

1 Article

2 Drilling of γ -TiAl intermetallic alloys. Mechanistic 3 model to predict cutting force and torque

4 Aitor Beranoagirre ^{1*}, Gorka Urbikain ¹, Amaia Calleja ² and L. N. López de Lacalle ³

5 ¹ Department of Mechanical Engineering, University of the Basque Country (UPV/EHU), Plaza Europa 1,
6 20018 San Sebastián, Spain; gorka.urbikain@ehu.es

7 ² Department of Mechanical Engineering, University of the Basque Country (UPV/EHU), Nieves Cano 12,
8 01006 Vitoria, Spain; amaia.calleja@ehu.es

9 ³ CFAA, University of the Basque Country (UPV/EHU), Parque Tecnológico de Zamudio 202, 48170 Bilbao,
10 Spain; norberto.lzlacalle@ehu.es

11 * Correspondence: aitor.beranoagirre@ehu.es; Tel.: +34-943-018-636

12

13 **Abstract:** Gamma titanium aluminides (γ -TiAl) present an excellent behaviour under high
14 temperature conditions, being a feasible alternative to nickel-based superalloy components in
15 aeroengine sector. However, considered as a difficult to cut material, process cutting parameters
16 require special study to guarantee components quality. In this work, developed drilling mechanistic
17 model is a useful tool in order to predict drilling force (F_z) and torque (T_c) for optimal drilling
18 conditions determination. The model is validated for three types of Gamma-TiAl alloys. Integral
19 hard metal end-drilling tools and different cutting parameters (feeds and cutting speeds) are tested
20 in three different sized holes for each alloy.

21 **Keywords:** Gamma-TiAl, superalloys, slight materials, drilling, titanium aluminides.

22

23 1. Introduction

24 Aeronautic is an emerging sector, aircraft manufacturing predictions estimate to double current
25 fleet by the year 2033, in order to satisfy passenger's increment demand that grows at a rate of 4.2%
26 per year [1]. Components manufacturing for the aeronautic sector is a high value added process, and,
27 concretely, motor components require special attention because they represent one of the most
28 expensive (20%) components [2]. Aviation industry will try to satisfy manufacturing demand
29 according to regulations requirements regarding efficiency, noise, fuel consumption and
30 contamination. Therefore, new materials and manufacturing processes are under development by
31 aeronautic sector manufacturers. In this sense, γ -TiAl alloys are a feasible replacement for nickel-
32 based alloys frequently used for compressor blades and stator in gas turbine aeroengines [3-5].
33 Moreover, titanium alloys are also interesting for bio-medical engineering, automobile sector and
34 chemical industries [6].

35 Gamma titanium aluminide intermetallics present excellent strength-weight ratio, and corrosion
36 resistance at high temperature [7]. However, it is considered a difficult to cut material [8, 9] due to its
37 poor tensile and low room temperature ductility (<2%). Besides, high heat values are generated
38 during γ -TiAl machining due to material low thermal conductivity. Consequently, tool and
39 workpiece wear mechanism are accelerated [10], tool life is reduced and workpiece integrity is
40 affected. Moreover, γ -TiAl reacts chemically with many materials causing material adhesion.

41

42 Several studies have considered γ -TiAl a difficult to cut material regarding turning [11, 12],
43 grinding [13], high speed milling [14-16], drilling and micro-drilling [17].
44 In this sense, high machining conditions can cause irreversible consequences such as surface cracking,
45 hardened layers and tensile residual stresses [18]. Optimal machining strategies and parameters are
46 studied for electrochemical machining [19], turning [20, 21], and milling [22] related to cutting
47 temperature techniques evaluation, and, appropriate machining conditions determination in order
48 to reduce tool wear and increase tool life. Milling studies have also investigated the effects of
49 operating parameters and conditions on tool life and surface integrity when milling γ -TiAl alloys [23-
50 25].

51 In relation to lubrication techniques, some studies [26] focus on cryogenic lubrication (liquid
52 nitrogen) obtaining cutting feed values increment while tool life is maintained.
53 Machining processes modelling is also a well-known field for difficult to cut materials. In this sense,
54 two-dimensional models [27] are developed for cutting parameters such as cutting speed and feed
55 influence determination. Empirical models are also used for thrust and torque values prediction in
56 composites. Other techniques such as response surface methodology (RSM) and finite element
57 analysis (FEA) are implemented for milling [28] and drilling [29] process parameters analysis. In
58 addition finite element models are also developed for performance characteristics such as hole
59 quality prediction [30].

