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[Seongwoo Woo](#)*, [Dennis L. O'Neal](#), [Yimer Mohammed Hassen](#), Gezae Mebrahtu

Posted Date: 28 February 2023

doi: 10.20944/preprints202302.0517.v1

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Article

Improving the fatigue design of mechanical products such as bearings based on a (generalized) life-stress prototype and sample size

Seongwoo Woo ^{1,*}, Dennis L. O'Neal ², Yimer Mohammed Hassen ¹, and Gezae Mebrahtu ¹

¹ Manufacturing Technology, Mechanical Technology Faculty, Ethiopian Technical University, Addis Ababa P.O. Box 190310, Ethiopia

² Department of Mechanical Engineering, School of Engineering and Computer Science, Baylor University, Waco, TX 76798-7356, USA; dennis_oneal@baylor.edu

* Correspondence: twinwoo@yahoo.com; Tel. : +251-90-047-6711

Abstract: To prolong the fatigue life of a product handled by machine, parametric accelerated life testing (ALT) is mentioned as an organized technique to pinpoint design flaws and minimize fatigue-connected failures. It requires (1) an ALT procedure, (2) fatigue model, (3) parametric ALTs with adjustment, and (4) an approximate of whether present product complete the BX life. The use of a quantum-transported time-to-failure archetype and a sample size are advised. The improvements in the reliability of a refrigerator ice-maker, comprising an auger motor with bearing, were utilized as a case investigation. In the 1st ALT, a steel rolling bearing cracked due to repeated loads under cold circumstances (below -20°C) in the freezer compartment. The bearing was altered by adjusting the matter from AISI 52100 alloy steel with 1.30-1.60% chromium to lubricated sliding bearing with sintered and hardened steel (FLC 4608-110HT) because of its high fatigue strength at lower temperatures. In the 2nd ALT, a helix made of polycarbonates (PCs) fractured. As an adjustment, a reinforced rib of the helix was thickened. As no troubles in the 3rd ALT occur, the lifetime of an ice-maker was proved to be B1 life 10 years.

Keywords: mechanical system; fatigue design; parametric ALT; ice-maker; bearing; design defects

1. Introduction

To have good qualities competitive with rival companies in the market, mechanical systems such as refrigerators shall be enhanced with new scientific knowledge and performance to satisfy the desires of purchaser. If these new attributes are supplied to the field with insufficient testing, there is the possibility for early failure of these new properties. These untimely failures undesirably influence the recognition of the aspect of the products. To avoid discovering unanticipated design defects in the market, the new attributes for a product should be judged in the development process before being launched into the end-user. Assessing the reliability of a new mechanical system should include a systematic method with reliability quantitative (RQ) specifications [1].

Achieving decreasing speed through several gears engaged with a driving gear mounted in the shaft, an ice-maker including auger motor with bearing is designed to attain a sufficient torque that can squash the through a large torque. Under the low temperature condition (-20°C ↓) in the freezing compartment, the auger motor is subjected to repeated stresses supported by bearings. A common material utilized in ball bearing rings is the alloy steel AISI 52100 because it is effortlessly forged, heat treated, and machined. To stop the cancellations of mechanical systems from the market which have structural imperfections [2-5], it might be designed to outlast the normal functioning circumstances executed by customers who acquire and utilize the product.

For example, after 346 passengers died in a crash, the Boeing 737 MAX airline from March, 2019 to December, 2020 was prohibited from flying. The airplane used the CFM

International LEAP-1B engines adopting the most effective 68-inch fan design. They were 12% more fuel efficient and 7% lightweight than previous engine [6]. Inspectors had conjectured that the accident was produced by the engine in the aircraft. As a result, the whole economy experiences the elimination of lots of parts (or wastes) due to improper design. Possible troublesome components thus need to be confirmed by laboratory testing that shall cause reliability quantitative (RQ) expression [7-9].

Fatigue is the main origin of metallic failure in structural elements, elucidating more or less 80–95% of all failures [10]. It displays itself in the form of cracks which usually start from stress raisers, such as holes, edges, slender surfaces, grooves, etc., on the structure of systems. Fatigue is the withering of a matter that is frequently produced by cyclic loading. Notable concern concentrates on the (low-cycle) fatigue of components, specifically in the area of turbine-engine that is nickel-built polycrystalline matter [11,12]. It is also measured as a quantity element, such as the stress proportion, $R (= \sigma_{\min}/\sigma_{\max})$, which shall be explained as the correlation of the greatest cyclic stress to the least cyclic stress [13]. Utilizing a stress proportion, R , which shall be presented in an ALT, shall identify the design defects in the mechanical system.

Designers have frequently recognized design imperfections and have fixed them by utilizing technique such as Taguchi's method [14]. Especially, design of experiments (DOE) [15] is an organized method to decide the connection between the factor affecting a procedure and its production. The aim is to secure that the factors are placed in the most successful manner for functioning (or environmental) situations. DOE is executed for related factors which influence the product designs. Their functionality is revealed by analysis of variance (ANOVA). Because a person who operates DOE may not know which factors are the most influential in a failure, there is no fatigue failure in the process to be followed in calculations. Thus, DOE may require a large number of mathematical calculations and may not identify a potential source of failure.

Designers have frequently utilized the strength of materials as a solution to conventional design [16-18]. A crucial element in fracture mechanics [19] is the toughness as a material attribute of strength. With the implementation of quantum mechanics, engineers have pinpointed that structural failures occur from nanoscale or microscale voids, which may occur in metallic alloys or engineering plastics. As finite samples and limited testing periods are utilized [20–24], this method cannot reproduce the design flaws in a complex form or identify the fatigue problem which occurs to the elements by consumer in the market. To identify the fatigue phenomena in a system functioned by machine, a life-stress type [25,26] can be integrated with a (quantum) mechanics way to distinguish a prevailing defect or crack form in matter because unsuccessfulness stochastically happens in the region of particularly big stress.

The finite element method (FEM) [27] is utilized as a different way. Designers think that failures may be determined by (1) an appropriate mathematical (Lagrangian or Newtonian) formulation; (2) deriving the time response for (dynamic) loads, which generates the stress/strain on the part structure; (3) employing the generally accepted way of rain-flow counts with von Mises stress [28]; and (4) evaluating system effectiveness by Palmgren–Miner's principle [29]. Deploying this methodology shall give closed-form answers. However, this method cannot pinpoint fatigue failures in a complex system produced by structural defects such as micro-voids, sharp edges, slender surfaces, contacts, etc.

This investigation proposes parametric ALT as a straightforward way that can be employed to identify structural flaws of a new product and improve them. It involves the following: (1) a parametric ALT plan developed on BX life, (2) load study, (3) ALTs with the structure modifications, and (4) an appraisal of whether the system structure fulfills the objective BX life. This procedure for recognizing the root causes and enhancing design examines properness in the mechanical system such as appliance, automobile, airplane, etc. The quantum-transported failure type and sample size also are advocated. A new refrigerator ice-maker involving an auger motor with a bearing is deployed to explain it.

2. Parametric ALT for a System functioned by machine

2.1. Meaning of BX life

Product operated by machine utilizes (generated) power to achieve a desired motion by adapting an appropriate mechanism [30]. Forces are utilized to supply movement of mechanisms in the system. This movement signifies that the system shall be subjected to repeated loads. In a mechanical product, fatigue falls when there are structural imperfections such as notches, sharp-edged, grooves, slender surfaces, etc., in a component.

For instance, a refrigerator uses the heat pump cycle that consists of condenser, capillary tube, evaporator, and compressor. In a heat exchanger, chilled air is generated so that it keeps not having decayed for the food in the refrigerator and freezer sections. Figure 1 manifests that a refrigerator covers some complete subsystems (or modules) – the cupboard and door, shelves and boxes, compressor or (electric) motor, evaporator and condenser, water supplying and ice-maker apparatus, controller, and diverse parts. A domestic refrigerator includes the same number of 2000 elements. It can be divided into up to 20 units (or 8~10 modules) possessing roughly 100 elements.

