

1 Article

# 2 Distributed deformation monitoring for a single-cell 3 box girder based on distributed long-gage fiber bragg 4 grating (LFBG) sensors

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13 **Abstract:** Distributed deformation based on Fiber Bragg Grating sensors or other kinds of strain sensors can  
14 be used to evaluate safety in operating periods of bridges. However, most of the published researches about  
15 distributed deformation monitoring are focused on solid rectangular beam rather than box girder—a kind of  
16 typical hollow beam widely employed in actual bridges. Considering that the entire deformation of a  
17 single-cell box girder contains not only bending deflection but also two additional deformations respectively  
18 caused by shear lag and shearing action, this paper again revises the improved conjugated beam method  
19 (ICBM) based on the LFBG sensors to satisfy the requirements for monitoring two mentioned additional  
20 deformations. The best choice for the LFBG sensor placement in box girder is also proposed in this paper due  
21 to strain fluctuation on flange caused by shear lag effect. Results from numerical simulations show that most  
22 of the theoretical monitoring errors of the revised ICBM are 0.3%-1.5%, and the maximum error is 2.4%. A  
23 loading experiment for a single-cell box girder monitored by LFBG sensors show that most of the practical  
24 monitoring errors are 6%-8%, and the maximum error is 11%.

25 **Keywords:** deformation monitoring; distributed monitoring; single-cell box girder; long-gage  
26 strain; long-gage Fiber Bragg Grating; strain distribution; shear lag effect; shear action

27

## 28 1. Introduction

29 Monitored deformation is usually used as an effective index not only to evaluate the overall  
30 health and safety of the in-service bridges but also to prevent some abnormal states due to the  
31 inextricable relationship between the deformation and the stress/strain distribution. Geodetic  
32 survey using digital level or total station has been widely applied to directly measuring bridge  
33 deformation [1,2] because of its low cost and easy operation. The main disadvantages of geodetic  
34 survey are possible obstruction to public traffic when the survey is ongoing and measurement error  
35 from manual observation. Recently, some automatic monitoring techniques, such as Global  
36 Positioning System [3,4], displacement sensors[5], hydrostatic leveling system[6] and laser  
37 measurements[7] are applied to gain bridge deformation. However, these sensors may be also  
38 disturbed by some environmental factors including bad weather, accidental vibration or satellite  
39 ephemeris error. In addition, these techniques used in deformation monitoring are criticized as the  
40 characteristics of “point” sensing, which implies that they can only collect displacements of a few  
41 pre-designed points. In practical monitoring, these “point” sensing techniques may ignore some  
42 damages occurred in other positions. Therefore, installing large numbers of “point” sensors to  
43 obtain distributed deformation may result in cost overruns in long-term monitoring. The best  
44 solution to keep balance between comprehensive monitoring and its cost is to replace these “point”  
45 sensors with some kind of distributed sensors.

46 In recent years, some indexes including slope variation [8] or strain distribution [9] before or  
47 after applied loads are introduced to indirectly calculate bridge deformation. Considering that the  
48 slope is the first derivative of bending deflection in an Euler–Bernoulli beam, an  $n$  degree  
49 polynomial used to approximate to the bridge deflection can be differentiated once to an  $n-1$  degree  
50 polynomial. Then the  $n$  monitored slopes and their position coordinates can be substituted to the  
51  $n-1$  degree polynomial to form  $n-1$  degree polynomial equations. The answer of the equations is the  
52 bridge deflection. This method is only applicable to small and single-span bridges. In case of  
53 long-span continue beam bridges, it still needs to deploy numbers of expensive and high-precise  
54 inclinometers. Moreover, the double integration method(DIM) can also achieve bending deflection  
55 curve of an Euler–Bernoulli beam by double integrating strain distribution and the cost of  
56 distributed strain sensors are obviously lower than that of high-precise inclinometers. Results from  
57 model tests of simple-supported beam show that the maximum difference between the monitoring  
58 displacement and the true value is only about 3% [10–12]. Nevertheless, according to the data from  
59 a practical deflection monitoring on a multi-span beam bridge, the monitoring error in the second  
60 span can rise to over 15%, which is clearly higher than the difference about 3% in the first  
61 span[13,14], because measurement errors accumulate in double integrating process. In order to  
62 solve this problem, an improved conjugated beam method(ICBM) has been proposed to deduce the  
63 influence of error accumulation[15].

64 Though the mentioned methods for deflection monitoring are suitable for solid rectangular  
65 beam suffering bending moment, the applicability may be challenged in case of box girder seen as a  
66 kind of typical hollow beam widely employed in long-span bridges because of two remarkable  
67 additional deformations caused by shear lag effect and shearing action. For one thing, similar  
68 results from different researches [16–18] illustrate that the first additional deformation (AD1) caused  
69 by shear lag effect at the mid-span of a simply supported box girder can easily approach 10% of the  
70 bending deflection when the height/span ratio exceeds 0.1. For another, the second additional  
71 deformation (AD2) caused by shear action can also reach 10% of the bending deflection when the  
72 shear span-depth ratio is lower than 1/20[19]. In brief, the existing methods based on the distributed  
73 strain measurements, including DIM or ICBM, have to be revised to obtain AD1 or AD2 owing to  
74 the differences between the box girder and the Euler–Bernoulli beam.

75 Moreover, there is another significant argument of obtaining enough strain data by installing  
76 minimized quantity of strain sensors to cover the entire box girder. For this problem, long-gage  
77 fiber bragg grating strain sensor [20,21] is proved to be an acceptable solution. The most notable  
78 advantage of this sensor is that it can achieve the average strain of a long distance which can reach  
79 0.1m to 10m. It implies that the entire beam can be covered by a few of these sensors. Based on the  
80 concept of long-gage strain sensor, a packaged design of LFBG sensor [22] and sensitivity-improved  
81 LFBG sensor [23] are proposed to catch slight strain variation in practical monitoring. The LFBG  
82 sensor is also verified to be applicable to measure not only dynamic strain [24] but also dynamic  
83 displacement [25]. Therefore, the LFBG sensor is seen as a useful tool for high-precision strain  
84 measurement with relatively low cost.

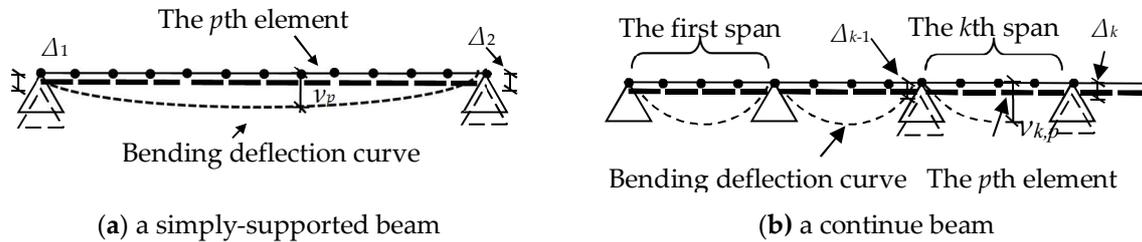
85 This paper is organized as follows. Section 2 introduces the ICBM proposed in our previous  
86 research and the LFBG strain sensor used in the experiment in Section 5. Section 3 revises ICBM to  
87 monitor AD1 and AD2 based on long-gage strain measuring. Section 4 gives the best position for  
88 the LFBG sensor placement. At last, experiments using numerical models and a concrete  
89 reduce-scale box girder monitored by LFBG sensors are shown Section 5 to evaluate the theoretical  
90 and actual precision of the revised ICBM.

