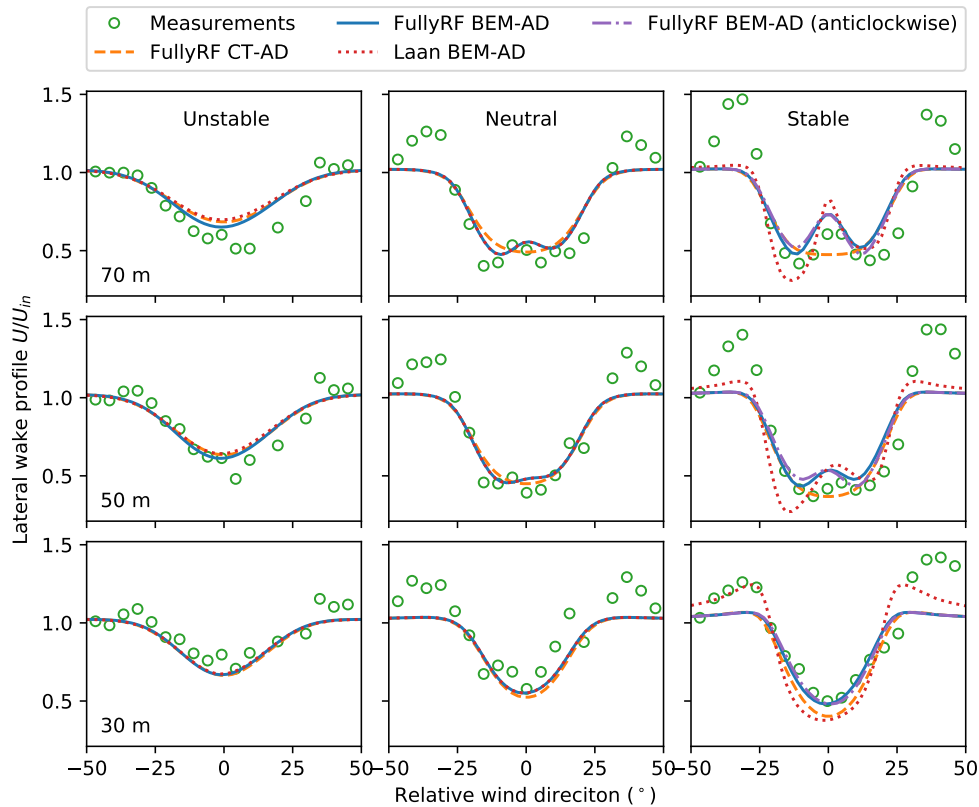


Figure 7. Wake profile at $1.45 D$ and U_{in} is the mean inlet velocity.

249 The FullyRF BEM-AD model has a good agreement with measurements except for multiple wake
 250 cases. At $1.45 D$, the inlet wind speed at M1 could be affected of wakes of #12 and #15, which results
 251 in a significant discrepancy between the measurements and simulations around the relative wind
 252 direction $\pm 30^\circ$ especially under neutral and stable conditions, see Appendix A for the definition of
 253 the relative wind direction in a wake. In the stable case, the FullyRF CT-AD model fails to predict the
 254 double-bell near-wake shape and tends to overestimate the near-wake deficit at lower part of wakes
 255 (30 m at 1.45 , 50 m and 30 m at $2.15 D$) while the Laan BEM-AD model yields an enhanced asymmetric
 256 double-bell near-wake shape and overestimates the wake deficit despite of the longitudinal distance
 257 to the rotor. Since the similarity functions of the Laan model and the FullyRF model are very similar
 258 under unstable conditions, there are no major differences on the wake deficit predictions between the
 259 three AD models.