60 Considering γ -TiAl promising application possibilities, drilling machinability studies are a focus
61 of interest due to presented machinability problems being drilling one of the most frequent processes
62 for components assembling. In this work, developed drilling mechanistic model is a useful tool in
63 order to predict drilling force (F_z) and torque (T_c) for optimal drilling conditions determination. The
64 model is validated for three types of Gamma-TiAl alloys. Integral hard metal end-drilling tools and
65 different cutting parameters (feeds and cutting speeds) are tested in three different sized holes for
66 each alloy.

67 2. Experimental procedure

68 2.1. Material

69 Four different titanium alloys are tested in performed drilling tests: Ti-6Al-4V and three different
70 γ -TiAl alloys (TNB, extruded MoCuSi and ingot MoCuSi). Table 1 shows tested alloys mechanical
71 properties.

72 γ -TiAl alloys present higher aluminum percentage in comparison to other titanium such as Ti-
73 6Al-4V, 43-48% in γ -TiAl and 6% in Ti-6Al-4V, improving thermal conductivity in γ -TiAl. On the
74 other hand, ductile transition temperature occurring between 600-800°C, depending on the
75 microstructure and grain size alloys, is aggravated for alloys with higher titanium percentage [11].
76 The intermetallic TiAl provides low density [31] as well as high mechanical strength under high
77 temperatures and corrosive environments. The intermetallic γ -TiAl superalloys offer excellent
78 mechanical properties [32], with low density (4gr/cm³), high resistance at high temperatures, low
79 electrical and thermal conductivity, oxidation resistance, ultimate strength of 1000 MPa and Young's
80 modulus of 160 GPa.

81 MoCuSi alloy [Ti-(43-46) Al-(1-2) Mo-(0.2) Si-Cu] is used at low temperatures with high
82 resistance below 650°C, and, TNB alloy [Ti-(44-45) Al-(5-10)Nb-(0.2-0.4)C] resists very high
83 temperatures maintaining high resistance and oxidation values. Regarding MoCuSi alloys, the
84 material is presented in both, extruded o ingot structure. Extruded alloys present an oriented
85 structure oriented in the extrusion direction whereas melted alloys present a structure without any
86 preferable orientation, typical of no extruded or laminated materials. The ingot structure is directly
87 obtained from VAR (Vacuum Arc Remelting) process. For extruded structures, material is extruded
88 at 1200°C and smaller sized grains are obtained with superior creep strain, yield strength and KIC.

89

90

Table 1. Mechanical properties comparison between TiAl alloys.

PROPERTY	TNB	Ti-6Al-4V (annealed)	MoCuSi extruded	MoCuSi ingot
Density (g/cm ³)	3.86	4.49	3.74	3.88
Specific modulus (GPa/Mg/m ³)	43	24	43	37
Tensile strength (MPa)	683	1087	607	689
Specific strength (MPa/g/cm ³)	192	947	198	180
Yield strength (MPa)	589	942	589	570
Ductility (%)	1.9	7.8	1.7	2.4
Fracture toughness (MPa m ^{1/2})	23	52	23	20
Thermal conductivity (W/mK)	24	8.6	24	19
Maximum operating temperature (°C)	900	615	900	865

91 2.2. *Equipment*

92 Machining tests were performed in a vertical CNC milling machine, Kondia® model B640, with
 93 maximum rotational speed of 10,000 rpm and 25 kW. During the tests, process-cutting conditions are
 94 measured and recorded. For the axial/thrust (F_z) and radial (F_x , F_y) cutting forces, and, Z axis (T_c)
 95 torque measurements, a dynamometric Kistler® equipment 9257B was used.
 96



97

98 **Figure 1.** Kondia® model B640 (left) and Kistler® 9257B dynamometer (right)99 2.3. *Machining conditions*

100 Drilling tests were carried out for the described four titanium alloys (Table 1). As it can be seen
 101 in Table 2, for each material, tested feed values are 0.05 and 0.1 mm/rev, and, $D = 3-5-7$ and 8.5 mm
 102 holes were drilled in each case. The drilled depth is 20 mm.
 103

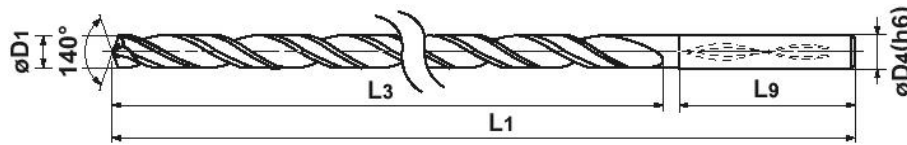
104

Table 2. Machining conditions for the 8.5 mm hole

Material	V_c [m/min]	N [rev/min]	f [mm/rev]	f [mm/rev]
Ti-6Al-4V	50	1874	0.05	0.1
TNB	15	1874	0.05	0.1
MoCuSi Extruded	15	562	0.05	0.1
MoCuSi Ingot	15	562	0.05	0.1

105

106 The selected tool is a solid carbide drill (Mitsubishi®, MPS0850S-DIN-C, Figure 2). During the
107 machining operations cutting forces, torque and power consumption were recorded.