## 91 2. ICBM and LFBG sensor

### 92 2.1. Improved conjugated beam method

93 Based on the long-gage strains, ICBM [15] is proposed to provide a linear and explicit  
94 relationship between bending deflection and strain distribution, which can be seen as the most  
95 attractive benefit rather than DIM which can only give an implicit and double-integrated function.

96 Therefore in ICBM, it is easy to predict error accumulation from each monitoring parameter. A  
 97 simply-supported solid beam is shown in Figure 1(a). It has length  $L$  and uniform flexural rigidity  $EI$ .  
 98 The beam is uniformly divided into  $m$  elements artificially, denoted as  $E_1 \sim E_m$ . The height of the beam  
 99 is  $h$ . The average strains at the top surface and the bottom surface of the  $E_i$  are  $\varepsilon_i^B$  and  $\omega_i^B$  ( $1 \leq i \leq m$ ),  
 100 respectively. Superscript B implies that the variable is used under the pure bending.



(a) a simply-supported beam

(b) a continue beam

101 **Figure 1.** Schematic diagram of a simply-supported solid beam and a continuous multi-spans beam

102 Without support settlement, the vertical displacement  $v_p^B$  at the boundary point between  $E_p$  and  
 103  $E_{p+1}$  is:

$$104 \quad v_p^B = -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\kappa}_i^B \left( m-i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\kappa}_i^B \left( p-i + \frac{1}{2} \right) \right] \quad (1)$$

105 where  $\bar{\kappa}_i^B$  is the average curvature of  $E_i$ , which can be calculated by Equation(2). Tensile strain and  
 106 upward deflection are defined to be positive in this paper.

$$107 \quad \bar{\kappa}_i^B = \frac{\bar{\omega}_i^B - \bar{\varepsilon}_i^B}{h} \quad (2)$$

108 Considering combined action of arbitrary loads and support settlements, the vertical  
 109 displacement  $v_p^B$  can be revised as follows:

$$110 \quad v_p^B = -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\kappa}_i^B \left( m-i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\kappa}_i^B \left( p-i + \frac{1}{2} \right) \right] + \frac{m-p}{m} \Delta_0 + \frac{p}{m} \Delta_1 \quad (3)$$

111 where  $\Delta_0$  and  $\Delta_1$  are two support settlements respectively, which can be measured by displacement  
 112 meters.

113 ICBM is also adaptive to monitor bending deflection in multi-span bridge shown in Figure 1(b).  
 114  $\Delta_{k-1}$  and  $\Delta_k$  represent settlements occurred at two supports of the  $k$ th span of this bridge. The  $k$ th span  
 115 is equally divided into  $n$  elements named as  $E_{k,1} \sim E_{k,n}$ . The length and the height of the  $k$ th span are  $L_k$   
 116 and  $h_k$ , respectively. The average strains at the top surface and the bottom surface of the  $E_{k,i}$  are  $\bar{\varepsilon}_{k,i}^B$   
 117 and  $\bar{\omega}_{k,i}^B$  ( $1 \leq i \leq n$ ), respectively. So the vertical displacement  $v_{k,p}^B$  at the boundary point between  $E_{k,p}$  and  
 118  $E_{k,p+1}$  in the  $k$ th span is:

$$119 \quad v_{k,p}^B = -\frac{L_k^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\kappa}_{k,i}^B \left( n-i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\kappa}_{k,i}^B \left( p-i + \frac{1}{2} \right) \right] + \frac{m-p}{m} \Delta_{k-1} + \frac{p}{m} \Delta_k \quad (4)$$

120 where  $\bar{\kappa}_{k,i}^B$  is the average curvature of  $E_{k,i}$ , which can be calculated by Equation (5).

$$121 \quad \bar{\kappa}_{k,i}^B = \frac{\bar{\omega}_{k,i}^B - \bar{\varepsilon}_{k,i}^B}{h_k} \quad (5)$$

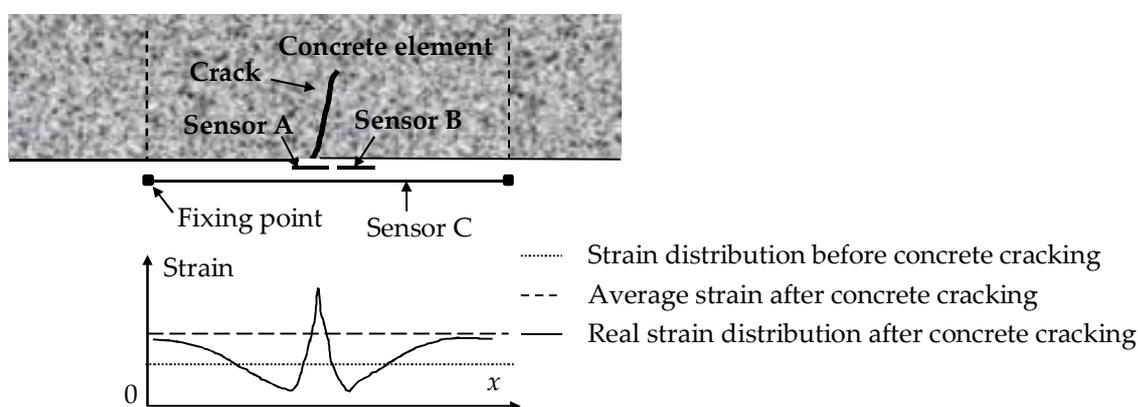
122 From Equation (1) to Equation (5), two remarkable features of ICBM can be summarized as  
 123 follows:

- 124 I. The formula of ICBM is linear and explicit. All parameters are free from actual load patterns or  
 125 flexural rigidity of the monitored beam.
- 126 II. Precision of bending deflection monitoring in one span is just related to the measurement  
 127 errors of strain distribution in the same span. It implies that measurement error accumulation  
 128 of one span cannot affect monitoring results of bending deflection in other spans.