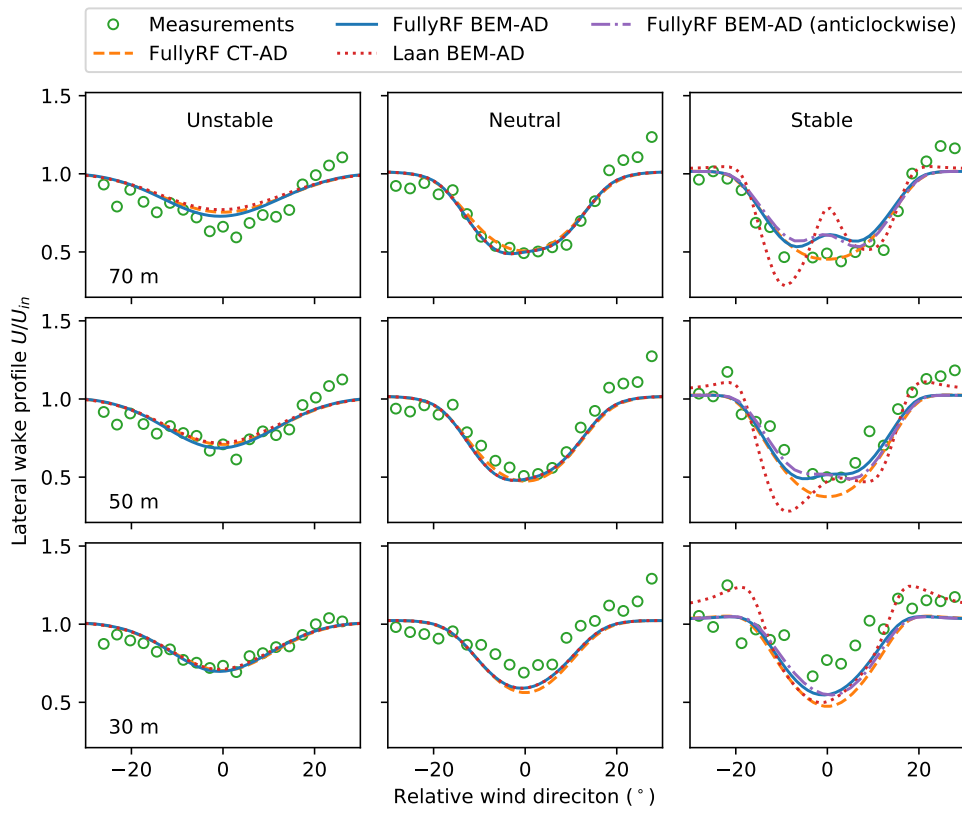


Figure 8. Wake profile at $2.15 D$

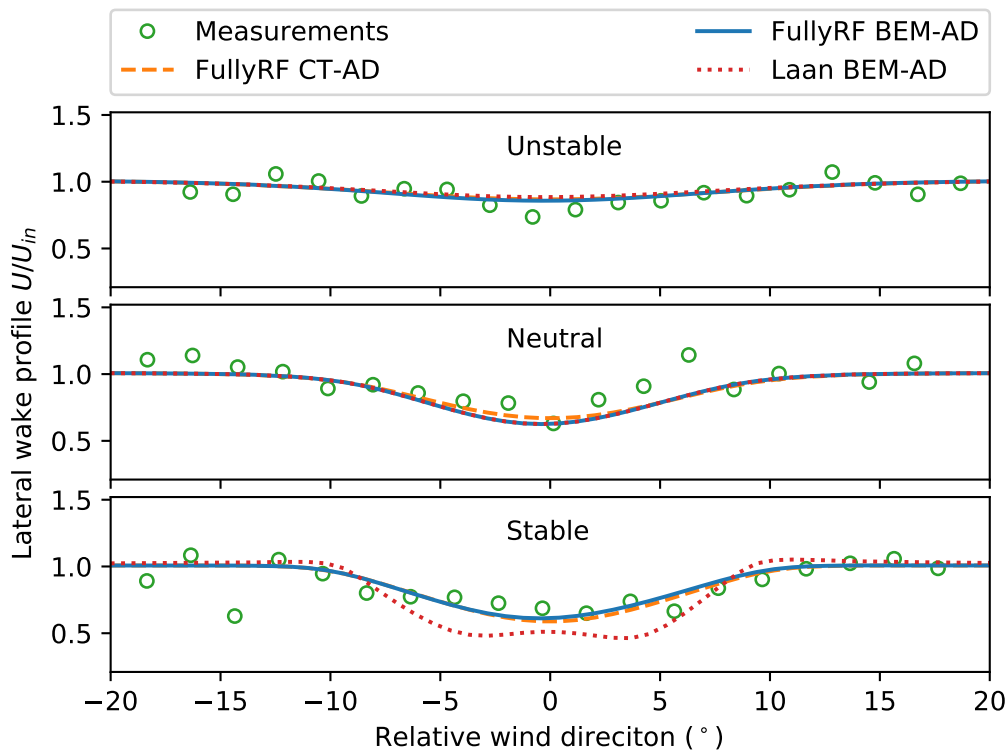


Figure 9. Wake profile at $5 D$

260 Both measurements and simulations show that the double bell is not symmetric to the wake
 261 center although it is slight in measurements. Churchfield et al. [40] relates this phenomenon to the
 262 asymmetric vertical profile due to wind shear and the wake rotation that counters the blade rotation.
 263 For the clockwise-rotation turbine, the wake rotation is anticlockwise and transfers low-speed flow
 264 from below to the left side of the wake and high-speed flow from above to the right when looking at
 265 the turbine from downwind. In this study, the high wind shear of the Laan BEM-AD model tends to
 266 enhance this asymmetry and countering the blade rotation to anticlockwise does reverses the local
 267 maximum of wake deficit from the left side to the right side of the wake.

268 The asymmetric double-bell near-wake shape is also clearly observed for the turbulence intensity
 269 increase ΔI in the wind turbine wakes at $1.45 D$ under neutral and stable conditions. The proposed
 270 model predicts the double-bell near-wake shape and underestimates the maximum increase of
 271 turbulence intensity. However, this underestimation decreases with atmospheric stability and wake
 272 recovery. In cases without the double-bell near-wake shape, the increase of turbulence intensity
 273 predicted by the AD models show good agreements with measurements.