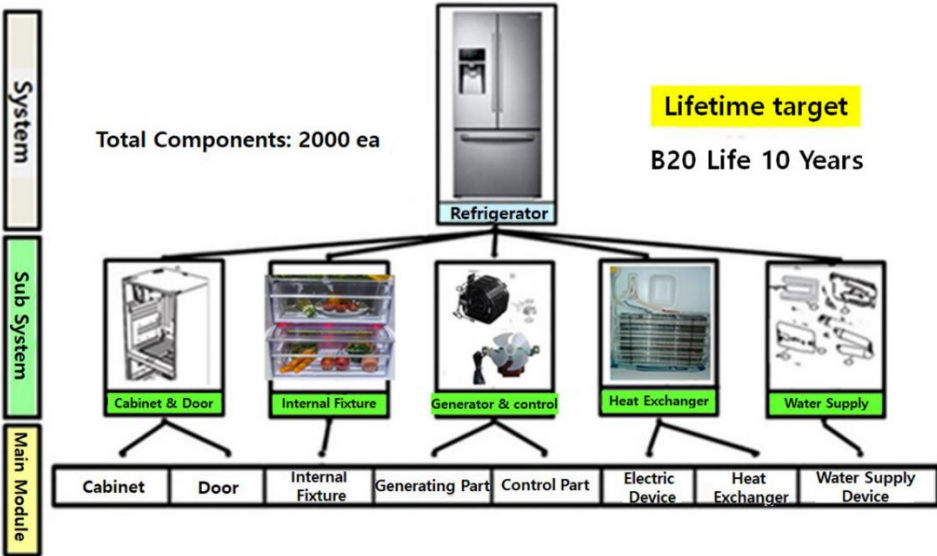


Figure 1. Classification of a domestic refrigerator with multiple subsystems

As the objective of system life is presumed to have B20 life 10 years, the lifetime objective of all units should have B1 life 10 years. As a new subsystem, named Module #3 in Figure 2, has a design flaw, it resolves the lifetime of the total refrigerator.

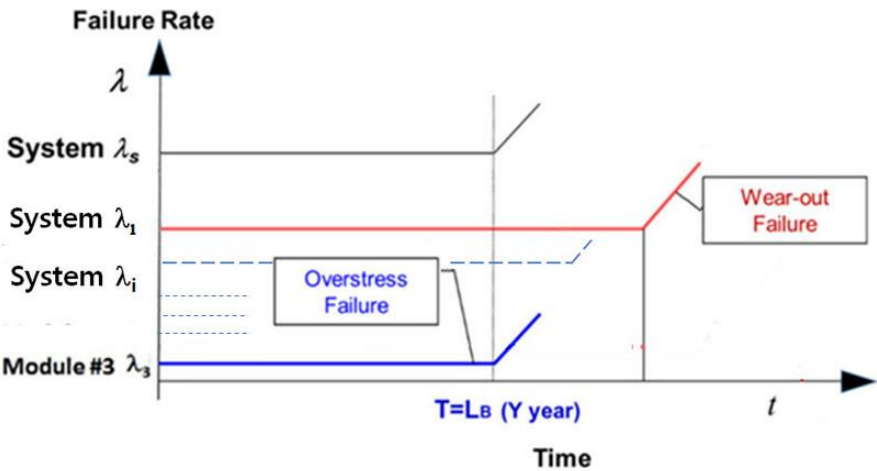


Figure 2. Complex product lifetime determined by the new module

BX lifetime, L_B , could be explained as a quantity of lifetime that X percent of the population has been unsuccessful. 'BX life Y years' therefore is a further reasonable clause.

That is, as the life of a mechanical part is B20 life 10 years, 20% of the concern parts shall have been failed for ten-year. Otherwise, as the reverse of the failure rate, the B60 life, denoting the mean time to failure (MTTF), might not be utilized for system lifetime as it is too lengthy for 60% of system to be unsuccessful. BX lifetime shall assign an acceptable indicator for system life.

2.2. Posing an Entire ALT procedure

Reliability might be defined as the potential to function under a set of prescribed operational/environmental circumstances for a required period of time [31]. It is mainly demonstrated as bathtub curve, having three divisions. They may be expressed in accordance with the shape parameter in the Weibull chart. In the 1st division, there is a declining rate of failure in the premature section of the system's life ($\beta < 1$). In the 2nd division, there is a uniform rate of failure ($\beta = 1$) in the medium lifetime of the system, pursuing an exponential distribution. Finally, there is a growing rate of failure to the ending of the product life ($\beta > 1$), which pursues a Weibull distribution (Figure 3).

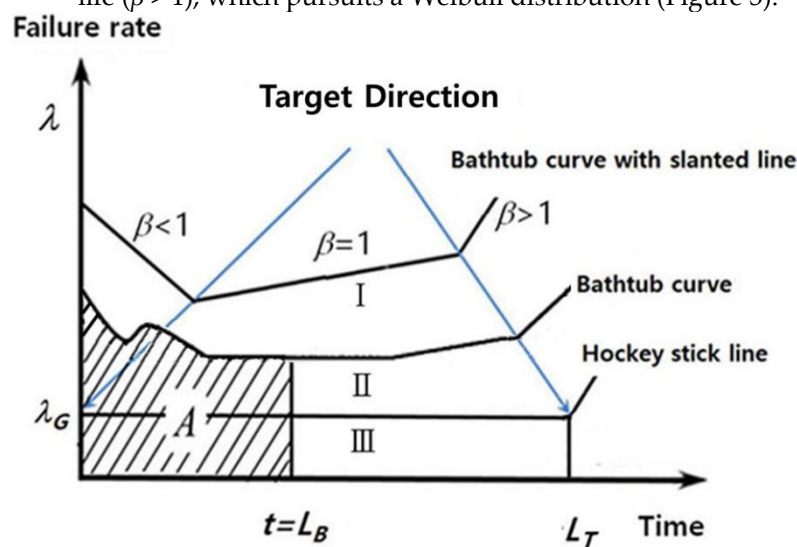


Figure 3. BX lifetime (L_B) on the (slanted) bathtub.

The reliability function, $R(t)$, is $R(t) = 1 - F(t)$ (or X). The unreliability or accumulative distribution function (CDF), $F(t)$ (or X) = $1 - R(t)$, is expressed by

$$F(t) = P(T < t) \quad (1)$$

On the (slanted) bathtub in Figure 3, the failure rate, λ , shall be defined as:

$$\lambda(t) = f(t)/R(t) = \frac{dF(t)/dt}{R(t)} = \frac{(1 - R(t))'}{R(t)} = \frac{-R'(t)}{R(t)} \quad (2)$$

where f is the failure density function.

If Equation (2) takes the integral, the life of $X\%$ cumulative failure, $F(L_B)$, at $t = L_B$ may be assessed. $F(L_B)$ shall be expressed:

$$F(t) = \int \lambda(t) dt = -\ln R(t) \quad (3)$$

$$\text{Or } A(\text{or } X) = \langle \lambda \rangle \cdot L_B = \int_0^{L_B} \lambda(t) \cdot dt = -\ln R(L_B) = -\ln(1 - F) \cong F(L_B) \quad (4)$$

As T_1 is presumed to have the period of the 1st failure in the 2nd division of the (slanted) bathtub, reliability, $R(t)$, shall be expressed:

$$R(t) = P(T_1 > t) = P(\text{no failure in } (0, t]) = \frac{(m)^0 e^{-m}}{0!} = e^{-m} = e^{-\lambda t} \quad (5)$$

As the failure rate of a system mimics the features on the (slanted) bathtub curve (Stage I or II), it shall be unsuccessful in the marketplace. Due to design flaws, the huge number of premature failures in the early part of the curve could spoil the brand name of the company with the product release. High failure rates in the initial product lifetime require warranty costs on the manufacturer, and market share would be anticipated to be negatively affected. The company would be required to enhance the system by (1) abolishing unpredicted premature failures, (2) lessening (random) failures for its function time, and (3) growing the product life.

As a structural design is enhanced, the system lifetime from the market should increase, and its failure rate should decrease. In a situation, the (slanted) bathtub might be altered to a curve with low failure rates and a long lifetime. Eventually, if the system is competently made, the accumulative failure rate, $F(t)$, is improved till the useful life is fulfilled. The product's bathtub shall be similar to the straight line categorized the "Hockey stick line" (stage I \rightarrow III) in Figure 3.

In Equation (5), the product reliability is simply expressed as the failure rate, λ , and lifetime, L_B . Namely,

$$R(L_B) = 1 - F(L_B) = e^{-\lambda L_B} \cong 1 - \lambda L_B \quad (6)$$

This correlation is adequate and less than just about 20% of the cumulative failure rate [32].

As an instance, an ice maker repetitively demands a straightforward mechanical operation: (1) water is provided to the flat and shallow container; (2) it then solidifies into ice by cooled air being blown over it; and (3) it is then harvested till the ice container is filled. Ice is retrieved by the consumer when the end-user applies force on a lever that allows the cubed (or crushed) ice to dispense. During the process, an ice-maker shall be subjected to repeated stresses. Failed parts from the field are decisive for comprehending and pinpointing the repetitive usage methods of end-users and picking out structural imperfections in the structure. From the marketplace statistics, the real cause(s) of the troublesome auger motor, including the bearing, was recognized. As setting the objective life, L_B , by employing an ALT, the part functioned by machine shall be altered by pinpointing the controversial component and improving it (Figure 4).

