

Inversion of lift-off distance and thickness for non-magnetic metal using eddy current testing

Xiaobai Meng, Mingyang Lu, Wuliang Yin, Abdeldjalil Bennecer*, Katherine J Kirk*

Abstract — For the electromagnetic eddy current testing, various methods have been proposed for reducing the lift-off error on the measurement of samples. In this paper, instead of eliminating the measurement error caused by the lift-off effect, an algorithm has been proposed to directly measure the lift-off distance between the sensor and non-magnetic conductive plates. The algorithm is based on a sample-independent inductance (SII) feature. That is, under high working frequencies, the inductance is found sensitive to the lift-off distance and independent of the test piece under an optimal single high working frequency (43.87 kHz). Furthermore, the predicted lift-off distance is used for the thickness prediction of the non-magnetic conductive samples using an iterative method. Considering the eddy current skin depth, the thickness prediction is operated under a single lower frequency (0.20 kHz). As the inductance has different sensitivities to the lift-off and thickness, the prediction error of the sample thickness is different from that of the lift-off distance. From the experiments on three different nonmagnetic samples – aluminium, copper, and brass, the maximum prediction error of the lift-off distance and sample thickness is 1.1 mm and 5.42 % respectively at the lift-off of 12.0 mm.

Index Terms— Eddy current sensor, lift-off measurement, thickness measurement, non-destructive testing, sample-independence.

I. INTRODUCTION

VARIOUS techniques including the electromagnetic, ultrasonic, acoustic and laser sensing methods have been introduced in the field of the non-destructive testing (NDT) or non-destructive evaluation (NDE) [1-6]. Owing to its high adaptability, electromagnetic (EM) sensing has been massively used for the detection and measurement of surface fatigue information, quality inspection, and reconstruction of sample parameters (including the magnetic permeability, electrical conductivity, and thickness profiles) [7-12].

The magnetic flux leakage (MFL), eddy current testing (ECT) and permanent magnet perturbation (PMP) are currently the three dominant methods of EM NDT [13]. PMP methods have advantages of easily installed sensor but can only be applied in the measurement of ferromagnetic materials [13]. Besides, PMP methods are significantly sensitive to the lift-off distance to the test magnetic samples. MFL is playing an increasing role in the metallic quality check, detecting and characterising defects [14], particularly for the safety inspection

of tubes and railway. Udpa and Kim have proposed several methods of improving the magnetic flux leakage methods in the remote region of the pipeline inspection [15-18].

ECT methods are generally applied for both magnetic and non-magnetic conductive materials. Diverse methods including the multi-frequency eddy current (MEC) sensing [20-29], pulsed eddy current (PEC) sensing [30-37], and numerical models [38,39] have been used for the measurement of surface crack information (using rotating field) [40,41], thickness, magnetic permeability and electrical conductivity of the specimen. For the MEC technique, Pinotti and Puppini have proposed a lock-in approach to retrieve both the electrical conductivity and thickness of the specimen by referring to the voltage phase feature [22]. Different methods have been proposed to eliminate the lift-off effect on the thickness measurement of the nonmagnetic conductive plates [12, 19, 23, 29]. However, these methods are mainly focused on peak frequency feature from the inductance spectrum.

In this paper, an eddy current (EC) sensor with two planar and co-axial coils has been used to directly reconstruct the lift-off distance and measure the thickness of the non-magnetic conductive plate. The selected EC sensor is less mechanically complicated and more accurate than conventional triple-coil sensors without concerning the error of the magnetic balance. Although the used EC sensor was originally designed decades ago, the previous technique on the lift-off retrieval is based on the calibration method - Lissajous diagram [44] or iterative method on the conventional analytical model [37], whereas the proposed technique is using the thin-skin regime via simplified model (which only needs single frequency for the lift-off restoration, and applies for the online measurement). Based on this EC sensor, a lift-off algorithm has been proposed, which originates from a sample-independent inductance (SII) feature under high working frequencies. That is, under an optimal high frequency – 43.87 kHz (for the sensor size used in this paper), the inductance is found sensitive to the lift-off distance and nearly immune to the test piece due to the (surface-restrained) eddy current skin effect. The error of the reconstructed lift-off reaches its smallest value 8.8 μ m at 2.0 mm and highest value 1.1 mm at 12.0 mm. Furthermore, by using an iterative method, the predicted lift-off is used for the thickness prediction of different non-magnetic samples under a lower single frequency.

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The error of the predicted sample thickness achieves its lowest percentage (0.11 %) at the lift-off distance of 2.0 mm, and highest one (5.42 %) at the lift-off distance of 12 mm. Owing to the different sensitivity of the inductance regarding to the lift-off distance, the prediction error of the thickness differs from that of the lift-off distance.