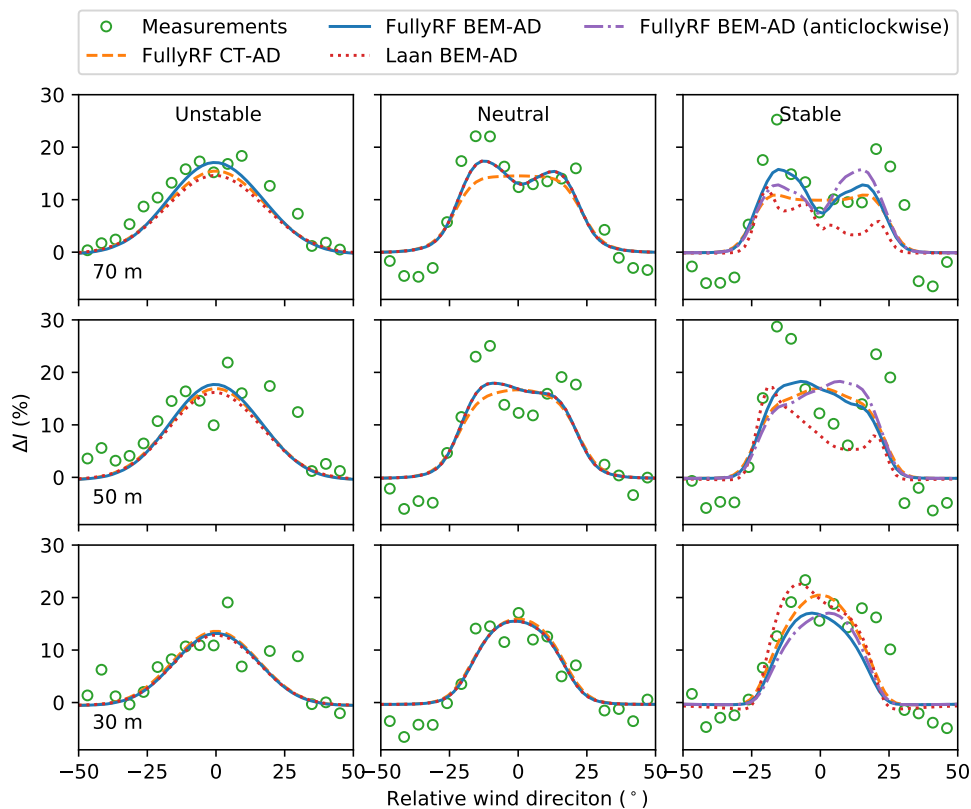


Figure 10. ΔI at $1.45 D$

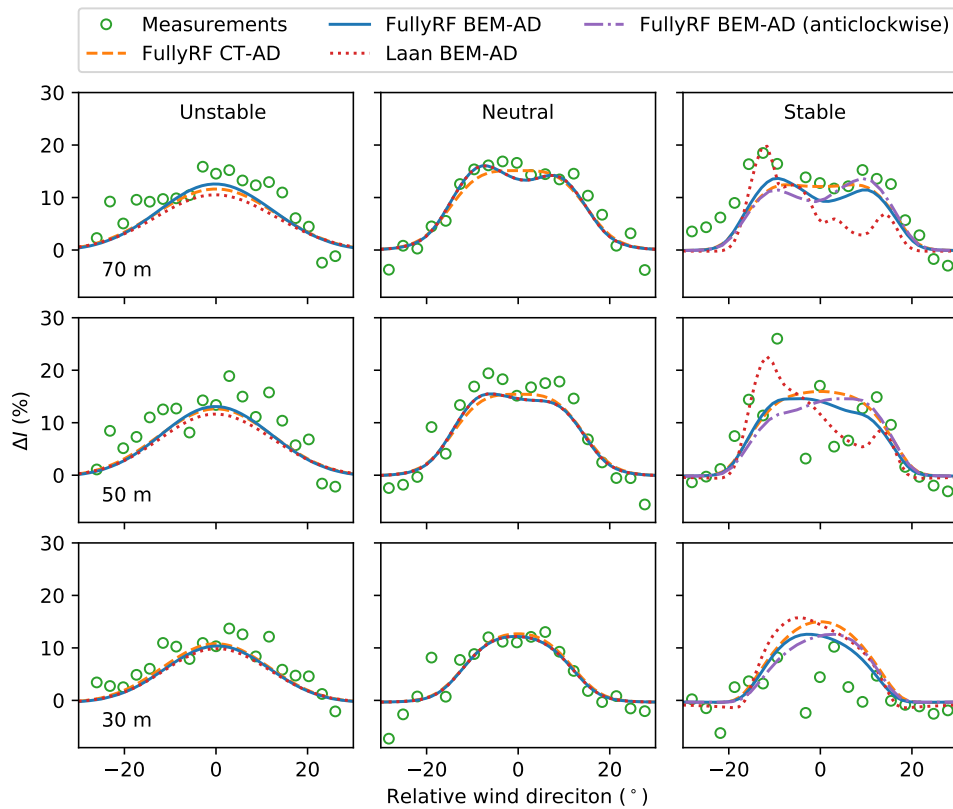
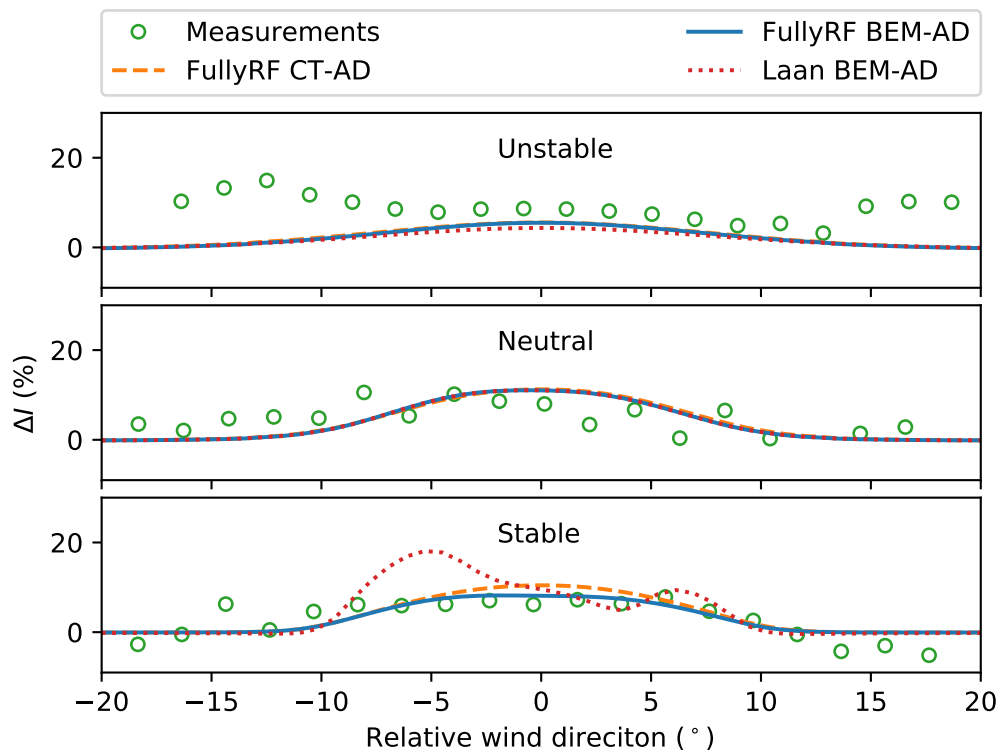


