

1 Article

2 The scientific information model of Chang'e-4 3 Visible and Near-IR Imaging Spectrometer(VNIS) 4 and in-orbit verification

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12 **Abstract:** The Chang'e-4 (CE-4) lunar rover, equipped with The Visible and Near-IR Imaging
13 Spectrometer(VNIS) which based on acousto-optic tunable filter spectroscopy, was launched to the
14 far side of the moon on December 8, 2018. The detection band of VNIS ranges from 0.45 to 2.4 μ m.
15 Because of the weak reflection of infrared radiation from the lunar surface, a static electronic phase-
16 locked acquisition method is adopted in the infrared channel for signal amplification. In this paper,
17 full-link simulations and modeling are conducted of the infrared channel information flow of the
18 instrument. The signal/noise characteristics of VNIS are analyzed in depth, and the signal-to-
19 noise(SNR) ratio prediction and laboratory verification are presented. On January 4, 2019, the VNIS
20 started working successfully and acquired high-resolution spectrum data of the far side of the moon
21 for the first time. Through analysis, the SNR ratio is in line with predictions, and the data obtained
22 by VNIS in orbit are consistent with the information model proposed in this paper.

23 **Keywords:** The Chang'e-4 lunar rover; Phase-locked; Signal-to-noise ratio; Spectral resolution;
24 Infrared focal plane components;

25 1. Introduction

26 The Chang'e-4 (CE-4) lunar rover is the first man-made aircraft launched to the far side of the
27 moon, and its rover (Yutu-2) is equipped with an Visible and Near-IR Imaging Spectrometer(VNIS),
28 which is used to analyze the composition of lunar surface minerals, and inherits form the Chang'E-3
29 Lunar Rover's Scientific Payloads[1]. The VNIS is a spectrum detector based on AOTF
30 spectroscopy[2], and it has two detection channels - visible near-infrared(VIS/NIR:450~950nm) and
31 short-wave infrared(SWIR:900~2400nm). It performs spectral analysis and imaging detection of
32 minerals on the lunar surface under appropriate solar illumination, and assists in the comprehensive
33 detection of minerals and chemical compositions in the patrol areas. The CE-4 spacecraft was
34 successfully launched on December 8, 2018, and it landed on the moon on January 3, 2019. Then the
35 Visible and Near-IR Imaging Spectrometer (VNIS) powered on and acquired infrared spectrum data
36 of the lunar surface. This paper focuses mainly on simulation analyses and laboratory tests based on
37 the infrared channel design features and information flow model of the instrument, and carries out
38 verification according to the acquired in-orbit data.

39 2. Instrument description

40 2.1. Basic principle of The Visible and Near-IR Imaging Spectrometer(VNIS)

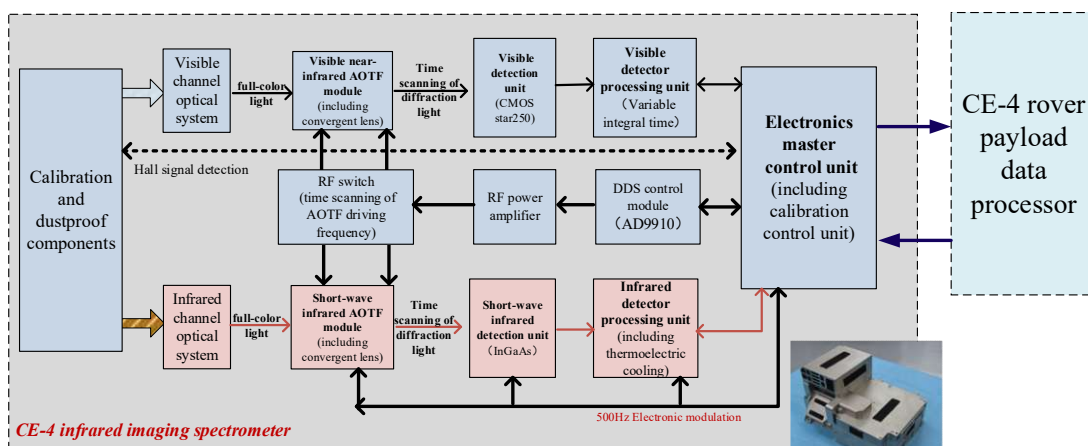
41 The VNIS uses an acousto-optic tunable filter (AOTF) for light splitting. When a beam of
42 multicolor light passes through an optically elastic crystal that vibrates at high frequencies, the
43 monochromatic light of a certain wavelength will be diffracted inside the crystal and transmitted

44 from it at an angle, and the non-diffracted light travels through the crystal directly in the original
 45 direction, thereby achieving the goal of light splitting[3]. The AOTF is a spectroscopic device based
 46 on the principle of acousto-optic diffraction, and is composed of a crystal and a transducer bonded
 47 on it. The transducer converts the electric signal into ultrasonic vibrations in the crystal, which
 48 generate spatially periodic modulations. When an incident light is irradiated to the grating, the
 49 wavelength of the diffracted light is related to the frequency of the driving electric signal, and it can
 50 be changed by varying the frequency of the electric signal[4].