108

109

Figure 2. Cutting tool geometry

110 One of the critical aspects when drilling γ -TiAl is chip evacuation and heat dissipation. Internal
111 lubrication is directly applied on the cutting edge, due to the poor thermal conductivity of these
112 alloys. Specially designed for low machinability materials, Rhenus® FU W (Table 3) coolant was
113 used. Internal coolant pressure is set to 8.5 bar.

114

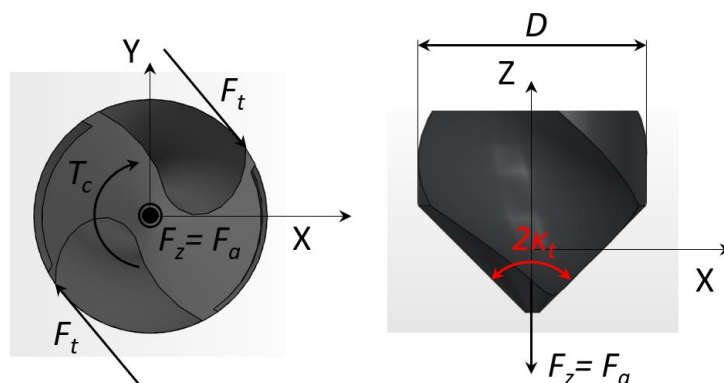
Table 3. Properties of Rhenus FU 70 W coolant

CONCENTRATED		EMULSION	
Viscosity 20 °C (mm ² /s)	Content of mineral oil %	pH Value 5% concentration	Protection against corrosion (DIN 51360/1)
Approx. 150	Approx. 33	Approx. 9,0	Note 0 al 2%

115 3. Cutting forces prediction mechanistic model

116 Predictive cutting forces mechanistic models help programmers with cutting parameters
117 selection. Especially when drilling low machinability titanium alloys, it is interesting to simulate
118 different machining conditions in order to decide whether the estimated cutting forces values led to
119 an optimum machining process

120 This section explains the modelling of cutting forces and torque in drilling. Since the cutting
121 speed varies along the drill's lips, the way the cutting force coefficients are introduced inside the
122 model is a crucial issue. There have been a number of attempts based on orthogonal-oblique
123 transformation [33] and mechanistic models [34].



124

125 **Figure 3.** Cutting or tangential force (F_t), thrust force (F_z) and cutting torque (T_c) in drilling process. Drill
126 bottom view (left) and front view (right).

127 The cutting edge is divided into discrete elements in the drill axis ($j=1$ to n elements) and the
128 global cutting force and torque are obtained by summing the elemental contributions along each lip
129 ($i=1$ to Z). As it can be seen in Figure 3, while F_x and F_y are coupled due to tool rotation leading to the

130 radial F_r and tangential F_t cutting force components, F_z is directly obtained from the axial component
 131 F_a . Here, we will focus on the thrust force (Eq.1) in Z direction and torque (Eq.2), which are critical
 132 parameters when characterizing tool life. These magnitudes are calculated as:

$$F_z(t) = \sum_{i=1}^Z \sum_{j=1}^n K_{c,z}(h,r) \cdot \Delta z \cdot h \quad (1)$$

$$T_c(t) = \sum_{i=1}^Z \sum_{j=1}^n K_{t,z}(h,r) \cdot \Delta z \cdot h \cdot r \quad (2)$$

133 where K_z and K_T are the corresponding thrust force and torque coefficients, Δz is the axial width
 134 of a differential element and h the chip thickness (here, $h = 0.5f \cdot \sin 70^\circ$). As usual, the thrust cutting
 135 force and torque coefficients need to be calibrated. Here, 4 different titanium alloys were tested: 1)
 136 Ti-6Al-4V, taken as reference; 2) TNB and MoCuSi types, 3) extruded and 4) ingot types. Under this
 137 approach the cutting force coefficients were assumed as function of feedrate (or h) and radial distance
 138 of the cutting point to drill axis as quadratic functions. First, a_0 - a_5 and b_0 - b_5 coefficients in Eqs. (3-4)
 139 are identified from a linear regression for the four types of materials from a set of experiments.

$$K_z(h,r) = a_0 + a_1 \cdot h + a_2 \cdot r + a_3 \cdot hr + a_4 \cdot r^2 + a_5 \cdot h^2 \quad (3)$$

$$K_T(h,r) = b_0 + b_1 \cdot h + b_2 \cdot r + b_3 \cdot hr + b_4 \cdot r^2 + b_5 \cdot h^2 \quad (4)$$

140 4. Results

141 The mechanistic model is valid for $V_c = 50$ m/min (Ti-6Al-4V), and $V_c = 15$ m/min (TNB, and
 142 extruded/ingot types), the cutting speed at the tool diameter D . To identify the cutting coefficients 10
 143 tests were done. After a pilot hole the hole was finished at $D = 8.5$ mm. Five different pilot hole
 144 diameters ($D_0 = 3-4-5-6-7$ mm) and two feed rates ($f = 0.05-0.1$ mm) were programmed.