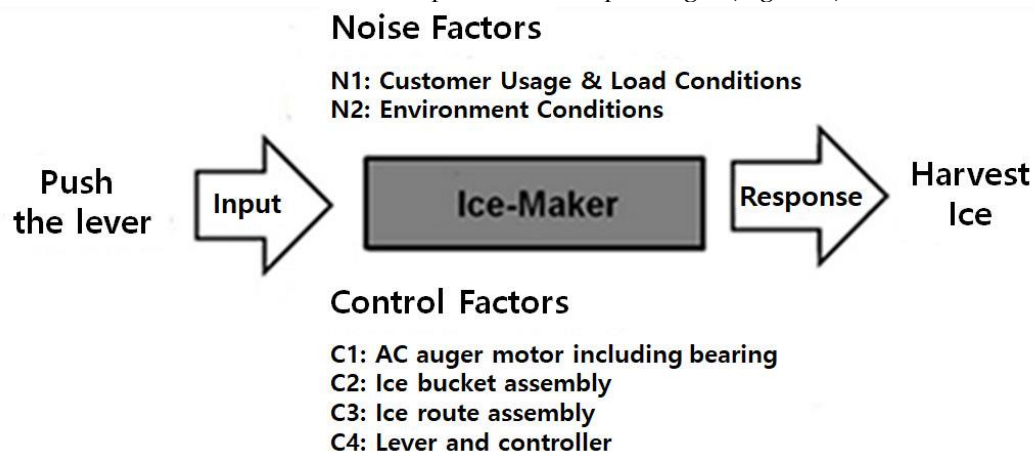


Figure 4. Parameter description of the ice maker (example).

From the market statistics – present lifetime and failure rate – the real cause(s) of the troublesome ice-maker failed from the end-user had been plainly recognized. To fulfill the desirable reliability from the objective life, L_B , and failure rate, λ , the possible design flaws of the component might be found and altered by utilizing an ALT.

To reach the target of a product life by ALT, three subsystems (or modules) were classified: (1) a modified system, (2) a newly designed system, and (3) the same system.

The subsystem such as ice-maker in a household refrigerator utilized as a test investigation here was a system which had design flaws to be corrected. End-users had been demanding for substitutions which had been failing too soon before the anticipated life of the system. System D (Table 1) from the field statistics had a failure rate of 0.20% per year and a B1 life 5.0 years. To reply to end-user appeal, an objective life for the ice maker was specified to be B1 life 10 years.

Table 1. Complete ALT idea of subsystems in a household refrigerator

Modules	Market Reliability		Predicted Reliability				Goal Reliability	
	Failure Rate Per Year, %/Year	BX Life, Year	Failure Rate Per Year, %/Year		BX Life, L_B (Year)		Failure Rate Per Year, %/Year	BX Life, Year
A	0.34	5.3	New	×5	1.70	1.1	0.15	12(BX = 1.8)
B	0.35	5.1	Same	×1	0.35	5.1	0.15	12(BX = 1.8)
C	0.25	4.8	Modified	×2	0.50	2.4	0.10	12(BX = 1.2)
D	0.20	6.0	Modified	×2	0.40	3.0	0.10	12(BX = 1.2)
E	0.15	8.0	Same	×1	0.15	8.0	0.10	12(BX = 1.2)
Miscellaneous	0.50	12.0	Same	×1	0.50	12.0	0.50	12(BX = 6.0)
System	1.79	7.4	-	-	3.60	3.7	1.10	12(BX = 13.2)

2.3. Deduction of Life-Stress Model

As a customer has a desire to have (cubed or crushed) ice, new function such as an ice-maker is included in a household refrigerator. The major components are the geared auger motor, helix support, bucket case, helix upper dispenser, blade, etc. An auger motor has two or more gears working together by interlocking their teeth and revolving each other to produce torque and speed. As motors are employed, the geared trains lessen the speed of the augers and grow torque. Namely, the auger motor operated by alternating current (AC) grows the torque by gearbox to crush it at the end of an ice-maker. As a consumer pushes the lever with a cup on the dispenser, (cubed/crushed) ice flows to fill the cup. Consequently, the ice maker shall be subjected to repeated stresses due to loading/unloading in the process of crushing ice. If there are structural flaws, such as an inadequate strength to withstand repeated loads, the ice-maker can be successful before satisfying its targeted life. That is, failure happens when the materials in the system parts are too fragile to withstand the exerted stress under environmental circumstance [33].

As reproducing the field failures by ALT, an engineer must understand and quantify the loading that is encountered by the ice maker in the field before designing the system shape and materials to achieve the objective reliability of the system. Once optimally redesigned, the product might be anticipated to endure the minimum repeated loading in its expected life so that it may extend the targeted life. From the relation between load and lifetime, the (generalized) life-stress prototype that will integrate with geometry and material as design solution should be derived, which can be described by the phenomena of void generation/transport from the level of quantum mechanics. Eventually, cracks and their propagation might be described by a sample size formulation (Figure 5).

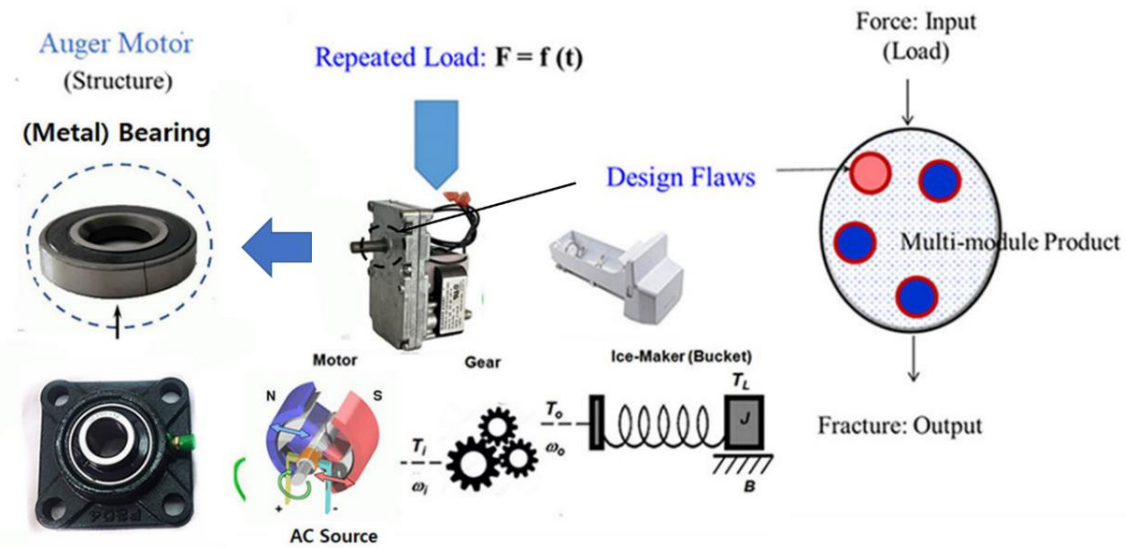


Figure 5. Fatigue produced by repeated loads and design deficiencies.

The motivation for ALT is to resolve how premature the anticipated failure mode might be pinpointed by mathematically employing the work for parametric modeling. That is why elevated tests need to be carried out. To depict the elevated testing time into actual usage time, it is requisite to arrange a straightforward failure expression and resolve the correct numerical method for the life type. The life-stress (LS) type, which requires quantifiable stresses and reaction factors, should be developed. Thus, it will express mechanical failure, such as structural fatigue. Fatigue on the surface of a structure can occur not only due to component stresses but also due to defects such as cracks.

It is presumed that fatigue shall arise from structural defects—electron/void—which appear in a Nano-range or microscopic. Think about a particle which is restrained to move only in the x orientation from $x = 0$ to $x = a$. The (potential) energy, $V(x)$, shall be expressed:

$$V(x) = \begin{cases} 0 & (0 \leq x \leq a) \\ \infty & (x < 0 \text{ or } x > a) \end{cases} \quad (7)$$

The Schrodinger governing equation shall be defined:

$$\hat{H}\psi = E\psi \quad (8)$$

where \hat{H} is the Hamiltonian operator, ψ is the wave function, and E is the (electron) energy.

If $\hat{H} = -\frac{h^2}{8\pi^2m} \frac{d^2}{dx^2} + V$ in Equation (8), it shall be set as follows,

$$\frac{d^2\psi}{dx^2} + \frac{8\pi^2m}{h^2}(E - V)\psi = 0 \quad (9)$$

where h is the Planck constant and m is the electron mass.

