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Swapnil Sarjerao Jagtap*, Peter R. N. Childs, Marc E.J. Stettler

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Brief Report

Data in brief: Comparative Life Cycle Evaluation of Alternative Fuels for a Futuristic Subsonic Long-Range Aircraft

Swapnil S. Jagtap 1,*, Peter R.N. Childs 2 and Marc E.J. Stettler 1

- Centre for Transport Studies, Department of Civil and Environmental Engineering, Imperial College London, London SW7 2AZ, United Kingdom
- ² Energy Futures Lab, Imperial College London, London SW7 2AZ, United Kingdom
- * Correspondence: author: swapniljagtap111@gmail.com

Abstract: Liquid hydrogen (LH₂) and 100% synthetic paraffinic kerosene (SPK), or sustainable aviation fuel (SAF), offer promising low-emission alternatives to conventional Jet-A fuel for long-distance flights—assuming they are produced through processes that support net-zero well-to-wake (WTWa) emissions. This study assesses the WTWa performance, including non-CO₂ effects, of a blended wing body aircraft designed to carry 300 passengers over a range of 13,890 km, using either LH2 or 100% SPK. The analysis quantifies emissions during the operational phase and evaluates fuel production impacts using the GREET model. Results from over 100 fuel production pathways show that LH2 can deliver net-zero or even negative WTWa CO₂-equivalent emissions when derived from biomass or produced via integrated fermentation combined with carbon capture and storage. Non-CO₂ emissions are found to be a major contributor to overall climate impacts. Using miscanthus as a feedstock, 100% SPK reduces WTWa CO₂-equivalent emissions by 70–85% relative to Jet-A. A high-level supply assessment suggests that by 2050, production of SAF and hydrogen could satisfy the energy needs of long-haul aviation, assuming a 4% annual increase in air traffic and widespread adoption of these alternative fuels. These findings offer valuable direction for future research, cost assessments, and policy-making aimed at enabling sustainable long-haul air travel.

Keywords: Life cycle assessment; decarbonising long-range aviation; climate neutral aviation; Hydrogen aircraft; alternative aviation fuel; sustainable aviation fuel

1. Literature Review of Life Cycle Assessment of Alternative Aviation Fuels

The climate impact of aircraft comprises CO_2 and non- CO_2 effects (soot, aerosol, sulphate, water vapour, oxides of nitrogen [NO₈], and contrails and cirrus clouds) [1–3]. Presently, aviation contributes 3.5% to the overall anthropogenic radiative forcing [1,2], of which approximately two-thirds is attributable to non- CO_2 effects [3]. The industry foresees a doubling in air travel demand over the next two decades (2024 - 2043) even considering the effects of the pandemic [4], which is expected to substantially increase its climate impact. It is projected that advancements in aircraft technology and the adoption of low-carbon fuels could collectively address 80% of the measures necessary for achieving carbon-neutral growth [5,6].

The total life cycle emissions associated with aircraft fuel performance encompass emissions from both the operational phase of the aircraft and the manufacturing stage of the fuel (from raw material extraction to fuel storage). According to research by Chester and Horvath [7], the greenhouse gas (GHG) emissions during the operational phase of aircraft account for approximately 70% of the total GHG emissions from the Jet-A aircraft. Presently, technological advancements and regulatory efforts in aviation are primarily focused on the use phase or direct operation of aircraft. The Greenhouse Gases, Regulated Emissions, and Energy Use in Technologies (GREET) model,

developed by Argonne National Lab, USA, is one of the many tools for comprehensive life cycle assessment.

Not all alternative fuel pathways are equally energy efficient when considering their embodied emissions. For instance, from a conventional standpoint focusing on direct emissions, liquid hydrogen (LH₂) appears promising for aviation due to its higher energy density and zero-carbon emissions during aircraft operation, compared to conventional Jet-A fuel. However, using the GREET model [8], it is observed that LH₂ derived from coal emits three times more CO₂ emissions over its life cycle compared to Jet-A. According to the aviation industry [9], evaluating the 'sustainability' of aviation fuels should include life cycle assessment, taking into account net emissions throughout the entire life cycle of the fuel.

Decarbonizing long-haul aviation poses significant challenges [6,10]. Based on the authors' previous work [6,11–13], it was observed that decarbonising the long-range aviation sector is challenging, and of the several alternative energy carriers and propulsion technologies examined (including batteries and fuel cells) for aircraft operational performance, LH₂ and 100% synthetic paraffin kerosene (SPK) or sustainable aviation fuel (SAF) are the only two alternatives to Jet-A for a large twin aisle (LTA) aircraft, whether tube-wing or blended wing body (BWB) airframes. Presently, 100% SPK is not permitted for use in the existing aircraft fleet. Approved drop-in fuels for civil aviation use include up to a 50% blend of alcohol-to-jet (ATJ), Fischer-Tropsch (FT), and hydroprocessed renewable or hydro-processed esters and fatty acids (HRJ or HEFA) SPK pathways, as well as a 10% blend of sugar-to-jet (STJ) SPK pathway [6,14–17].

Studies by Proesmans et al. [18,19] examine the climate impacts of aircraft powered by SPK and LH₂, including the impacts of contrails, but the analysis is restricted to the operational phase of the aircraft. The embodied energy/emissions associated with alternative fuel production also need to be considered for estimating greener production pathways. Therefore, a holistic approach needs to be used for evaluating the performance of future aircraft technology and energy vector combinations. This is also supported by review studies [20–22] which establish the need for an integrated methodological framework that should consider life-cycle impacts of aircraft performance towards the goal of sustainable aviation. Therefore, in the present study, a life cycle approach is used for identifying fuel feedstock and/or manufacturing pathway combination(s) that could enable a climateneutral long-range flight of an LTA aircraft powered by LH₂ and 100% SPK (separately). The scope of this work is limited to fuel life cycle assessment of combustion-based long-range LTA aircraft fuelled by LH₂ and 100% SPK (separately), and the following literature review will focus on it.