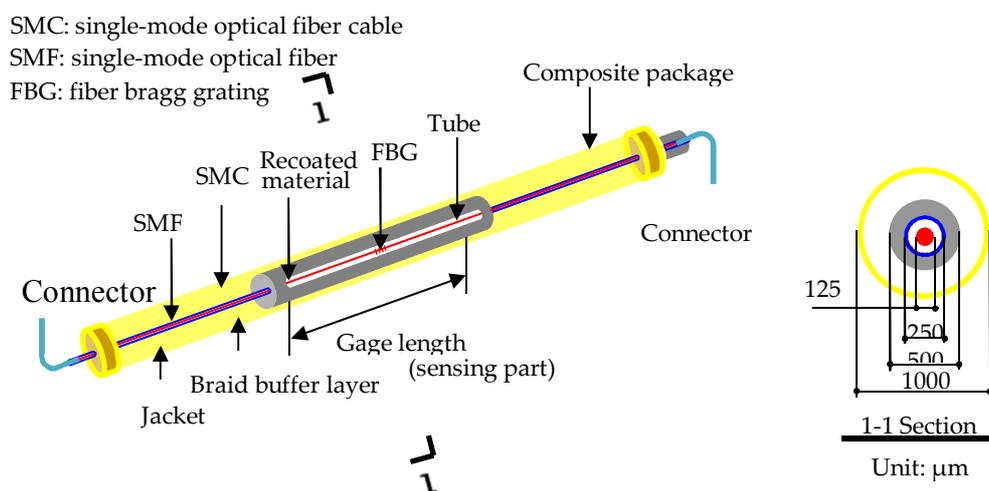
129 Moreover, it is noted that the ICBM is effective on a basic assumption of "solid beam". It  
 130 represents that the beam has enough rigidity to keep a changeless shape regardless of loading mode  
 131 (LM). However, when minor change occurs on the cross-section shape of box girder under the  
 132 action of LM changing, ICBM needs to be updated again.

## 133 2.2. LFBG strain sensor

134 In concrete structures, precise and long-term strain monitoring is quite difficult due to concrete  
 135 crack. Figure 2 gives a comprehensive illustration to show the traditional strain measuring by  
 136 short-gage sensors (Sensor A and Sensor B) which are entirely bonded on the surface of the  
 137 structure with resin. Before concrete cracking, both of them catch the true concrete strain.  
 138 Nevertheless, after crack occurring, Sensor A is over stressed to snap because its gauge just covers  
 139 the crack. And strain in Sensor B is almost released at the same time. It is obvious that the  
 140 measurements from such short-gage sensor cannot represent the true strain increasing. Compared  
 141 with short-gage sensor, long-gage sensor has aroused increasing concern for its point fixation,  
 142 which implies that only two ends of the sensor are bonded to the specimen to form a uniform strain  
 143 distribution in its gauge. The most notable advantage of long-gage sensor is the fact that the sensor  
 144 can avoid sudden rupture induced by concrete crack because the sensing part in the sensor has a  
 145 little distance away from the concrete surface. Therefore, it is found that long-gage sensor has more  
 146 applicability than traditional short-gage sensor in strain monitoring for concrete structures.



**Figure 2.** Different strain measurements from three sensors before and after concrete cracking



**Figure 3.** Structural design of packaged LFBG sensor [22]

147 On the other hand, it can obviously save monitoring and maintenance cost if all sensor can  
 148 constitute a sensing network to share the same signal source, connecting wire and demodulation  
 149 system. Fiber Bragg Grating (FBG) sensor, which is characterized by distributed sensing along a  
 150 single optical fiber, exactly meets the mentioned requirements. Based on FBG sensor, Li [22]  
 151 proposed Long-gage Fiber Bragg Grating (LFBG) sensor which successfully interweaves the  
 152 precision and distributed sensing characteristic of FBG sensor and the applicability of long-gage

153 sensor in long-term strain monitoring. Considering these benefits, LFBGS shown in Figure 3 is used  
154 in this paper to measure long-gage strain in each element of box girder.

### 155 3. Theoretical improvement for ICBM to monitor AD1 and AD2 in a single-cell box girder

156 Results from theoretical derivation and numerical simulation indicate that the total  
157 deformation of a box girder can be devoted to three parts – bending deflection which is the main  
158 portion in the total deformation, AD1 caused by shear lag effect and AD2 caused by shear action.  
159 Considering bending deflection can be obtained by ICBM, this part derives the relationship  
160 between AD1, Ad2 and distributed long-gage strain measurement. The deriving process is based on  
161 the following assumptions. First, material used is isotropic and elastic. Second, the stress-strain  
162 curve of material used is linear. Third, shear lag effect can only affect stress distribution on  
163 cross-section and it cannot influence stress distribution in the longitudinal direction. Fourth, total  
164 deformation of the single-cell box girder is still relatively small. At last, torsion, torsional warping  
165 and distortion are ignored in the derivation of this paper.

#### 166 3.1. AD1 calculating based on strain distribution

167 Shear lag effect represents a phenomenon on cross-section of box girder that the longitudinal  
168 stress on a flange near the web is much larger than that far from the web. Obviously, it is quite  
169 different from the uniform stresses distribution assumption in the elementary beam theory. This  
170 phenomenon also implies that the flange far from the web on the cross-section is barely contributed  
171 to flexural rigidity. Therefore, an additional curvature occurs on of the section due to the extra  
172 decrease in flexural rigidity calculated according to the elementary beam theory. AD1 is the  
173 macroscopic result from accumulation of additional curvature.

174 Results from numerical simulations [17] point out that the practical curvature of the section  
175 equals to the product of  $\lambda$  and curvature calculated according to the elementary beam theory,  
176 where  $\lambda$  is shear lag coefficient defined as the ratio of the normal stresses to those obtained  
177 according to the elementary beam theory. This conclusion has two meanings. One is that AD1 is a  
178 special bending deflection essentially and the equation (1) is applicable to describe the relationship  
179 between AD1 and the additional curvature increments in all elements. The other is that the actually  
180 measured average strains are also the products of  $\lambda$  and strains calculated according to the  
181 elementary beam theory, which is shown as  $\bar{\beta}_i (\bar{\beta}_i = \lambda_i \bar{\varepsilon}_i^B)$  and  $\bar{\alpha}_i (\bar{\alpha}_i = \lambda_i \bar{\omega}_i^B)$ . Therefore, the curvature  $\bar{\kappa}_i^B$   
182 in Equation (2) has to be replaced by  $\bar{\gamma}_i (\bar{\gamma}_i = \lambda_i \bar{\kappa}_i^B)$ , where  $\lambda_i$  is shear lag coefficient of  $E_i$ . Superscript  
183 SL implies that the variable is used under shear lag action. Then this expression of  $\bar{\gamma}_i$  is shown as  
184 follows:

$$185 \quad \bar{\gamma}_i = \lambda_i \bar{\kappa}_i^B = \bar{\kappa}_i^B + \bar{\kappa}_i^{SL} = \bar{\kappa}_i^B + (\lambda_i - 1) \bar{\kappa}_i^B = \frac{\bar{\omega}_i^B - \bar{\varepsilon}_i^B}{h} + (\lambda_i - 1) \frac{\bar{\omega}_i^B - \bar{\varepsilon}_i^B}{h} = \frac{\lambda_i \bar{\omega}_i^B - \lambda_i \bar{\varepsilon}_i^B}{h} = \frac{\bar{\alpha}_i - \bar{\beta}_i}{h} \quad (6)$$

186 As a result, the AD1 represented as  $v_p^{SL}$  at the boundary point between  $E_p$  and  $E_{p+1}$  is:

$$187 \quad \begin{aligned} v_p^{SL} &= -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\kappa}_i^{SL} \left( m - i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\kappa}_i^{SL} \left( p - i + \frac{1}{2} \right) \right] \\ &= -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m (\lambda_i - 1) \bar{\kappa}_i^B \left( m - i + \frac{1}{2} \right) - \sum_{i=1}^p (\lambda_i - 1) \bar{\kappa}_i^B \left( p - i + \frac{1}{2} \right) \right] \\ &= -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \lambda_i \bar{\kappa}_i^B \left( m - i + \frac{1}{2} \right) - \sum_{i=1}^p \lambda_i \bar{\kappa}_i^B \left( p - i + \frac{1}{2} \right) \right] + \frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\kappa}_i^B \left( m - i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\kappa}_i^B \left( p - i + \frac{1}{2} \right) \right] \\ &= -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\gamma}_i \left( m - i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\gamma}_i \left( p - i + \frac{1}{2} \right) \right] - v_p^B \end{aligned} \quad (7)$$