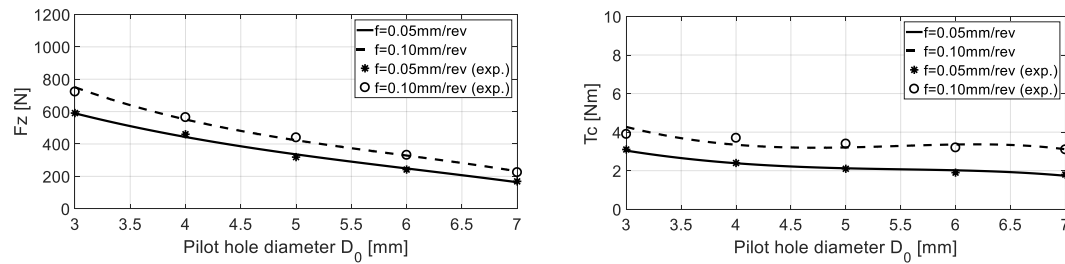
145 Results for Ti-6Al-4V alloy

146 As it can be seen, both experimental and the predicted thrust force and torque values are higher
 147 for higher feed rate values (Figure 4). Table 4 shows the experimental data for both variables during
 148 the calibration of the cutting coefficients. While the curve F_z reflects almost a linear tendency, the
 149 torque is approximated as a curve with an inflexion (third order). At the same time, the higher
 150 dispersion is found the lower is the difference between the pilot and final hole diameters.

151 **Table 4.** Experimental results for F_z [N] and T_c [Nm] in Ti-6Al-4V alloy.

D_0 [mm]	F_z [N]		T_c [Nm]	
	f [mm/rev]	f [mm/rev]	f [mm/rev]	f [mm/rev]
	0.05	0.1	0.05	0.1
3	591	723	3.1	3.9
4	460	565	2.4	3.7
5	319	440	2.1	3.4
6	242	332	1.9	3.2
7	169	225	1.8	3.1

159



160

161 **Figure 4.** F_z [N] (left) and T_c [Nm] (right) for Ti-6Al-4V alloy drilling experimental and mechanistic model
162 values

163 Results for TNB alloy

164 Thrust force values (Figure 5 left) show higher values for higher feed rate values but are not
165 doubled when doubling the feed rate. However, cutting tool behaved correctly despite the cutting
166 force is increased $\times 1.33$ and cutting torque is increased $\times 1.5-2$, with respect to Ti4Al6V. When feed
167 rates values are multiplied $\times 2$, torque values are multiplied $\times 1.3$ approximately. Table 5 shows again
168 the experimental data necessary to obtain the polynomials of the cutting coefficients. A higher
169 dispersion is seen for the cutting torque T_c as the pilot hole diameter approaches the final hole
170 diameter (Figure 5, right).

171

Table 5. Experimental results for F_z [N] and T_c [Nm] in TNB alloy.

172

173

174

175

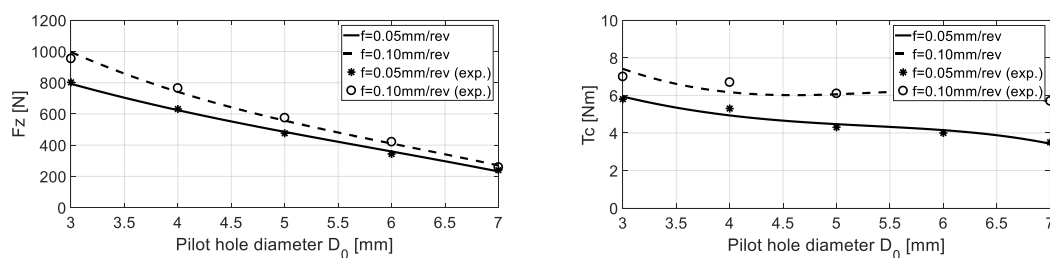
176

177

178

179

D_0 [mm]	F_z [N]		T_c [Nm]	
	f [mm/rev]		f [mm/rev]	
	0.05	0.1	0.05	0.1
3	802	954	5.8	7.0
4	631	765	5.3	6.7
5	476	575	4.3	6.1
6	343	421	4.0	6.0
7	240	259	3.5	5.7



180

181 **Figure 5.** F_z [N] (left) and T_c [Nm] (right) for TNB alloy drilling experimental and mechanistic model values

182 Results for ingot MoCuSi alloy

183 Ingot MoCuSi alloys are harder Ti-6Al-4V alloys and TNB alloys. This statement was verified on
184 sight of Figure 6. The axial force F_z and cutting torque T_c is higher regarding to previous alloys.
185 Focusing on cutting torque (Figure 6 right), when the feed rate value is multiplied $\times 2$, torque values
186 are multiplied $\times 1.1$ approximately.