II. METHODOLOGY – PREDICTION ALGORITHMS

A. Original formula of circular coil above conductive plate

Previously, different structures of eddy current sensors have been proposed for the lift-off compensation and thickness measurement [12,19,23,29], which mainly involves combining the measured signals from dual-sensing pairs (with three coils). In this paper, a simplified high-frequency regime model is extracted from a simple sensor setup (Fig. 1).

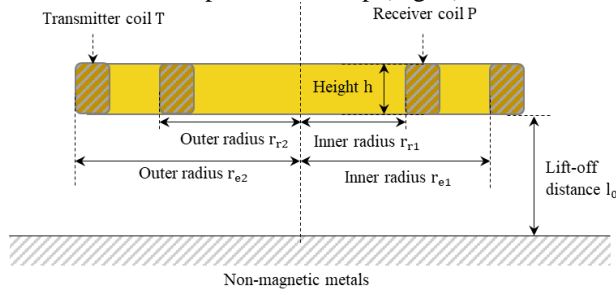


Fig. 1 Structure of two co-axial coils eddy current sensor

As shown in Fig. 1, the inductance change (minus the background signal when the sensor is in the free space) for two co-axial circular coils parallel but lifted with a small distance to the non-magnetic conductive plate can be derived from the Dodd-Deeds formulation [43],

$$L = K \int_0^{\infty} M e^{-2\alpha l_0} T d\alpha \quad (1)$$

Where in (1), l_0 is the lift-off distance between the eddy current sensor and test piece. α is the integration variable related to the wavenumber of the incident transverse electric (TE) plane EM wave [43]. M and T are the magnitude part and phase part of the integrand for the mutual inductance, which are determined by the sensor geometry and test piece respectively. M is defined as

$$M = \frac{P(\alpha)}{\alpha^6} (e^{-\alpha h} - 1)^2 \quad (2)$$

In (2), h is the height of the coils. P is the integration of the series of J_1 over the exciting and receiver coils.

$$P(\alpha) = \int_{\alpha r_{e1}}^{\alpha r_{e2}} \tau J_1(\tau) d\tau \int_{\alpha r_{r1}}^{\alpha r_{r2}} \tau J_1(\tau) d\tau \quad (3)$$

In (3), r_{e2} and r_{e1} are the outer and inner radii of the exciting coil, while r_{r2} and r_{r1} are that of the receiver coil. J_1 is the first-order Bessel function of the first kind. τ is the integration variable. In (1), K is a constant factor related to the geometry of the sensor and defined as

$$K = \frac{\pi(r_{r1} + r_{r2})\mu_0 N_e N_r}{2h^2(r_{e2} - r_{e1})(r_{r2} - r_{r1})} \quad (4)$$

where μ_0 is the vacuum magnetic permeability. N_e and N_r are the number of turns of the exciting and receiver coils. The phase term T in (1) is defined as

$$T = \text{Re}\left(\frac{(\alpha_1 + \alpha)(\alpha_1 - \alpha) - (\alpha_1 + \alpha)(\alpha_1 - \alpha)e^{2\alpha_1 c}}{-(\alpha_1 - \alpha)(\alpha_1 - \alpha) + (\alpha_1 + \alpha)(\alpha_1 + \alpha)e^{2\alpha_1 c}}\right) \quad (5)$$

In (5), c is the thickness of the non-magnetic test piece; α_1 is a complex number related to the wavenumber of the transmitted EM TE plane wave, which is defined as

$$\alpha_1 = \sqrt{\alpha^2 + j2\pi\sigma\mu_0 f} \quad (6)$$

where σ is the electrical conductivity of the test piece; f is the working frequency of the current flowing in the exciting coil.

B. Prediction of lift-off distance from quiescent inductance using simplified algorithm – high frequency-regime

As shown in Fig. 2, it has been found that the magnitude part of the integrand M can be approximated using $\sin^2\left(\frac{\alpha\pi}{2\alpha_0}\right)$, where α_0 is related to the height and the radii of the coils.

$$M = S \sin^2\left(\frac{\alpha\pi}{2\alpha_0}\right) \quad (7)$$

In (7), S is the ratio factor between $\frac{P(\alpha)}{\alpha^6} (e^{-\alpha h} - 1)^2$ and $\sin^2\left(\frac{\alpha\pi}{2\alpha_0}\right)$.

$$S = \frac{P(\alpha_0)}{\alpha_0^6} (e^{-\alpha_0 h} - 1)^2 \quad (8)$$

It can be observed from (8) that the ratio factor S is a coil-dependent constant. Substituting (7) into (1), the inductance in (1) becomes,

$$L = KS \int_0^{2\alpha_0} \sin^2\left(\frac{\alpha\pi}{2\alpha_0}\right) e^{-2\alpha l_0} T d\alpha \quad (9)$$

In (9), since the effective range of α for M is from 0 to $2\alpha_0$, the upper limit of the integration becomes $2\alpha_0$.