Figure 11. ΔI at $2.15 D$

274 Due to the absence of the turbulence intensity data at wind turbines, the turbulence intensity at
 275 #12 is approximated by measurements at M1 for the wake of wt12 to M1 and is set to 15%, 9%, and
 276 6% for unstable, neutral and stable conditions, respectively. At $5D$, ΔI predicted by the FullyRF AD
 277 models have good agreements with measurements while being overestimated by the Laan BEM-AD
 278 model (Fig. 12).

Figure 12. ΔI at 5 D

279 4.2. Wakes of the NTK500/41 Turbine

280 Fig. 13 shows the comparisons of normalized velocity contour in wakes under stable conditions.
 281 The simulation in LES show larger deficits in near wake and a slightly faster recovery wake than that
 282 in the proposed model, while the Laan BEM-AD model predicts a slower recovery wake with much
 283 larger deficit. For a given reference wind speed U_{hub} and a fixed roughness length z_0 , higher wind
 284 shear in the Laan model results in a smaller friction speed u_* and a lower turbulence intensity than the
 285 proposed model and thus weakens the wake recovery.

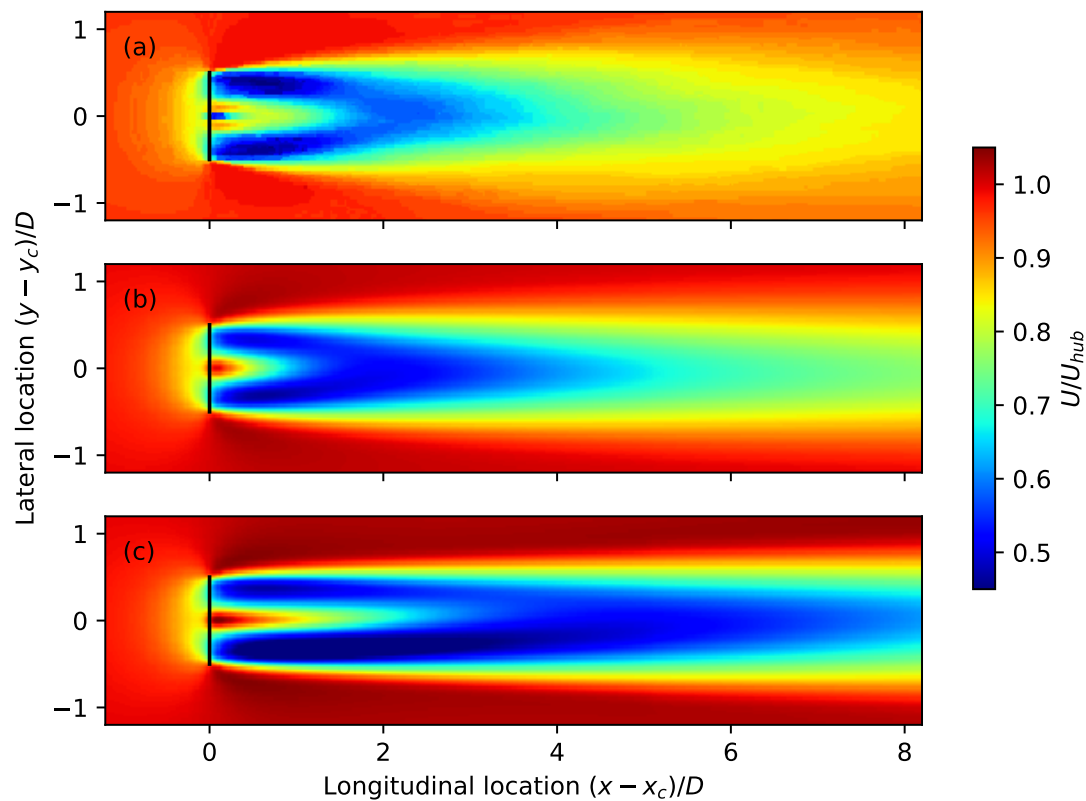


Figure 13. Normalized velocity distribution at hub height in the stable case where (x_c, y_c, z_c) is the center point of the disk.