Different studies conduct life cycle or well-to-wake (WTWa) emissions analysis for SPK and LH₂ fuel for specific aircraft range applications – fossil fuel based SPK fuel [23], bio-jet fuel [23–33,42–45], power-to-liquid (PtL) or electro-fuel [29,31,39–55], and LH₂ [30,31,37–63]. None of these studies [23–43,52–62], [27,28,32–63], examine different types of feedstocks and manufacturing pathway combination(s) that could enable a climate-neutral long-range flight of an LTA aircraft powered by LH₂ and 100% SPK (bio-jet and PtL fuel), and not all of these consider non-CO₂ emissions in their WTWa analysis. The important findings and/or limitations of these studies are concisely reviewed below, and a detailed review of some of these studies is included in Supplementary Information (SI) File SI §1. For details in terms of different manufacturing pathways and processes, properties, operability issues, and other miscellaneous aspects of these alternative fuels, the reader is advised to explore studies by Su-ungkavatin et al. [64], Cabrera et al. [65], Ansell [34], and Braun et al. [33].

Lau et al. [32] review HEFA and ATJ bio-jet fuel and find that the WTWa GHG emissions reduction potential of HEFA (from jatropha and palm oil) and ATJ (from wheat grain and wheat straw) could be 19% – 42% and 20% – 65%, respectively. According to Braun et al. [33], FT SPK fuel from miscanthus (silver grass), forestry and agricultural residues, and municipal solid waste, has the highest potential to decrease WTWa GHG emissions (90% – 100%). In some regions, GHG emissions reduction could be as high as 125%. The median reduction potentials of ATJ pathways are about 60%, while those of HEFA are below 60%. A review study by Ansell [34] examines different studies on biojet fuel and hydrogen produced from steam methane reforming (SMR) and electrolysis (using

renewable electricity) which do not consider climate impacts of non-CO₂ emissions in their life cycle assessment . Ansell finds that bio-jet fuel and renewable hydrogen has the potential to decrease the WTWa CO₂ emissions by 68% and ~80%, respectively, with a fully renewable grid. The above studies [32–34] do not consider the non-CO₂ emissions in their analysis. Afonso et al. [35] and Song et al. [36] observe that bio-jet fuel can reduce WTWa GHG emissions by up-to 80% (with non-CO₂ emissions) and 41–89% (without non-CO₂ emissions), compared to Jet-A. The reduction in WTWa GHG emissions depends on the feedstock and manufacturing pathways. The above studies consider limited/selective feedstocks for producing alternative fuels.

A study by Kolosz et al. [23] is limited to blended/drop-in SPK fuel, where a comparison of WTWa performance metrics is conducted for: fossil fuel based SPK fuel (from coal, oil sands, oil shale, natural gas, etc.), and bio-jet fuel (first, second, and third generation). Studies by Wei et al. [24], Pavlenko et al. [25] (also reviewed in [37]), and De Jong et al. [26] conduct WTWa emission of bio-jet fuel only for few selected biomass feedstocks. Similarly, studies [27], [28] are focussed only on bio-jet fuel and they explore only the feedstocks considered in CORSIA database (Carbon Offsetting and Reduction Scheme for International Aviation). A study by Van Der Sman et al. [29] reviews WTWa emission of SPK fuel (both bio-jet fuel and PtL) but this analysis is restricted to EU region. Similarly, Saad et al. [38] observe that PtL and bio-jet fuel could reduce ~50% WTWa emission, but the study is limited to Switzerland. All of the above studies [23–29,38] do not consider the climate impact of non-CO₂ emissions (from the aircraft's operational phase) in their WTWa analysis.

Grim et al. [48] observe that PtL could decrease WTWa GHG emissions by 66% – 94% (without non-CO₂ emissions). According to Sacchi et al. [49], PtL fuel from direct air capture and carbon storage could reduce WTWa GHG emissions by 65% – 100% (including non-CO₂ emissions). A study by Micheli et al. [51] examines PtL fuel produced from wind power and finds that the WTWa GHG emissions could be reduced by 27.6% – 46.2% (with non-CO₂ effects) and 52.6% – 88.9% (without non-CO₂ effects), but this study is limited to Germany's energy landscape. Similar observations are made by Papantoni et al. [50], where the authors find that inclusive of the non-CO₂ emissions, PtL from solar and wind energy could reduce WTWa GHG emissions by 32% and 42%, respectively. According to Klenner et al. [52], using wind power, the WTWa GHG emission reduction potential (including non-CO₂ emissions) of PtL fuel is 48% and of LH₂ fuel is 44% for short flights (<200 km), and for longer flights, the average WTWa GHG emission reduction potential increases to 52% (for PtL) and 54% (for LH₂). These findings are limited to Norway's energy mix. Studies by VanLandingham [53] and VanLandingham et al. [54], examine the performance of PtL and LH2 (both fuels are produced from renewable power) in a single aisle aircraft (Boeing 737) and find that the WTWa GHG emissions (including non-CO₂ emissions) could be reduced by 43% and 61%, respectively. Similarly, a study by Prashanth et al. [55] finds that the WTWa GHG emission reduction potential (including non-CO₂ emissions) of LH₂ could be 91% (from solar energy) to 98% (from wind energy) and PtL 84% (from solar energy) to 93% (from wind energy). Studies [46,47] are limited to WTWa emission of PtL or electro-fuel in Germany's energy landscape having ~100% WTWa GHG reduction potential in the future energy mix. The analysis in the above studies [46–55] are limited to a specific energy landscape and/or limited feedstocks are explored to produce the fuel under consideration. The life cycle GHG performance of PtL is dependent on the sourcing of CO₂ (direct air capture or point sourcing), and of PtL and LH₂ is dependent on the type of electricity mix.