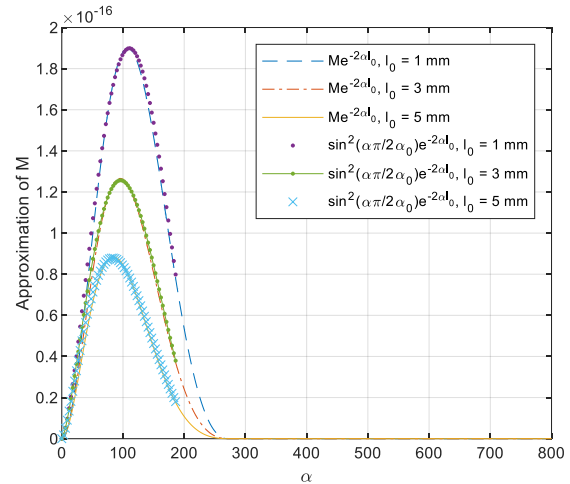


Fig. 2 Estimation for the magnitude part of the integrand – M under different lift-off distances

Since the eddy current is confined to the surface of the test piece under the high working frequency, the thickness of the sample barely affects the signal. That is, $\text{Re}(e^{2\alpha_1 c}) \geq 1$. Therefore, the phase part of the integrand - T in (5) can be estimated as

$$T = -\text{Re}\left(\frac{\alpha_1 - \alpha}{\alpha_1 + \alpha}\right) \quad (10)$$

Besides, it can be seen from Fig. 2 that, the maximum effective α for M is $2\alpha_0$ (around 250 for the sensor in Table 1). Consequently, $\alpha^2 \ll 2\pi\sigma\mu_0 f$ holds under high working frequencies. Then, T can be approximated as

$$T = -\text{Re}\left(\frac{\sqrt{\pi\sigma\mu_0 f} + j\sqrt{\pi\sigma\mu_0 f} - \alpha}{\sqrt{\pi\sigma\mu_0 f} + j\sqrt{\pi\sigma\mu_0 f} + \alpha}\right) \quad (11)$$

Since $\sqrt{\pi\sigma\mu_0 f} \gg \alpha$, equation (11) becomes

$$T = -1 \quad (12)$$

Substituting (12) into (9), the inductance under the high working frequency becomes

$$L = -KS \int_0^{2\alpha_0} e^{-2\alpha_0 l_0} \sin^2\left(\frac{\alpha\pi}{2\alpha_0}\right) d\alpha \quad (13)$$

After the integration, (13) becomes

$$L = -KS \frac{\pi^2 (1 - e^{-4\alpha_0 l_0})}{4l_0 (4\alpha_0^2 l_0^2 + \pi^2)} \quad (14)$$

The lift-off of the eddy current sensor - l_0 can be derived by finding the numerical solution of the non-linear equation (14). It can be observed that under high working frequencies, the inductance value is (almost) immune to the test piece. Moreover, the solution of the lift-off is merely determined by the parameters of the eddy current sensor (including h and spatial frequency α_0 - determined by the height h , and radii r_{r1} , r_{r2} , r_{e1} , r_{e2} of coils).

C. Prediction of sample thickness

The derived lift-off, l_0 , from equation (14) can be further used for the prediction of the thickness of non-magnetic conductive plates. The thickness can be reconstructed with the modified Newton-Raphson method in an iterative manner. Parameters including the predicted lift-off, l_0 , the electrical conductivity, σ , and a lower single frequency are served as input for the prediction in each iterative loop. The single working frequency should be lower enough that the actual thickness of the test piece is much smaller than the skin depth computed by the following equation.

$$\delta = \frac{1}{\sqrt{\pi\sigma\mu_0 f}} \quad (15)$$

The predicted thickness under the current iterative loop is

$$c = c_r + \Delta c \quad (16)$$

In (16), c_r is the reference value of the thickness, i.e. the predicted thickness from the last iterative loop. For the first iterative loop, c_r can be roughly predicted from the intuitive observations. Δc is the predicted change of the thickness and can be calculated as

$$\Delta c = J(L_m - L(c_r)) \quad (17)$$

In (17), L_m is the measured inductance. $L(c_r)$ is the computed inductance with the input of c_r using equation (1). J is the reciprocal of the inductance sensitivity with respect to the thickness of the test piece c_r .

$$J = \frac{c_r - c_0}{L(c_r) - L(c_0)} \quad (18)$$

In (18), c_0 is a value very close to c_r (it can be defined as $c_0 = 0.95 c_r$). $L(c_0)$ is the computed inductance with the input of c_0 using equation (1). The iterative loop can be stopped as $\frac{\Delta c}{c_r}$ approaches a small residual.

