Application of a Composite Model in the Analysis of Creep Deformation at Low and Intermediate Temperatures

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Abstract: The creep behaviors of TA2 and R60702 at low and intermediate temperature were presented and discussed in this paper. Experimental results indicated that an apparent threshold stress exhibited in the creep deformation of R60702. Meanwhile, the primary creep phase was found as the main pattern in the room temperature creep behavior of TA2. Compared with exponential law, the power law has been proved to be a proper constitutive model in the description of primary creep phase. It also showed that \(\theta\) projection method had its significant advantage in the evaluation of accelerated creep stage. Thus, a composite model which combined power law with \(\theta\) projection method was applied in the creep curves evaluation at low and intermediate temperature. Based on the multiaxial creep deformation results, the model was modified and discussed. A linear relationship existed between composite model parameters and applied load. Finally, the creep life could be accurately predicted and the composite model method is suitable for application in low and intermediate temperature creep life analysis.

Keywords: Creep; Composite constitutive model; \(\theta\) projection method; low and intermediate temperature

1. Introduction

High temperature failure, due to creep, fatigue and fracture, is a nonnegligible issue in the safety production of modern industry [1]. During the past few decades, a significant amount of research efforts have been done to estimate the high temperature strength of structure component [2]. With the deepening of the study, microstructure evolution, creep damage and constitutive model are gradually discussed and presented in the high temperature steels analysis [3-5]. Finally, a relatively compact framework system for evaluating the high temperature creep properties was established.

Nowadays, a wide variety of metals such as Ti, Cu and Zr alloys, have been applied in the design of key component, due to their outstanding corrosion resistance, high strength-to-weight ratio and machine deforming character [6]. Different from conventional metallic materials, evident creep deformation of these materials may take place at low and intermediate temperatures. How to describe the low temperature creep characteristics accurately and effectively has attracted more and more attention in the past decade. Kassner et al. presented the significant creep plasticity of 304 stainless steel and proved that the room temperature creep is partially attributed to the high strain-rate sensitivity of material [7]. Peng et al. investigated the primary creep constitutive model of CP-Ti and introduced a partition method in order to analyze the deformation at different stresses and temperatures [8]. With the purpose of evaluating the relaxation mechanism of Zn specimens, Matsunaga et al. analyzed the role of grain boundary and discussed its influence on the low
temperature creep [9]. In addition to creep deformation character, activation energy was also extensively evaluated and proved to be a variate with the increase of temperature [10].

It can be found that creep deformation mechanism and micro damage evolution is the main focus of previous researches. However, low temperature constitutive model, which has guiding significance for the safety design, has not caused widespread attention. How to describe the deformation curve at low and intermediate temperatures accurately, in order to realize the accurate life extrapolation, is still a practical problem need to be solved.

In this paper, the creep characters of TA2 and R60702 were investigated under different stresses and temperatures. According to the creep evolution feature, the primary creep phase and threshold stress were found and discussed. Then, a composite model was introduced and modified, which can be widely used in the creep deformation analysis at the low temperature. Finally, the composite model was used to analysis the low temperature creep behavior of materials under multiaxial deformation states successfully.

2. Experiment

TA2 and R60702 analyzed in this paper were widely used materials in aerospace, transport and petrochemical field. The chemical composition of R60702 is Zr+Hf≥99.2, Hf≤4.5, 0.2Fe+Cr, 0.16O, 0.005H, 0.025N, 0.05C. The chemical composition of TA2 is 0.12Fe, 0.01C, 0.03N, 0.14O, balance Ti. Prior to the experiment, uniaxial creep specimens were wire-electrode cutting and polished to avoid the influence of surface defect. Then, Rectangular cross sections creep specimens with 10 mm in width and 3 mm in depth were obtained and used. According to the service conditions, two temperature conditions (293K and 473K) were selected. Three groups of thermocouple were placed near specimens and the error was controlled in ±1K in order to ensure the accuracy of creep temperature. Finally, the tensile creep tests were accomplished and the creep strain was real-time measured by two extensometers.