188 Consequently, the sum of  $v_p^B$  and  $v_p^{SL}$  is given by following:

$$189 \quad v_p^B + v_p^{SL} = -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\gamma}_i \left( m - i + \frac{1}{2} \right) - \bar{\gamma}_i \left( p - i + \frac{1}{2} \right) \right] \quad (8)$$

190 Considering combined action of arbitrary loads and support settlements, the sum of  $v_p^B$  and  $v_p^{SL}$  is  
191 revised as follows:

$$v_p^B + v_p^{SL} = -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\gamma}_i \left( m-i + \frac{1}{2} \right) - \sum_{i=1}^p \bar{\gamma}_i \left( p-i + \frac{1}{2} \right) \right] + \frac{(m-p)}{m} \Delta_0 + \frac{p}{m} \Delta_1 \quad (9)$$

### 3.2. AD2 calculating based on strain distribution

Before the deriving process of AD2's expression, it is worth discussing a relative problem which is whether shear action brings extra longitudinal strain. It is proposed in the material mechanics that the extra longitudinal strain can be ignored in a slender beam subjected to a uniformly distributed load. When the beam is subjected to a concentrated load, the influence in strain from shear action is still approximate to zero except for the area near to the supporting points. Therefore, it is conclude that the longitudinal strain is free from shear action in the deriving process of AD2's expression.

Timoshenko [26] points out that the first derivative of  $v^S(x)$  with respect to  $x$  equals to the shear strain in neutral axis of cross-section. Superscript S implies that the variable is used under shear action. This point can be expressed as follows:

$$\frac{dv^S(x)}{dx} = -\frac{\eta V(x)}{AG} \quad (10)$$

where  $x$  is the longitudinal coordinate.  $G$  and  $A$  are shear modulus of material and cross-section area, respectively.  $V(x)$  is shear force along the section.  $\eta$  is shear correction factor which equals to the ratio of shear stress  $\tau_{NA}$  on neutral axis to average shear stress  $\bar{\tau}$  on the entire section.

Because shear force  $V(x)$  is the first derivative of moment  $M(x)$ , Equation (10) can be transformed into:

$$\frac{dv^S(x)}{dx} = -\frac{\eta}{AG} \frac{dM(x)}{dx} \quad (11)$$

Integrating on both sides of Equation (11) from 0 to  $x$  and considering both  $v^S(0)$  and  $M(0)$  equal to 0 in case of simply-supported condition, it is obtained as follows:

$$v^S(x) = -\frac{\eta}{AG} M(x) \quad (12)$$

Shear correction factor  $\eta$  can be expressed by following:

$$\eta = \frac{\tau_{NA}}{\bar{\tau}} = \frac{VS_y}{2t_{NA}I\bar{\tau}} = \frac{AS_y}{2t_{NA}I} \quad (13)$$

where  $S_y$  is the first moment with respect to the neutral axis of the area on one side of the neutral axis.  $V$  and  $I$  respectively represent shear force on neutral axis and moment of inertia of the entire cross-sectional area.  $t_{NA}$  is width of web, which is measured on the same height of neutral axis.

Substituting Equation (13) into Equation (12), Equation (12) can be simplified as follows:

$$v^S(x) = -\frac{S_y}{2Gt_{NA}I} \cdot M(x) = -\frac{(1+\mu)S_y}{t_{NA}} \cdot \frac{M(x)}{EI} = -\frac{(1+\mu)S_y}{t_{NA}} \cdot \kappa^B(x) = -\frac{(1+\mu)S_y}{\lambda t_{NA}} \cdot \gamma(x) \quad (14)$$

where  $E$  and  $\mu$  respectively represent the elastic module and Poisson's ratio of material.

Therefore, the AD2 named as  $v_p^S$  at the boundary point between  $E_p$  and  $E_{p+1}$  is:

$$v_p^S = -\frac{(1+\mu)S_y}{(\lambda_p + \lambda_{p+1})t_{NA}} \cdot (\bar{\gamma}_p + \bar{\gamma}_{p+1}) \quad (15)$$

It is noted that the shear lag coefficient  $\lambda$  of each element is predetermined before calculating Equation (13). However, the value of  $\lambda$  fluctuates according to different LMs and different positions. Besides, it is difficult to identify the accurate LM in the real structure in practical monitoring. As a result, keeping a constant value for  $\lambda$  in entire calculating process is a simple and feasible solution. On the basis of references [16,27], Table 1 gives recommended value of  $\lambda$  for a single-cell beam under simply-supported condition.

**Table 1.** Recommended value of  $\lambda$  for a single-cell beam under simply-supported condition [27]

$L/b^1$	6	8	$\geq 10$
$\lambda$	1.22	1.15	1.10

231 <sup>1</sup>  $L$  and  $b$  are the entire length of the beam and the width of flange, respectively.

232 So Equation (13) can be simplified as follows:

$$233 \quad v_p^S = -\frac{(1+\mu)S_y}{2\lambda t_{NA}} \cdot (\bar{\gamma}_p + \bar{\gamma}_{p+1}) \quad (16)$$

234 Finally, the entire deformation  $v$  including bending deflection, AD1 and AD2 are obtained by  
235 following:

$$236 \quad v = v_p^B + v_p^{SL} + v_p^S = -\frac{L^2}{m^2} \left[ \frac{p}{m} \sum_{i=1}^m \bar{\gamma}_i \left( m-i+\frac{1}{2} \right) - \sum_{i=1}^p \bar{\gamma}_i \left( p-i+\frac{1}{2} \right) \right] - \frac{(1+\mu)S_y}{2\lambda t_{NA}} \cdot (\bar{\gamma}_p + \bar{\gamma}_{p+1}) + \frac{(m-p)}{m} \Delta_0 + \frac{p}{m} \Delta_1 \quad (17)$$

#### 237 4. The best choice for LFBG sensor placement

238 As shown in Figure 4(a), strain distribution on the flange of a box girder is nonuniform due to  
239 action of shear lag effect. It illustrates that strain sensor placement in the box girder deserve to be  
240 deliberately discussed. As illustrated in Figure 4(b), there are five possible locations from A to E for  
241 placement of LFBG sensor. Nevertheless, two preconditions are worth considering. The first one is  
242 that fixing sensor on the outer surface of a box girder is usually more available than that on the  
243 inner surface of a box girder. The other one is that the strain distribution around the practical fixing  
244 location needs to follow the plane-section assumption and to be far from influence of shear lag.  
245 According to the two preconditions, it is obvious that C+D may be the most suitable choice for the  
246 LFBG sensor placement in practice.

