

On the Anisotropic Mechanical Properties of Selective Laser Melted Stainless Steel

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Abstract: The thorough description of the peculiarities of additively manufactured structures represents a current challenge for aspiring freeform fabrication methods, such as the selective laser melting (SLM). All of which have an immense advantage in the fast fabrication (no special tooling or moulds required), the geometrical flexibility in the design of components, and their efficiency when only low quantities are required. However, designs demand the precise knowledge of the material properties, which in case of additively manufactured structures are anisotropic and, under certain circumstances, in addition of an inhomogeneous nature. Furthermore, these characteristics are highly dependent on the fabrication settings. Within this study, the anisotropic tensile properties of selective laser melted stainless steel (1.4404, 316L) are investigated: The Young's modulus ranged from 148 GPa to 227 GPa, the ultimate tensile strength from 512 MPa to 699 MPa and the breaking elongation ranged, respectively, from 12% to 43%. The results were compared to related studies, in order to classify the influence of the fabrication settings. Furthermore, the influence of the chosen raw material was addressed by comparing deviations on the directional dependencies reasoned by differing microstructural developments during manufacture. Stainless steel was found to possess its maximum strength at a 45° layer versus loading offset, which is precisely where AlSi10Mg was previously reported to be at its weakest.

Keywords: Tensile Strength, Hardness, Microstructure, Grain Morphology, Epitaxial Grain Growth, Scan Strategy, Directional Dependencies

1. Introduction

Additive manufacturing (AM) methods, such as the selective laser melting (SLM), represent powerful freeform fabrication techniques. All of which have an immense advantage in the fast fabrication (no special tooling or moulds required), the geometrical flexibility in the design of components, and their efficiency, when only low quantities are required [1-3]. Since the full melting of the raw metal powder enables the generation of fully dense parts within a single production step, with mechanical properties exceeding the specifications of the conventional material, the fabrication of highly specialized components (like tools, moulds, ultra-lightweight components or medical implants) via AM is on the rise [4-7]. One

of the major challenges is, to date, the characterization and prediction of the properties of additively manufactured structures and their linkage with the selected fabrication settings [8]. The most utilized approach to describe the manufacturing process is via the energy input of the laser beam per unit volume, commonly referred to as the energy density [9].

$$\text{Energy density } W_s = \frac{\text{laser power [W]}}{\text{layer thickness [m]} \times \text{hatch distance [m]} \times \text{scan speed [m/s]}}$$

Unfortunately, it appears that this convenient approach via the characterization with a single number is not able to sufficiently express the entire complexity of powder-bed based AM processes, like SLM [10-12]. Thus, at this stage, a proper description of the manufacturing process still requires the listing of the individual irradiation parameters. In addition to the pure irradiation, the information of the raw metal powder, mainly its size and the distribution, is also of great importance and should not be neglected. Spierings, *et al.* [13] pointed out the necessity of having both small and large powder particles: Fine particles are easily molten and favour a good relative density and surface quality; whereas bigger powder particles benefit the ductility. On this note, the mechanical properties, such as hardness and tensile strength, greatly correspond with the relative density, which is, without a doubt, the most utilized characteristic to evaluate the quality of fabricated components [10]. To illustrate its importance, the aeronautic industry introduced a minimum relative density of 99% as a standardized quality requirement [14].

The properties of AM fabricated components are known to be anisotropic and, under certain circumstances, in addition of an inhomogeneous nature. In a previous study on a precipitation-hardenable aluminium-silicium alloy, it was documented that the anisotropic tensile strength characteristics were superimposed by varying age-hardening stages [15]. Varying statements and conclusions were reported in former studies on stainless steel. For example, the outcomes about the directional dependencies of the tensile strength differ widely. For the polar angle (the inclination to the layers) the findings range from the predominantly accepted formulation (the highest tensile strength and highest breaking elongation are found in a parallel layer to loading direction scenario, and the lowest results, with an almost linear tendency, are found when the loading direction is in-built direction, i.e. perpendicular to the layers), to the following particular findings [4,16]: Sehrt and Witt [17] reported the opposite case, with the highest breaking elongation being obtained in the perpendicular loading scenario. Rehme and Emmelmann [18] stated that the lowest results for both the ultimate tensile strength and the breaking elongation occurred under a 75° angle to the layers, whereas the maximum values for both were examined under a 15° angle. Tolosa, *et al.* [19] found an increase in strength by increasing the inclination angle from 30° to 45°, whereas Guan, *et al.* [20], on the other hand, reported a minimum strength occurring at an 45° inclination. Similarly differing findings also were reported for the in-plane dependencies. In general, the effects occurring in-plane are less pronounced; Sehrt and Witt [17] even described the influence as being negligible. However, there are various findings and it can be concluded that the in-plane dependencies are highly influenced by the chosen scan strategy, and thus, the need of their consideration is dependent on the individual settings [1]. Niendorf, *et al.* [21] stated that, in addition to the irradiation settings, the dimensions of the structure also affected the mechanical strength and altered the microstructure.

Within this study, the anisotropic material properties of stainless steel are examined with destructive material tests, since the characterisation of the anisotropic material properties via non-destructive procedures was found to be inadequate [22]. Moreover, the findings of these material tests were compared with the reported results in the literature to acquire a comprehensive overview about the inherent directional dependencies and their variation amongst various machinery and irradiation settings. Special consideration is given to the scan strategy settings and the microstructural development throughout the process.

2. Methodology

2.1. Manufacturing conditions

In this study, a SLM 280HL machine (SLM Solutions GmbH, Lübeck, Germany) equipped with a 400 W Yb-fibre-laser was utilized to manufacture the specimens. It features an available build space of 280 x 280 x 320 mm³ and includes a preheating system. The selected fabrication parameters are summarized in Table 1 and brief explication is provided in Figure 1. The low carbon stainless steel type EN 1.4404, US 316L (also known as X2CrNiMo17-12-2) was chosen as the raw material, which was supplied by SLM Solutions and had the following properties; a mean particle diameter of 35.5 µm and an apparent powder density of 3.85 g/cm³ [23].