51 **Table 1.** The characteristics of the Chang'e-4 Visible and Near-IR Imaging Spectrometer(VNIS)

DESCRIPTION	SPECIFICATION	
	VIS/NIR CHANNEL	SWIR CHANNEL
Spectral range (nm)	450~950	900~2400
Spectral resolution (nm)	2~10	3~12
Field of view (°)	8.5×8.5	φ3.58
Number of valid pixels	≥256×256	1
Quantized value (bit)	10	16
S/N ratio (dB)	≥43 (maximum SNR) ≥33 (albedo 0.09, solar elevation angle 45°)	≥46(maximum SNR) ≥31 (albedo 0.09, solar elevation angle 15°)
Detection range (m)		0.7~1.3
Infrared bands acquisition time		~2 min

52 According to this principle, by changing the driving frequency of AOTF crystal through rapid
 53 scanning of the time dimension, the wavelength of the diffracted monochromatic light passing
 54 through the AOTF changes sequentially, which is functionally equivalent to rapidly switching (in
 55 the order of 10μs) the transmission wavelength filter by electronic control. In this way, the spectral
 56 information of the target can be obtained by acquiring signals in the time dimension [5]. The VNIS
 57 can obtain a spectral image in the VIS/NIR channel and spectral data in the SWIR channel
 58 simultaneously. Figure .1 shows the components and basic principle of the VNIS, and Table 1
 59 presents the main Specification of the VNIS. In order to improve the sensitivity of the SWIR channel
 60 of the VNIS, a static electronic phase-locked acquisition method is adopted to realize the high
 61 sensitivity in the SWIR channel.



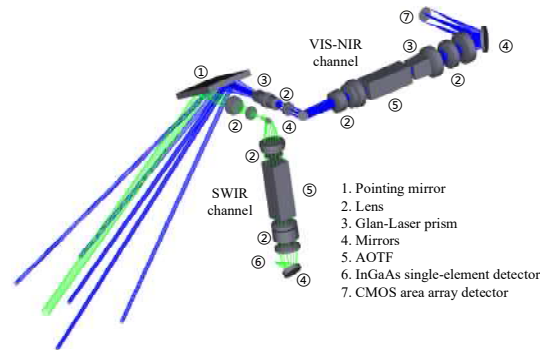
62
 63 **Figure 1.** The components and basic principle of the VNIS

64 2.2. Optical design of the SWIR channel

65 The optical system of the VNIS in the infrared channel is shown in figure. 2. The target input
 66 rays enters into the instrument through an imaging lens and, after being collimated into a parallel
 67 beam, drives the AOTF to work, so that the emitted light passing through the AOTF forms an infrared
 68 monochromatic light of a specific wavelength and then converges to the detector through a
 69 convergent lens.

70 According to the basic principle of AOTF spectrophotometry, after passing through the AOTF,
 71 three lights are formed through the convergent lens, which are +1 level diffraction light, -1 level

72 diffraction light and 0 level light[7]. When the system is designed, an InGaAs detector with a diameter
 73 of 1mm is placed at the convergence of the +1 level diffraction light, and a light filter is designed for
 74 the 0 level light, thereby suppressing the stray light.