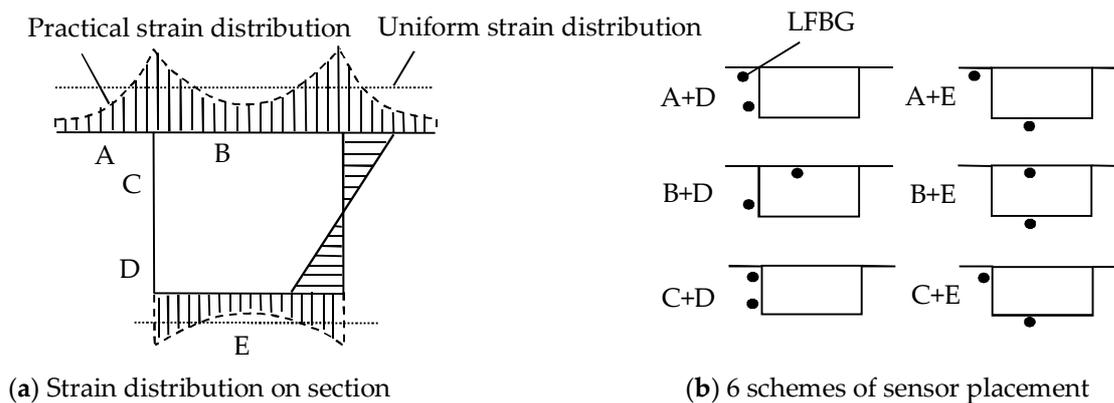


Figure 4. Several schemes of sensor placement base on strain distribution

#### 247 5. Verification of revised ICBM: Numerical simulation

248 There are two types of errors which can decrease the accuracy of monitored deformation. One  
249 is algorithm error from inaccuracy of simulation for real structures, and the other is measuring  
250 error in practical strain monitoring. In this verification, a numerical model simulating a real  
251 single-cell box girder under different loading modes (LMs) is used to evaluate the influence of  
252 algorithm error from revised ICBM. Evaluation for effect of measuring error accumulation in  
253 experimental test is carried out in the next part.

##### 254 5.1. Detailed design for the numerical model, sensor placement and loading mode

255 The numerical model is built based on the solid45 element in ANSYS software. Details about a  
256 single-cell concrete box girder with single-supported boundary condition are illustrated in Figure 5.  
257 The dimensions of the cross-section are as follows:  $b=400\text{mm}$ ,  $t_u=t_w=t_b=t_{NA}=50\text{mm}$ ,  $h=300\text{mm}$ ,  
258  $L=3600\text{mm}$ . The compressive strength of concrete is  $23.1\text{N/mm}^2$ . The elastic module and Poisson's  
259 ratio of concrete are  $34.5\text{GPa}$  and  $0.2$ , respectively. This beam is uniformly divided into 18 elements,  
260 denoted as  $E_1 \sim E_{18}$ .

261 There are three different LMs applied to the model, which are uniform loading, loading at  
262 midpoint and loading at trisection points, respectively. According to Table 1,  $\lambda$  is 1.1 due to

263  $L/b=3600/400=9$ . Surface load  $q$  is  $20\text{kN/m}^2$ . Linear loads include  $f_0$ ,  $f_1$  and  $f_2$ , whose values are  $41.25$   
 264  $\text{kN/m}$ ,  $41.25\text{ kN/m}$  and  $20.625\text{ kN/m}$ , respectively.

265 A series of long-gage strain sensors are simulated to fixed at the two horizontal edges of each  
 266 element in the web. The distance  $d$  between the upper part and lower part of sensors is  $250\text{mm}$ .

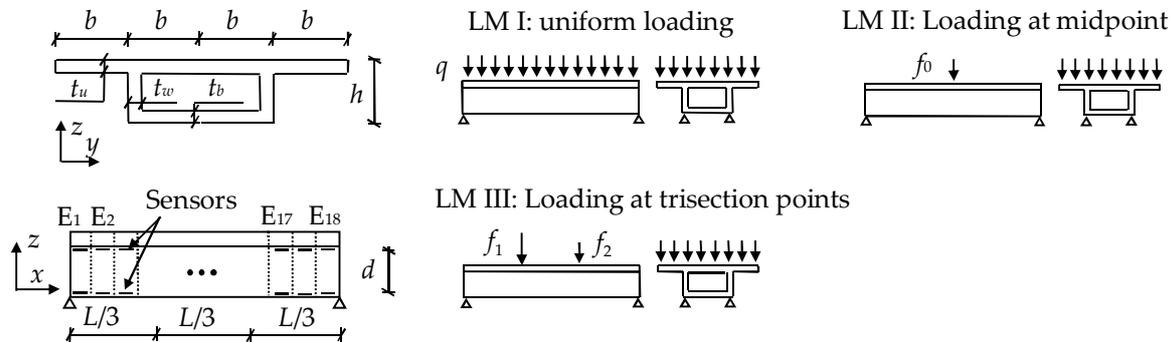


Figure 5. Detailed design for the numerical model of a single-cell box girder in 3 different LMs

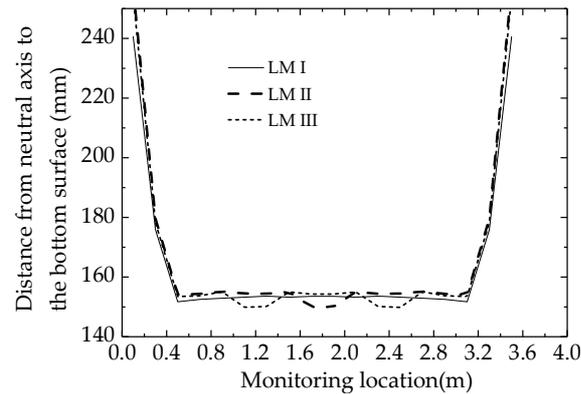
267 5.2. Results and Analysis

268 Long-gage strain measurements of 18 elements in different LMs are given in Table 2. It is noted  
 269 that in each element, compressive strain occurs in upper part and tensile strain occurs in lower part.  
 270 Figure 6 gives the distances between the neutral axis and the bottom surface of each element in 3  
 271 different LMs. Calculation formula is given in Equation (18). It is clear that the height of neutral  
 272 axes of E3~E16 keep almost in the range of  $152\text{mm}\sim 153\text{mm}$  except for small variation occurring in  
 273 some elements near concentrated loads such as E9, E10 in LM II and E6, E7, E12, E13 in LM III. Due  
 274 to boundary restraints, tensile strains of E1 and E18 are larger than those of other elements. This  
 275 leads to the result that the height of neutral axes of E1, E2, E17 and E18 are obvious larger than  
 276 those of other elements. Therefore, the partial and small variation about height of neutral axes  
 277 shows the approximate application of the plane-section assumption in web.