3. Results and discussion

3.1 Creep characters

At the room temperature, typical primary creep stages were observed in the creep strain - time curves of TA2 as shown in Figure 1. Comparing with the results at different stresses, the profiles are similar and the creep strain rate decreases with the increase of deformation at 293K. It should be noted that the primary creep behavior lasted more than 2500 hours when the stress is 300MPa. It means that there is no clear steady creep stage in the room temperature creep of TA2 when the stress varies from 240MPa to 330MPa. Meanwhile, a large number of deformation twins were found in the deformation region. The twinning deformation is an important factor for the deformation of TA2. It has been proved that, besides dislocation slip, the low temperature creep of HCP (hexagonal close packed) material is also correlated with twinning deformation [11]. The stress to activate the dislocation slip is temperature dependent. While, the stress to activate twinning deformation is temperature independent, and high stress is acquired to activate the twinning deformation mechanism [8]. All these indicate that the mechanism of low temperature creep cannot be equivalent to that of high temperature creep although the primary creep stage is similar.

On the other hand, an apparent threshold stress was also presented in the creep deformation of R60702 [8]. After elastic-plastic deformation, the creep strain is not obvious when the stress is less than 130MPa. However, typical creep phase can be found when the stress is equal or greater than 145MPa. Different from the primary creep behavior of TA2, all of the three creep deformation phases can be found which proves that the typical creep failure can also show up at low and intermediate temperatures. High stress leads to a larger deformation combined with slips of dislocation which may induce fracture of specimen. Thus, the percentage of primary creep phase in the whole deformation of high stress is apparently larger than that of lower stress in this region. Along with the balance between damage and strain-hardening, the second creep phase will apparently shorten and the creep strain rate will increase with the increase of the stress.
3.2 Description of primary creep phase

In the traditional creep analysis, the basic deformation stages of creep curves are well known: primary creep stage, steady creep stage and accelerated creep stage [12]. However, particular creep deformation character is found at low and intermediate temperature in the previous section. Except slip of dislocation, twinning deformation induced by high stress plays an important role in the creep damage. Especially in the analysis of TA2, only primary creep stage is presented in our creep conditions. Thus, how to describe the constitutive model and deformation feature of primary creep stage is the issue need to be discussed in this paper.

In the traditional high temperature creep deformation study, several models and modifications have been reported to predict the creep behavior. The most known are the Norton model, the Graham-Walles model and the theta model [13-15]. It can be found that the first part of Graham-Walles model is defined as the power law. Meanwhile, the first part of theta model is similar to exponential law as listed below:

\[ \varepsilon = \varepsilon_0 + b_1 t^{\gamma_1} \]  
\[ \varepsilon = \varepsilon_0 + b_2 [1 - \exp(-c_2 t)] \]

Where \( \varepsilon \) is the total strain, \( \varepsilon_0 \) is the initial elastic-plastic strain, \( t \) is the creep time, \( b_1, b_2, \gamma_1 \) and \( c_2 \) are the material parameters.

Based on the variation of parameters, the primary parts of two models are both used in the analysis of primary creep phase in this paper. As shown in Figure 3, primary creep deformation behavior can be described by Graham-Walles model properly and accurately. Different from the power law of Graham-Walles model, the primary phase of creep-strain curve can also be predicted by the theta model when the stress level is not high. With the increase of stress, the prediction precision would descend gradually. For R60702, two models can both be used in the analysis when the stress is approximate or lower than the threshold stress. Once the stress is higher than the threshold stress, apparent three creep stages can be found and the advantage of power law is presented in Figure 3(c). This indicates that the first part of Graham-Walles model is approximate in the analysis of low and intermediate temperature creep.
In terms of engineering assessment requirement, simple form of power law is another important advantage in the deformation analysis. Thus, the power law can be defined as the foundation of low temperature model in order to estimate primary creep behavior accurately.

![Graphs of experimental data and fitting curves for different materials under various conditions.](image1)

**Figure 3.** Analysis of primary creep phase: (a) TA2 at 300MPa; (b) TA2 at 270MPa; (c) R60702 at 145MPa

### 3.3 The composite model analysis

In the previous section, the primary creep model has been discussed and described reasonably. However, most failures of creep take place at the accelerated creep stage in practice engineering. With the increase of strain, the micro-damage and thickness reduction can be perceived in this stage along with the increase of creep strain rate. Thus, more and more constitutive models were proposed to evaluate the accelerated creep stage under different conditions.