187

188

Table 6. Experimental results for F_z [N] and T_c [Nm] in ingot MoCuSi alloy.

189

190

191

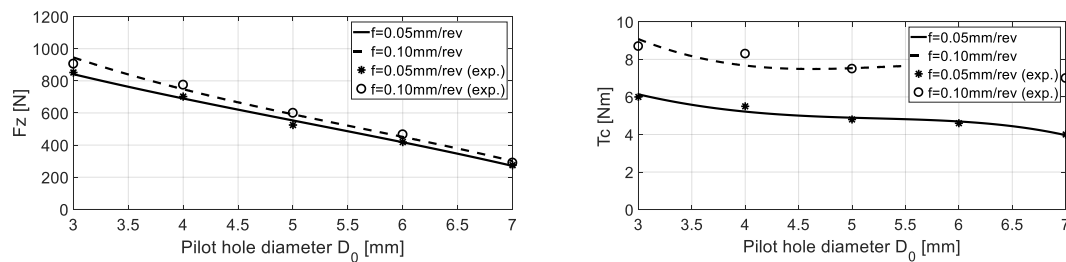
192

193

194

195

D_0 [mm]	F_z [N]		T_c [Nm]	
	f [mm/rev]		f [mm/rev]	
	0.05	0.1	0.05	0.1
3	852	906	6.0	8.7
4	702	775	5.5	8.3
5	525	600	4.8	7.5
6	419	466	4.6	7.3
7	275	290	4.0	7.0



196

197

198

Figure 6. F_z [N] (left) and T_c [Nm] (right) for ingot MoCuSi alloy drilling experimental and mechanistic model values

199

Results for extruded MoCuSi alloy

200

201

202

203

204

MoCuSi extruded alloys presents the highest hardness values in comparison to the latter ones. While the results for the cutting torque T_c are quite similar to the ingot MoCuSi alloy (both results are almost identical), the thrust force is more dramatic to the extruded MoCuSi alloy. Regarding cutting torque (Figure 7 right), when the feed rate value is multiplied by two, torque values are multiplied $\times 1.2$ approximately.

205

Table 7. Experimental results for F_z [N] and T_c [Nm] in extruded MoCuSi alloy

206

207

208

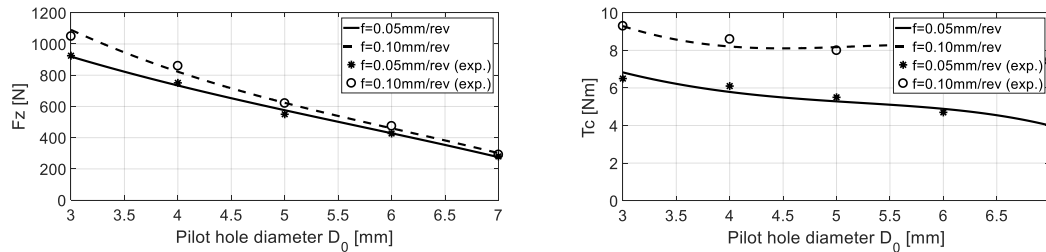
209

210

211

212

D_0 [mm]	F_z [N]		T_c [Nm]	
	f [mm/rev]		f [mm/rev]	
	0.05	0.1	0.05	0.1
3	925	1050	6.5	9.3
4	751	860	6.1	8.6
5	550	621	5.5	8.0
6	427	475	4.7	7.7
7	280	292	4.0	7.5



213

214 **Figure 7.** F_z [N] (left) and T_c [Nm] (right) for extruded MoCuSi alloy drilling experimental and mechanistic
 215 model values.

216 Using the above experimental results, the coefficients in Eqs.3-4 are obtained and so, the
 217 corresponding polynomial functions are built (Table 8).

218

Table 8. Obtained coefficients for the polynomials in Eqs. (3-4).