278 Distance in each element =  $\frac{\bar{\alpha}_i/\bar{\beta}_i}{1+\bar{\alpha}_i/\bar{\beta}_i} \cdot d$  (18)

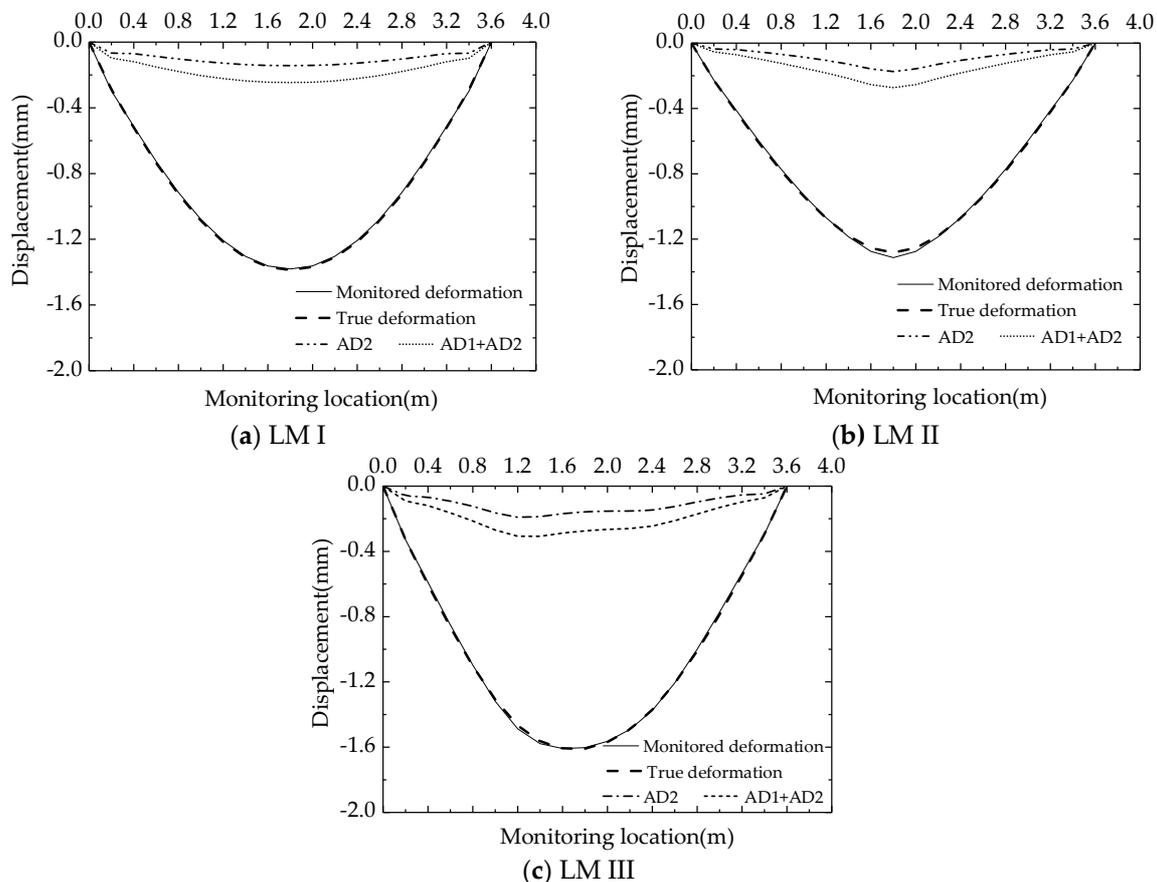
279 Table 2. Long-gage strains at the top and bottom of each element on the finite element model of the  
 280 box girder (Unit:  $\mu\epsilon$ )

LM	E1	E2	E3	E4	E5	E6	E7	E8	E9	E10	E11	E12	E13	E14	E15	E16	E17	E18	
1	$\bar{\beta}_i$	-5	-35	-57	-69	-80	-89	-96	-101	-103	-103	-101	-96	-89	-80	-69	-57	-35	-5
	$\bar{\alpha}_i$	128	83	88	108	126	141	153	160	164	164	160	153	141	126	108	88	83	130
2	$\bar{\beta}_i$	2	-18	-33	-44	-55	-68	-83	-101	-130	-130	-101	-83	-68	-55	-44	-33	-18	2
	$\bar{\alpha}_i$	71	46	53	71	90	110	134	165	195	195	165	134	110	90	71	53	46	73
3	$\bar{\beta}_i$	2	-31	-56	-77	-101	-139	-147	-125	-116	-110	-108	-113	-103	-78	-60	-44	-24	2
	$\bar{\alpha}_i$	119	77	89	123	165	208	221	204	187	179	177	175	159	128	97	71	61	97



**Figure 6.** Distance from the neutral axis to the bottom surface of each element in different LMs

281 A comparison is made between monitored deformations and true deformations under the  
 282 different LMs, which are shown in Figure 7(a)~(c). The dash lines and the solid lines respectively  
 283 represent the true deformations and the monitored deformations calculated by substituting strain  
 284 data in Table 2 to Equation 17 ( $\lambda=1.1$ ). The dot dash lines and the dot lines in Figure 7 are AD2 and  
 285 sum of AD1 and AD2, respectively. It is found that either AD1 or AD2 accounts for about 10% of the  
 286 total deformations. These proportions illustrate that neither of AD1 or AD2 can be ignored in this  
 287 model. Moreover, most of deviations between monitored deformations and true deformations are  
 288 about 0.3%~1.5% except those of E9, E10 in LM II which are 2.4%. In fact, the accurate value of  $\lambda$   
 289 under the condition of uniform loading is about 1.069~1.079, which is quite close to the constant  
 290 value of 1.10. The accurate value of  $\lambda$  in E9, E10 under the condition of concentrated loading is 1.266,  
 291 which is larger than 1.10. This deviation may be the main reason leading to the error of 2.5% in  
 292 calculated deformation of E9, E10 in LM II.



**Figure 7.** Comparison between monitored deformations and true deformations in case of different LMs

293 Table 3 gives monitoring errors between monitoring displacements and true displacements in  
 294 1/3 span, mid span and 2/3 span. It is evident that the monitoring errors may slightly reduce if  $\lambda$  is  
 295 substituted to its accurate value. The maximum error decreases from 2.4% to 0.6% at the mid span  
 296 in LM II. Because the LM in actual bridge is usually difficult to measure, it is an applicable solution  
 297 to give a constant value to  $\lambda$  in the calculating process with little influence to the precision of  
 298 deformation monitoring for a single-cell box girder.

299 Table 3 Monitoring errors between monitored displacements and true displacements in three  
 300 positions (unit :%)

LM	Value of $\lambda$	Positions		
		1/3 span	Mid span	2/3 span
I	$\lambda=1.1$	-0.6	-0.4	-0.6
	$\lambda=\text{accurate value}$	-0.5	-0.1	-0.5
II	$\lambda=1.1$	-0.3	2.4	-0.3
	$\lambda=\text{accurate value}$	0.5	0.6	0.5
III	$\lambda=1.1$	1.5	-0.3	0.3
	$\lambda=\text{accurate value}$	0.3	0.4	-0.6

## 301 6. Verification of revised ICBM: Experiment

302 This experiment has two main purposes. One is to show the effectiveness of revised ICBM to  
 303 obtain accurate deformation in actual single-cell concrete box girder. The other is to investigate the  
 304 possibility of replacing the true  $\lambda$  with a constant value in practical calculating process.