286 As compared with measurements, the proposed model shows better performance than other AD
 287 approaches in RANS and is comparable to the LES approach (Fig. 14). For the longitudinal distance
 288 above $2D$, the Laan BEM-AD model overestimates the wake deficits in both vertical and lateral
 289 directions. Since the wake tends to follow terrain under very stable conditions, the wind speed for the
 290 longitudinal distance above $5D$ at the hub height is actually the value of 5.5 m above the wake center.
 291 Considering such an effect, the proposed model shows a good agreement with measurements at $5D$.
 292 Due to the absence of wake measured data above $5D$, it is difficult to distinguish which is better for
 293 wake predictions between the proposed model and the LES approach. The slower wake recovery over
 294 $5D$ predicted by the proposed model in Fig. 13 as compared to the LES approach, should be studied in
 295 further research.

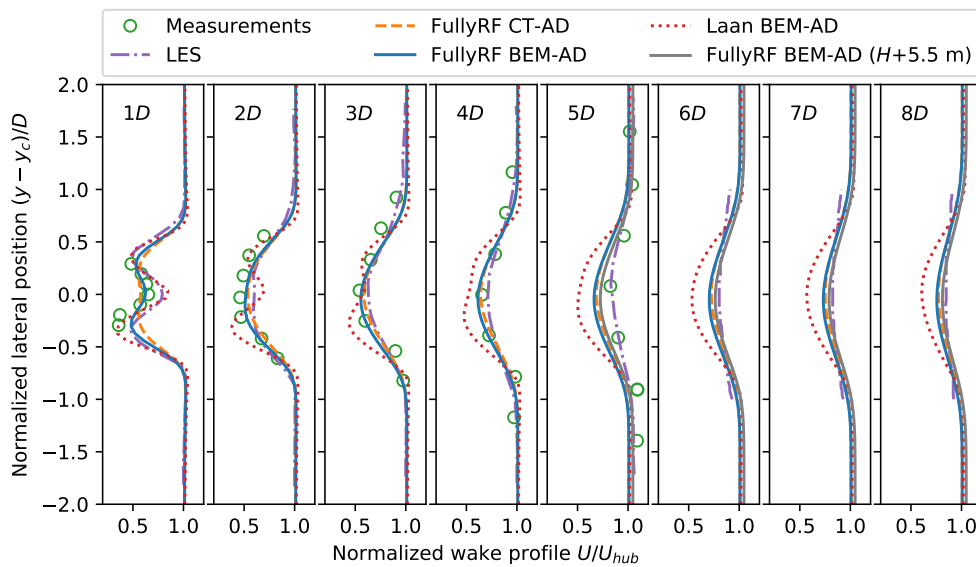


Figure 14. Wake deficit at hub height in the stable case.

296 In the stable case, the proposed model underestimates wake deficits above the hub height at
 297 1D and 2D while overestimating wake deficits below the hub height at 5D (Fig. 15). There are no
 298 major differences between the vertical wake profiles predicted by the FullyRF CT-AD model and the
 299 proposed model above 2D. However, the Laan BEM-AD model significantly overestimates wake
 300 deficits at 4D and 5D. As the LES model [2] applies slip wall conditions at the bottom of the domain
 301 without considering the roughness effect, large discrepancies of wind profile are observed near the
 302 ground.

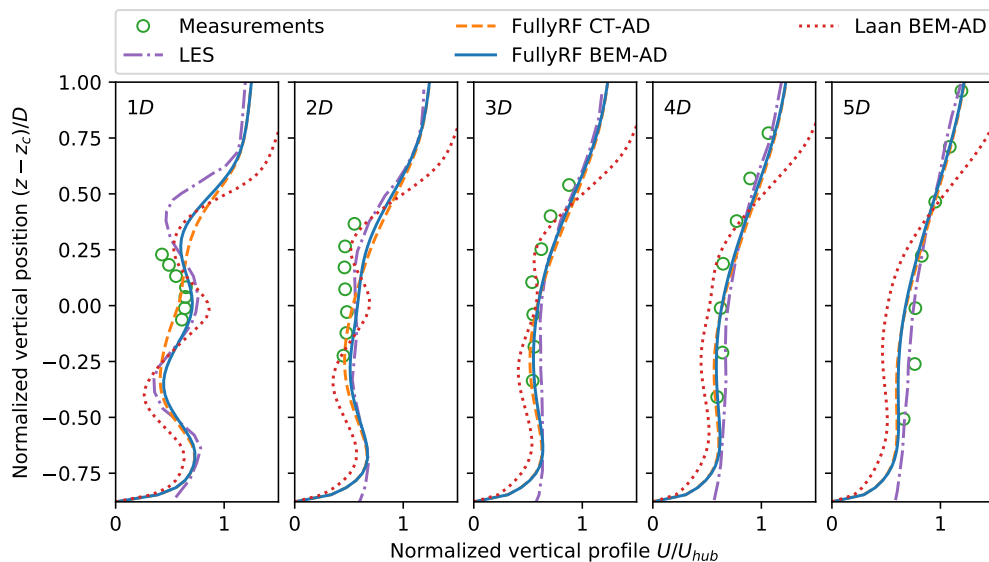


Figure 15. Vertical wake deficit profile in the stable case.