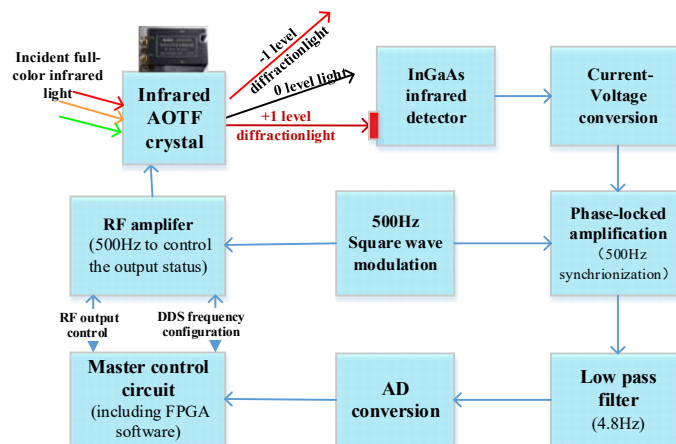


75
76 **Figure 2.** The optical design diagram of the VNIS

77 2.3. The information link of infrared channel

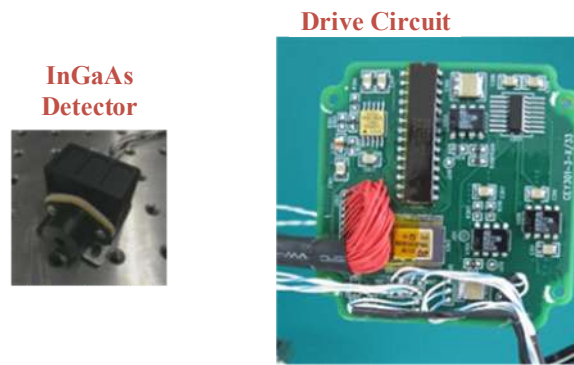
78 Figure 3 shows the block diagram of the SWIR channel of the VNIS, which is the basis for
 79 building the infrared information flow model. In order to make the AOTF crystal work normal,
 80 a radio frequency (RF) signal of a specific frequency should be applied on it, which is generated by the
 81 digital display scope (DDS) chip configured with FPGA and then amplified by the RF power
 82 amplifier. After the AOTF crystal is driven by the RF signal, the multicolor infrared light entering the
 83 AOTF penetrates to produce two channels (+1 level diffraction light and -1 level diffraction light) of
 84 monochromatic light and one channel of multicolor 0 level light. In the actual design, we choose to
 85 detect the +1 level diffraction light.

86 Due to the low albedo of the lunar surface, the two channels of infrared monochromatic light
 87 after AOTF diffraction are very weak. In order to improve the ability to detect such weak light, a
 88 static electronic phase-locked acquisition method is adopted.



89
90 **Figure 3.** Schematic diagram of the phase-locked processing circuit of infrared channel.

91 As shown in Figure. 3, by controlling the output RF signal amplified by the power amplifier, the
 92 periodic modulation of the output two channels of monochromatic infrared light at the AOTF outlet
 93 can be controlled. The modulation frequency is 500 Hz. The +1 level diffraction light after modulation
 94 is received by the InGaAs infrared detector, and the corresponding voltage signal is obtained through
 95 current-voltage conversion. The signal is processed by a phase-locked amplifier and after low-pass
 96 filtering, sent to the AD conversion chip to be converted into a digital signal. It is processed by the
 97 FPGA control circuit, which performs on-chip multiple accumulative averaging and then uploads it
 98 to the load processor. Figure. 4 shows the picture of the SWIR channel processing circuit and the
 99 InGaAs infrared detector.



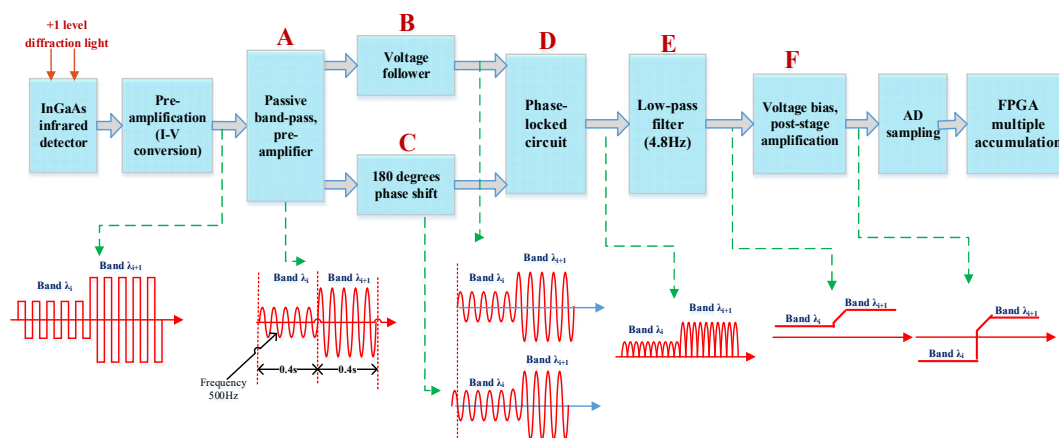
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 101 **Figure 4.** The picture of the SWIR channel processing circuit and the detector. The infrared detector
 102 type is J23TE2-66C-R01M-2.6, and is manufactured by Judson, with a peak current response rate of
 103 1.2A/W.