The tensile specimens were designed in accordance to the German standard DIN 50125:2009-07 as flat specimen type E 5 x 10 x 40 and fabricated in seven distinct orientations (Table 2, Figure 2), subsequently referred to as configurations (a) to (g). The azimuth angle (Θ) describes the inclination of the sample to the x-axis and the polar angle (Φ) describes the inclination in relation to the xy-plane. Since the rough as-built surface was reported to lower the tensile strength, the samples were fabricated with an oversize of 0.4 mm in width and thickness and milled to their final shape [17,24]. Detailed studies on the effects of the positioning and inclination on the as-built surface roughness can be found elsewhere [23,25].

Table 1: Parameter sets utilized for irradiation in SLM for the processing of 1.4404; see description of the parameters and the setup in Figure 1

Parameter set	Scan speed [mm/s]	Laser power [W]	Hatch distance [mm]	Rotation angle increment [°]	Energy density [J/mm ³]
Contour	400	100	0.09	-	92.6
Core	800	200	0.12	33	69.4
Final layer	400	300	0.1	-	250.0
Support	875	200	-	-	-
Common	Layer thickness of 30 µm				
	Mounting plate temperature of 200°C				
	Nitrogen is employed as the inert gas				
	Contour is irradiated first, followed by the core, utilising the line scanning strategy with a scan vector length of 10 mm				
	Limitation window of 90°, respectively ± 45° to the y-axis				

Table 2: Summary of positioning details for all considered configurations, grouping of individual manufacturing jobs and corresponding time per job

Config.	Polar angle Φ ; α_{XY} [°]	Azimuth angle Θ ; α_X [°]	Total runtime [h]
(a)	0	0	39.5
(b)	0	90	
(c)	15	0	86.5
(d)	45	0	
(e)	75	0	
(f)	90	0	
(g)	90	90	

*Slight deviations from 0° and 90° angles were introduced for the azimuth angle of the in-plane orientated configurations ((a) and (b)) to improve the recoating process by ensuring that its blade does not abruptly hit the entire edge at once.

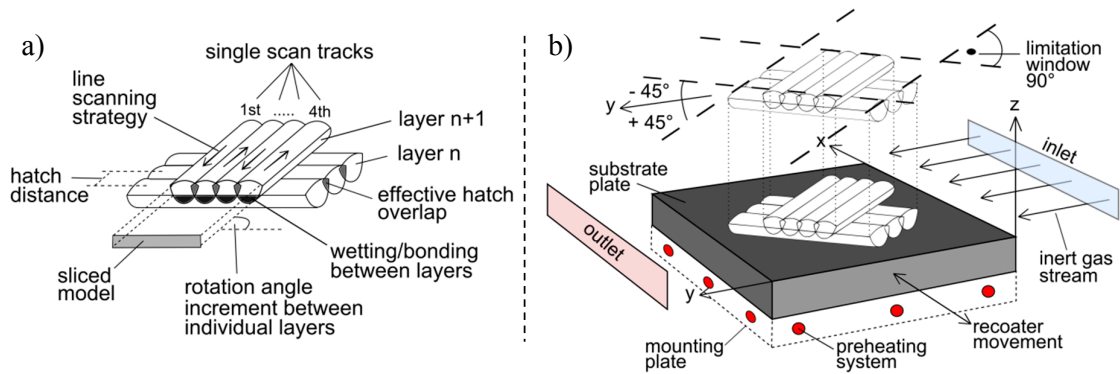


Figure 1: Schematic depiction of the SLM process; a) the representation of the geometry via single scan tracks and layers; b) the build environment; adapted from [8]

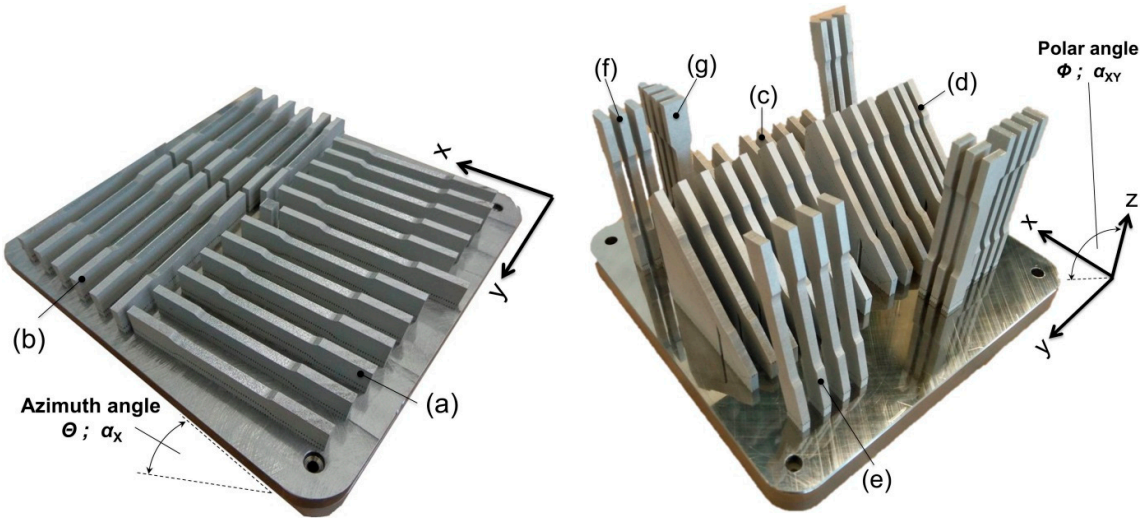


Figure 2: Tensile samples on the substrate plate, overview of the positioning and arrangement

2.2. Material testing

2.2.1. Composition and density

The material composition was determined with optical emission spectrometry (Q4 TASMAN, Bruker Corp., Billerica, Massachusetts, USA) on the machined samples. Based on the determined composition and the corresponding theoretical density the relative density was obtained via the Archimedes method.

2.2.2. Hardness

Surface hardness tests were systematically conducted on the clamping areas of the as-built and machined samples. Four indentations were evaluated on each sample. The hardness measurements were undertaken with a Reicherter KF hardness tester (Reicherter Georg GmbH Co Kg, Esslingen, Germany) in accordance to the DIN EN ISO 6507-2 standard. The testing force was set to 294.1 N and the hardness was obtained in HV30.

2.2.3. Tensile testing

For the destructive material tests, a tensile testing machine (Zwick/Roell, Ulm, Germany) with an inbuilt extensometer was utilized. The maximum load for this machine and the employed load cell was 100 kN and the initial distance of the extensometer was set to 50 mm. The testing procedure was carried out in accordance to the German standard DIN EN ISO 6892-1:2009-12 with a constant crosshead speed of 5 mm/min.