TABLE I
PARAMETERS OF EDDY CURRENT SENSOR

	Exciting coil	Receiver coil
Inner radii - r_{e1} or r_{r1} / mm	19.7	14.4
Outer radii - r_{e2} or r_{r2} / mm	13.2	13.9
Height - h / mm	5.0	
Turns	30	
Lift-off distance - l_0 / mm	1.0 - 12.0	

III. EFFECT ANALYSIS AND MEASUREMENT SETUP

A. Analysis of measurement uncertainty – effect of thickness

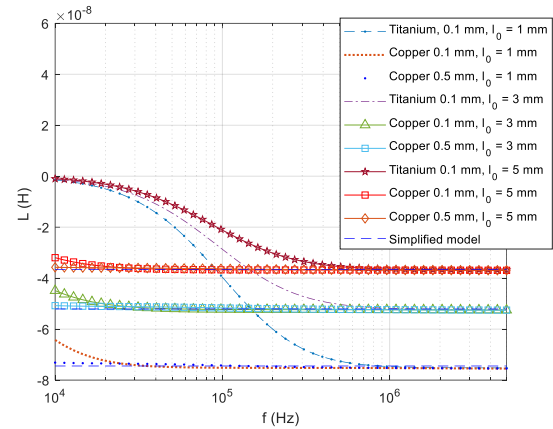


Fig. 3 Simulation of high-frequency inductance change for the eddy-current sensor (size of Table 1) above the Titanium (electrical conductivity of 2.4 MS/m) plate and copper plates with different thicknesses using both conventional formula in (1) and simplified analytical formula in (14)

To retrieve the lift-off information, it is necessary to analyse the coupling effect of different thicknesses. Titanium and copper are selected as the general limit of most non-magnetic samples. Fig. 3 shows the simulation of swept-frequency inductance change for the sensor above the Titanium (electrical conductivity of 2.4 MS/m) plate and copper plates (electrical conductivity of 58.7 MS/m) with different thicknesses. It can be found that under high working frequencies (over 1 MHz), the inductance computed from the conventional analytical model in (1) is insensitive to the thickness variations, and (converges at) matches well with the quiescent inductance from the simplified model in (14). Previously, due to the measurement uncertainty of the specimen (i.e. thickness effect), the Lissajous diagram [44] is proposed for the establishment of a lift-off curve under high frequencies using a calibration specimen of nominal thickness. Regarding the differing lift-off conditions, the proposed method is continuous in lift-off and thickness without calibration. The conventional "Lissajous" method employs a set of calibration specimens that span the thickness and lift-off range of interest to the application. The number of thicknesses and lift-off distances are chosen such that adequate accuracy is achieved using interpolation over a 2D calibration surface in R-L (resistance-inductance) space for the pair of impedance measurements. Hence, both methods deal with changing lift-off and thickness conditions for online application. However, for the potential high-temperature measurement (e.g. testing of hot strip steel mill [45]), the sensor dimensions (e.g. coil height, radius, spacing, etc.) will slightly vary with the temperature. Changing of sensor dimensions will result in different impedance responses and 2D R-L calibration diagrams. Thus, a changing step of each sensor dimensions (coil height, radius, spacing, etc.) can lead to a new trivial series of diagram mapping. The proposed method is analytically based, which can easily be adapted to sensors with different dimensions or varying dimensions (e.g. with different temperatures). Overall, the proposed embedded algorithms save time and effort, are inexpensive (as the calibration blocks are manufactured to the highest standard of quality), eliminate the

calibration need (As potential human factors could highly affect the reliability of the NDT inspection in the calibration process). Examples of analytical based R-L space diagram for the inductive coil above the half-space conductor, tilted effect of sensors, and thin-skin regime of inductive coil above long surface crack model are reported in [43,46,47]. Moreover, with the analytical based algorithm, multiple properties (including thickness, magnetic permeability, electrical conductivity, and lift-off) could be simultaneously retrieved from multi-frequency impedance [9, 26, 29] (e.g. the Newton-Raphson based inverse solver [29]). The capability of proposed method compared to [29] is only a single-frequency inductance is required for the lift-off retrieval. Besides, the lift-off range is much extended up to 12 mm. However, the proposed method is limited to non-magnetic plates with an accurate measurement of electrical conductivity. The electrical conductivity of the test piece is derived from four-terminal measurement. With the simplified algorithm in (14), the lift-off distance is retrieved without the calibrated specimen and nominal thickness profile. Moreover, compared to the conventional formula in (1) (Dodd-Deeds model), the simplified (high-frequency regime) one is already after the integral. Therefore, the simplified algorithm can apply for the online lift-off retrieval as it only takes milliseconds (compared to the conventional formula in (1), which takes over 10 seconds due to the integration over the whole wavenumber - α domain).