### 305 6.1 Test Setup and Sensor Placement

306 Details regarding the beam dimensions and reinforcement configuration of a simply-supported  
 307 single-cell box girder are illustrated in Figure 8. It can be found that the dimensions of the section  
 308 and the entire length of the beam are as same as the simulated model shown in Figure 5. The  
 309 compressive strength of concrete is about 39N/mm<sup>2</sup>. The elastic modulus and Poisson's ratio of  
 310 concrete are 3.03×10<sup>4</sup> N/mm<sup>2</sup> and 0.19, respectively. In Figure 8, 20 reinforcements 6 mm in diameter  
 311 and 11 passive reinforcements 12 mm in diameter are used for longitudinal bars, located 20 mm  
 312 away from the edges of the beam. There are also 2 passive reinforcements 6 mm in diameter in each  
 313 web of the beam. Stirrups are deployed throughout the entire length of the beam, with 6 mm  
 314 diameter and 100 mm distance in two adjacent vertical bars. The yield strength of the bars is about  
 315 380N/mm<sup>2</sup>. All mentioned material parameters are determinate by standard experiments. Moreover,  
 316 there are four steel baffles placed at two trisection points and two ends of the beam to prevent  
 317 torsional warping and distortion.

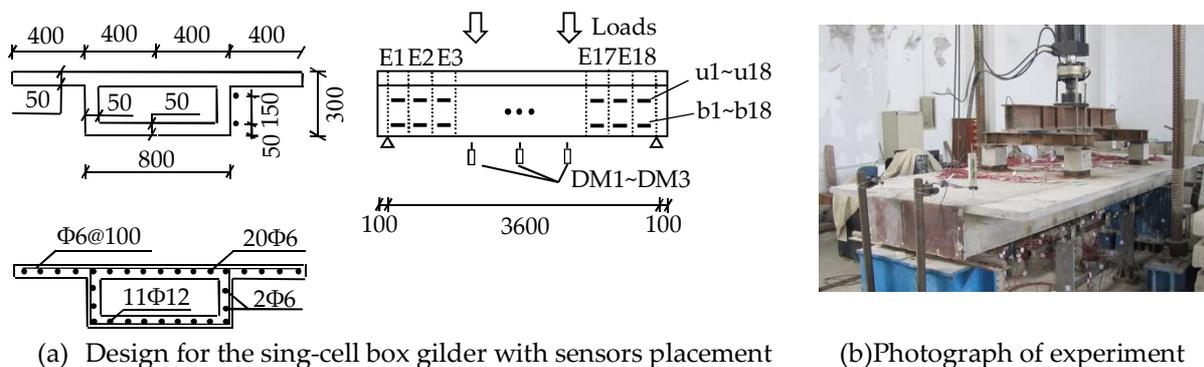


Figure 8. Outline of the sing-cell box girder and the experiment (unit: mm)

318 The single-cell box girder is artificially divided into 18 zones with a uniform length of 200 mm,  
 319 as denoted by Element1-Element18 (E1~E18). 36 LFBG sensors with a uniform length of 180mm are

320 installed on the surface of one web of the beam. Half of LFBG sensors named as b1~b18 are fixed at  
 321 the position 50mm higher than the bottom of beam. By contrast, the other half named as u1~u18 are  
 322 fixed at the position 200mm higher than the bottom of beam. Obviously, it has a distance of 150mm  
 323 between the two parts of sensors. Three displacement meters (DM1~DM3) are installed at the two  
 324 trisection points (point A, point C) and the midpoint B of the beam. Point A, B and C are the  
 325 boundary points between E6 and E7, E9 and E10, E12 and E13, respectively.

326 The load is divided equivalently into two parts by using a transferred steel board landing at two  
 327 points 1200 mm from each support. The increasing load is continuously applied by 5 successive  
 328 loading steps (LSs) from 0 to 10KN, 15KN, 20KN, 25KN and 30KN. The maximum measured strain  
 329 is ensured to be lower than  $100\mu\epsilon$  considered as the ultimate tensile strain of concrete. All strain  
 330 measurements are revised by temperature compensation.

## 331 6.2. Results and Analysis

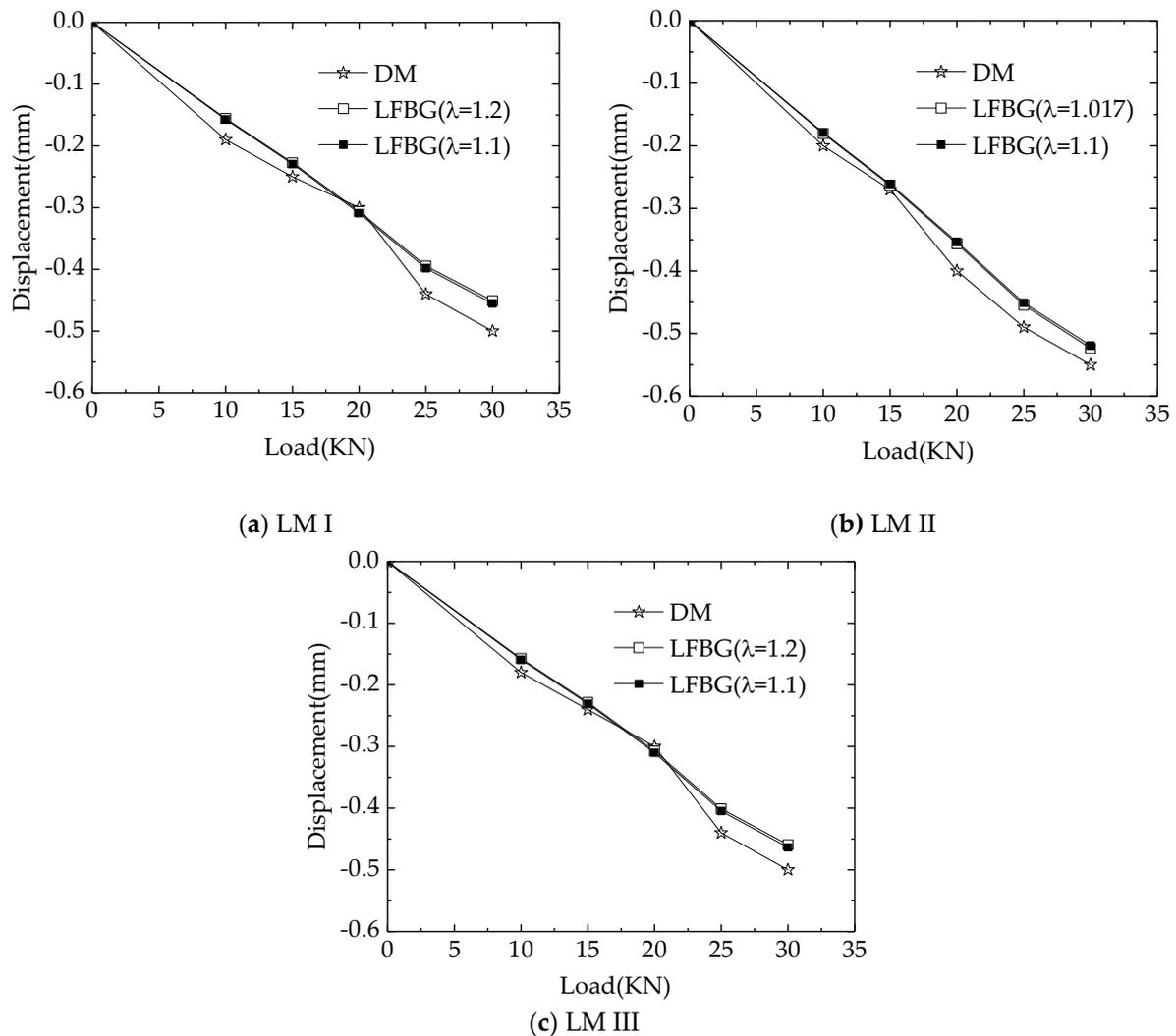
332 Table 4 gives the long-gage strain measurements from LFBG sensors placed on the surface of  
 333 each element in different LSs. It is evident that in each element, the measured strains from b1~b18  
 334 are positive, whereas the measured strains from u1~u18 are negative. This trend illustrates that the  
 335 testing single-cell box girder mainly suffers increasing bending moments as rising loads. The  
 336 bending moments brings tension to the lower part of the beam and compression to the upper part of  
 337 the beam. Figure 9(a)~(c) respectively gives the comparisons between the monitored displacements  
 338 of point A, B and C in different LSs, which are calculated from the strain measurements given in  
 339 Table 4. Upward deformation is defined to be positive. The symbol  $\blacksquare$  means that  $\lambda$  equals to a  
 340 constant value of 1.1 according to Table 1 during the calculating process. The symbol  $\square$  represents  
 341 that  $\lambda$  equals to the accurate values at corresponding monitoring points. At point A, C and point B,  
 342 the accurate value of  $\lambda$  is 1.2 and 1.017, respectively. Table 5 shows the comparison of monitoring  
 343 error percentages between the monitored displacements and the true displacements in each LS. It  
 344 can be noted that the monitored displacements agree well with the true displacements. Most of the  
 345 errors range from 6%~8%, and the maximum monitoring error is only 11.0%. This agreement  
 346 implies the applicability of revised ICBM for deformation monitoring to a single-cell box girder  
 347 based on LFBG sensors. In addition, the fact that influence from different values of  $\lambda$  in deformation  
 348 monitoring is almost lower than 1% shows that  $\lambda$  can be determinate approximately to a constant  
 349 value according to Table 1 to not only reduce the difficulty of parameter determining but also  
 350 ensure accuracy of monitored deformations. Moreover, comparison between monitoring errors  
 351 from Table 3 and Table 5 shows that the precision of monitored deformations is depended on the  
 352 precision of sensors rather than algorithm error. It illustrates the importance of long-term  
 353 monitoring precision and durability of long-gage sensors in practical deformation monitoring.