104 3. Signal flow model simulation and testing

105 3.1 The signal acquisition model of infrared spectral

106 The phase-locked amplification method of infrared channel electronics is introduced in section
 107 2.3. In this section, we will further discuss the infrared channel information flow model and signal
 108 characteristics. The infrared spectral signal acquisition model is established based on AOTF
 109 spectroscopic system, as shown in figure 5.

110 For weak infrared signal processing technology with phase-locked amplification, moving parts
 111 (such as the modulating reticle and chopper) are generally used for modulation between the signal
 112 light and background, such as spatial target infrared spectroscopy system [6]. In the VNIS, the special
 113 point is that a phase-locked amplification model without moving parts is designed according to the
 114 characteristics of AOTF, as shown in figure. 5. By controlling the output RF signal amplified by the
 115 power amplifier, the model can achieve monochromatic infrared light modulation after AOTF
 116 spectrophotometry. The modulation frequency is designed to be 500Hz, which is a purely electronic
 117 modulation method, without any moving parts. With the characteristics of great stability and high
 118 reliability, it is especially applicable in space.



119
 120 **Figure 5.** The information flow model of infrared spectrum acquisition based on AOTF modulation

121 The +1 level diffraction light after passing through the AOTF crystal is received by the InGaAs
 122 detector. After current-voltage conversion of a preamplifier, it is sent to the subsequent stage for
 123 signal conditioning. As shown in Figure. 5, for two consecutive wavelength signals (band λ_i and band
 124 λ_{i+1} in the figure), after 500Hz electronic modulation, what emerges is a periodic (500Hz) infrared
 125 analog signal similar to a square wave after passing through the preamplifier. After passive band-

126 pass and pre-amplification, the square wave appears to be an approximate sinusoidal signal, and the
 127 peak of the waveform reflects the intensity of the infrared monochromatic light of the wavelength.
 128 Then convert the signal into two signals of the same frequency and the same amplitude but their phase
 129 differs by 180 degrees.

130 Specifically, one signal is designed as a voltage follower circuit and the other is designed as a
 131 reverse circuit, thus realizing a phase shift of 180 degrees. The phase-locked circuit in figure. 5 realizes
 132 the chip selection of two signals. It uses a high-speed analog switch (ADG409) to realize phase-locked
 133 output so that the signal becomes half the sinusoid shown in the figure. After low pass filtering
 134 (4.8Hz), the signal is converted into a corresponding DC level signal, which is fed for AD sampling
 135 after voltage bias and post-stage amplification. In order to further improve the detection sensitivity
 136 of the system, 8 times of accumulative average processing of single wavelength analog signals are
 137 conducted within the FPGA. For infrared full-spectrum detection of lunar surface targets, the time-
 138 division detection of different wavelengths is achieved by varying the frequency of the RF signal
 139 applied to the AOTF crystal. The VNIS has 300 infrared sampling bands. The acquisition time of a
 140 single band is 0.4s, so the acquisition time of the full spectrum channel is approximately 2 minutes.

141 The above model describes the mechanism of the infrared channel signal generation of the VNIS.
 142 At the forefront of the model, the infrared detector produces a photo-generated signal current $I(\lambda)$
 143 due to receiving infrared light energy $P(\lambda)$. The model aims to enhance the sensitivity of the infrared
 144 channel. Based on the above information flow model, the source of the photo-generated current $I(\lambda)$
 145 at the input end can be further analyzed so as to obtain the signal-to-noise ratio of the system.

146 For the lunar surface spectral detection model, the solar radiation is transmitted to the surface
 147 of the moon through the space, and the surface of the moon can be approximated as a Lambert body.
 148 The target energy $P(\lambda)$ received by the VNIS can be expressed as[9]:

$$P(\lambda) = \frac{1}{4} \cdot E \cdot D_0^2 \cdot \Omega \cdot \tau_o \cdot \rho \cdot \sin \theta \quad (1)$$

149 where E represents the spectral irradiance of the sun near the lunar surface, D_0 is the optical
 150 aperture, Ω is the instantaneous solid angle of observation, τ_o is the optical system efficiency, θ is the
 151 solar elevation angle, $\Delta\lambda$ is the spectral resolution, and ρ is the lunar albedo.