B. Measurement system and sensor setup

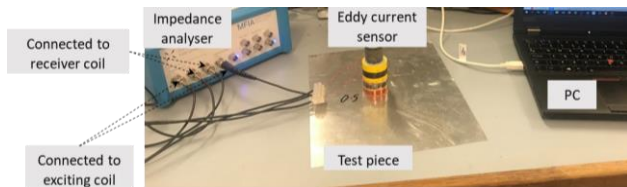


Fig. 4 Measurement system

For the experimental verification, the eddy current sensor is connected to the Zurich impedance analyser (Fig. 4). Considering the effect of non-ideal electromagnetic interferences, the measured data is refined by subtracting to the background/ambient signal (noise) when the sensor is in the free space (without the test piece). The refined measurement data, i.e. the inductance change due to the test piece can be used for further signal processing on the retrieval of sample parameters using analytical solvers in (14). The sensor consists of two co-axial coils, including the circular exciting coil and receiver coil. As listed in Table 1, the two coils have different radii but the same height, and are wound at the same lift-off plane. The coil is wound using the enamelled copper wire. To test the inductance under different lift-off distance, 12 spacers with a thickness of 1.0 mm are used between the sensor and test piece. Besides, the inductance is measured under multi frequencies to test the performance of the proposed method (for the prediction of both lift-off distance and thickness of the non-magnetic conductive plate) under different working frequencies. The working frequency of the measurement ranges from 0.1 to 500 kHz. Since the measured inductance has been found sensitive to lift-off distance and almost nearly independent of the test piece, the higher-frequency inductance

is used for the prediction of the lift-off distance. Moreover, as the inductance is less affected by the lift-off distance and more sensitive to the test piece (particular the thickness without concerning the eddy current skin effect), the lower-frequency inductance is applied for the further prediction of the thickness of non-magnetic conductive plates.

Since the thickness of the non-magnetic sample should be much smaller than the skin depth (can be calculated using equation 15), three non-magnetic thin plates – aluminium, copper, and brass with thicknesses of less than 1 mm have been used for the measurement. As listed in Table 2, the selected samples have a wide range of electrical conductivity, which aims to test the performance of the proposed prediction method for different properties of the (non-magnetic conductive) sample.

TABLE II
PROPERTIES OF NON-MAGNETIC SAMPLES

	Aluminium	Copper	Brass
Electrical conductivity / MSm^{-1}	36.9	58.7	15.9
Thickness / mm	0.5	0.8	0.4

IV. RESULTS AND DISCUSSIONS

A. Experimental result and analysis

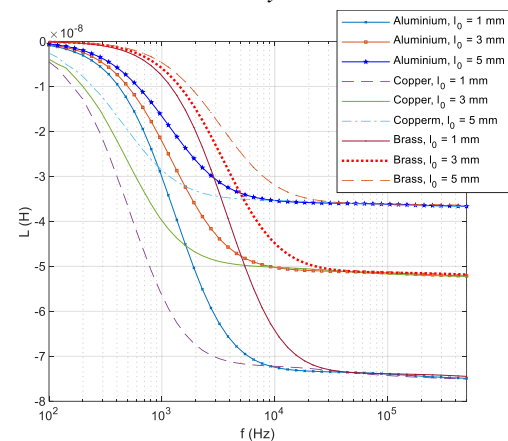


Fig. 5 Comparison of the measured inductance change (due to test piece) for different non-magnetic samples and lift-off distance (frequency ranges from 0.1 to 500 kHz)

Fig. 5 shows the measured inductance change (due to test piece) of three samples under frequencies from 0.1 to 500 kHz with different lift-off distances. The magnitude of the inductance curve increases rapidly and gradually nearly stable with the working frequencies owing to an enhanced coupled effect between the coils and test piece. It can be observed that for a specific lift-off distance, the inductance curve will shift laterally towards either left or right for different non-magnetic samples.

For a specific non-magnetic sample, the inductance curve is less affected by the lift-off distance between the eddy current sensor and test piece under lower working frequencies. Moreover, the inductance curve for a specific non-magnetic sample will significantly reduce with the lift-off (particularly under the high working frequency), which agrees with the lift-off decay factor - $e^{-2\alpha l_0}$ in equation (1) (that is $e^{-2\alpha l_0}$ approaches its maximum value - 1 for a smaller lift-off value). Besides, it can be found that under the high working