As $V = \infty$ at the wall exteriors, it is feasible when $\psi = 0$. Electron is not at the wall exteriors. As $V = 0$ at the wall interiors, Equation (9) shall be expressed:

$$\frac{d^2\psi}{dx^2} + K^2\psi = 0 \quad (10)$$

where $K^2 = \frac{8\pi^2mE}{h^2}$

The answer in Equation (10) shall be presumed:

$$\psi(x) = A\sin Kx + B\cos Kx \quad (11)$$

where A and B are constant

As $x = 0$ or $x = a$ at the barriers, $\psi(0) = \psi(a) = 0$, $B = 0$, $K = \frac{n\pi}{a}$, and $E = \frac{n^2 h^2}{8ma^2}$, $n = 1, 2, 3, 4$
Thus, Equation (11) shall be expressed:

$$\psi(x) = A \sin\left(\frac{n\pi}{a}\right)x$$

(12)

The chance of discovering the electron in a limited room between x and $x + dx$ is expressed:

$$\int_0^a \psi^2(x)dx = 1 \text{ or } \int_0^a \left(A \sin\left(\frac{n\pi}{a}\right)x\right)^2 dx = 1$$

(13)

Thus, the answer of Equation (12) shall be attained:

$$\psi(x) = \sqrt{\frac{2}{a}} \sin\left(\frac{n\pi}{a}\right)x$$

(14)

where $\psi(x + a) = \psi(x)$, a is the (periodic) interval, and n is the main quantum number.

The transport diffusion procedure can be expressed (Table 2) [34,35]:

$$J = LD$$

(15)

where J is the diffusion flux, D is the driving force, and L is the transport quantity.

Table 2. Linear transport phenomena.

In particular, as an electromagnetic force, ξ , is exerted, the metal impurities, caused by electronic motion, easily float to the right-hand as the magnitude of the junction energy is lowered. Expressing solid-state diffusion of impurities of silicon in a semiconductor can be shortened: (1) electro-migration-induced voiding; (2) build-up of chloride ions; and (3) trapping of electrons or holes. The transport diffusion process, J , might be defined as [36]:

$$J = [aC(x - a)] \cdot \exp\left[-\frac{q}{kT}\left(W - \frac{1}{2}a\xi\right)\right] \cdot v$$

(16)

$$= -[a^2ve^{-qw/kT}] \cdot \cosh\frac{qa\xi}{2kT}\frac{\partial C}{\partial x} + [2ave^{-qw/kT}]C \sinh\frac{qa\xi}{2kT}$$

where A is constant, C is the concentration quantity, q is the amount of accumulated electrical energy, v is the jump frequency, a is the atomic interval, ξ is the applied field, k is Boltzmann's quantity, Q is the energy quantity, and T is the (absolute) temperature.

Ohm's Law: $j = -\sigma \nabla V$		
J = current density, j (quantity: A/cm ²)	D = electric field, $-\nabla V$ (quantity: V/cm, V = potential)	L = conductivity, $\sigma = 1/\rho$ (quantity: ρ = resistivity (Ω cm))
Fourier's Law: $q = -\kappa \nabla T$		
J = heat flux, q (quantity: W/cm ²)	D = thermal force, $-\nabla T$ (quantity: °K/cm, T = temperature)	L = thermal conductivity, κ (quantity : W/°K cm)
Fick's Law: $F = -D \nabla C$		
J = material flux, F (quantity:/sec cm ²)	D = diffusion force, $-\nabla C$ (quantity:/cm ⁴ , C = concentration)	L = diffusivity, D (quantity: cm ² /sec)
Newton's Law: $F_u = -\mu \nabla u$		
J = fluid velocity flux, F_u (quantity:/sec ² cm)	D = viscous force, $-\nabla u$ (quantity:/sec, u = fluid velocity)	L = viscosity, μ (quantity:/sec cm)

Unless the electric field is relatively small, i.e., $\xi \ll \frac{qa}{2kT}$, Equation (16) might be redefined as follows:

$$J = \Phi(x, t, T) \sinh(a\xi) \exp\left(-\frac{Q}{kT}\right) = B \sinh(a\xi) \exp\left(-\frac{Q}{kT}\right) \quad (17)$$

where Q is the energy, $\Phi()$ & B are constant.

On the other hand, the chemical process which relies on speed shall be expressed as

$$K = K^+ - K^- = a \frac{kT}{h} e^{-\frac{\Delta E - aS}{kT}} - a \frac{kT}{h} e^{-\frac{\Delta E + aS}{kT}} = A \sinh(aS) \exp\left(-\frac{\Delta E}{kT}\right) \quad (18)$$

where K is the reaction speed, S is the (chemical) field result, E is the (activation) energy, Δ is the difference, and A is constant.

Equations (17) and (18) could be shortened as

$$K = B \sinh(aS) \exp\left(-\frac{Q}{kT}\right) \quad (19)$$

If Equation (19) captures a reverted formulation, the life-stress (LS) type shall be clarified as:

$$TF = A [\sinh(aS)]^{-1} \exp\left(\frac{E_a}{kT}\right) \quad (20)$$

As a life-stress (LS) prototype, Equation (20) is clarified as a general expression because the sine hyperbolic expression $[\sinh(aS)]^{-1}$ designating stress shall be exchanged into a power (or exponential) formulation. It then may outline most of the LS prototypes about some failure, such as fatigue in the system. It can be conveyed: (1) first, $(S)^{-1}$ has a nearly straight line, (2) second, $(S)^{-n}$ has what is viewed, and (3) third, $(e^{aS})^{-1}$ is largely developed (Figure 6).

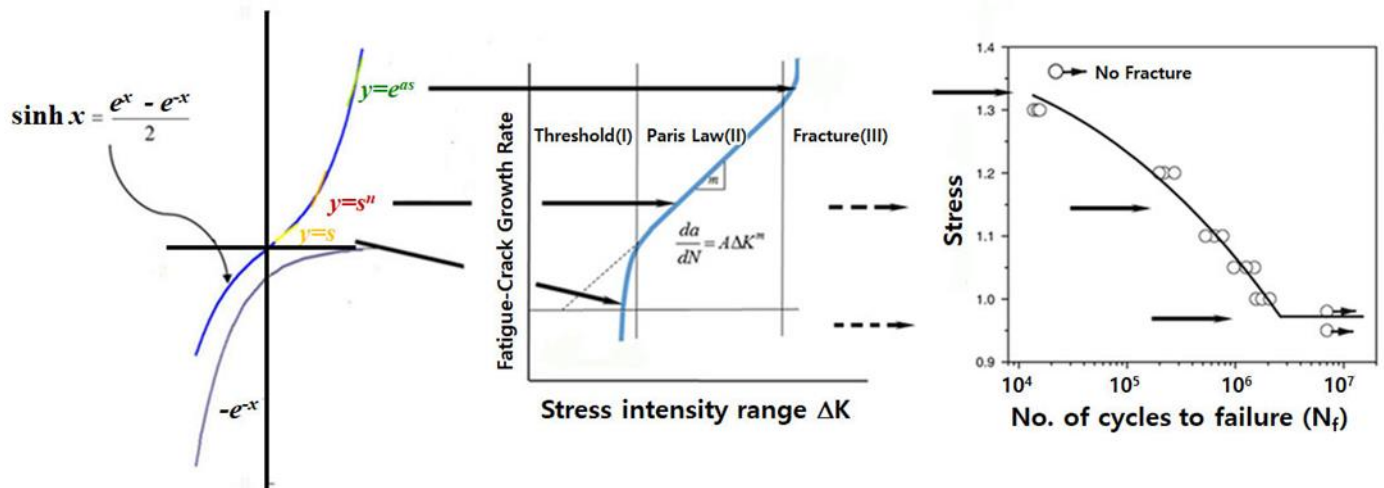


Figure 6. Meaning of the hyperbolic sine stress in the S-N curve and Paris law.

Because ALT is frequently carried out in the span of midst stress, Equation (20) shall be expressed as:

$$TF = A(S)^{-n} \exp\left(\frac{E_a}{kT}\right) \quad (21)$$

$$\text{where } n = -\left[\frac{\partial \ln(TF)}{\partial \ln(S)}\right]_T$$

For an expressed crack and structural form, Equation (21) can be redefined as

$$TF = B(\Delta K)^{-n} \exp\left(\frac{Q}{kT}\right) \quad (22)$$

where B is constant, $\Delta K = YS(or \Delta\sigma)\sqrt{\pi a}$

As stress intensity factor, ΔK , is exerted on a material, the crack will produce to a specific amount Δa , which relies on the crack growth speed, $\Delta a/\Delta N$, in component shapes such as crack tip such as grooves, sharp-edged, slender areas, holes, etc. It therefore propagates to a risky magnitude. As loads are exerted till the targeted lifetime, L_B or mission time, the stress raisers (or material) in a component can be discovered.