303 In the neutral case, the double-bell near-wake shape is only observed at 1D and captured by
 304 the proposed model and the LES approach (Fig. 16). Measurements show good agreements with
 305 predictions of the proposed model at 1D to 3D. Since the wake also follows the slope slightly under
 306 neutral conditions [2], the proposed model show a slightly decay recovery wake for the longitudinal
 307 distance above 4D. In Fig. 17, effects of slope are considered by rising the measured data slightly, and
 308 the predictions of the proposed model tends to approach the measurements.

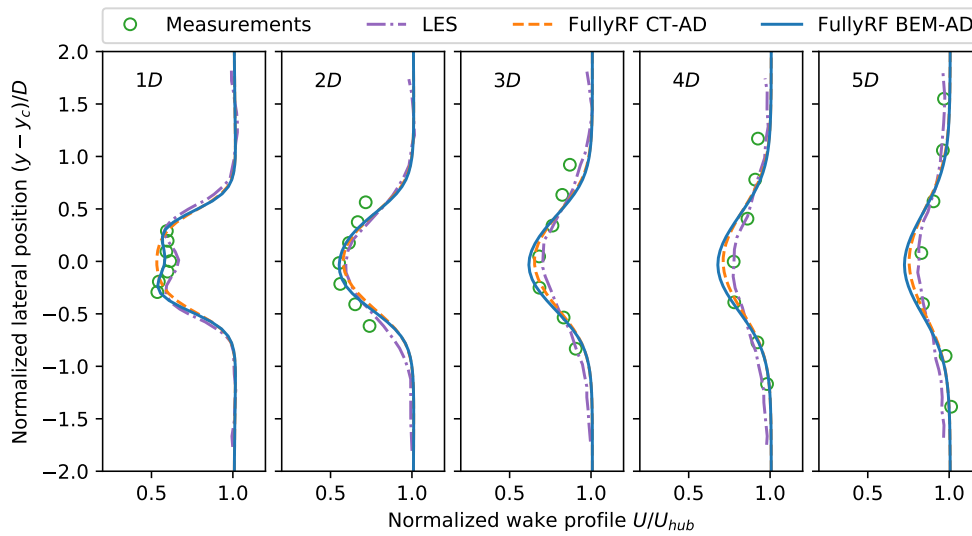


Figure 16. Wake deficit at hub height in the neutral case.

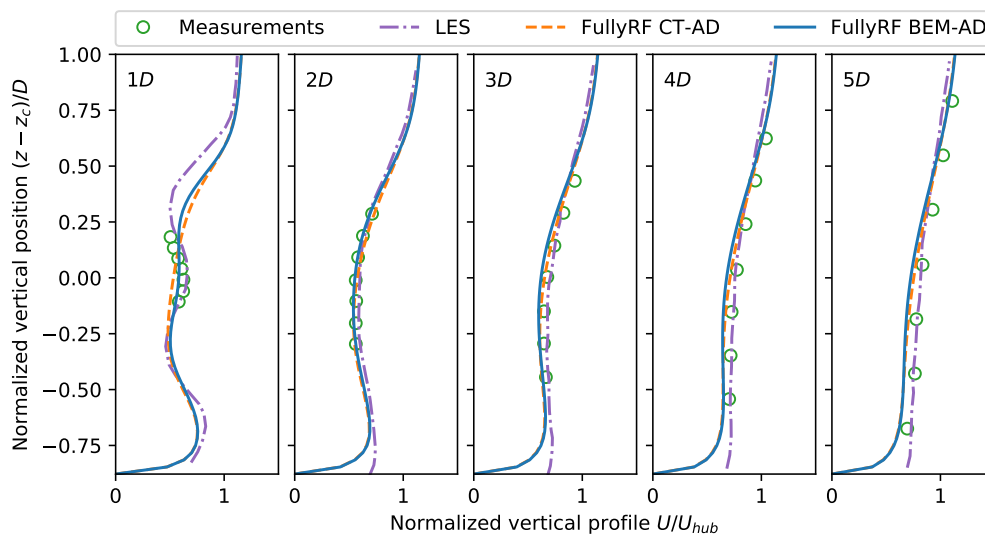


Figure 17. Vertical wake deficit profile in the neutral case.

309 Fig. 18 shows the comparisons of normalized wind speed contour in wakes under unstable
 310 conditions from simulations using different approaches. The simulation in LES show larger deficits in
 311 near wake and a slightly slower wake recovery than that predicted by the proposed model or the Laan
 312 BEM-AD model.