In addition, 3 out of 6 samples per configuration were equipped with an additional strain gauge (Type FCB-2-17-1L, Tokyo Sokki Kenkyujo Co., Ltd., Tokyo, Japan), comprising two individual measurement grids in perpendicular arrangement to each other (each with a size of 1.5 x 2.5 mm²). The tensile setup is depicted in detail in Figure 3.

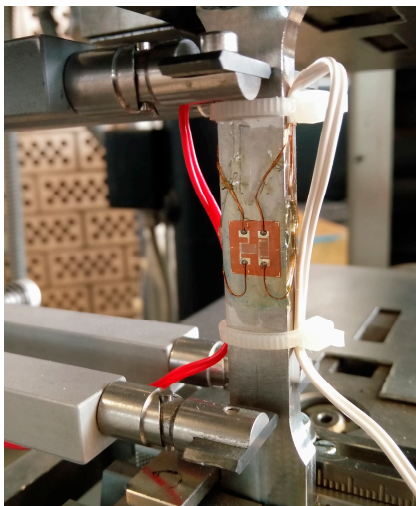


Figure 3: Tensile testing setup with extensometer and strain gauge

2.2.4. Microstructure

For investigations of the microstructure, segments were taken from the tensile samples, which were embedded in a hot mounting resin. Various grinding and mechanical polishing steps were performed to expose the metallurgical structure. The visibility of the scan track pattern and the inherent grain structure was enhanced by a subsequent etching process. High resolution images of the etched micro-sections were taken with an optical light microscope (Carl Zeiss Microscopy GmbH, Jena, Germany).

3. Results and discussion

3.1. Density and composition

The chemical composition has been examined on numerous samples of both batches and the averaged compositions are shown in Table 3. In short, the deviations amongst the two batches are negligible and are within the specifications of 1.4404. With the Archimedes method a consistent relative density greater than 99% (machined condition) was determined.

Table 3: Chemical composition of the specimens, all values in mass-percent

Config.	Fe	C	Si	Mn	P	S	Cr	Ni	Mo	N
(a)-(b)	Bal	0.308	0.564	1.044	<0.005	0.007	16.837	11.691	2.371	-
(c)-(g)	Bal	0.0235	0.585	1.051	<0.005	<0.005	16.994	11.257	2.390	-
DIN EN 10088-3	Bal	<0.03	<1	<2	<0.045	<0.03	16.5-18.5	10-13	2-2.5	<0.1

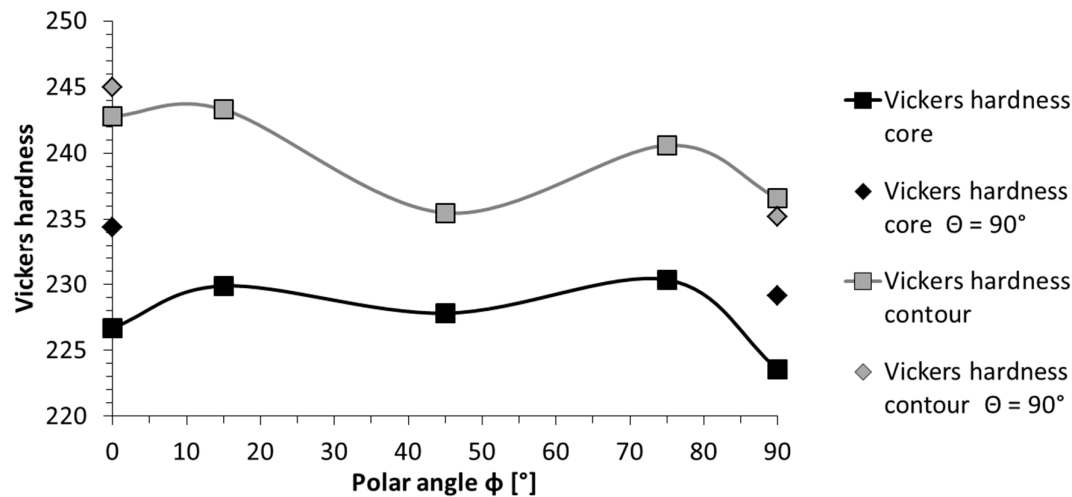
3.2. Hardness

The hardness results were consistent along all directions and the obtained results ranged from 223 HV to 234 HV for the machined condition and, respectively, from 235 HV to 245 HV in the as-built condition (Table 4), which is in perfect agreement to the documented values in the literature (Table 5). Throughout all configurations the surface hardness in the as-built condition exhibited an increased hardness, on average around 11 HV higher.

Unlike the previous investigation on AlSi10Mg, the hardness measurements did not exhibit a noteworthy deviation along the specimens, indicating a homogeneous structure of the 1.4404 samples [15]. The reason for the inhomogeneities encountered in the AlSi10Mg samples was related to the alloy being age-hardenable. A precipitation hardening occurred within the process, triggered by the dwell times at elevated temperatures in the build chamber. In the instance of stainless steel, the difference between the present temperatures in the build chamber in comparison to its melting temperature is, by far, greater as it was the case for AlSi10Mg. Thus, for steel samples it can be concluded that the risks of inducing microstructural inhomogeneities due to the dwell times are negligible at this point and are only of interest when considerably higher preheating temperatures are applied. However, one noteworthy and consistent relation stood out; comparing the configurations with identical polar angles, the configurations with a 90° azimuth angle revealed higher core hardness results (Figure 4). This finding will be addressed at a later point together with the findings of the tensile test and the micro-sections.