Delbecq et al. [40] evaluate the WTWa performance of bio-jet fuel, PtL, and hydrogen with different scenarios at aviation system level for a small aircraft. The individual potential of each of the fuels in terms of WTWa GHG reduction is not known. Additionally, the sourcing of feedstock for bio-jet fuel is not known, and PtL and hydrogen are sourced only from renewable energy. Fantuzzi et al. [41] examine different alternative aviation fuels but consider limited feedstock/pathways for bio-jet fuel (HEFA and ATJ), PtL, and hydrogen (SMR and electrolysis using renewable electricity) and these offer up-to 70% savings in WTWa GHG (without non-CO₂ emissions). Additionally, the study is limited to the UK energy landscape. Another study by Dray et al. [42] evaluates different alternative aviation fuels and their WTWa effects including non-CO₂ emissions. The authors examine bio-jet fuel

(ATJ, HEFA, and FT), PtL (using direct air capture), and hydrogen (electrolysis using renewable power). They observe that considering the ongoing efficiency improvements and efforts to avoid contrails, but excluding offsets, the energy transition could cut WTWa GHG emissions by 89-94% compared to 2019 levels. This reduction would be achieved while considering 2 – 3 times increase in demand by 2050. Despite being a detailed analysis with 2050 forecast, the study examines limited/selective feedstocks for each of the fuel examined. Quante et al. [43] examine alternative aviation fuels and observe that FT SPK, PtL, HEFA SPK, ATJ SPK, STJ SPK, and hydrogen could reduce WTWa GHG emissions (without non-CO₂ emissions) by 85%, 100%, 54%, 62%, 61%, and 80%, respectively. However, the specific feedstocks used for producing bio-jet fuels are unknown. Another study by Penke et al. [39] observe that PtL, bio-jet fuel (HEFA) from soy oil, and renewable hydrogen could reduce WTWa GHG emission (without non-CO₂ emissions) by 77%, 30%, and 95%, respectively. Similarly, Kossarev et al. [44] (continuing their previous work [45]) restrict their study to renewable hydrogen, algae-based HEFA bio-jet fuel, and hydrogenated vegetable oil, which have WTWa GHG emission reduction potential (including non-CO₂ emissions) of 59.5%, 35.8%, and 112%, respectively. The above studies [39,41-45] focus on a particular energy landscape and/or explore limited feedstocks for producing the fuel in consideration.

Studies [34,37,41,43,57-62,66,67] on hydrogen as an aviation fuel do not consider non-CO2 emissions in their WTWa analysis which are important, and studies [56,61,62,66-69] limit their selection of feedstock/pathway for fuel production only to renewable power. The aircraft use-phase energy consumption and/or emissions have a significant impact on the WTWa performance. Studies [34,37,39,43–62] are focussed on the life cycle effects of hydrogen use in aircraft, it is either not clear from the information supplied or these do not consider the poor volumetric energy density characteristic of LH2 which penalises aircraft energy performance, which are their limitations. A study by Koroneos et al. [56] that examines an A320 (or single aisle) type of aircraft though considers realistic effects of LH₂ on aircraft design, is less recent (from early 2000s) and considers fewer ways of LH₂ production. Another study by Mukhopadhaya et al. [61] examines PtL, and LH₂ from electrolysis (renewable energy) and observes that ~100% reduction in WTWa GHG emission could be achieved (without non-CO₂ emissions). Tveitan [66] examines green hydrogen which has a 58% WTWa GHG emission (excluding non-CO₂ emissions) reduction potential. Chan et al. [67] evaluate green hydrogen and bio-jet fuel (feedstock sourcing unknown), and find that bio-jet fuel and hydrogen have the best potential and can achieve up to 88% WTWa GHG emissions reduction. It is also unclear, or the above studies do not consider the non-CO₂ emissions in their WTWa analysis.

Studies by Miller [30] and Miller et al. [63] include more number of feedstocks and pathways than the above studies, as consideration is given to tens of feedstocks and/or pathways for LH₂ and bio-jet fuel along with the effect of contrails cirrus in WTWa analysis. However, PtL or electro-fuels and STJ pathway are omitted, and the WTWa results are limited only to a smaller aircraft with shorter range compared to LTA aircraft which is the focus of this work. Lastly, the FlyZero report [31] accounts the performance penalty due to cryogenic tank installation and non-CO₂ emissions in their WTWa analysis of LH₂, PtL and bio-jet fuel SPK for small to mid-size aircraft. However, their analysis is limited to a few feedstocks and/or pathways of manufacturing LH₂, PtL and bio-jet fuel SPK.

Not all of the above studies [23–43,52–63,70–72] consider non-CO₂ emissions in their WTWa analysis and none of these examine different types of feedstocks and manufacturing pathway combination(s) that could enable a climate-neutral long-range flight of an LTA aircraft powered by LH₂ and 100% SPK (bio-jet and PtL fuel). According to the International Air Transport Association (IATA) [73], reaching net-zero CO₂ emissions by 2050 could involve the following contributions: SAF at 65%, new aircraft technology, including electric and hydrogen aircraft, at 13%, infrastructure and operational efficiencies at 3%, and offsets and carbon capture at 19%. Carbon removal is identified as an important strategy to reduce GHG emission. None of the above studies consider fuel production routes that employ carbon capture and storage (CCS), except a study by Fantuzzi et al. [41] which shows that by employing carbon capture and storage for SMR LH₂ production route can reduce the WTWa GHG emissions by 60% (median value). Also, according to the study by Pavlenko et al. [74],

hydrogen is a crucial component in the SPK fuel manufacturing process and thus has a significant impact on the life cycle GHG emissions of SPKs (bio-jet and PtL fuel) and using green hydrogen in SPK production could help reducing the WTWa GHG emissions. None of the studies reviewed above considers the sensitivity of hydrogen production to the WTWa GHG emissions of SPKs. Furthermore, above studies do not provide any estimation of the energy demand in 2050 for the long-haul aviation sector, and whether or not that energy demand can be met with 100% SPK (or SAF) and/or LH₂.