As a widely used creep model, \( \theta \) projection method believes that creep curve contains vast information which can be presented by the function of temperature and stress:

\[
\varepsilon = \varepsilon_0 + \theta_1 (1 - e^{-\theta_2 t}) + \theta_3 (e^{\theta_4 t} - 1)
\]

where \( \varepsilon \) is the creep strain at deformation time \( t \), \( \varepsilon_0 \) is the instantaneous deformation, \( \theta_1 \), \( \theta_2 \), \( \theta_3 \) and \( \theta_4 \) are material parameters.

With the future development, the focus of research was presented in the steady and accelerated creep stages and a more simplified form which ignored the first creep stage was presented:

\[
\varepsilon = \varepsilon_0 + \theta t + \theta_3 (e^{\theta t} - 1)
\]

According to the simple form, more and more researches used this model to estimate deformation process. The predicted precision was gradually proved, especially in the third creep stage [16]. Jiang et al. [17] combined the third part of theta model and the first part of G-W model in the prediction of heat resistant steel as shown in equation (5):

\[
\varepsilon = \varepsilon_0 + A t^B + \theta_3 (e^{\theta t} - 1)
\]

Meanwhile, the minimum creep strain rate can be defined as:

\[
\dot{\varepsilon}_m = ABt_m^{B-1} + \theta_3 \theta_4 e^{\theta t_m}
\]

Where the \( t_m \) is the time that reaches the minimum creep strain rate, which can be defined as follows:

\[
AB(B-1)t_m^{B-2} + \theta_3 \theta_4^2 \exp(\theta t_m) = 0
\]

Based on the previous discussion, it can be found that \( \theta \) projection method has its apparent advantage in the evaluation of accelerated creep stage. Meanwhile, the power law can be used in the prediction of primary creep stage at low and intermediate temperature environment. Thus, the composite model is also applied in this paper in order to estimate low and intermediate temperature creep deformation.

Considering that the elastic-plastic deformation is small, the influence is negligible here. Then, fitting curves of R60702 at 473K is plotted in Figure 4 on the basis of Equation (6). The results show that the fitting curve is highly consistent with the experiment data, which proves that the composite model can be used in the analysis of creep behavior under low temperature.
On the other hand, it should be noted that required stress of dislocation slip would be affected by the temperature. Deformation induced by slip of dislocation is easier to happen as the temperature becomes higher. Consequently, to evaluate the applicability of composite model at intermediate temperature, the results of low-temperature sintered nano-silver lap shear joints at 600K [18] are also introduced. Different from the traditional creep analysis, the creep behavior of nano-silver sintered lap shear joints will happen in a shorter time, which is sensitive to the stress and temperature. In spite of the fact that the creep character is diverse, the model has also been applied to estimate creep behavior of nano-silver sintered lap shear joints successfully as shown in Figure 5. It proves that the suitability of composite model will not be affected by the creep deformation mechanism.

Since the focus of creep deformation analysis is to avoid the failure in the actual engineering assessment, how to establish the relationship between model parameters and experiment condition is another main issue needed to be discussed. The analyzed results of $\theta_3$, $\theta_4$, $\theta_3$ and $\theta_4$ are plotted in Figure 6. It is well known that there exists a linear relationship between the theta parameters and applied stress in the $\theta$ projection method [19]. A similar analysis is also completed here, which indicates that a similar linear relationship can also be found between the composite model parameters and applied stress as follows:

$$\log(m) = a + b\sigma \quad (m = A, B, \theta_3, \theta_4)$$  \hspace{1cm} (8)

Where $a$ and $b$ are material constants.

**Figure 4.** Fitting results of composite creep model for R60702 at 473K

**Figure 5.** Fitting results of composite creep model for nano-silver lap shear joints at 600K

$$\log(A, B, \theta_3, \theta_4)$$
Figure 6. The relationship between material parameters of composite model and stress

With the increase of stress, the log($A$), log($\theta_3$), log($\theta_4$) increase and the log($B$) decreases. Thus, the four parameters under low temperature can be determined and the creep curve can be extrapolated to any temperature and stress [20].

3.4 Multiaxial creep deformation analysis

To reveal the deformation and failure character of high temperature creep behavior, conventional uniaxial creep tests are always used in the experimental analysis. However, the deformation and stress state of actual structure is complex and various in the practice engineering. How to describe the multiaxial creep behavior and establish the rational constitutive model gives rise to a new question. In this paper, the creep results of R60702 under multiaxial creep state are obtained by small punch creep tests as shown in Figure 7.