Table 4: Surface hardness results per configuration obtained on the machined samples

Config.	Vickers hardness core [HV30]	Standard deviation core [HV30]	Vickers hardness contour [HV30]	Standard deviation contour [HV30]
(a)	226.7	6.1	242.8	18.4
(b)	234.4	8.0	245.0	12.2
(c)	229.9	5.6	243.3	26.9
(d)	227.8	7.2	235.5	11.0
(e)	230.4	6.2	240.6	14.3
(f)	223.6	7.9	236.6	13.7
(g)	229.2	8.3	235.2	9.3

**Figure 4: Orientation dependency of the surface hardness in both, the as-build and machined condition****Table 5: Hardness results 1.4404, comparison with literature and supplier specification; relative densities ($\geq 99\%$)**

Reference	Vickers hardness	Machine	Max. laser power [W]
This work	223-245 HV30	SLM 280HL	400
Cherry, et al. [10]	220-225 HV	Renishaw AM250	200
Tolosa, et al. [19]	215-255 HV mean of 235 HV	SLM 250 Realizer	-
Kruth, et al. [26]	220-250 HV0.1	-	-
Montani, et al. [27]	245 HV	Prototype, not further specified	1000
SLM Solutions brochure [28]	209 HV	-	-
Sheet metal, typical value [29-31]	~220 HV (212-217 HB)	-	-

3.3. Tensile strength

The samples throughout all configurations revealed arbitrary occurrence of failure along the gauge length, thus, confirming the presence of a holistic homogeneous structure. Hence, the subsequently discussed deviations originated solely from the inherent directional dependencies, yet, differentiations were necessary for both the polar and azimuth angle. The averaged results, as well as the according standard deviations, of the tensile test are presented

in Table 6. These are drawn from six samples each, except Poisson's ratio, which is based on three measurements each. The latter will be investigated more in detail in a separate publication, since the encountered findings are scattered within the entire possible range of Poisson's ratio (i.e. between 0 and 0.5 [32]), with one exception even being outside this range. Thus, requiring to consider theories known from porous and composite material, which can exhibit Poisson's ratios greater than 0.5 [33].

Similarly as in the hardness evaluation, the samples with an azimuth angle of 90° stood out. Considering the in-plane orientated configurations (i.e. (a) and (b)), the Young's modulus differed by more than 30%, whereas the deviations in the yield strength and UTS were marginal. Interestingly, this huge dependency of Young's modulus on the azimuth angle was only present for the samples with a polar angle of zero degree. This finding contradicts Sehr and Witt [17], who reported that the in-plane orientation can be neglected. However, this simplification was extended; Sehr [1] added that the in-plane tendencies correspond with the irradiation strategy and especially the rotation angle of subsequent layers. He reported that the negligible case corresponds with a 67° increment between layers, which was the only case the Young's modulus was independent from the azimuth angle.

The breaking elongation was considerably higher for both cases with $\Theta = 90^\circ$, which respectively increased by 28.5% (config. (a) to (b)) and 48.5% (config. (f) to (g)). These findings coincide with those of Meier and Haberland [16], which also reported fluctuations of the breaking elongation with an varying azimuth angle.

Yet, the results of this investigation are in perfect agreement with the polar angle being the major directional dependency, influencing all tensile characteristics. The polar angle dependencies of each single characteristic are depicted in detail in the Figures 5-8 and compared with the results of related studies. Interestingly, the maxima for the Young's modulus and the tensile strength was evident for $\Phi = 45^\circ$. On a side note, given this, superimposed with the azimuth angle dependency, it can be anticipated that in the case at hand the combination of $\Phi = 45^\circ$ and $\Theta = 90^\circ$, which was not investigated in the present study, would yield the highest results with the utilized manufacturing settings. Fluctuations in the strength (yield and ultimate tensile strength) and, especially, in the breaking elongation by alterations of the polar angle were frequently reported. However, alterations in the linear elastic behaviour are far less investigated and the few existing studies are not in agreement. In addition to the depicted results (Figure 5), Rehme and Emmelmann [18] stated that there is no dependency of the Young's modulus on the polar angle evident. Based on the results of this work, however, there were remarkable deviations present.

The specific progress of the tendencies in the tensile strength values also differ noticeably between independent studies (Figures 6 and 7). Considering the general formulation about the occurrence of the highest strength and breaking elongation in the parallel layer to loading orientation, it can be concluded that this is indeed correct in many instances, but only partially correct when considering the big picture and far away from being applicable as a general rule. The tensile strength, no matter if considering yield or ultimate strength, showed a general tendency of being higher for the parallel loading to layer case. However, the progression in between the extrema (i.e. parallel and perpendicular) did not follow a general rule. It differed greatly by chosen manufacturing settings and setup applied

amongst these various studies and it is also highly volatile to the individual material characteristics. The latter will be explained in detail on the microstructural development.

Proceeding with the breaking elongation, which appears to be, by far, the most volatile characteristic to alterations in the orientation (Figure 8). In this instance there is no clear tendency evident, the range of reported results scatters greatly and the progression behaviour appears random at first glance, leading to the question of how this can be the case. Clearly, to holistically clarify this question more work needs to be done. For now, the major causes for these deviations are anticipated to be inherent residual stresses and incomplete fusion between scan tracks and layers. Both resulting in a weakening of the material in a predominant direction, but depending on where the defect occurs, the weakening varies in its predominant direction. Furthermore, both of these effects are greatly influenced by the utilized laser power and the ability to control the thermal environment, e.g. by the range of available preheating temperatures, which alter the microstructural development. On a side note, Wang, *et al.* [34] reported that defects and pores in the as-built state can be overcome to a great extent by applying a subsequent hot isostatic pressing (HIP) treatment. Leuders, *et al.* [35] has found that HIP increases the ductility of 1.4404, but due to the reduction in strength and the already very good properties in the as-built state a post-heat treatment is in most cases not required.

Table 6 Averaged results for the tensile properties of 1.4404

Conf. ig.	Young's modulus E [GPa]		Yield strength $R_{p0.2}$ [MPa]		Ultimate tensile strength R_m [MPa]		Elongation at failure A_t [%]		Poisson's ratio ν [-]	
	Average	STDEV	Average	STDEV	Average	STDEV	Average	STDEV	Average	STDEV
(a)	151.01	25.56	516.51	7.16	634.43	7.39	33.24	0.57	0.444	0.031
(b)	207.57	24.22	539.47	3.29	643.67	3.25	42.74	0.82	0.155	0.014
(c)	147.87	23.59	501.32	7.70	624.65	4.36	34.09	1.12	0.479	0.058
(d)	227.35	25.12	589.89	11.86	698.98	23.65	32.56	10.17	0.203	0.024
(e)	151.43	18.80	485.65	11.93	571.23	18.63	22.84	7.27	0.558	0.020
(f)	137.78	14.25	438.60	9.69	511.99	17.95	11.76	5.38	0.453	0.005
(g)	137.83	16.25	457.21	17.29	530.22	8.09	17.46	4.42	0.170	0.085