152 Generally, the target energy $P(\lambda)$ is determined after the instrument system parameters are
 153 determined. That is, given the target radiant power received by the InGaAs infrared detector in the
 154 information flow model in figure. 5, the signal current $I(\lambda)$ of the infrared detector response can be
 155 expressed as:

$$I_{s\lambda} = P_{\lambda} \times A \quad (2)$$

156 The VNIS uses the J23TE2-66C-R01M-2.6 infrared detector manufactured by Judson, with a peak
 157 current response rate of 1.2A/W. After the signal current of the detector response is determined, the
 158 noise current $I_{n\lambda}$ can be calculated by the following equation.

$$I_{n\lambda} = \sqrt{2 \times q \times (I_{s\lambda} + I_{dark} + I_{black}) \Delta f} \quad (3)$$

159 where q represents the electron charge, I_{dark} denotes the dark current of the detector, I_{black} denotes
 160 the current caused by the thermal background radiation, and Δf is the bandwidth of the information
 161 processing circuit. Due to the rapid development of the detector technology, the dark current of the
 162 InGaAs infrared detector at a low temperature of around -40°C is basically zero. The SNR can be
 163 simplified as:

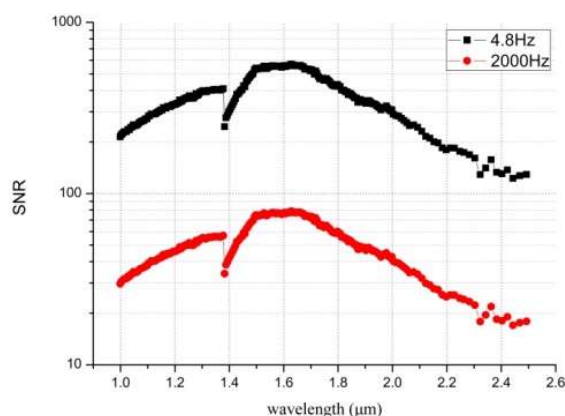
$$\text{SNR}(\lambda) = \frac{I_{s\lambda}}{I_{n\lambda}} = \frac{I_{s\lambda}}{\sqrt{2 \times q \times I_{s\lambda} \Delta f}} = \sqrt{\frac{I_{s\lambda}}{2 \times q}} \frac{1}{\sqrt{\Delta f}} \quad (4)$$

164 As can be seen, given the system design parameters and the selected infrared detector, the
 165 system SNR is directly related to the information processing bandwidth Δf of the circuit. The system

166 adopts the information flow model shown in figure 5. Here, Δf is actually the low-pass filtering 4.8Hz
 167 mentioned in the figure.5, and the system noise beyond $((500-2.4)\text{Hz}-(500+2.4)\text{Hz})$ is filtered out.

168 In contrast, if the direct signal acquisition method (without phase lock-in method) is adopted,
 169 the bandwidth Δf is generally around 2000Hz, which is the mechanism of increasing the SNR ratio.
 170 The figure. 6 shows the estimated SNR curve based on the static electronic phase-locked acquisition
 171 method (bandwidth is about 4.8Hz), which is compared with the direct signal acquisition method
 172 (bandwidth is about 2000Hz).

173 As shown in figure. 6, there exists a step at $1.4\mu\text{m}$. The reason is that the AOTF crystal has two
 174 channels, and the $1.4\mu\text{m}$ is exactly the switching point of the two channels. The input conditions for
 175 the above evaluation are a lunar albedo of 0.09 and a solar elevation angle of 15 degrees. It can be
 176 seen from the figure 6 that the SNR is about 600 in the $1.7\mu\text{m}$.



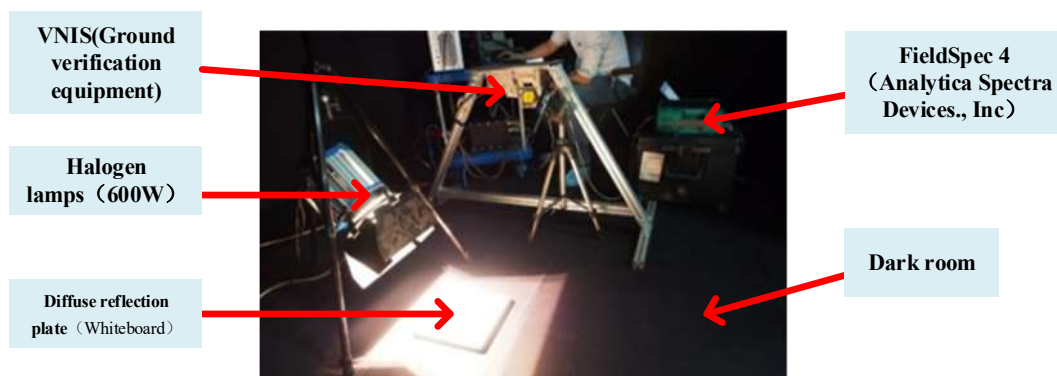
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178 **Figure 6.** The Signal-to-noise ratio (SNR) curve comparison between the static electronic phase-
 179 locked acquisition method(black) and the direct signal acquisition method(red).