The stress of a product functioned by machinery is a complex quantity to formulate in a raised testing. Because the energy is clarified as the product of flow and effort, the stress comes from effort in an energy transport system [37]. Thus, Equation (21) or (22) can be stated as follows:

$$TF = A(S)^{-n} \exp\left(\frac{E_a}{kT}\right) = C(e)^{-\lambda} \exp\left(\frac{E_a}{kT}\right) \quad (23)$$

where C is constant.

The acceleration factor (AF) is defined as the proportion between the raised stress and typical functioning situation. AF from Equation (23) shall be modified to merge the effort idea:

$$AF = \left(\frac{S_1}{S_0}\right)^n \left[\frac{E_a}{k} \left(\frac{1}{T_0} - \frac{1}{T_1}\right)\right] = \left(\frac{e_1}{e_0}\right)^\lambda \left[\frac{E_a}{k} \left(\frac{1}{T_0} - \frac{1}{T_1}\right)\right] \quad (24)$$

2.4. Obtaining of sample size formulation for ALT

To attain the desired mission time of ALT from the targeted BX lifetime on the testing plan, expressed in sections 2.1 and 2.2, the sample size formulation integrated with AF in section 2.3 might be derived.

Each testing time Bernoulli test has one of the pair yields, such as failure or success. The cumulative probability, which pursuits a binomial distribution, shall be defined:

$$L(p) = \sum_{r=0}^c \binom{n}{r} p^r \cdot (1-p)^{n-r} \leq \alpha \quad (25)$$

where n is the sample amount and c is the presumed unsuccessful amount.

If chance p is minute and n is big enough, Equation (25), which pursues a Poisson distribution, will be redefined:

$$L(n \cdot p) = \sum_{r=0}^c \frac{1}{r!} (n \cdot p)^r \cdot e^{-(np)} = \sum_{r=0}^c \frac{1}{r!} m^r \cdot e^{-m} \leq \alpha \quad (26)$$

where $m = \text{parameter} = n \cdot p$.

As the p amount is α from Equation (26), parameter m pursuits the chi-square distribution, $\chi_\alpha^2()$. That is,

$$m = n \cdot p \sim \frac{\chi_\alpha^2(2r+2)}{2} \quad (27)$$

The Weibull distribution for system lifetime is extensively employed because it is defined as an expression of the characteristic life, η , and shape parameter, β . Therefore, if the system pursuits the Weibull distribution, the accumulative failure rate, $F(t)$, in Equation (1) is defined as

$$F(t) = 1 - e^{-\left(\frac{t}{\eta}\right)^\beta} \quad (28)$$

where t is the (elapsed) time.

In the occasion of unreliability, $p = F(t)$, and reliability, $1 - p = R(t)$, Equation (28) shall be placed into Equation (25). That is,

$$L(p) = \sum_{r=0}^c \binom{n}{r} \left(1 - e^{-\left(\frac{t}{\eta}\right)^\beta}\right)^r \cdot \left(e^{-\left(\frac{t}{\eta}\right)^\beta}\right)^{n-r} \leq \alpha \quad (29)$$

Because $e^{-\left(\frac{t}{\eta}\right)^\beta} \cong 1 - \left(\frac{t}{\eta}\right)^\beta$, Equation (29) can be closed as follows:

$$L(p) \cong \sum_{r=0}^c \frac{1}{r!} \left(\frac{t}{\eta}\right)^{\beta r} \cdot \left(1 - \left(\frac{t}{\eta}\right)^\beta\right)^{n-r} \leq \alpha \quad (30)$$

As Equations (26) & (30) have a close shape, the characteristic life with a confidence level of 100 (1- α) may be clarified:

$$m = n \cdot p = n \cdot \left(\frac{t}{\eta}\right)^\beta \sim \frac{\chi_{\alpha}^2(2r+2)}{2} \text{ or } \eta_{\alpha}^{\beta} = \frac{2}{\chi_{\alpha}^2(2r+2)} \cdot n \cdot t^{\beta} \quad (31)$$

At BX life, L_B , in Equation (28), testing time, t , becomes h .

$$L_B^{\beta} \cong x \cdot \eta_{\alpha}^{\beta} = x \cdot \frac{2}{\chi_{\alpha}^2(2r+2)} \cdot n \cdot t^{\beta} = x \cdot \frac{2}{\chi_{\alpha}^2(2r+2)} \cdot n \cdot h^{\beta} \geq L_B^{*\beta} \text{ for } x \leq 0.2 \quad (32)$$

where $x = 0.01F(t)$

If Equation (32) is reordered, the sample size expression is found as:

$$n \geq \frac{\chi_{\alpha}^2(2r+2)}{2} \times \frac{1}{x} \times \left(\frac{L_B^*}{h}\right)^{\beta} \quad (33)$$

As the 1st term $\frac{\chi_{\alpha}^2(2r+2)}{2}$ in a 60% confidence level is approximated to $(r+1)$, Equation (33) shall be redefined as:

$$n \geq (r+1) \times \frac{1}{x} \times \left(\frac{L_B^*}{h}\right)^{\beta} \quad (34)$$

As AF in Equation (24) is replaced into the testing time, h , Equation (34) shall be re-defined as:

$$n \geq (r+1) \times \frac{1}{x} \times \left(\frac{L_B^*}{AF \cdot h_a}\right)^{\beta} \quad (35)$$

where Equation (35) will be clarified as $n \sim (\text{failed samples} + 1) \cdot (1/\text{cumulative failure rate}) \cdot ((\text{objective life}/(\text{test time}))^{\beta})$.

Equation (35) shall be affirmed as [1,38]. Namely, for $n \gg r$, the sample size shall be expressed as:

$$n = \frac{\chi_{\alpha}^2(2r+2)}{2m^{\beta} \ln R_L^{-1}} = \frac{\chi_{\alpha}^2(2r+2)}{2} \times \frac{1}{\ln(1-F_L)^{-1}} \times \left(\frac{L_B}{h}\right)^{\beta} \quad (36)$$

wherem $\cong h/L_B$

If the life objective of a system such as ice maker is presumed to have B1 life 10 years, the allocated test shall be computed for the assigned parts under raised circumstances. In performing parametric ALTs, the structural imperfections of a mechanical product will be found and altered to obtain the intended system life.

2.5. Case Investigation—Magnifying Lifetime of an Ice-Maker Incorporating Auger Motor with a Bearing in a Household Refrigerator

Because customers want to be the convenience of (cubed or crushed) ice being dispensed from a domestic refrigerator, an ice-making system was designed in a refrigerator. As a consumer utilizes a cup to apply force on the dispenser lever, (crushed or cubed) ice is distributed. The major parts are composed of an auger motor, incorporating a geared system and bearing, helix support, helix upper dispenser, etc. They are required to have high-strength fatigue because of the repetitive stresses under the consumer's operation environment (Figure 7).

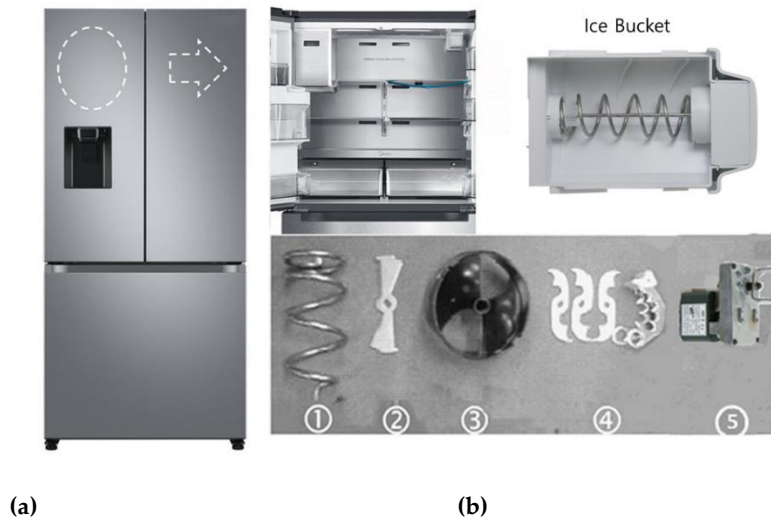


Figure 7. A household refrigerator equipped with ice-maker. (a) Household refrigerator; (b) parts of a domestic ice-maker: helix support ①, blade dispenser ②, helix upper dispenser ③, blade ④, and auger motor ⑤.