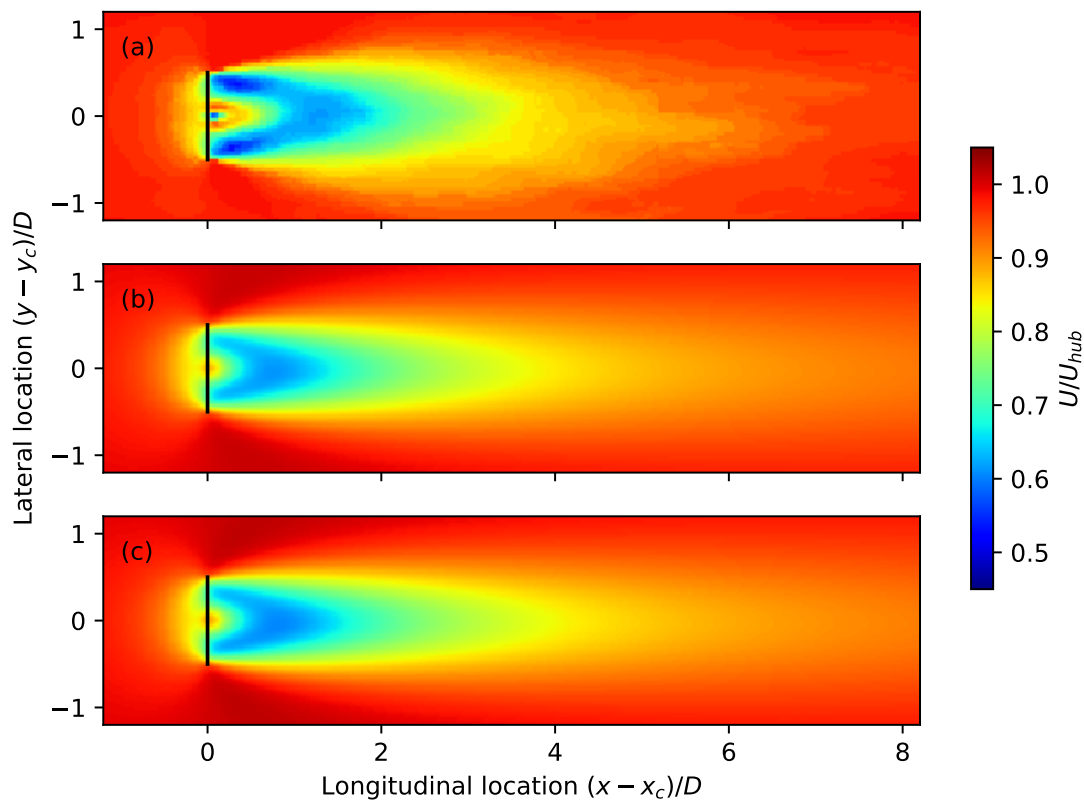


Figure 18. Normalized velocity distribution at hub height in the unstable case.

313 All the models in this paper, fail in predicting near wake under unstable conditions (Fig. 19 and
 314 Fig. 20): the measurements show a larger deficit than simulations at $1 D$ and at $2 D$. However, for the
 315 longitudinal distance above $2 D$, all the simulations show good agreements with the measurements.
 316 Compared with the good agreements between measurements using anemometers in the H93-2.0MW
 317 cases and predictions of the proposed model, the near-wake measurements using lidars could introduce
 318 much uncertainties due to wake meandering under unstable conditions and the terrain effects [2].

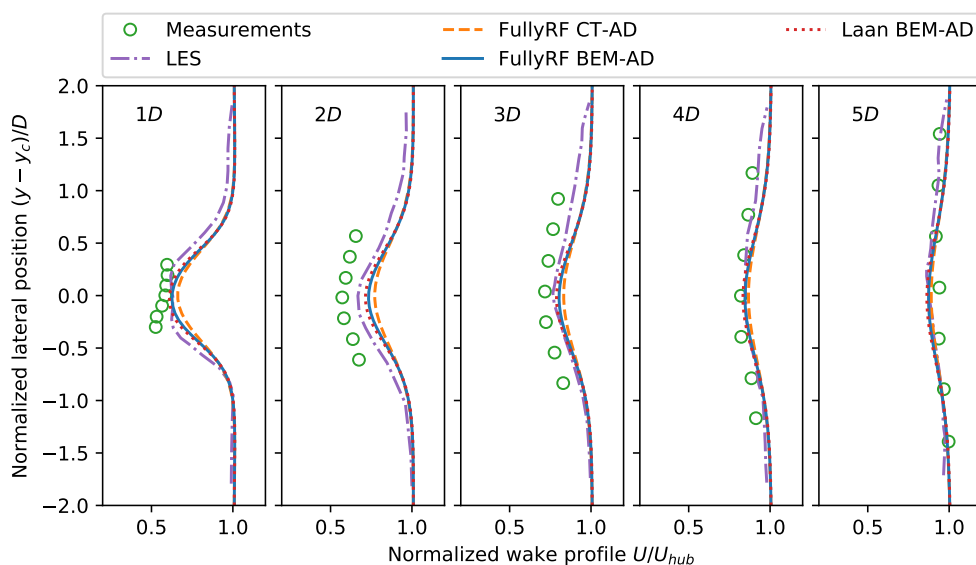


Figure 19. Wake deficit at hub height in the unstable case.