180 3.2 Laboratory testing and evaluation

181 After the information flow model is established and the system signal-to-noise ratio is analyzed,
 182 the signal characteristics and signal-to-noise ratio of the instrument are tested in the laboratory.
 183 Shown in Fig. 7 is a photo of laboratory testing, where the light source is a searchlight and the target
 184 is 90% reflectivity calibration plate. Fig. 8 shows the original digital number (DN)values of 300
 185 infrared spectral bands, and the signal-to-noise ratio at $1.7\mu\text{m}$ is about 500(the signal DN is 1600, and
 186 the noise DN is about 3.2). This is slightly different from the value (600) given in Section 3.1, because
 187 the light source here is not the sunlight with an equivalent elevation angle of 15 degrees and its target
 188 albedo is not 0.09. In order to further validate the information flow model, the analog signal of the
 189 $1.7\mu\text{m}$ spectrum of the infrared signal in the whole spectrum acquisition process of the system
 190 is measured by an oscilloscope. Figure. 9 shows the signal waveforms of A, B, C and D in the model in
 191 figure 5, and the characteristics of the waveforms are consistent with the predictions of the
 192 established information flow model.

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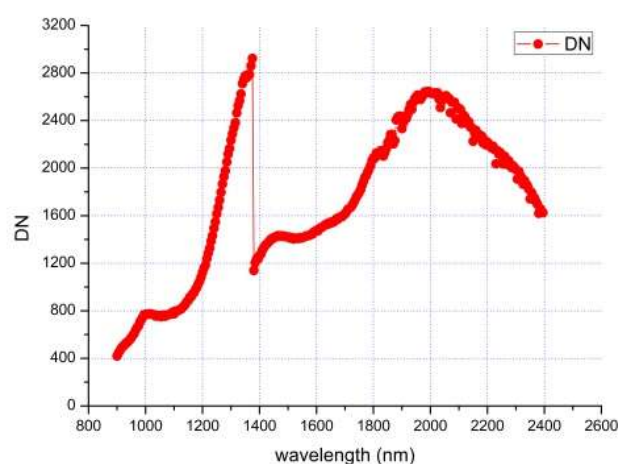
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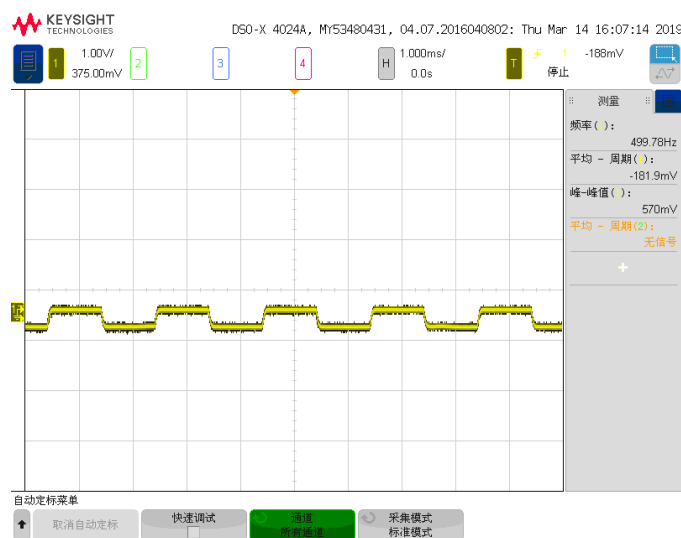
Figure 7. The experimental test was carried out in a dark room, using halogen lamps (600W) to illuminate the diffuse reflective whiteboard (reflectance $>90\%$), and then at the same angle, using both VNIS (ground verification equipment) and ground spectrometer (FieldSpec 4, Analytica Spectra Devices, Inc) measures the diffuse reflectance spectrum[10].



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Figure 8. The original infrared full spectrum signal measured in the laboratory.