Materials	$K_z(h,r)$	$K_T(h,r)$
Ti-6Al-4V	$a_0= 13,159; a_1= -93,861; a_2= -3,225; a_3= -10,986;$ $a_4= 977.11; a_5= -7,056.0$	$b_0= 39.646; b_1= -68.280; b_2= -21.597; b_3= -$ $41.882; b_4= 5.253; b_5= -5.171$
TNB	$a_0= 19,860; a_1= -217,630; a_2= -3,830; a_3= 17,235;$ $a_4= 855.42; a_5= -16,347.1$	$b_0= 77.853; b_1= -283.109; b_2= -36.204; b_3= -$ $61.168; b_4= 8.334; b_5= -21.331$
MoCuSi (ingot)	$a_0= 23,445; a_1= -270,543; a_2= -4,028; a_3= 20,962;$ $a_4= 688.07; a_5= -20,320$	$b_0= 90.152; b_1= -274.538; b_2= -45.120; b_3= -$ $17.808; b_4= 9.663; b_5= -20.703$
MoCuSi (extruded)	$a_0= 24,016; a_1= -282,289; a_2= -4,270; a_3= 24,114;$ $a_4= 860.73; a_5= -21,202$	$b_0= 84.201; b_1= -120.895; b_2= -45.120; b_3= -$ $17.808; b_4= 9.663; b_5= -20.703$

219

220 For model validation, different combinations of parameters were tested within the window
 221 parameters. New pilot hole drills and final drill of $D = 8.5$ mm were used for these tests. 8 different
 222 tests were programmed to put the model to work. Table 9 presents the conditions for these validation
 223 tests.

224

Table 9. Cutting conditions for the validation tests

Materials	V_c [m/min]	D_0 [m/min]	f [mm/rev]	# Test
Ti-6Al-4V	50	4	0.06	1
		6	0.08	2
TNB	15	4	0.06	3
		6	0.08	4
MoCuSi (ingot)	15	4	0.06	5
		6	0.08	6
MoCuSi (extruded)	15	4	0.06	7
		6	0.08	8

225

226 Figure 8 shows the results from the simulated values using the model described and the
 227 experimentally measured values. The model presents a good correlation with the real thrust forces
 228 and cutting torques. For the cutting forces the maximum relative error between the predicted and
 229 measured values was 8% while for the cutting torque the worst case was found at 13%.

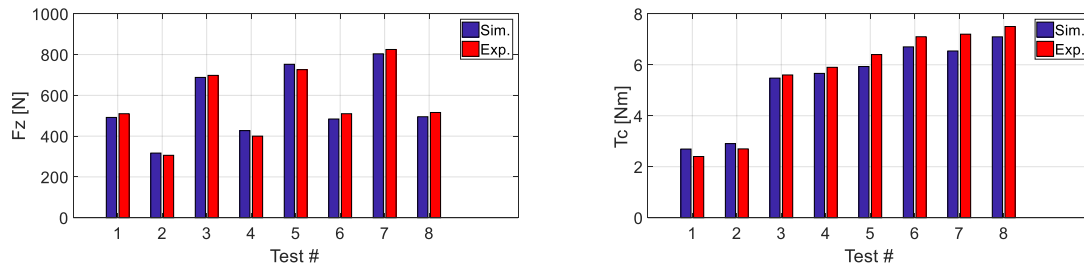


Figure 8. Model validation for F_z and T_c magnitudes.

6. Conclusions

This work presents the machinability results for 4 different titanium alloys. Currently there are few data about machining of γ -TiAl. The obtained results reflect that these alloys present a much smaller machinability than conventional titanium alloys. Three manufactured presentations for these alloys are in the market, the TNB alloy, the alloy solidified as ingot, and the alloy extruded after solidification. These are very brittle materials, so a special care must be paid to avoid the chipping and cracking of components during machining processes [35–36]. Besides, two serious disadvantages during their processing are their great sensibility to the impurity during the foundry process and so, high production costs.

This work presents a mechanistic stationary model capable of simulating the thrust force and cutting torque in the drilling process. The influence of vibratory effects, lateral vibrations or runout was not considered. After model calibration, validation tests were done showing a good agreement.

Experimental drilling tests were done to evaluate the machinability of these difficult-to-cut materials. The high cost of the tested materials (approx. 400 €/kg) limited the number of tests and a stronger experimental design. In sight of the obtained cutting coefficients, a quasi-linear dependence on the pilot hole D_0 was found for the axial force F_z while a cubic is more suitable in the case of T_c .

The experiments demonstrated big differences between the more conventional Ti6Al4V alloy and the three γ -TiAl intermetallic alloys. As the mechanical properties of extruded/ingot alloys are higher than the TNB alloy, this trend is inversely true for the machinability grade. Tests proved that MoCuSi alloys are the toughest materials.

Despite the good results during the validation, a statistical approach would be desirable to test any differences in material series, machine tools, etc. as well as to build more complete models. Wear studies including process damping effects can be subjects for future research works.

Author Contributions: A.B. designed and performed the experiments. G.U. and A.C. analyzed obtained experimental and mechanistic model predicted results. Finally, L.N.L.d.L. contributed with the resources (machine, tools, material, etc.) and general supervision of the work.