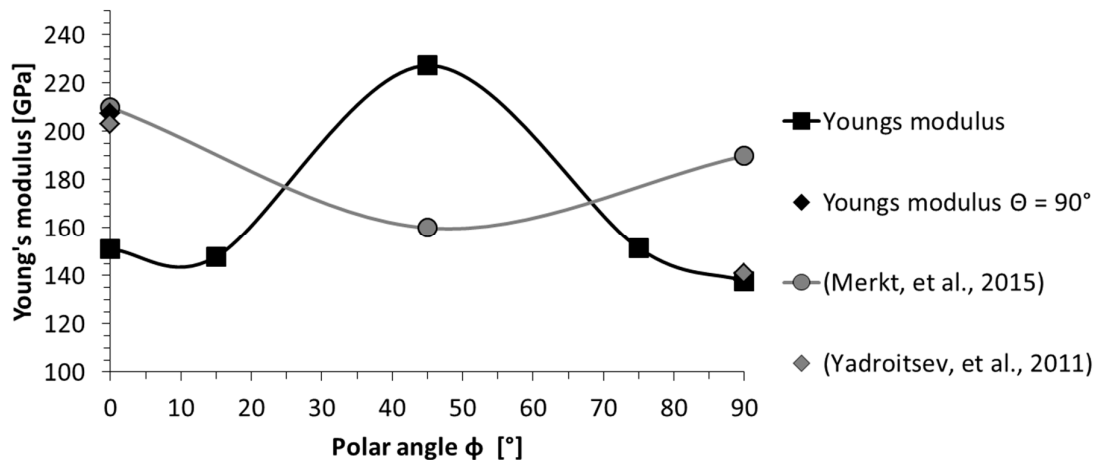


Figure 5: Orientation dependency of Young's modulus; comparison with reported results [4,36]

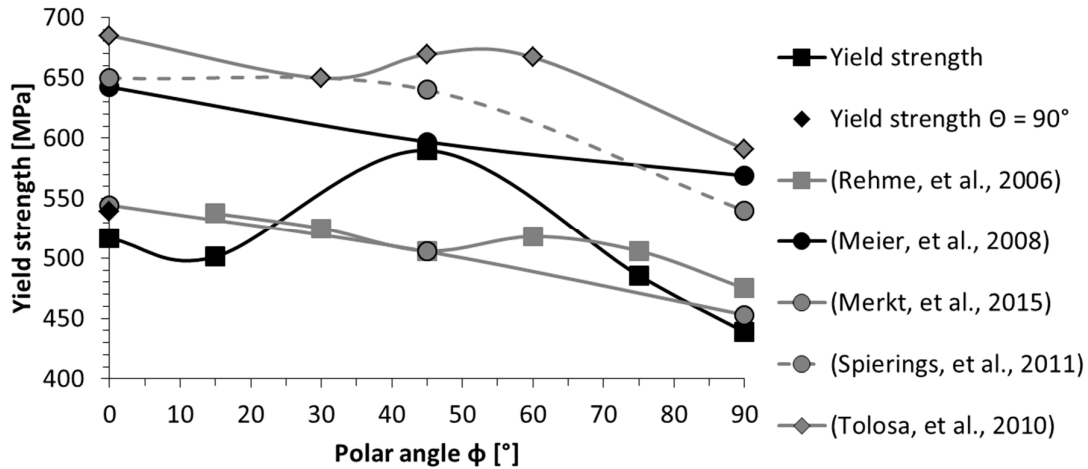


Figure 6: Orientation dependency of the yield strength; comparison with reported results [4,13,16,18,19]

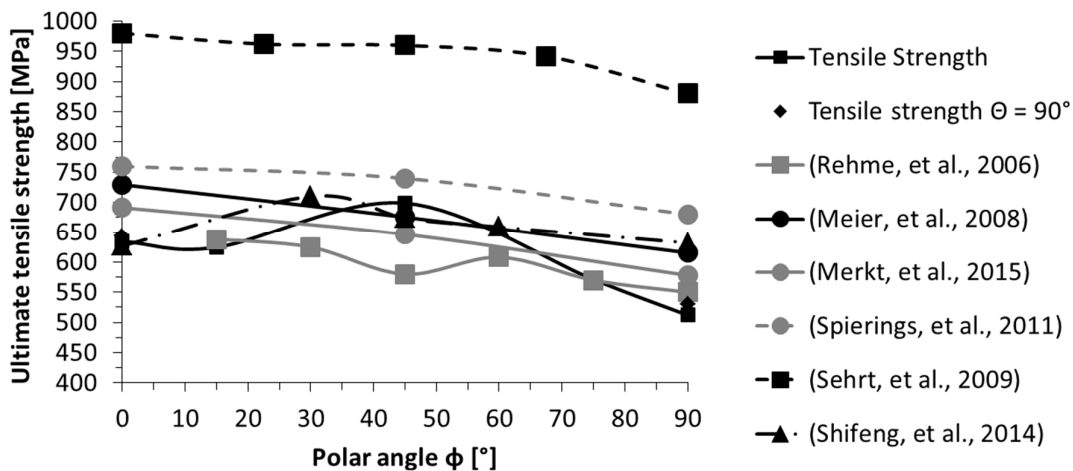


Figure 7: Orientation dependency of the ultimate tensile stress; comparison with reported results [4,13,16-18,37]

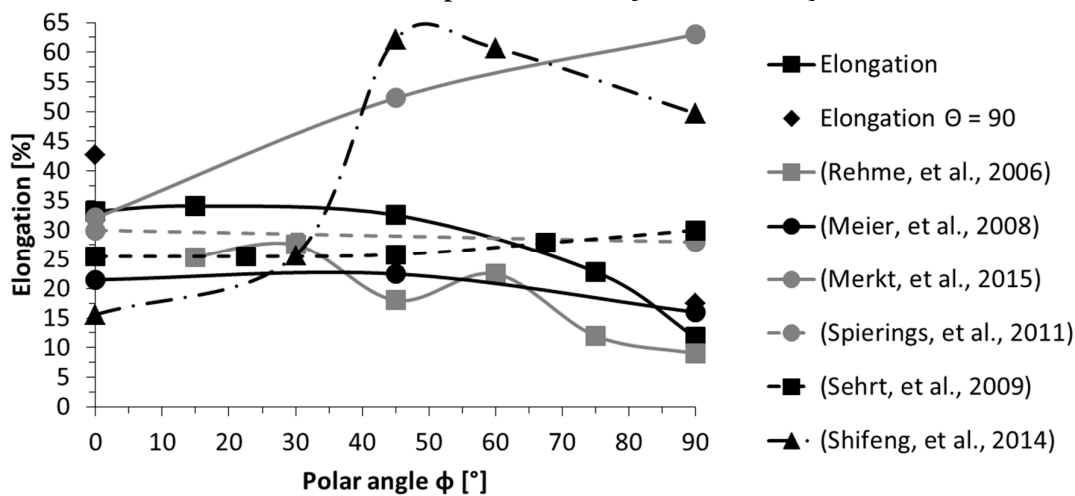


Figure 8: Orientation dependency of the elongation at fracture; comparison with reported results [4,13,16-18,37]