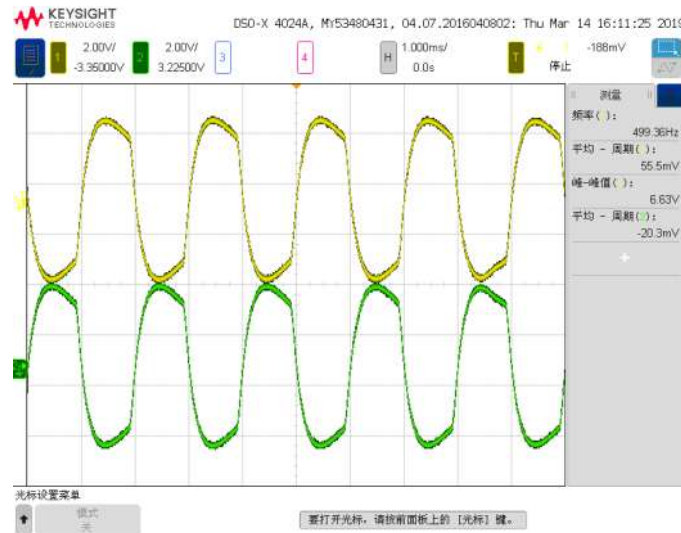


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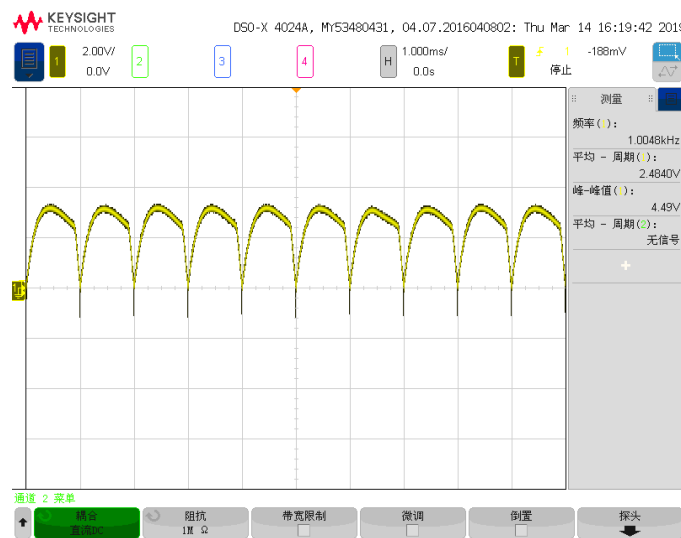
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(a) The analog displayed by oscilloscope of infrared signal after I-V conversion (The point A in figure 5.)



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(b) The analog displayed by oscilloscope of the infrared signals before phase-locked circuit
(The yellow is for point B in figure 5, and the green is for point C in figure 5)



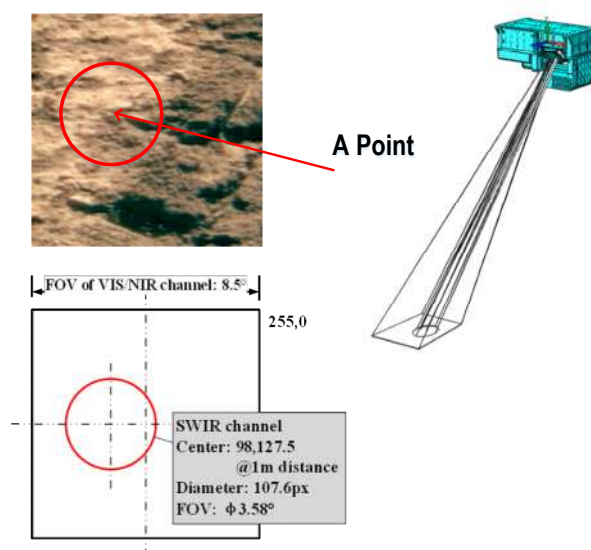
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(c) The analog displayed by oscilloscope of the infrared signal after phase-locked circuit
(The point D in figure 5.)

Figure 9. The signals of VNIS at band $1.7\mu\text{m}$ displayed by the oscilloscope (corresponding to A, B, C and D in figure. 5)

212 4. In-orbit testing

213 On January 4, 2019, the instrument obtained the infrared spectrum data of the first scene on the
214 far side of the moon. Figure 10 shows the scene of the first infrared spectrum on the far side of the
215 moon. The DN values of the original infrared spectrum data of A Point and its full-spectrum SNR are
216 show in figure 11 and figure 12, and the figure 13 gives the infrared spectral reflectance curve of the
217 moon surface at A point [8]. In the test, the solar elevation angle is about 15 degrees and the lunar
218 albedo is about 9%. According to the obtained raw data, the signal-to-noise ratio at $1.7\mu\text{m}$ is
219 calculated to be around 470, which is basically consistent with the prediction in Section 3.1 (Fig. 6).
220 This test validates the proposed AOTF system infrared spectral information processing model based
221 on phase-locked amplification technology and also verifies the high sensitivity of the lunar surface
222 weak infrared signal detector.