The above limitations in literature motivate this current work. In this work, the primary research effort is to estimate the life cycle or WTWa GHG performance while addressing non-CO₂ emissions, for a long-range LTA aircraft (separately) powered by LH₂ and 100% SPK (bio-jet and PtL fuel) manufactured from over 100 different feedstocks and/or pathways, where in some cases carbon capture and storage is also employed. Additionally, the sensitivity of hydrogen sourcing to SPK production, and of biomass sourcing for some hydrogen production pathways, are investigated in this work. Similarly, this work explores the sensitivity of energy mix (2020 vs 2050) to alternative fuel production. Furthermore, the energy demand and supply (of 100% SPK [or SAF] and LH₂) in future (2050) for long-haul aviation is considered. These are novel contributions of the present work.

In authors' previous studies [12,13], the engine (ultrahigh bypass ratio geared turbofan) and aircraft operational energy performance modelling of a 2030+ (N+2 timeframe) BWB LTA aircraft powered by LH₂ and 100% SPK (separately) are conducted considering penalty due to cryogenic tank installation for LH₂ aircraft modelling. This enables the estimation of GHG emissions in the aircraft use-phase. Knowing the fuel manufacturing emissions, the WTWa or life cycle emissions can be calculated. In this work, over 100 different ways in total (including fuel manufacturing unit that employs carbon capture and storage) for producing LH₂, PtL, and bio-jet fuel SPK are examined (for use in the N+2 BWB aircraft).

Not all bio-jet fuels (i.e. fuel from different production pathways) can be directly used in the present aircraft (called 'drop-in' fuels), because some of fuel properties and chemical contents are not same as that of the conventional jet fuel [75]. Some systems within the aircraft are designed considering the properties of conventional jet fuel. For example: The aromatic content of conventional jet fuel causes rubber seals used in the high-pressure fuel system to swell, thereby preventing fuel leakage during aircraft operation at different altitudes; and synthetic paraffin kerosene (SPK) jet fuel cannot be used in neat form (100%) without modifications to aircraft or without addition of synthetic aromatics/additives [75]. The American Society for Testing and Materials (ASTM) has approved four bio-jet fuel pathways which can be used in aircraft as 'drop-in' fuels [76]. These are:

- 1. Fischer Tropsch (FT) SPK (FT-SPK) with maximum 50% blend and syngas FT with aromatic alkylation (FT-SPK/A) with maximum 50% blend
- 2. Hydro-processed renewable jet fuel (HRJ) or hydro-processed ester and fatty acids (HEFA) (HRJ/HEFA-SPK) with maximum 50% blend
- 3. Bio-chem sugars or hydro-processed fermented sugars to synthetic iso-paraffins with maximum 10% blend (also referred to as sugar-to-jet fuel or STJ), and
- Alcohol-to-jet (ATJ-SPK) fuel with maximum 50% blend (recently revised from 30% blending)
- 5. where the blending is done with the conventional jet fuel. 100% SPK (biomass-based and electrofuel) is not strictly a drop-in fuel as it has not yet been approved.

A study by Wei et al. [24] reviews different bio-jet fuel types produced from different feedstocks. The authors carry out a comparison of holistic life cycle (well-to-wake [WTWa]) greenhouse gas (GHG) emission for the fuel type and feedstock combinations, which is summarised in Table 1 (also includes use-phase GHG emissions [without the effect of contrails]). The GHG life cycle assessment is carried out at an industrial level and the GHG emission comprise of following process components:

- Feedstock cultivation, harvesting, and transportation
- Land use effect
- Production and transportation of ancillary chemicals
- Refining
- Transport, distribution, storage, and the combustion of the fuel

Additionally, the authors carry out a cost comparison of some of the fuel type and feedstock combinations (from Table 1) and this is summarised in Table 2. It can be observed from Tables 1 and 2 i.e., considering both holistic life cycle GHG emission [carbon dioxide (CO₂) equivalent] and fuel cost, that HEFA-camelina oil and HEFA-waste oils and animal fat are the preferred candidates of fuel type and feedstock combinations followed by ATJ-sugarcane (biochemistry) and ATJ-switchgrass (lignocellulose-biochemistry).

Table 1. Comparison of GHG emission for the fuel type and feedstock combinations

Use	e-phase GHG emissions (without the effect o	of contrails) from GREET 2021 [8]
	Fuel	Use-phase GHG (g CO ₂ -eq/MJ)
Conventional jet f	uel (petroleum)	73.2
100% SPK		70.4
	WTWa GHG emissions (inform	ation source [24])
	Fuel	GHG (g CO ₂ -eq/MJ)
Conventional jet f	uel (petroleum)	90
Fuel type	Feedstock	GHG (g CO ₂ -eq/MJ)
FT	Lignocellulose	-1.60 to 18.20
HEFA	Microalgae	27.00 to 38.00
	Camelina oil	3.06 to 53.10
	Waste oils and animal fat	16.80 to 21.40
	Jatropha	33.00 to 40.00
ATJ	Corn grain	47.5 to 117.50
	Sugar cane	-27.00 to 19.70
	Poplar	32.00 to 73.00
	Switchgrass	11.70 to 89.80
STJ	Sugar cane	22.00 to 80.00