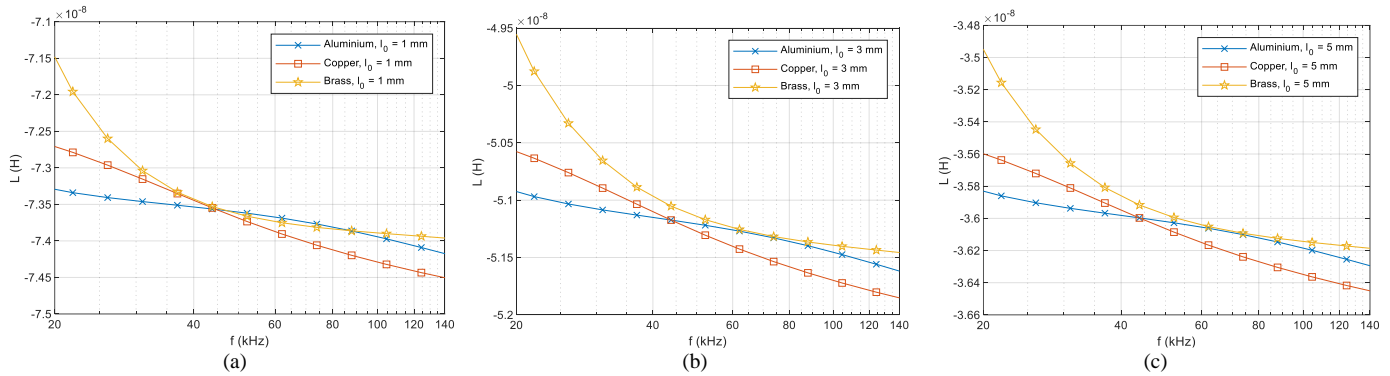


Fig. 6 Zooming-in comparison of the measured inductance change (due to test piece) for different non-magnetic samples (frequency ranges from 20 to 140 kHz) (a) lift-off distance of 1.0 mm (b) lift-off distance of 3.0 mm (c) lift-off distance of 5.0 mm

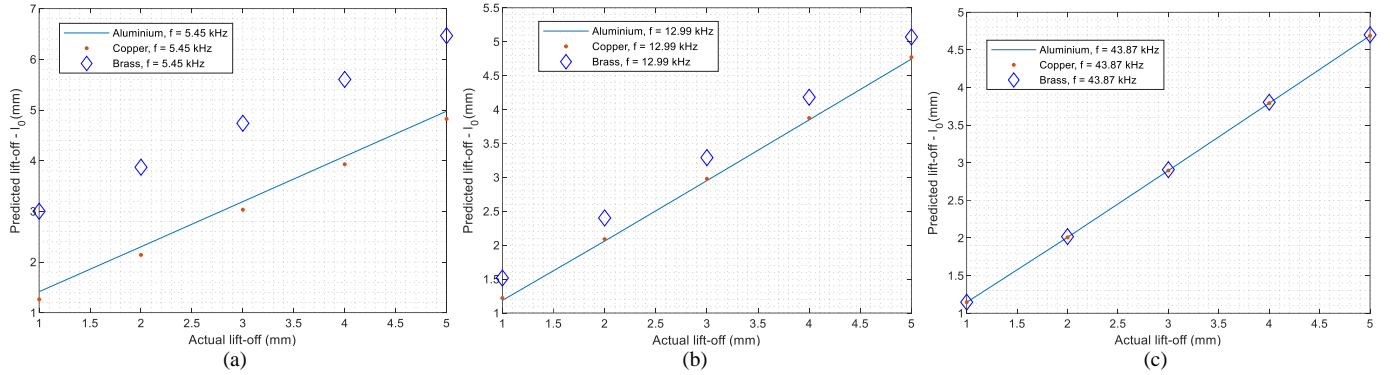


Fig. 7 Comparison of the predicted lift-off distance for different non-magnetic samples (lift-off ranges from 1.0 to 5.0 mm) (a) under the working frequency of 5.45 kHz (b) under the working frequency of 12.99 kHz (c) under the working frequency of 43.87 kHz

frequencies, the inductance gradually becomes nearly independent of the test piece. The independence of the sample is due to the skin effect of the eddy current within the sample. The eddy current is constrained beneath the surface of the test piece under high working frequencies. Moreover, the high-frequency eddy current is almost delayed 180 degrees with respect to the induced voltage on the sensor. Therefore, the test piece can be treated as a pure inductor or inverter under high frequencies, which agrees with equation (12) ($T = -1$ under high working frequencies).

Fig. 6 depicts the zooming-in frequency range for the inductance curves under a specific lift-off distance between the eddy current sensor and sample. It can be seen from Fig. 6 (a) that under the high working frequency and lift-off distance of 1.0 mm, the difference between the inductance of three non-magnetic samples gradually decreases and arrives at its lowest value (around 0.07 % compared to the magnitude) around 43.87 kHz. After 43.87 kHz, the difference of the inductance for three samples slightly increases and becomes a constant value with the working frequency. Similar trends can be observed for the lift-off distance of 3.0 mm in Fig. 6 (b) and 5.0 mm in Fig. 5 (c), where the difference is around 0.28 % and 0.25 % respectively. Therefore, the optimal frequency for the prediction of the lift-off distance is around 43.87 kHz for different non-magnetic samples.

B. Prediction of lift-off distance

As shown in Fig. 7, the lift-off distance has been predicted by finding the solution of the proposed non-linear equation in (14) under the working frequency of 5.45, 12.99, and 43.87 kHz. From (14), the lift-off distance can be estimated without

the variables (parameters) of the test piece – electrical conductivity. It can be observed that for a small range of lift-off distance (from 1.0 to 5.0 mm), the predicted lift-off distance is linear with but higher than the actual value under frequencies lower than 43.87 kHz. This is because the inductance curve is shown more sensitive to the test piece under lower frequencies (as depicted in Fig. 5). Besides, the proposed method in (14) is based on the high-frequency approximation ($T = -1$) where the inductance is nearly independent of the test piece. As the working frequency approaches 43.87 kHz, the prediction curve for three non-magnetic samples gradually overlaps (independent of test piece). For a small lift-off range up to 5.0 mm, the maximum error of the predicted lift-off is around 0.3 mm (at 5.0 mm) under 43.87 kHz.