Similar to the uniaxial creep tests results, an obvious threshold stress can be found. Meanwhile, a typical creep deformation curves is obtained when the load is larger than 493N. When the test temperature increases to 573K, the threshold load decreases. And the typical creep failure takes place when the load is 454N. High load leads to a larger deflection which may induce fracture of specimen. So, the percentage of primary creep phase in the whole deformation of high load is apparently larger than that of lower load in this region. The second creep phase would apparently shorten and the creep rate increases with the increase of the load, which is also in accordance with the results in uniaxial creep tests. All these indicate that an approximate equivalent relation can be estimated between the multiaxial creep load and the uniaxial creep stress. Theoretical analysis is shown in Figure 8.

Figure 7. The creep deflection – time curves of R60702 at different temperatures and loads

![Figure 7](image)

Figure 8. Deformation model of round specimen in the multiaxial deformation status

In the vertical direction, the deformation load $F$ can be described as the function of uniform membrane pressure $p$ [21]:

$$ F = \int_0^{\alpha} 2\piqr p \cos \varphi d\varphi $$  \hspace{1cm} (9)
Where the \( r \) is ball radius, \( \varphi \) is the deformation angle of any point, \( \alpha \) is the deformation angle of contact boundary.

The Equation (9) is integrated and then:

\[ F = \pi r^2 p \sin^2 \alpha \]  

(10)

Meanwhile, the radial stress \( \phi_1 \) and circumferential stress \( \phi_2 \) can be described as the function of membrane pressure:

\[ \phi_1 = \phi_2 = \frac{pr}{2h} \]  

(11)

Thus, the equivalent stress is:

\[ \phi_m = \frac{\sqrt{2}}{2} \sqrt{\left(\phi_1 - \phi_2\right)^2 + \left(\phi_2 - \phi_3\right)^2 + \left(\phi_3 - \phi_1\right)^2} \]  

(12)

Where \( \phi_3 \) is the stress along the thickness direction of the specimen, and it is very small compared with \( \phi_1 \) and \( \phi_2 \), thus

\[ \phi_m \approx \frac{pr}{2h} \]  

(13)

As a result, the equivalent stress can be described as the function of applied load and current thickness. On the other hand, Hyde et al [22] proved that specimen thickness changes with the deformation and can be approximated as:

\[ \frac{h}{h_0} = \frac{1}{\sqrt{1+1/\tan^2 \beta}} \]  

(14)

In the composite model, the creep strain can be replaced by the deflection of small punch creep test as discussed by Yang [23]. Meanwhile, an approximate linear relationship between equivalent stress and creep load can be obtained based on the Equation (9), Equation (13) and Equation (14). Thus, the composite model is modified and the parameters can be replaced as listed below:

\[ L = L_0 + A \beta + \Theta_3 (e^{\beta t} - 1) \]  

(15)

\[ \lg(m) = a + bF (m = A, B, \Theta_3, \Theta_4) \]  

(16)

The fitting results are shown in Figure 9 based on Equation (16). The results show that the composite model can be used in the analysis of multiaxial creep deformation. According to the thin specimen, a shorter creep second phase exists in the small punch creep test. Thus, the modified composite model can predict the deformation behavior more accurately.

![Figure 9](image-url)  

**Figure 9.** Fitting results of composite creep model for multiaxial creep results of R60702

### 4. Conclusions

In this paper, the creep behaviors of TA2 and R60702 at different stresses and temperatures were analyzed. The low and intermediate temperature creep characters were discussed and a composite model was applied in the prediction. Some important conclusions have been obtained as follows:

1. Obvious threshold stress and primary creep phase were found in the creep deformation of TA2 at 293K and R60702 at 473K respectively, which proved that the creep mechanism under low...
and intermediate temperature was different from that of high temperature creep deformation. The reason may due to the twinning deformation induced by high stress.

2. Two constitutive models were introduced and compared. It proved that the power law could be used as a proper constitutive model in the description of primary creep phase at low and intermediate temperature.

3. Based on the power law and $\theta$ projection method, a composite model was applied in the creep curve evaluation at low and intermediate temperatures. A linear relationship was found between model parameters and applied stress. Thus, the creep life can be accurately predicted by composite model in practice engineering.

4. Under multiaxial creep deformation stage, the composite model was modified and proved that it could be used to predict the creep behavior accurately.

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