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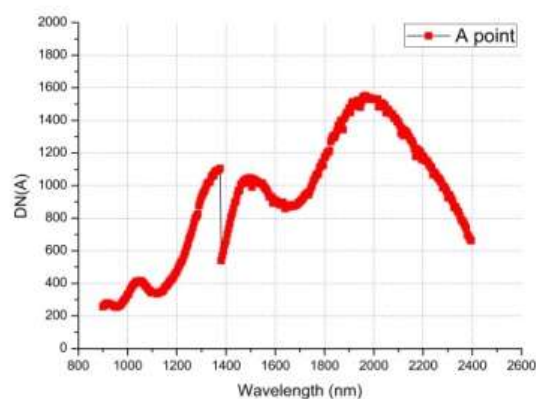
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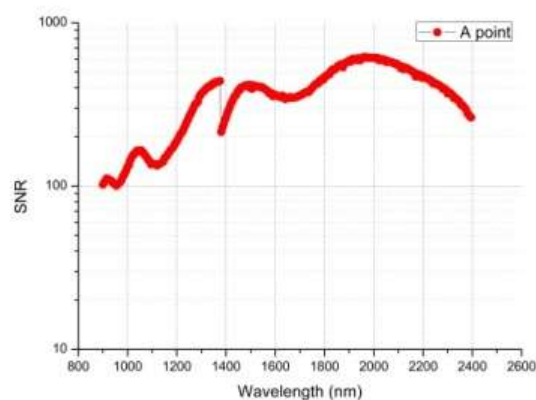
Figure 10. The first scene (which is defined as A point) obtained by VNIS on the far side of the moon. The VNIS is used to detect lunar surface objects and the optical axis of the VIS/NIR channel and SWIR channel is parallel to each other at an 18 mm distance [10, 11]. The FOVs in the VIS/NIR and SWIR are $8.5^\circ \times 8.5^\circ$ and $\Phi 3.58^\circ$, respectively. The circle represents the SWIR channel's FOV, which has a diameter of 107.6 pixels and is centered at the coordinate (98, 127.5) of the VIS/NIR image in 1 m detection distance typically.



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Figure 11. The original DN values of the far side of the moon surface at the A point.



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Figure 12. The full-spectrum SNR curve of the far side of the moon surface at the A point.

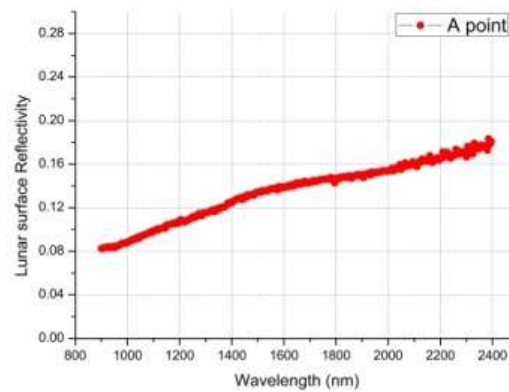


Figure 13. The spectral reflectance curve of the far side of the moon surface at the A point.

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236 5. Conclusions

237 In view of the extremely weak infrared reflectance spectral signals on the lunar surface, a method
 238 of a static electronic phase-locked acquisition in infrared channel is proposed. In addition, the paper
 239 establishes an information flow model, on the basis of which the signal-to-noise ratio has been
 240 predicted. As of the end of February 2019, the VNIS has operated 6 times on the lunar surface, and
 241 acquired 12 sets of infrared spectral data. It is found that an average SNR of 300 can still be obtained
 242 when the lunar albedo is around 9% and the solar elevation angle is 15 degrees, which further verifies
 243 the effectiveness of the proposed static electronic phase-locked acquisition method for weak infrared
 244 targets. The SNR values of in-orbit testing are basically consistent with the predictions given in this
 245 paper. In China's future deep space exploration programs, the method proposed in this paper will
 246 be helpful to further study the acquisition of weak infrared spectral information on the surface of
 247 planets.

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 252 Sciences for its contributions to the development and testing of the instruments.

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