Funding: Acknowledgments: Thanks are addressed to the UFI in Mechanical Engineering of the UPV/EHU for its support to this project, and to Spanish project DPI2016-74845-R, ESTRATEGIAS AVANZADAS DE DEFINICION DE FRESADO EN PIEZAS ROTATIVAS INTEGRALES, CON ASEGURAMIENTO DE REQUISITO DE FIABILIDAD Y PRODUCTIVIDAD and project RTC-2014-1861-4, INNPACTO DESAFIO II.

Acknowledgments: The authors acknowledge the technical assistance from Eng. Garikoitz Goikoetxea at UPV/EHU.

Conflicts of Interest: The authors declare no conflict of interest. The funders had no role in the design of the study; in the collection, analyses, or interpretation of data; in the writing of the manuscript, or in the decision to publish the results.

References

1. www.boeing.com

- 270 2. Martin, R.; Evans, D. Reducing Costs in Aircraft: The Metals Affordability Initiative Consortium.
271 *JOM* **2000**, *52* (3), 24-28.
- 272 3. Voice W.E.; Henderson M.; Shelton E.F.J.; Wu, X. Gamma titanium aluminide-TiNb. *Intermetallics*
273 **2005**, *13*, 959-964.
- 274 4. Boyer, R.R. An overview on the use of titanium in the aerospace industry. *Mater. Sci. Eng. A* **1996**,
275 *213*, 103-114.
- 276 5. Jha, A.K.; Singh, S.K.; Kiranmayee, M.S.; Sreekumar, K.; Sinha, P.P. Failure analysis of titanium
277 alloy (Ti6Al4V) fastener used in aerospace application. *Eng. Fail. Anal.* **2010**, *17*, 1457-1465.
- 278 6. Murr, L.E.; Quinones, S.A.; Gaytan, S.M.; López, M.I.; Rodela, A.; Martinez, E.Y.; Hernandez,
279 D.H.; Martinez, E.; Medina, F.; Wicker, R.B. Microstructure and mechanical behavior of Ti-6Al-
280 4 V produced by rapid-layer manufacturing for biomedical applications. *J. Mech. Behav. Biomed.*
281 *Mater.* **2009**, *2*, 20-32.
- 282 7. Yamaguchi, M.; Inui, H.; Ito, K. High temperature structural intermetallics. *Acta Mater* **2000**, *48*,
283 307-322.
- 284 8. Kim Y.W.; Dimiduk, D.M. Gamma TiAl alloys: emerging new structural metallic materials.
285 Proceeding of ICETS 2000-ISAM, 2000, Beijing.
- 286 9. Semiatin S.L.; Seetharaman, V.; Weiss, I. Hot workability of titanium and titanium aluminide
287 alloys. *Mater Sci Eng.* **1998**, *243*, 1-24.
- 288 10. López de Lacalle, L.N.; Perez, J.; Llorente, J.I.; Sanchez, J.A. Advanced cutting conditions for the
289 milling of aeronautical alloys. *J. Mater. Process. Technol.* **2000**, *100*, 1-11.
- 290 11. Aspinwall D.K.; Dewes, R.C.; Mantle, A.L. The Machining of γ -TiAl Intermetallic Alloys. *CIRP*
291 *Ann. - Manuf. Tech.* **2005**, *54*(1), 99-104.
- 292 12. Arrazola, P.J.; Garay, A.; Iriarte, L.M.; Armendia, M.; Marya, S.; Maitre, F.L. Machinability of
293 titanium alloys (Ti6Al4V and Ti555.3). *J. Mater. Process. Technol.* **2009**, *209*, 2223-2230.
- 294 13. Beranoagirre, A.; López de Lacalle, L.N. Grinding of Gamma TiAl Intermetallic Alloys. *Proc.*
295 *Eng.* **2013**, *63*, 489-498.
- 296 14. Lindemann, J.; Glavatskikh, M.; Leyens, C. Surface effects on the mechanical properties of
297 gamma titanium aluminides. 7th THERMEC, 2011, Canada.
- 298 15. Hood, T.; Aspinwall, D.K.; Sage, C.; Voice, W. High speed ball nose and milling of γ -TiAl alloys.
299 *Intermetallics* **2013**, *32*, 284-291.
- 300 16. Thepsonthi, T.; Ozel, T. Experimental and Finite Element Simulation based Investigations on
301 Micro-Milling Ti-6Al-4V Titanium Alloy: Effects of cBN Coating on Tool Wear. *J. Mater. Process.*
302 *Technol.* **2013**, *213*, 532-542.
- 303 17. Cantero, J.L.; Tardio, M.M.; Canteli, J.A.; Marcos, M.; Miguelez, M.H. Dry drilling of alloy
304 Ti6Al4V. *Int. J. Mach. Tools Manuf.* **2005**, *45*, 1246-1255.
- 305 18. Hood, R.; Aspinwall, D.K.; Soo, S.L.; Mantle, A.L.; Novovic, D. Workpiece surface integrity when
306 slot milling γ -TiAl intermetallic alloy. *CIRP Ann.* **2014**, *63*(1), 53-56.
- 307 19. Liu, J.; Zhu, D.; Zhao, L.; Xu, Z. Experimental Investigation on Electrochemical Machining of γ -
308 TiAl Intermetallic. *Proc. CIRP* **2015**, *35*, 20-24.
- 309 20. Kosaraju, S.; Anne, V.G. Optimal machining conditions for turning Ti-6Al-4 V using response
310 surface methodology, *Adv. Manuf. Technol.* **2013**, *1*(4), 329-339.
- 311 21. Mantle, A.L.; Aspinwall, D.K. Temperature Measurement and Tool Wear When turning gamma
312 titanium aluminide intermetallic, Proceedings of the 13th Conference of the Irish manufacturing
313 Committee (IMC-13). Re-Engineering for World Class manufacturing, Limerick, Ireland, (1996),
314 427-446.
- 315 22. Aspinwall, D.K.; Mantle, A.L.; Chan, W.K.; Hood, R.; Soo S.L. Cutting Temperatures when Ball
316 Nose End Milling γ -TiAl Intermetallic Alloys. *CIRP Ann.* **2013**, *62*(1), 75-78.
- 317 23. Mantle, A.L.; Aspinwall D.K. Surface integrity of a high speed milled gamma titanium
318 aluminide. *J. Mater. Process. Technol.* **2001**, *118*, 143-150.
- 319 24. Olvera, D.; Urbicain, G.; Beranoagirre, A. López de Lacalle, L.N. Hole Making in Gamma TiAl.
320 DAAAM International scientific book, 337-347, 2010.