Table 2. Cost comparison of fuel type and feedstock combinations (information source [24])

	Fuel	Cost (€/litre)	
Conventional jet f	uel (petroleum)	0.47 (in 2018)	
Fuel type	Feedstock	Cost MJSP (€/litre)	
FT	Lignocellulose	1.24 – 1.51	
HEFA	Microalgae	6.36	
	Camelina oil	0.33 - 0.92	
	Waste oils and animal fat	0.47 - 0.94	
	Soybean oil	0.76 - 0.88	
	Jatropha	1.1 – 1.15	
ATJ	Lignocellulose (thermochemistry)	1.46 - 1.56	
	Sugar cane (biochemistry)	0.73 - 1.61	
	Corn grain (biochemistry)	0.77 - 1.32	

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	Lignocellulose (biochemistry)	0.86 - 2.17
STJ	Lignocellulose	3.61 – 4.86
	Sugar cane	1.43

Another study by Pavlenko et al. [25] examines different alternative jet fuel types produced from different feedstocks. The authors use three different parameters that measure embodied carbon for a given fuel. The first measure is called direct life cycle emissions (i.e., WTWa). This is attributable to upstream phase of fuel manufacturing (feedstock production and fuel conversion), transport, and the use-phase of the fuel. The second measure is indirect land-use change (ILUC) that comprises of indirect GHG emissions considering land-use effects. The ILUC emissions are associated with cropbased feedstocks. The ILUC emissions may also be attributable to by-products, waste, and residues if these are deflected from the present utilisation. Such impacts may be significant, especially if the economic relationships of the feedstocks are closely related with vegetable oils.

Table 3. Comparison of complete life cycle GHG emissions (three measures) and levelized production cost for the fuel type and feedstock combinations (source [25])

					Approximate		
Fuel		Carbon i	Carbon intensity (gCO ₂ -eq./MJ)				
					(€/litre)		
Conventional jet fuel				87	0.4		
		Direct life cycle	ILUC		Approximate		
Fuel	Ess data di	emissions		Carbon intensity	levelized costs		
type	Feedstock	(WTWa)	emissions		of production		
			(gCO ₂ -eq./MJ)		(€/litre)		
FT	Agricultural residue	6.3	_	6.3	1.75		
	Municipal solid waste	14.8		14.8	1.34		
	(MSW)	14.0	_	14.0	1.34		
	Power-to-liquids (solar)	1.0	12.5*	13.5	2.35		
	Energy crop	11.7	- 12.0	- 0.3	1.87		
HEFA	Palm oil	30.8 to 36.5	231.0	216.8 to 267.5	1.01		
	Palm fatty acid distillate	10.4	212.0	222.4	0.00		
	(PFAD)	19.4	213.0	232.4	0.98		
	Soy oil	27.9 to 34.9	150.0	177.8 to 184.9	1.09		
	Used cooking oil (UCO)	19.4	_	19.4	0.88		
ATJ	Agricultural residue	14.9	_	14.9	2.33		
	Corn grain	65.0	14.0	79.0	1.67		
	Sugarcane	48.1	17.0	65.1	1.65		
	Energy crop	20.3	_	20.3	2.48		
STJ	Molasses	47.0	_	47.0	3.85		

^{*} emissions attributable to infrastructure required for new renewable electricity generation

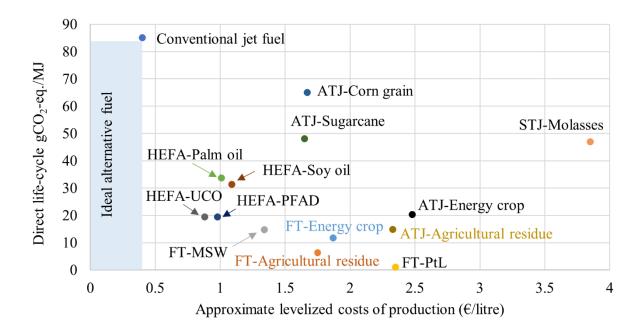


Figure 1. Comparison of feedstock and fuel type considering the levelized production cost and (WTWa) direct life cycle emissions (data source [25]).

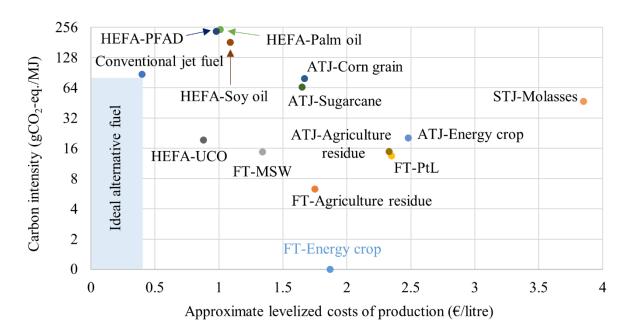


Figure 2. Comparison of feedstock and fuel type considering the levelized production cost and carbon intensity (data source [25]).

The third measure is called carbon intensity, which is a sum of the first and the second measure i.e. sum of direct life cycle (WTWa) emissions and ILUC emissions [25]. Additionally, the authors carry out a cost comparison of some of the fuel type and feedstock combinations, where they compare the levelized cost of the production process. The levelized cost comprises of capital, feedstock, and operating cost. The comparison of complete life cycle GHG emissions (three measures) and levelized production cost for the fuel type and feedstock combinations is summarised in Table 3.