Fig. 8 (a) illustrates the error of the lift-off prediction for a wide range up to 12.0 mm under the working frequency of 43.87 kHz. It can be found the error curves of the lift-off predicted from the inductance of three non-magnetic samples almost overlap on the whole lift-off range. Moreover, as the actual lift-off increases, the predicted lift-off begins from a slightly overestimated value and almost linearly varies to an underestimated value. The estimation error for a wide range of lift-off (up to 12.0 mm) reaches its lowest value $-8.8 \mu\text{m}$ at 2.0 mm and the highest value $-1.1 \mu\text{m}$ at 12.0 mm under 43.87 kHz.

It can be seen from Fig. 8 (b) that the maximum error of the restored lift-off becomes more sample-dependent but changes very slowly for frequencies higher than 43.87 kHz. Thus, any frequency higher than 43.87 kHz is acceptable for lift-off retrieval. For thinner and less-conductive samples, the inductance curve shifts to higher frequencies [12]. Thus, the

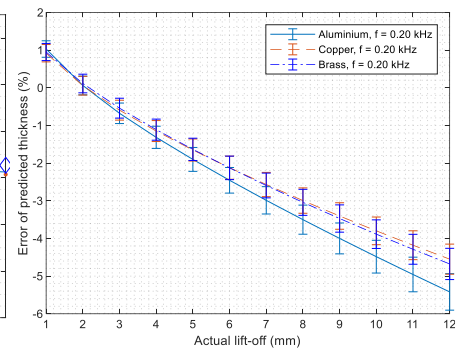
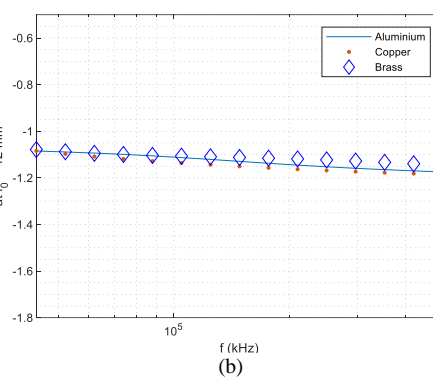
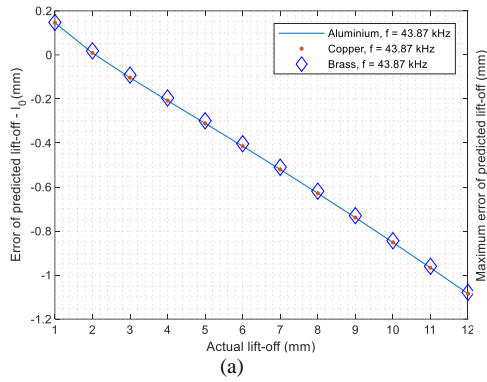


Fig. 8 (a) Comparison of the predicted lift-off distance for different non-magnetic samples (lift-off ranges from 1.0 to 12.0 mm) under the working frequency of 43.87 kHz (b) Maximum error of predicted lift-off at actual value of $l_0 = 12$ mm under different working frequencies

Fig. 10 Comparison for the error of the predicted thickness of different non-magnetic conductive plates (lift-off ranges from 1.0 to 12.0 mm) under the working frequency of 0.20 kHz

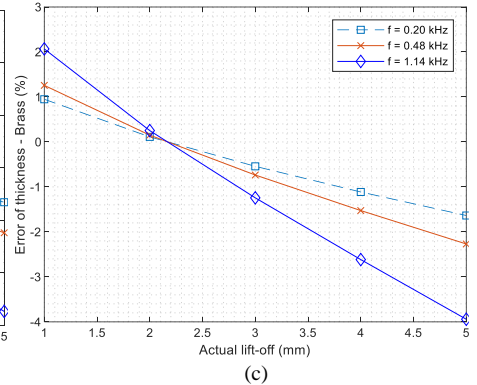
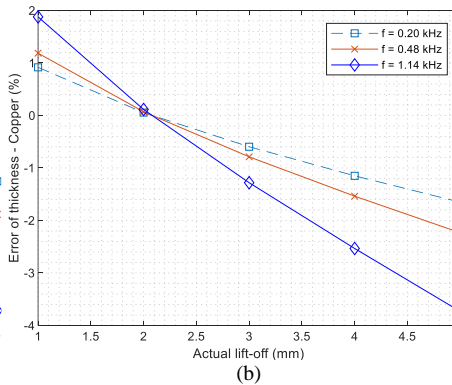
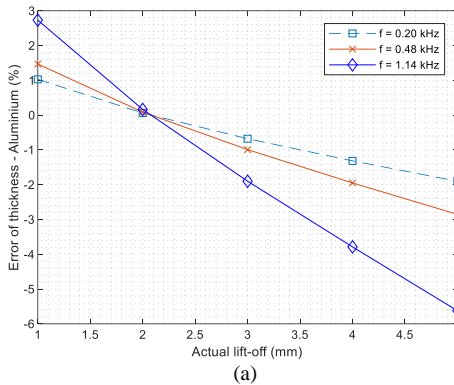