- 321 25. Priarone, P.C.; Rizzuti, S.; Settineri, L.; Vergnano, G. Effects of Cutting Angle, Edge Preparation
322 and Nano-Structured Coating on Milling Performance of a Gamma Titanium Aluminide. *J.*
323 *Mater. Process. Technol.* **2012**, *212*, 2619–2628.
- 324 26. Hong, S.Y.; Ding, Y. Cooling approaches and cutting temperatures in cryogenic machining of
325 Ti-6Al-4 V. *Int. J. Mach. Tools Manuf.* **2001**, *41*, 1417–1437.
- 326 27. Marasi, A. Modeling the effects of cutting parameters on the main cutting force of Ti6Al4V alloy
327 by using hybrid approach. *Int. J. Adv. Eng. Appl.* **2013**, *2(5)*, 6–14.
- 328 28. Kadirgama, K.; Abou-El-Hossein, K.A.; Mohammad, B.; Habeeb, A.; Noor, M.M. Cutting force
329 prediction model by FEA and RSM when machining Hastelloy C-22HS with 90 °holder. *J. Sci.*
330 *Ind. Res.* **2008**, *67*, 421–427.
- 331 29. Bagci, E.; Ozcelik, B. Finite element and experimental investigation of temperature changes on a
332 twist drill in sequential dry drilling. *Int. J. Adv. Manuf. Technol.* **2006**, *28*, 680–687.
- 333 30. Chatterjee, S.; Mahapatra, S.S.; Abhishek, K. Simulation and optimization of machining
334 parameters in drilling of titanium alloys. *Sim. Mod. Prac. Th.* **2016**, *62*, 31–48.
- 335 31. Azeem, A.; Feng H.Y.; Wang, L. Simplified and efficient calibration of a mechanistic cutting force
336 model for ball-end milling. *Int. J. Mach. Tools Manuf.* **2004**, *44*, 291–298.
- 337 32. Altintas, Y. (2000) Manufacturing automation. Cambridge University press.
- 338 33. Budak, E.; Altintas, Y.; Armarego, E.J.A. Prediction of milling force coefficients from orthogonal
339 cutting data. *ASME J. Manuf. Sci. Eng.* **1996**, *118(2)*, 216–224
- 340 34. Roukema, J.C.; Altintas, Y. Time domain simulation of torsional–axial vibrations in drilling. *Int.*
341 *J. Mach. Tools Manuf.* **2006**, *46(15)*, 2073–2085.
- 342 35. Beranoagirre, A.; Olvera, D.; López de Lacalle, L.N.; Urbicain, G. Drilling of intermetallic alloys
343 gamma TiAl, AIP Conference Proceedings 1315 (1), 1023–1028.
- 344 36. Beranoagirre, A.; Urbicain, G.; Calleja, A.; López de Lacalle, L.N. Hole Making by Electrical
345 Discharge Machining (EDM) of γ -TiAl Intermetallic Alloys. *Metals* **2018**, *8(7)*, 543.