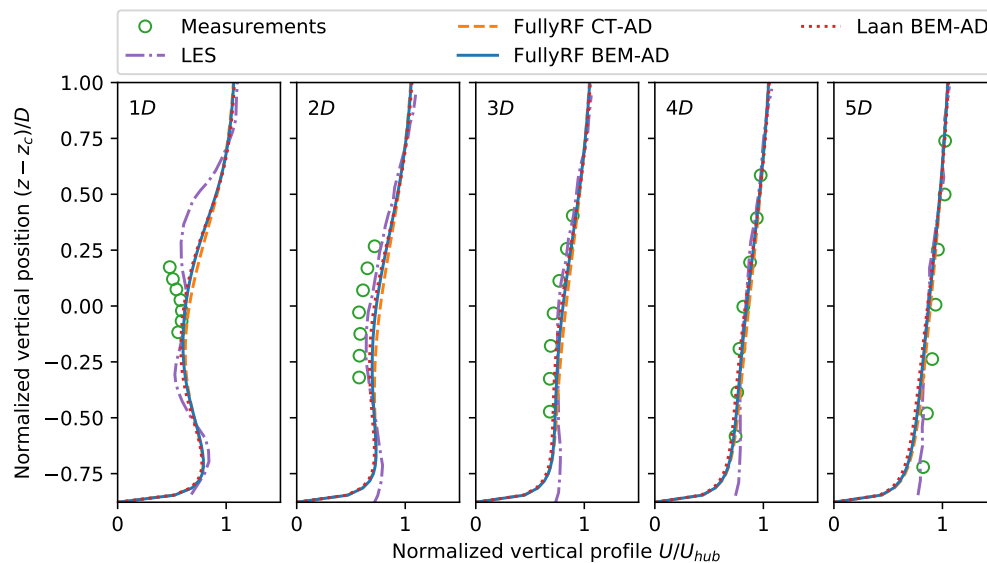


Figure 20. Vertical wake deficit profile in the unstable case.

319 5. Conclusions

320 In the present paper, the AD modeling of wind turbine wakes in RANS under varying atmospheric
 321 stability conditions are studied by focusing the effects of the distributed blade force based on BEM and
 322 the effects of the limited wind shear in the FullyRF model.

323 Wake simulations of two types of turbines under different stability conditions are carried out and
 324 compared with measurements from cup anemometers and a lidar. According to measurements, the
 325 wake tends to be enhanced and recovers slower in stable cases than in unstable cases, as a consequence
 326 of the depressed turbulence process. Asymmetric double-bell wake profiles are observed in near wake,
 327 especially for stable cases, which is probably caused by the lower extraction of energy at the rotor hub
 328 location. This asymmetric wake could be strengthened by high wind shear under stable conditions and
 329 converted by countering the blade rotation.

330 By distributing the blade force through the rotor, momentum effects of rotor to the atmospheric
 331 boundary layer are simulated in more detail way than the uniformly distributed blade force applied
 332 by the CT-AD method. This enables the proposed model to capture the double-bell near-wake shape.
 333 For the cases without the double-bell shape, predictions of the FullyRF models show a fair agreement
 334 with measurements. However, the Laan BEM-AD model significantly overestimates wake deficits
 335 and predicts a fairly slow recovery wake as compared to measurements, which is mainly due to the
 336 higher wind shear than measured data and much lower turbulence intensity than the FullyRF models.
 337 For cases that applied a lidar to measure the wake, predictions of the LES approach show a good
 338 agreement with measurements except for near wake under unstable conditions. And the LES approach
 339 predicts that the wake recovers slightly slower under unstable conditions and faster under stable
 340 conditions, as compared to the proposed model. Due to the fair agreement between the predictions of
 341 wakes in the proposed model and the measurements and the much lower computational resources
 342 than the LES approach, the proposed model is suitable for wind energy engineering applications.

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 344 Han; Project administration, De You Liu; Software, Xing Xing Han and Lin Min Li; Supervision, De You Liu;
 345 Validation, Xing Xing Han and Fei Fei Xue; Visualization, Xing Xing Han; Writing – original draft, Xing Xing Han;
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352 Abbreviations

353 The following abbreviations are used in this paper:

354	CFD	Computational Fluid Dynamics
	ASL	Atmospheric Surface Layer
	MOST	Moin-Obukhov Similarity Theory
	BEM	Blade Element Theory
355	RANS	Reynolds-averaged Navier–Stokes Equations
	LES	Large Eddy Simulation
	AD	Actuator Disk
	AL	Actuator Line

356 Appendix A Relative Wind Direction in Wake

357 The relative direction in a wake is defined as the angle between the inlet wind direction and the
 358 line from the upstream wind turbine to the mast. When looking at the turbine from downwind, this
 359 relative direction is negative at the left side and positive at the right side of the wake.

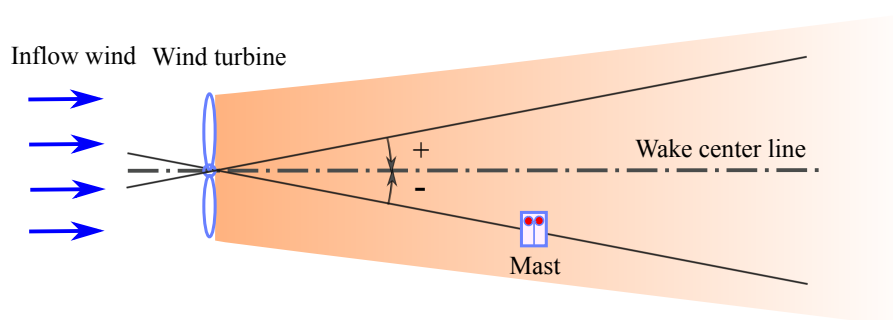


Figure A1. Definition of the relative direction.

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