354 **Table 4** Average strain measurements at the bottom and the top of each element (Unit:  $\mu\epsilon$ )

		E1	E2	E3	E4	E5	E6	E7	E8	E9	E10	E11	E12	E13	E14	E15	E16	E17	E18
1	$\bar{\beta}_i$	-2	-2	-2	-3	-5	-2	-3	-6	-5	-6	-3	-2	-5	-3	-2	-2	-3	1
	$\bar{\alpha}_i$	1	3	5	7	10	13	14	13	12	11	13	15	13	11	7	6	3	2
2	$\bar{\beta}_i$	-2	-2	-4	-5	-6	-4	-5	-8	-7	-7	-8	-5	-4	-5	-6	-4	-2	0
	$\bar{\alpha}_i$	1	2	7	11	15	18	20	19	18	17	19	21	19	13	12	7	2	0
3	$\bar{\beta}_i$	-3	-3	-5	-7	-9	-6	-7	-11	-10	-10	-11	-7	-9	-8	-4	-3	-2	-2
	$\bar{\alpha}_i$	2	5	10	14	19	24	26	25	24	25	24	26	24	19	15	9	6	1
4	$\bar{\beta}_i$	-4	-4	-6	-8	-7	-8	-9	-15	-9	-12	-12	-14	-9	-10	-8	-6	-4	-2
	$\bar{\alpha}_i$	4	7	11	21	20	33	35	33	27	33	34	35	28	30	22	10	9	5
5	$\bar{\beta}_i$	-4	-5	-7	-9	-7	-9	-10	-17	-10	-15	-16	-9	-9	-13	-9	-7	-5	-2
	$\bar{\alpha}_i$	6	8	15	22	27	37	40	33	34	38	39	45	36	30	22	15	8	5

355  $\bar{\tau}\alpha_i$  and  $\bar{\tau}\beta_i$  are the practical strain measurements from LFBG  $b_i$  and LFBG  $u_i$  ( $i=1\sim 18$ ), respectively.

356



**Figure 9.** Comparison between monitored displacements and true displacements in case of different LMs

357  
358

**Table 5** Monitoring errors between monitored displacements and true displacements in different points (unit :%)

Loading step		1	2	3	4	5
Point A	$\lambda=1.1$	-10.2	-8.2	-6.4	-9.5	-9.0
	$\lambda=1.2$	-11.0	-9.1	-7.2	-10.4	-9.8
Point B	$\lambda=1.1$	-10.5	-3.3	-11.6	-7.9	-5.6
	$\lambda=1.017$	-9.8	-2.5	-10.9	-7.2	-4.8
Point C	$\lambda=1.1$	-9.1	-3.9	-6.0	-8.0	-7.3
	$\lambda=1.2$	-10.0	-4.8	-6.9	-8.9	-8.2

359

## 7. Conclusions

360  
361  
362  
363  
364

Based on the previous researches about ICBM used to monitor the deformation of solid beam, a revised ICBM is proposed in this paper to accurately gain the entire deformation of single-cell box girder with simply-supported boundary condition. The best position for sensor placement is also given for practical monitoring. Verifications using numerical simulations and a reduce-scale box girder monitored by a series of LFBG sensors are carried out to show the theoretical and practical

365 precision of the revised ICBM. After theoretical and experimental investigation, the following  
366 conclusions can be drawn:

367 (1) For a single-cell box girder, it is verified that revised ICBM, which can still present a linear  
368 and explicit function between the deformation and the long-gage strain distribution, is applicable to  
369 monitor the entire deformation which contains the bending deflection, AD1 caused by shear lag  
370 and AD2 caused by shear action.

371 (2) The LFBG sensor seen as a typical long-gage strain sensor is useful to not only achieve the  
372 strain distributing on the structural surface but also keep a balance between measurements and  
373 cost.

374 (3) In calculating process, the shear lag coefficient  $\lambda$  can be determinate to a constant value to  
375 not only avoid the difficulty of investigating loading mode but also ensure the practical precision of  
376 monitored deformation.

377 (4) Results from numerical simulations show that the most of algorithm errors are about 0.3%  
378 to 1.5%, and the maximum error is about 2.4%. Results from testing a single-cell box girder  
379 monitored by a series of LFBG sensors show that most of the practical errors range from 6%~8%,  
380 and the maximum error is about 11%. It implies that in practical monitoring, errors in monitored  
381 deformation are mainly induced by errors in strain measurements rather than algorithm error from  
382 revised ICBM.

383

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395

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