In ice-making, the parts undergo repeated mechanical loads and need to be strong enough to not fracture due to fatigue before the expected life. A household refrigerator in the United States is equipped to harvest ice at a rate of 10 cubes per usage and 200 cubes per day. Ice harvest may also be affected by individual end-user usage patterns, such as ice usage, (tap) water pressure, notch positions in refrigerators, and the cycles of doors opening. When set to the crushed mode, the ice-maker is repetitively subjected to (impact) loads in crushing ice. In the market, icemakers, including auger motors, were unsuccessful under unidentified consumer usage in a refrigerator. Field statistics also manifested that the ice makers returned from the market had structural defects such as material problems (high carbon alloy steel with 1.30-1.65% chromium) under the typical freezing temperatures (below -20°C) found in the refrigerator. For the customer, the ice-maker system experienced a sudden failure and no longer functioned. Engineer was required to discover the basic causes by failure analysis (or laboratory test) and then modify the ice maker (Figure 8).



Figure 8. Failed auger motor in the market.

By utilizing failure analysis (and laboratory tests) for failed market parts, under typical freezer temperatures (below -20°C), a crack began in the outer ring of the bearing and propagated to the end. To work it for its expected life, company should redesign the failed product such as bearing fractures in an auger motor. Namely, if there are structural defects — improper bearing material in the auger motor — where repeated loads are exerted in the freezer section, it will fail in its expected life. To reproduce the troublesome

(a)

(b)

Because $fM_1 = fM_2 = fM_3 = \omega$ and $i = fE_1 = fE_2 = fE_3 = i_a$ from Equations (27) and (28),

$$eE_2 = e_a - R_a \times fE_3 \quad (32)$$

$$fE_2 = fE_3 = i_a \quad (33)$$

If Equations (32) and (33) are substituted into Equation (25), then

$$di_a/dt = 1/L_a \times (e_a - R_a \times i_a) \quad (34)$$

From Equations (29)–(31), we can attain

$$eM_2 = [(K_a \times i) - T_L] - B \times fM_3 \quad (35)$$

$$i = i_a \quad (36)$$

$$fM_3 = fM_2 = \omega \quad (37)$$

If Equations (35)–(37) are substituted into (26), then

$$d\omega/dt = 1/J \times [(K_a \times i) - T_L] - B \times \omega \quad (38)$$

From Equations (34) and (38), the state equations can be attained as follows:

$$\begin{bmatrix} di_a/dt \\ d\omega/dt \end{bmatrix} = \begin{bmatrix} -R_a/L_a & 0 \\ mk_a & -B/J \end{bmatrix} \begin{bmatrix} i_a \\ \omega \end{bmatrix} + \begin{bmatrix} 1/L_a \\ 0 \end{bmatrix} e_a + \begin{bmatrix} 1 \\ -1/J \end{bmatrix} T_L \quad (39)$$

As the differential equation in Equation (39) find the integral, the output harvested by the ice-maker is obtained as follows:

$$y_p = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} i_a \\ \omega \end{bmatrix} \quad (40)$$

From Equation (39), the lifetime of ice-maker depends on the required torque to harvest the crushed ice. By altering the torque, the ALT can be performed. The life-stress prototype in Equation (19) shall be adjusted as

$$TF = A(S)^{-n} = AT_L^{-\lambda} = A(F_c \times R)^{-\lambda} = B(F_c)^{-\lambda} \quad (41)$$

where A and B are constants

Therefore, the AF in Equations (20) & (21) shall be defined as

$$AF = \left(\frac{S_1}{S_0}\right)^n = \left(\frac{T_1}{T_0}\right)^\lambda = \left(\frac{F_1 \times R}{F_0 \times R}\right)^\lambda = \left(\frac{F_1}{F_0}\right)^\lambda \quad (42)$$

ALT from Equation (22) can be carried out till the mission time which satisfies the life objective—B1 life 10 years—are attained.

The surrounding situations of an ice-maker in a household refrigerator can change from almost -15 to -30 °C with a relative humidity altering from 0% to 20%. Relying on end-user use circumstances, an ice-maker is expected to operate from three to eighteen cycles per day. Under the largest utilization for ten years, 65,700 life cycles may occur.

To settle the stress amount for ALT, established on the permitted utilization span of the Auger motor manufacturer in bench-marked statistics, achieved from different main companies, the step-stress lifetime testing was applied, which shall judge the life under a constant usage circumstance for some accelerated loads, such as 0.8 kN-cm, 1.0 kN-cm, and 1.47 kN-cm [39]. As the different stress quantity was changed because the common torque is 0.69 kN-cm, the failure time of an auger motor at specific stress quantities might be noticed.

Engineering statistics from the company of auger motor showed that the common torque was 0.69 kN-cm and the maximum torque was 1.47 kN-cm. If the cumulative damage factor, λ , is 2, AF in Equation (42) was almost five.

For lifetime target – a B1 life 10 years, the number of mission time for ten components (attained by employing Equation (22)) was 42,000 cycles if the shape parameter was supposed to be 2.0. The ALT was set to assure a life objective—B1 life 10 years—if it shall be unsuccessful less than once for 42,000 cycles. Figure 10 manifests the test equipment of an

ALT for reproducing the failed ice-maker, involving the auger motor in the field. Figure 11 shows the duty cycles applied by the crushing torque T_L .

The evaluated life L_B in every ALT stage is expressed as

$$L_B^\beta \cong x \cdot \frac{n \cdot (h_a \cdot AF)^\beta}{r + 1} \quad (43)$$

where h_a is the real testing time

Let $x = \lambda \cdot L_B$. The approximated failure rate λ of the selected parts shall be expressed as

$$\lambda \cong \frac{1}{L_B} \cdot (r + 1) \cdot \frac{L_B^\beta}{n \cdot (h_a \cdot AF)^\beta} \quad (44)$$

In every ALT stage, by measuring the approximated L_B life and failure rate λ , the reliability of the design for a system operated by machinery can be secured.

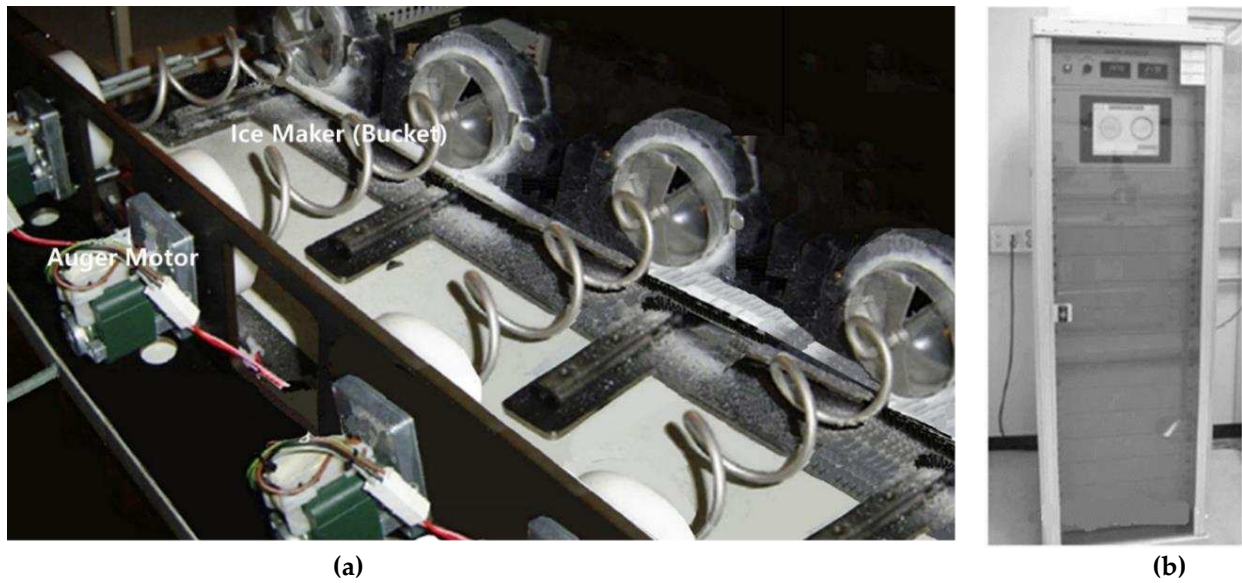


Figure 9. ALT system; (a) Equipment, (b) Controller

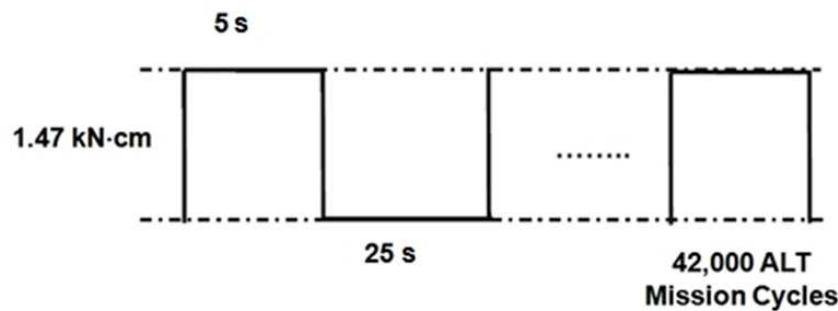


Figure 11. Duty cycles applied by ice-squashing torque T_L exerted in the band clasper.