3.4. Comparison AM and bulk material

Additively manufactured material generally exhibits higher strength, coupled with a reduced ductility. This holds true for the study at hand as well, bulk 1.4404 is listed with a minimum yield strength of 170 MPa and, respectively, a minimum ultimate tensile strength of 485 MPa [29,31]. Both values were exceeded by far, yet the 40% minimum breaking elongation of bulk 1.4404 is a criterion which was only achieved by one out of seven configurations. One other aspect which should be emphasised is the greatly reduced difference between the yield point and the ultimate tensile strength caused by AM [8]. In Table 7, several achieved ratios are compared with the bulk material. For safety aspects it is beneficial if the difference between these measures is large. Thus, this peculiarity is to be noted for components designed for fabrication via AM. It should be pointed out that this behaviour is not particularly negative; it simply is something which needs to be considered. Amongst materials with high strength it is very common and due to AM this behaviour is also introduced in materials which do not show this behaviour when fabricated with conventional procedures.

Table 7: Ratio between the yield point and the ultimate tensile strength for 1.4404

Reference	Configurations	Range R_e/R_m [-]	Averaged* ratio R_e/R_m [-]
this work	7	0.8026-0.8623	0.8383
Meier and Haberland [16]	5	0.8621-0.9261	0.8889
Merkt [4]	3	0.7819-0.7877	0.7844
Rehme and Emmelmann [18]	150	0.8400-0.8877**	0.8597**
Riemer, et al. [38]	1	0.8177	-
	heat treated 2 h, 650°C	0.7445	-
Spierings, et al. [13]	3	0.7941-0.8648	0.8380
Tolosa, et al. [19]	15	0.9163-0.9967	0.9565
bulk 1.4404 [29,31]	-	0.35-0.37	-

* Investigated configurations are valued equally, sample sizes neglected

** only $\theta = 0^\circ$ and $\phi = 0^\circ - 90^\circ$ considered

3.5. Comprehensive analysis of the directional dependencies and their origin

At next the inherent directional dependencies are investigated further and to emphasise that their forming are greatly influenced by the chosen material, the direct comparison with a previous study on an aluminium based die-cast alloy (AlSi10Mg) is drawn [15]. When comparing the polar angle dependencies of 1.4404 with the dependencies of AlSi10Mg a clear influence of the material can be shown. AlSi10Mg exhibited the lowest tensile strength under $\phi = 45^\circ$, which is precisely where 1.4404 had its peak strength. The explanation for this opposing finding is based on the microstructural development during AM fabrication. The microstructure of AlSi10Mg revealed clear traces of every single scan track, which inherits an inhomogeneous nature. The Al grain boundaries are surrounded with Si particle segregations, which mainly takes places in the connection between scan tracks and layers, namely hatch overlaps and wetting areas (see Figure 1a) [39,40]. These segregations on the grain boundaries have a stabilization effect and prevent secondary grain coarsening or growth. Furthermore, the Si-rich areas along the boundaries are comparably brittle, thus representing a weak spot for fracture. Given that the line scanning approach is most commonly coupled with a rotation angle in between subsequent layers, these brittle predetermined breaking points are far more emphasised between layers (in z-direction), as it

is the case in either x- or y-direction. Applying a tensile loading under 45° to the layers results in the maximum shear stress acting parallel to the layers, thus shearing the layers apart along these embrittled areas, which are rich in Si segregations.

Considering the microstructure of 1.4404, there are no obstacles for ongoing grain growth. Thus, subsequent heat inputs, i.e. due to the neighbouring scan track or subsequent layer being generated, the grains of the already solidified materials become altered by the secondary heat input [38]. This behaviour is commonly referred to as epitaxial grain growth, describing the tendency of needle-like grain growth towards the heat source [41,42]. Due to this tendency, the grains of 1.4404 grew through the individual layers (in direction of the heat source), causing an interlocking of the individual layers (Figure 9). This interlock occurs in all directions, i.e. through layers and also through neighbouring scan tracks. Since the scan track direction is altered between layers this interlocking mechanism can be seen best in the build-up direction. An exemplified depiction of these underlying mechanisms is depicted in Figure 10 and a direct comparison of the obtained polar angle dependencies is provided as well, which clearly points out the resulting opposing, material dependent, progressions. Further examples for the material dependency on the development of directional anisotropies in AM are found in the literature. In the work of Sehart [1] it was seen that the NiCr alloy (Hastelloy X) developed a more emphasised polar angle dependency as the stainless steel GP1 did. The results of Spierings, *et al.* [43] on a AlMgSc alloy (Scalmalloy) suggested that for this material the polar angle dependency can be neglected entirely, since the reported deviations are below 3.4%. Given this, there cannot be a true generalized statement on the inherent anisotropic character of AM fabricated components, at the least differentiations on the underlying material have to be made.