The information in Table 3 is important, particularly considering the first and third measure. This information is helpful for analysing the 'sustainability' aspect of bio-jet fuels with the limited feedstocks evaluated in the study by Pavlenko et al. [25]. The levelized production cost and the first measure i.e., direct life cycle emissions for feedstock and fuel type is analysed through Figure 1.

Similarly, the levelized production cost and the third measure i.e., carbon intensity for feedstock and fuel type is analysed through Figure 2. Figure 2 absorbs the unintended impacts of crop-based feedstocks i.e., ILUC emissions. Typically, ILUC emissions are not calculated in all life cycle studies.

It can be observed from Figure 1 and Figure 2 that none of the feedstock and fuel type combination fall in the 'ideal alternative fuel zone' (or sustainable alternative fuel zone) i.e., the combination that has lower GHG emission and cost compared to the conventional jet fuel. Referring to Figure 1, HEFA-UCO and HEFA-PFAD (food crop) are the preferred candidates of fuel type and feedstock combinations followed by HEFA-palm oil (food crop) and HEFA-soy oil (food crop). After considering the land-use impacts of the crop-based feedstocks i.e., Figure 2, particularly for food-crops, HEFA-PFAD (food crop), HEFA-palm oil (food crop) and HEFA-soy oil (food crop) are no longer preferred candidates. Referring to Figure 2, HEFA-UCO and FT-MSW are the top two preferred candidates. Figures 1 and 2, especially the latter convey the importance of a holistic evaluation. Though 'direct life cycle emission' or WTWa is a holistic measure, it does not capture the unintended impacts of certain feedstocks for fuel production, especially feedstocks that compete with food-crops.

A study by Schmidt et al. [46] demonstrates the development of a relatively novel fuel called 'power-to-liquid' (PtL) jet fuel. Electricity produced from renewable sources like solar and wind energy is used in the electrolysis of water for hydrogen production. After carbon (CO₂) capture, hydrogen and CO₂ undergo chemical process to form hydrocarbon fuel (PtL). This study provides information on different pathways of producing PtL fuel, and it estimates the life cycle GHG from the production pathways. PtL has significantly lower/near-zero life cycle GHG emission and about 55% lower water consumption compared to conventional jet fuel. It is not currently approved for civil aviation. The life cycle GHGs of PtL from two paths [46,47]; are as follows:

- i. PtL (wind/photovoltaics [PV] in Germany, renewable world embedding) = ~1 g/MJ
- ii. PtL (wind/PV in Germany, today's energy landscape in material sourcing and construction) = 11 to 28 g/MJ

One of the processes involved in fuel refining is the FT process. It is to be noted that 50% blended FT jet fuel is approved by ASTM, and 50% blend of PtL from FT process can be used directly. It has higher fuel productivity per hectare land compared to all bio-jet fuels. Its thermo-physical and fuel handling properties are like conventional jet fuel, which means that PtL can be potentially used as a 'drop-in' fuel. This enables status-quo in aircraft powerplants. Presently, it costs 7.3 – 10 times more than conventional jet fuel. Similar information is revealed in the report by the German Environmental agency [47], where details of PtL fuel production, life cycle GHG, and its current and 2050 production costs are provided. In 2050, the cost of PtL is predicted to be 1.4 – 4.5 times the cost of conventional jet fuel [46,47]. Table 3 and Figure 2 provide the ILUC emissions of PtL. The ILUC emissions are attributable to the infrastructure required for new renewable electricity generation. The direct life cycle GHG emissions for PtL in future renewable energy landscape is 1 g/MJ and after considering the ILUC emission of 12.5 g/MJ results in a carbon intensity of 13.5 g/MJ for PtL. Referring to Table 3, the levelized cost of production of FT-PtL (using future energy landscape) is 5.9 times the present cost of the conventional jet fuel.

A study by De Jong et al. [26] carries out WTWa CO $_2$ equivalent emission performance of several renewable jet fuel pathways and analyses the effect of different co-product allocation process. The WTWa CO $_2$ equivalent values for different renewable jet pathways are summarised in Table 4. The authors find that FT pathway enables the highest reduction of GHG emission (86 – 104%) compared to conventional jet fuel. The authors highlight that renewable jet fuel could significantly reduce aviation related GHG emissions if correct pathways and/or feedstock are used. Additionally, the authors note that the GHG emission performance of renewable jet fuel could be further improved by employing carbon capture and storage, and using sustainable hydrogen sources, in the fuel production.

Table 4. WTWa CO₂ equivalent values for different renewable jet pathways (source [26])

Technology	Feedstock	CO ₂ equivalent (g/MJ)
FT	Forestry residues	6
	Poplar	-6 to 10
	Corn Stover	13
	Willow	0
HEFA/HRJ	Camelina	44 to 47
	Used cooking oil	28
	Jatropha	21 to 55
STJ (10% blend)	Sugarcane	47 to 49
STJ (increased blend level)	Sugarcane	76 to 79
ATJ	Sugarcane	31
	Corn Stover	22 to 35
	Corn	54 to 71
Pyrolysis (in situ)	Forestry residues	22
Pyrolysis (ex situ)	Forestry residues	37 to 41
Hydrothermal Liquefaction (HTL) (in situ)	Forestry residues	18
HTL (ex situ)	Forestry residues	21

Table 5. CORSIA default values based on the core life cycle emissions and indirect land use change emissions (source [27], [28])

Fuel Conversion Process	Region	Fuel Feedstock	Core LCA Value	Indirect land use change LCA Value	Net emissions (gCO ₂ e/MJ)
HEFA	Global	Tallow	22.5	0	22.5
	Global	Used cooking oil	13.9		13.9
	Global	Palm fatty acid	20.7		20.7
		distillate			
	Global	Corn oil (from	17.2		17.2
		dry mill ethanol			
		plant)			
	USA	Soybean oil	40.4	24.5	64.9
	Brazil	Soybean oil	40.4	27	67.4
	EU	Rapeseed oil	47.4	24.1	71.5
	Malaysia	Palm oil – closed	37.4	39.1	76.5
	and	pond			
	Indonesia				