3. Results and Discussion

From Equation (41), the life of an ice-maker relies on the exerted impact torque T_L . To rapidly identify the failure time of an ice maker, the torque was enhanced from Equation (42). Putting a scale of stress quantity due to applied load through the step-stress life testing, the failure cycles was investigated at the successive stress levels: 0.8 kN-cm, 1.0 kN-cm, and 1.47 kN-cm (torque for ALT). For 0.8 kN-cm, the ice maker stopped at approximately 11,000 cycles. For 1.0 kN-cm, the ice-maker stopped at approximately 9,000 cycles and 13,000 cycles. And, for 1.47 kN-cm, the ice-maker stopped at approximately 5000

cycles and 8000 cycles. Therefore, the stress level as 1.47 kN-cm was set for ALT as it had a comparatively excellent linearity in the Weibull chart, compared to the dissimilar stress quantities.

In the 1st ALT, the fractured bearings of auger motor were occurred at approximately 6500 cycles and 6900 cycles as failed ice-makers were disassembled. Figure 12 illustrates a photograph compared with the product failed from the field and that from the 1st ALT. By employing scanning electron microscopy (SEM), fractured image was carried out on the outer ring. The fracture surface on the cross-section had an intergranular (IG) crack and fatigue due to repeated impact under severe environmental conditions.

Because failed samples were indistinguishable in shape through ALT, we might reproduce the fractured outer ring of a bearing in the market. There was material design defect—AISI 52100 Alloy Steel with 1.30 – 1.60% chromium, which cannot endure stress due to repetitive torque at the freezing temperatures (-20°C below) in the refrigerator. As the bearing and shaft in an ice maker repetitively struck together, they began to crack and ultimately fractured because the material (AISI 52100 Alloy Steel) was too brittle under these circumstances – repeated impacts and severe cold temperature (-20°C ↓). Figure 13 shows the visual inspection of the parametric ALT outcomes and field data on a Weibull chart. The shape parameter in the 1st ALT which relied on loading circumstances was estimated to be 2.0. For the test, the shape parameter was affirmed to be 4.38 in the Weibull chart (Figure 14).

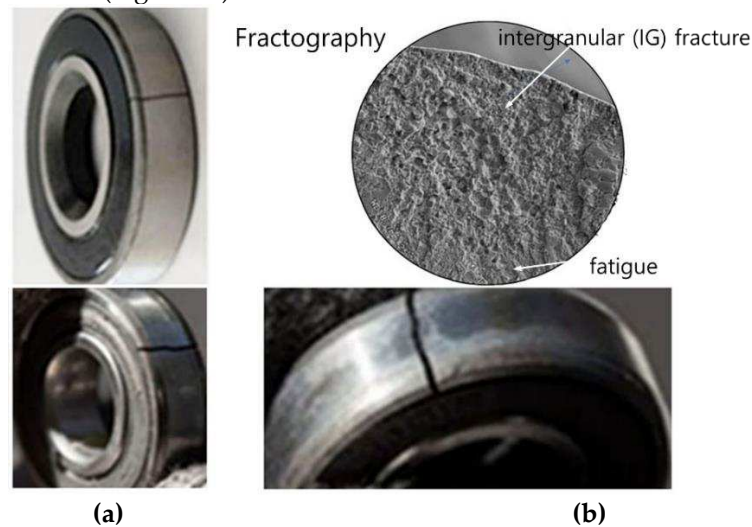


Figure 12. Failed bearing in auger motor from the marketplace and in the 1st ALT; (a) Unsuccessful part from the marketplace, (b) Failed bearing with in 1st ALT – crack origin at outer ring

To withstand the repetitive impact torque, the material of the troublesome bearing in an auger motor was modified from AISI 52100 Alloy Steel with 1.30-1.65% chromium to lubricated sliding bearing with sintered and hardened steel (FLC 4608-110HT).

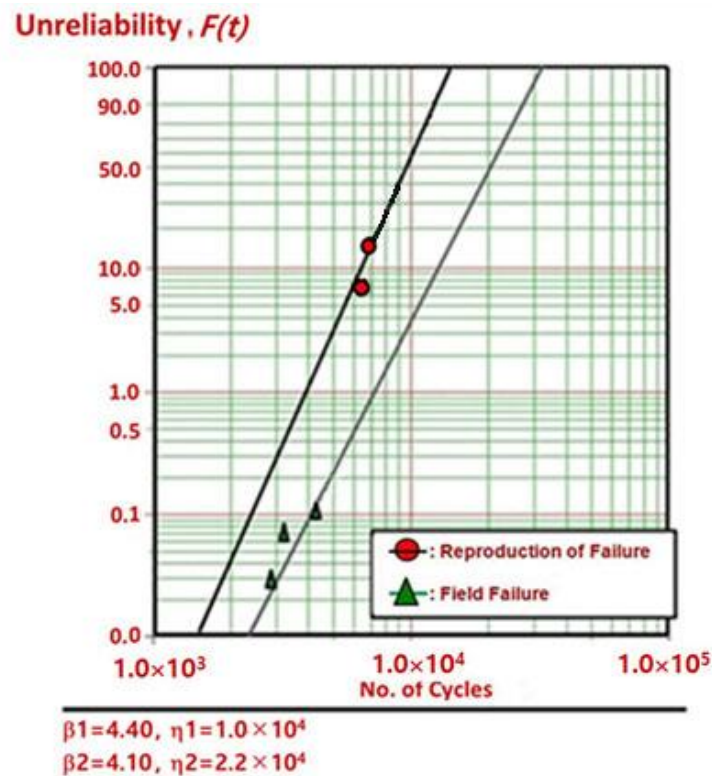


Figure 13. Failed products on the Weibull chart

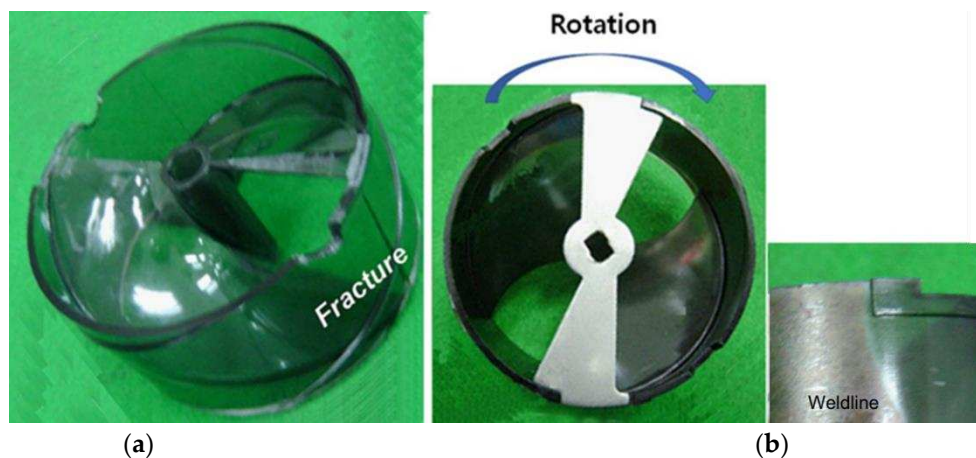


Figure 14. Failed parts in the 2nd ALT: (a) failed parts, (b) the basic cause of fractured parts.