Going back to the micro-sections (Figure 9), in these, there is also the answer for the encountered deviations regarding the azimuth angle. A closer look at the shape of the single scan tracks yielded the following. The cross-section of the configuration (a) samples mainly revealed oblong scan tracks, thus indicating that the majority of the scan tracks were fabricated around a perpendicular orientation to the longitudinal axis. For the configuration (b), the other case was evident, single bead cross-sections can be seen clearly, indicating that the single scan tracks were predominantly fabricated parallel to the specimen's longitudinal axis. These findings yield the following constellation: In configuration (a), the loading occurred predominantly perpendicular to the scan tracks, whereas in the case of configuration (b), the loading occurred predominantly parallel to the scan tracks, with the latter yielding the higher strength. These differentiations in the present structures are based on the applied irradiation pattern, more precisely on the limitation window, which limits the admissible range of irradiation directions. Its main purpose is to exclude all scan directions which result in a laser irradiation movement towards the inert gas stream. The reason for this limitation is to prevent any interaction of emerging weld splashes and smoke with the ongoing irradiation [44,45]. These particles get transported out of the build chamber via the inert gas flow; hence, a laser movement in direction of the inert gas stream is likely to cause unwanted interactions with these particles and can cause defects in the fabrication process due to a lack of highly focussed power. However, on the other hand this precaution limits the range of possible rotation increments of subsequent layers and causes the occurrence of predominant

directions. In this study the allowed range of possible track vectors was 90° , which limits the possible track vector range to $\pm 45^\circ$ with the track vectors always pointing in negative y-direction (opposed to the inert gas stream), which respectively results in the lower increment boarder at 135° and the upper boarder at 225° (Figure 11). With the rotation increment set to 33° , the sequence of track vectors and scan track vector angles, as outlined in Table 8, arises. Starting at the lower border of the limitation window as the initial track vector direction, with the scan track vectors placed in altering perpendicular directions. The track vector direction is continuously changed by constantly adding the rotation increment after every layer, until the upper limitation window is exceeded. In this case, the new vector track is defined by the lower limitation window border plus the former excess. From there, the pattern continues by adding the rotation increment after every layer until the upper limit is exceeded again.

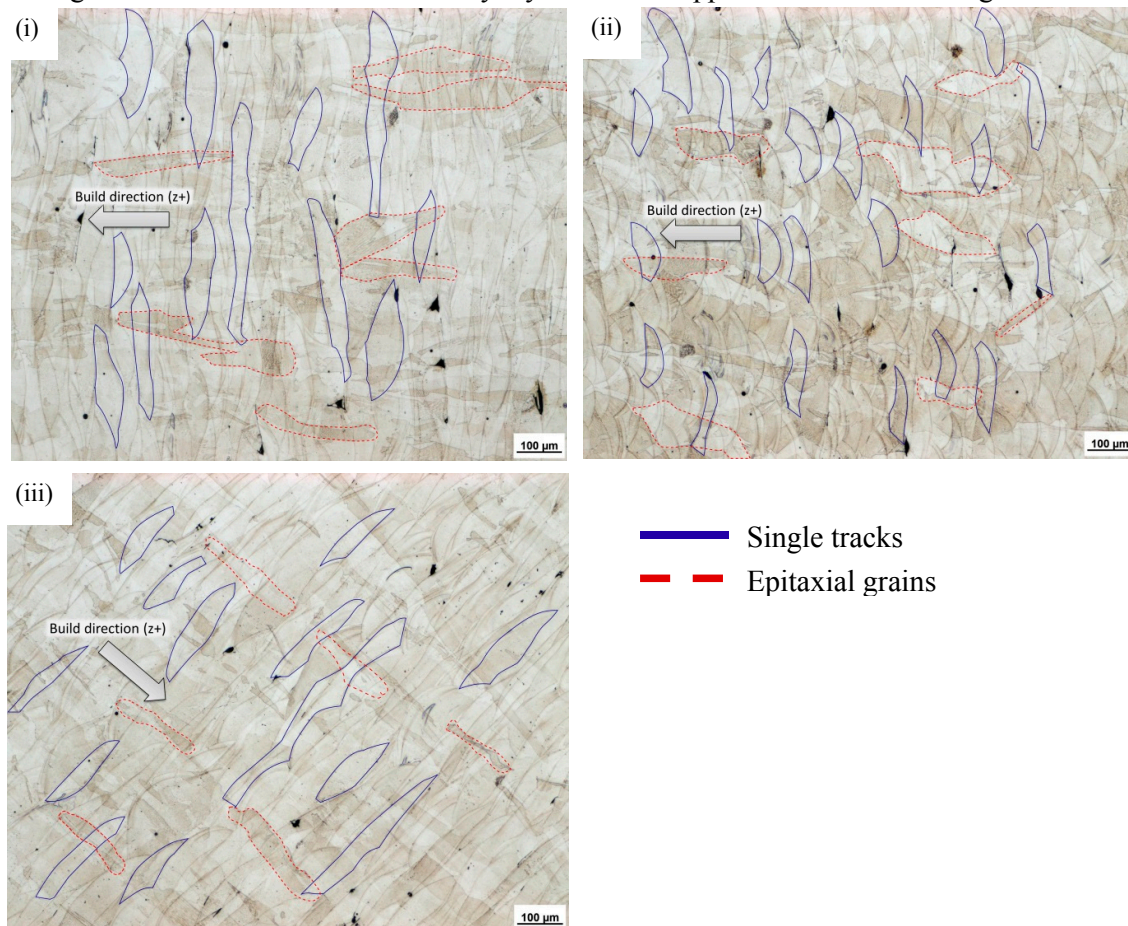


Figure 9: Microstructure of 1.4404, taken from the cross-section of the tensile samples of (i) configuration (a), (ii) configuration (b) and (iii) configuration (d)