Malays	a Palm oil – open	60	39.1	99.1
ar	d pond			
Indones	a			
Braz	il Brassica carinata	34.4	-20.4	14
	(grown as a			
	secondary crop			
	that avoids other			
	crops			
	displacement)			
US	A Brassica carinata	34.4	-21.4	13
	(grown as a			
	secondary crop			
	that avoids other			
	crops			
	displacement)			
FT Glob	al Agricultural	7.7	0	7.7
	residues			
Glob	al Forestry	8.3		8.3
	residues			
Glob	al Municipal solid	5.2		5.2
	waste (MSW),			
	0% non-biogenic			
	carbon (NBC)			
Glob	al MSW (NBC	NBC*170.5 +		NBC*170.5 + 5.2
	given as a	5.2		
	percentage of			
	the non-biogenic			
	carbon content)			
US	A Poplar (short-	12.2	-5.2	7
	rotation woody			
	crops)			
US	A Miscanthus	10.4	-32.9	-22.5
	(herbaceous			
	energy crops)			
Е		10.4	-22	-11.6
	(herbaceous			
	energy crops)			

12 of 39 USA Switchgrass 10.4 6.6 -3.8 (herbaceous energy crops) Alcohol Brazil 32.8 Sugarcane 24.1 8.7 USA 90.8 (ethanol) to jet Corn grain 65.7 25.1 Global Agricultural 39.7 0 39.7 residues (standalone conversion design) Global Agricultural 24.6 24.6 residues (integrated conversion design) Global 40 40 Forestry residues (standalone conversion design) Global Forestry 24.9 24.9 residues (integrated conversion design) USA Miscanthus -42.6 0.7 43.3 (herbaceous energy crops) standalone conversion design USA Miscanthus 28.3 -42.6 -14.3 (herbaceous energy crops) integrated conversion design

		EU	Miscanthus	43.3	-23.3	20
			(herbaceous			
			energy crops)			
			standalone			
			conversion			
			design			
		EU	Miscanthus	28.3	-23.3	5
			(herbaceous			
			energy crops)			
			integrated			
			conversion			
			design			
		USA	Switchgrass	43.9	-10.7	33.2
			(herbaceous			
			energy crops)			
			standalone			
			conversion			
			design			
		USA	Switchgrass	28.9	-10.7	18.2
			(herbaceous			
			energy crops)			
			integrated			
			conversion			
			design			
Alcohol		Global	Agricultural	29.3	0	29.3
(isobutanol)	to		residues			
jet (ATJ)		Global	Forestry	23.8		23.8
			residues			
		Brazil	Sugarcane	24	7.3	31.3
		USA	Corn grain	55.8	22.1	77.9
		USA	Miscanthus	43.4	-54.1	-10.7
			(herbaceous			
			energy crops)			
		EU	Miscanthus	43.4	-31	12.4
			(herbaceous			
			energy crops)			

	USA	Switchgrass	43.4	-14.5	28.9
		(herbaceous			
		energy crops)			
Synthesised iso-	Brazil	Sugarcane	32.8	11.3	44.1
paraffins	EU	Sugar beet	32.4	20.2	52.6

In 2021, the international civil aviation organization (ICAO) published/updated the default life cycle emissions values for carbon offsetting and reduction scheme for international aviation (CORSIA) eligible fuels [28], based on study [27]. For these CORSIA values, three different parameters are used that measure embodied carbon for a given fuel. The first measure is called direct or core life cycle emissions which is attributable to upstream phase of fuel manufacturing (feedstock production and fuel conversion), transport, and the use-phase of the fuel. The second measure is ILUC that comprises of indirect GHG emissions considering land-use effects. The ILUC emissions are associated with crop-based feedstocks which maybe attributable to by-products, waste, and residues if these are deflected from the present utilisation. Such effects may be important, especially if the economic relationships of the feedstocks are closely related with vegetable oils. The third measure is called carbon intensity or net emissions, which is a sum of the first and the second measure i.e., sum of direct life cycle emissions and ILUC emissions. These default values are calculated based on the core life cycle emissions and ILUC emissions, for selected feedstocks and pathways in different country, region and/or globally, and are summarised in Table 5. The lowest net emissions (WTWa) are observed for FT fuel (-22.5 g/M] CO₂ equivalent) produced from miscanthus in the USA.

A review study by Kolosz et al. [23] compares the WTWa performance metrics of different studies on 'drop in' fuels which include biofuels (first, second, and third generation) and different fossil fuel based jet fuel (from coal, oil sands, oil shale, natural gas, etc.). The review study highlights the uncertainty during fuel production and combustion at high altitudes. The study points out next generation fuels such as oleaginous yeasts, alcohols, and low-carbon fuels – liquid hydrogen (LH₂), liquid natural gas (LNG), and liquid ammonia (LNH₃), and the need to examine these fuels on WTWa basis, in future studies.

A report by Van Der Sman et al. [29] reviews cost of sustainable aviation fuels (biofuels and PtL) and their carbon reduction potential in EU, in addition to reviewing literature on other industry strategies for decarbonising aviation. The authors report that PtL has higher embodied carbon reduction potential as compared to biofuels but has significantly higher cost. For example, jet fuel produced from used cooking oil (HEFA) and hydrothermal liquefaction (separately) will cost 1.9 – 2.7 times higher and PtL will cost 2.2 – 6.4 times higher, compared to Jet-A.