In the 2nd ALT, about 10,000 cycles and 12,000 cycles, the helix upper dispenser (polycarbonates (PC)) fractured in the exposure area of the blade dispenser (Figure 14). To understand the basic cause of the failed system, it was examined. It was discovered that there was a constructional flaw—the weld line between the helix upper dispenser and the blade dispenser—that had countless micro-voids generated in the plastic injection procedure. As the blade dispenser (stainless-steel) stroke the helix upper dispenser (PC) under extreme freezing circumstances, it started to crack and finally fractured at the weld line (Figure 14b). As an alternative, a strengthened rib of the helix was thickened after the plastic injection procedure was altered. Then, finite element analysis (FEA), integrated with ALT, was carried out. As the helix upper dispenser was fastened against the barrier, a simple impact torque (1.47 kN-cm), as shown in Figure 9, was applied. Utilizing materials and processing circumstances close to those of the helix upper dispenser, the constitutive material properties such as PC (helix upper dispenser) were decided. As a

consequence, the stress of the parts by FEA inspection was lessened from 45.0 kPa to 20.0 kPa (Figure 15) [40,41].

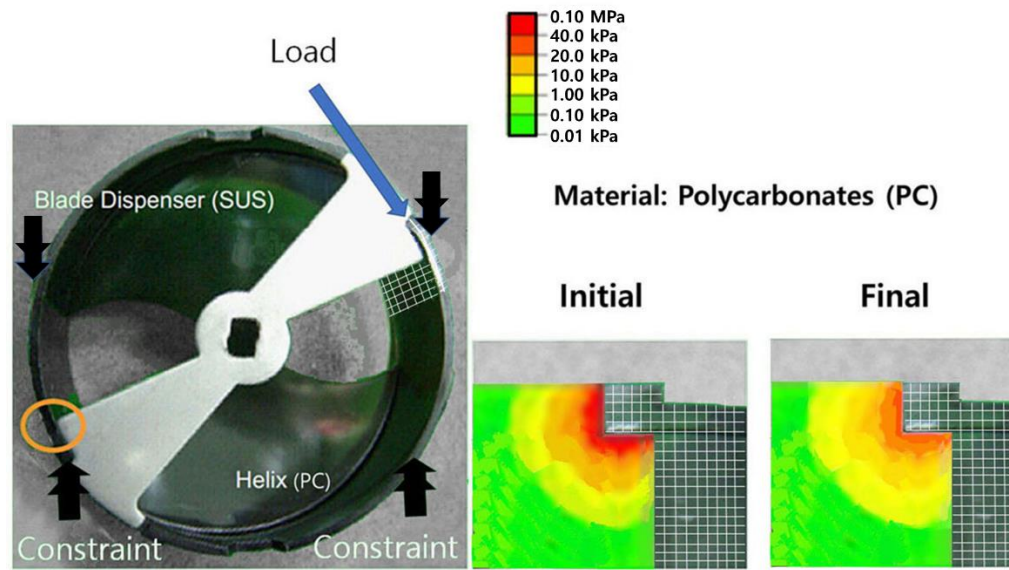


Figure 15. Outcome of stress by employing finite element analysis (FEA)

As the matter of the bearing in an auger motor was modified and the strengthened rib of the helix upper dispenser was thickened, the lifetime of an ice-maker was expanded. However, as 42,000 mission time in the 2nd ALT was not yet achieved, the 3rd ALT was performed to assure the structural alternation of an ice-maker.

There was no problem in the 3rd ALT till 60,000 cycles. Over the route of three ALTs with alternations, the ice-maker was assured to be B1 life 10 years with an accumulated failure rate of 1% from Equations (43) and (44) when the real cycles, $h_a = 60,000\text{ cycles}$, in inserting in lifetime target, $x = 0.01$, sample size, $n = 10$, accelerated factor, $AF = 5.0$, shape parameter, $\beta = 4.38$, and failure number, $r = 0.0$. Table 3 shows a curtailed outcome of the parametric ALTs.

Table 3. ALT results for ice-makers.

Parametric ALT	1st ALT	2nd ALT	3rd ALT
	Draft Design	-	Final Design
Over the route of 42,000 cycles, the ice maker system has no issues	6500 cycles: 1/10 fail 6900 cycles: 1/10 fail (Unsuccessful bearing samples)	10,000 cycles: 1/10 fail 12,000 cycles: 1/10 fail (Unsuccessful helix samples)	42,000 cycles: 10/10 60,000 cycles: 10/10 OK
Structure			
Action plans	C1: from AISI 52100 Alloy Steel to lubricated sliding bearing with sintered and hardened steel (FLC 4608-110HT) C2: thickened reinforced rib on the side of helix		

4. Summary and Conclusions

To strengthen the lifetime of a system operated by machinery such as an ice-maker, incorporating an auger motor with a bearing, utilized in a domestic refrigerator and failed from the field, a reliability structured way was employed, which involved a (generalized)

life-stress prototype by a transport process and a sample size formulation. It covered as follows: (1) the system BX lifetime formed the ALT scheme, (2) ALTs with design modifications, and (3) resolve if the product design obtained the targeted cycles. The ice-maker was examined as a case investigation.

- In the 1st ALT, the auger motor in an ice maker stopped near 6600 cycles and 6900 cycles as exerted for torque -1.47 kN-cm under the freezing temperatures (-20 °C below) in the refrigerator. After disassembling the troublesome samples, we found the fractured outer ring of the bearing in an auger motor. As an action plan, the bearing matter in an ice-maker was altered from AISI 52100 Alloy Steel with 1.30-1.60% chromium to lubricated sliding bearing with sintered and hardened steel (FLC 4608-110HT).
- In the 2nd ALT, the helix (polycarbonates) at 10,000 cycles and 12,000 cycles was fractured because the ice-maker had insufficient fatigue strength for repetitive stress in the freezer department. As an alternative, a reinforced rib of the helix was thickened after plastic injection procedure was modified.
- In the 3rd ALT, no problems were found. The ice-maker, involving an auger motor, will satisfy the life objective—B1 life 10 years. Inspecting controversial field parts and carrying out ALTs with modifications could grow the life of an ice maker, involving an auger motor with a bearing.
- By comprehending the design problems of products failed from the market, we might carry out parametric ALTs with system modifications. As reproducing the market failures, the design defects could be recognized and alter them. Ultimately, we approximated whether the system reached the life targets. In the process, the (generalized) time-to-failure type and sample size also were utilized.

This structured method has been relevant to different products operated by machinery. Engineer is required to grasp why multimodule system fails in their life. Namely, if there are structural defects where it is subjected to repeated loads in its working, the system shall be unsuccessful during its expected life. Designer might be required to know the (dynamic) loads of a mechanical product so that testing shall be carried out till the mission time. Therefore, this ALT will be deployed to pinpoint the structural imperfection and adjust it.

Author Contributions: S.W. carried out the idea development, method, investigation, and experiment and wrote the manuscript. D.L.O. edited the original documents and provided comments on the methods. Y.M.H. and G.M. altered the draft. All authors have understood and consented to the ready and published version of the manuscript.

Funding: This study received without outside funding.

Institutional Review Board Statement: Not relevant.

Informed Consent Statement: Not relevant.

Data Availability Statement: The data utilized in this study may be acquired on demand from the corresponding author.

Conflicts of Interest: The authors proclaim no conflict of interest.

Abbreviations

BX	Time which is a cumulated failure rate of $X\%$
E_a	Activation energy, eV
e	Effort
e_b	Counter electromotive force
e_f	Field voltage, V
f	Flow
F_c	Ice crushing force, kN
$F(t)$	Unreliability
h	Testing time

h^*	Nondimensional testing time, $h^* = h/L_B \geq 1$
i_f	Field current, A
J	Momentum of inertia, kg m ²
k	Boltzmann's numerical quantity, 8.62×10^{-5} eV/deg
L_B	Objective BX lifetime and $x = 0.01 X$, on the conditions that $x \leq 0.2$
m	Gear ratio
MGY	Gyrator in causal forms for basic 2-ports and 3-ports
n	Sample number
Q	Entire number of dopants per unit area
R	Ratio for minimum stress to greatest stress in stress cycle, $\sigma_{\min}/\sigma_{\max}$
r	Failed numbers
r	Coefficient of gyrator
S	Stress
T	Temperature, K
t_i	Test time for each sample
TF	Time to failure
T_L	Ice-crushing torque in bucket, kN cm
X	Cumulated failure rate, %
x	$x = 0.01 X$, on condition that $x \leq 0.2$.
Greek symbols	
ξ	Electrical field exerted
η	Characteristic life
λ	Cumulative damage quantity in Palmgren–Miner's rule
χ^2	Chi-square distribution
α	Confidence level
ω	Angular velocity in ice bucket, rad/s
Superscripts	
β	Shape parameter on the Weibull chart
n	Stress dependence, $n = - \left[\frac{\partial \ln(T_f)}{\partial \ln(S)} \right]_T$
Subscripts	
0	Usual stress conditions
1	Accelerated stress conditions

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