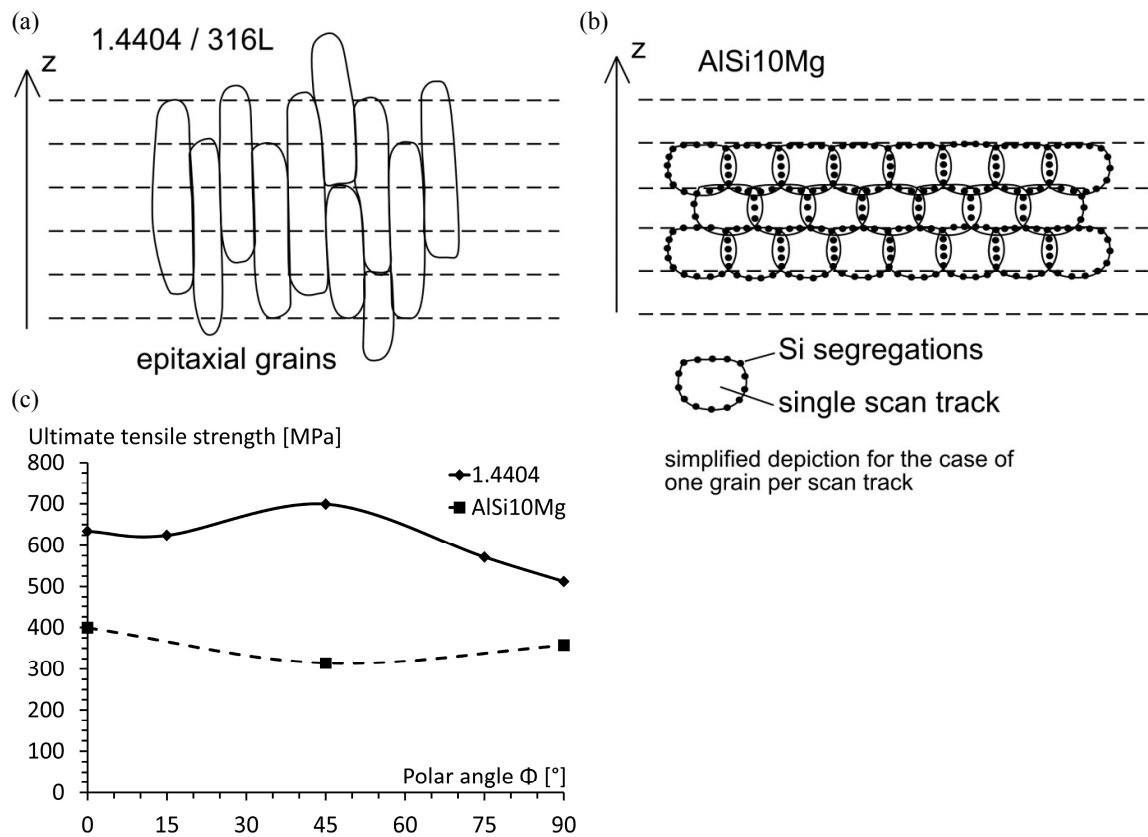


Figure 10: Comparison of the microstructural characteristics of a) 1.4404 stainless steel and b) AlSi10Mg, as well as their c) strength dependency on the loading versus layer orientation [15]

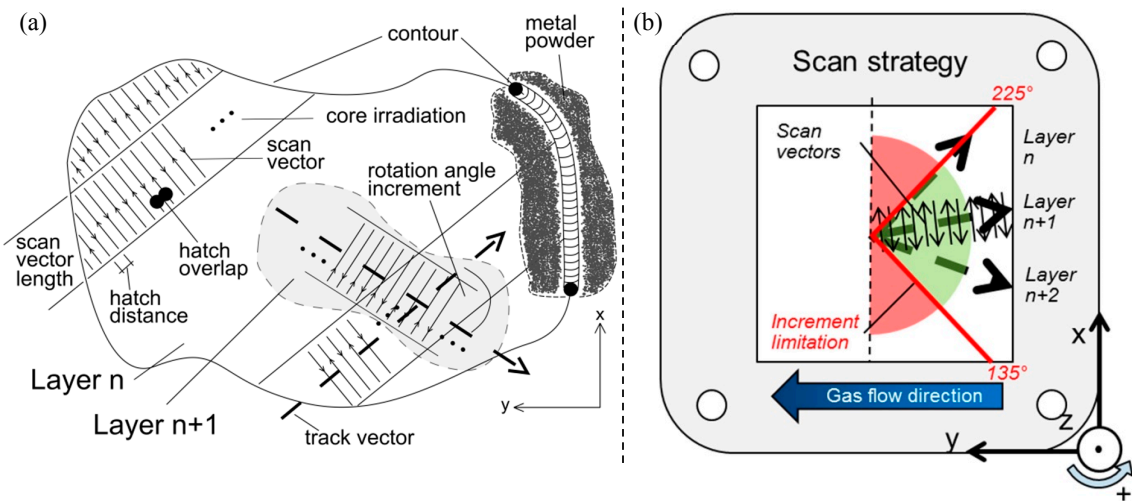


Figure 11: a) Irradiation strategy and b) limitation window; adapted from [8,45]

Table 8: Exemplary calculation of the direction of subsequent irradiation tracks

Layer	Track vector angle	Scan vector angle
1	[bottom increment limitation border] = 135°	[track vector angle] $\pm 90^\circ = 45^\circ; 225^\circ$
2	[previous track vector angle] + [rotation angle increment] = $135^\circ + 33^\circ = 168^\circ$	[track vector angle] $\pm 90^\circ = 78^\circ; 258^\circ$
3	[previous track vector angle] + [rotation angle increment] = $168^\circ + 33^\circ = 201^\circ$	[track vector angle] $\pm 90^\circ = 111^\circ; 291^\circ$
4	would be outside the limitation window!, thus: [previous track vector angle] + [rotation angle increment] – [top Increment limitation border] + [bottom Increment limitation border] = $201^\circ + 33^\circ - 225^\circ + 135^\circ = 144^\circ$	[track vector angle] $\pm 90^\circ = 54^\circ; 234^\circ$
5	[previous track vector angle] + [rotation angle increment] = $144^\circ + 33^\circ = 177^\circ$	[track vector angle] $\pm 90^\circ = 87^\circ; 267^\circ$

4. Conclusion

In this study, the peculiarities of additively manufactured material were addressed on the example of stainless steel. It was shown that homogeneous structures can be fabricated and preheating temperatures of up to 200°C do not cause location dependent alteration of the microstructure. The scan strategy was found to massively influence the material characteristics and even simple precautions, such as limiting the irradiation pathways to avoid possible interactions between emerging particles with the laser beam, promote inherent directional dependencies. In addition, the general rule of higher strength occurring in a parallel layer to loading direction, in comparison to the perpendicular layer to loading scenario, was proven accurate. However, the progression of the mechanical characteristics with altering the inclination between the loading and the layers differed and was shown to be highly material dependent. Stainless steel exhibited its peak strength and maximum Young's modulus under a 45° offset between the layer and loading direction, whereas the aluminium-silicium alloy AlSi10Mg revealed the lowest strength in this instance. In regard to the breaking elongation, the tested specimen showed a noteworthy drop in ductility past an inclination offset of 45° . Considering the disparate tendencies found in related studies, it can be concluded that the orientation dependency of the ductility in AM is, to date, not fully understood and further in-depth investigations need to be undertaken.

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