Fig. 9 Comparison for the error of the predicted thickness under different lower working frequencies. (lift-off ranges from 1.0 to 5.0 mm) (a) Aluminium plate (b) Copper plate (c) Brass plate

minimum value of the optimal frequency range increases with reduced thickness (as can be observed from Fig. 3). Overall, for majorities of non-magnetic samples (with electrical conductivities range from 2.4 (Titanium) to 100 MS/m), frequencies over 1 MHz are within the optimal frequency range for the thickness of over 0.1 mm.

C. Prediction of sample thickness

As shown in Fig. 9, the reconstructed lift-off can be used for the further prediction of the sample thickness. Parameters including the coil height, coil radii, number of turns (from the eddy current sensor in Table 1), electrical conductivity (from samples in Table 2), and reconstructed lift-off can be used for the initial reference value of the iterative method. The iterative loop can be terminated (under the condition/criterion) as the thickness change Δc in (17) becomes a stable residual.

Since the skin depth will decrease with the working frequency (according to equation 15) and may close to or even lower than the thickness of the test piece, the prediction of the sample thickness is operated under lower frequencies – 0.20, 0.48, and 1.14 kHz. It can be seen from Fig. 8 that the prediction reaches its lowest error under the working frequency of 0.20 kHz. This is because the inductance curve is more sensitive to the test piece and less affected by the lift-off distance under a lower working frequency (as can be seen from Fig. 5). Moreover, the skin depth for a lower working frequency is much larger than the thickness of the test piece, where the eddy current is more restrained by the bottom surface of the test piece. It can be seen that for a small lift-off range up to 5.0 mm,

the maximum error of the predicted sample thickness is around 2 % (at 5 mm) under 0.20 kHz.

Fig. 10 illustrates the error of the thickness prediction for a wide lift-off range up to 12.0 mm under the working frequency of 0.20 kHz. As the actual lift-off increases, the predicted thickness starts from an almost intersected and overestimated value, then becomes more sample-dependent and underestimated. Although the electrical conductivity is accurately measured (within 1 % deviation of the actual value on the datasheet) through the four-terminal measurement method, error bars are included to present variations of the predicted thickness considering a maximum deviation of 10 % of actual electrical conductivity (for three test pieces). As the error of the predicted lift-off reaches its lowest value at 2.0 mm (Fig. 8), the mean error of the predicted sample thickness achieved its lowest value (0.11 %) at the same actual lift-off 2.0 mm. Besides, the maximum mean error of the predicted thickness is 5.42 % at the lift-off distance of 12 mm. Due to the different sensitivity of the inductance regarding the lift-off, the prediction error of the thickness differs from that of the lift-off distance. Moreover, the influence of conductivity variations on the predicted thickness is less than 1 %, which is due to the low sensitivity of inductance regarding the conductivity (More details on the multi-frequency sensitivities are reported in [26]). For the measurement uncertainty of electrical conductivity, impedance values under more different frequencies are required to simultaneously retrieve the thickness and electrical conductivity of test pieces [9, 26, 29].

V. CONCLUSIONS

This paper has introduced an algorithm for the measurement of the lift-off distance between the eddy current sensor and non-magnetic conductive plate. According to the experimental result, the inductance for the sensor above the sample has been found sensitive to the lift-off and nearly independent of different non-magnetic samples (with an error of 0.28 %) under a single high frequency 43.87 kHz. Based on this effect (termed as the sample-independent insurance (SII) feature), the lift-off distance can be predicted with a maximum absolute error of 1.1 mm at the actual distance of 12.0 mm. Moreover, it has been found that any frequency higher than 43.87 kHz is acceptable for the lift-off retrieval. Considering the effect of measurement uncertainty (especially the thickness of the test piece), working frequencies over 1 MHz apply to most non-ferrous metals with thicknesses of over 0.1 mm.

Moreover, the predicted lift-off distance is used as the reference value for the prediction of the sample thickness. The prediction is operated under the iterative manner. Concerning the skin depth of the eddy current, a lower single frequency is used for the thickness prediction of the non-magnetic plates. From the measurement of different materials including aluminium, copper, and brass, the thickness can be accurately predicted with a maximum relative error of 5.42 % at the lift-off distance of 12.0 mm.

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