The study by Koroneos et al. [56] (based on the Cryoplane project) conducted a comparative WTWa performance examination of a small aircraft from the A320 family (3,360 km range, 124 passengers) powered by LH₂ and Jet-A (separately) considering effects such as greenhouse, acidification, eutrophication, and winter smog. This analysis is conducted using GEMIS (global emission model for integrated systems) database. The authors examine a smaller aircraft, though it is relevant to this work (large twin aisle aircraft) in terms of high-level objectives, is less recent (early 2000s) and considers fewer ways of producing LH₂. Also, the authors do not consider the effect of contrails cirrus in their WTWa analysis. The authors consider energy source like solar photovoltaics, solar thermal, wind, biomass, hydropower, and natural gas, for hydrogen production. The authors emphasise the significance of producing LH₂ from renewable energy sources such as wind energy and hydropower, especially considering their negligible environmental impacts. However, the authors also point the need to produce LH₂ from these sources cost-effectively while ensuring the fuel demands are met.

A study by Pereira et al. [57] compares LH₂, LNG, and Jet-A on a WTWa basis, where LH₂ is produced from four separate pathways – steam methane reformation, solar PV, wind energy, and hydropower. This comparison is carried out for both small (A320 type) and large aircraft (A340 type).

The authors use a combination of existing models (GREET and GEMIS) for the well-to-pump (WTP) phase or fuel manufacturing phase emissions modelling and existing EMEP/EEA dataset for the pump-to-wake (PTWa) phase or use-phase. The methodology does not consider the poor volumetric energy density characteristics of LH₂ which penalises aircraft energy performance. The aircraft use-phase energy consumption and/or emissions have a significant impact on the WTWa performance. Also, the authors do not consider the effect of contrails cirrus in their WTWa analysis. The authors observe that renewable hydrogen is the less polluting option, particularly from hydropower or wind energy. Additionally, manufacturing LH₂ from renewable energy sources has benefits both in terms of WTWa energy consumption, and environmental and social impacts. The authors point out the need to examine renewable LH₂ production cost, since the market penetration of this fuel depends on its cost effectiveness.

Bicer et al. [58] conducts a WTWa evaluation of a small twin aisle aircraft of conventional tubewing architecture (such as the Boeing 767) with a flight range of 5,600 km, where the aircraft is operated by Jet-A, LH₂, LNG, LNH₃, ethanol, and methanol (separately). It is to be noted that the authors use SimaPro software (with Ecoinvent database), which is a life cycle assessment software, for their analysis. The methodology does not consider the poor volumetric energy density characteristics of LH₂ which penalises aircraft energy performance that has a significant impact on the WTWa performance. Also, the authors do not consider the effect of contrails cirrus in their WTWa analysis. The fuels are examined considering aspects such as land use, global warming potential (GWP), ozone layer depletion, abiotic depletion, and human toxicity. For both LH₂ and LNH₃, the authors evaluate renewable energy sources for fuel manufacturing that include hydropower, geothermal, solar, and wind energy, and for LH₂ the production pathway of underground coal gasification with carbon capture storage is also examined. The authors find that LH₂ when manufactured from geothermal energy could be preferred route than LNH₃ (from geothermal), other alternatives, and Jet-A on WTWa basis in terms of GWP.

Similarly, a study by Ratner et al. [59] provides the WTWa performance of a small aircraft (1,667 km range, 190 passengers) powered by battery and fuel cell (separately), where the electricity and LH₂ are produced from different sources. The authors use Ecoinvent life cycle assessment (LCA) dataset for their analysis, and it is unclear (not explicit) whether the (negative) volumetric effects of LH₂ are considered in the aircraft energy modelling. The methodology is either not clear from the information supplied or do not consider the poor volumetric energy density characteristic of LH₂ which penalises aircraft energy performance. The aircraft use-phase energy consumption and/or emissions have a significant impact on the WTWa performance. Also, the authors do not consider the effect of contrails cirrus in their WTWa analysis. The authors provide comparison of 13 different alternative cases with metrics such as oxidation and eutrophication potential, climate impact, ecotoxicity, and land use, along with their cost. The authors find that electric aircraft powered by electricity produced from wind energy could reduce the aircraft's climate impact, compared to Jet-A, while being the most cost-effective solution of all alternatives and of similar magnitude as that of Jet-A. The analysis is simplistic in nature and is limited to fewer pathways/sources of producing electricity and LH₂, for a small aircraft with short range.

A study by Siddiqui et al. [60] conducts comparative WTWa performance examination of a passenger aircraft (range and passengers not known) powered by LH₂, LNH₃, ethanol, methanol, dimethyl ether, biodiesel, and Jet-A (separately). The fuels are examined considering aspects such as greenhouse effect, ionising radiation potential, terrestrial acidification, freshwater eutrophication, photochemical ozone formation, particulate matter, freshwater ecotoxicity, human carcinogenic toxicity, and land use occupation. For LH₂ and LNH₃, the authors examine renewable energy sources for fuel production such as hydropower, geothermal, solar, and wind energy. This analysis is conducted using SimaPro software using Ecoinvent database. The methodology is either not clear from the information supplied or do not consider the poor volumetric energy density characteristic of LH₂ which penalises aircraft energy performance. The aircraft use-phase energy consumption and/or emissions have a significant impact on the WTWa performance. Also, the authors do not

consider the effect of contrails cirrus in their WTWa analysis. The authors find that LH₂ when produced from geothermal energy, performs better than LNH₃ (from geothermal), other alternatives, and Jet-A, on WTWa basis in terms of greenhouse effect. The authors point out the need of manufacturing LH₂ from renewable energy sources cost-effectively while simultaneously ensuring that the required fuel demand is met.

In a study by Miller [30], a slightly more detailed WTWa analysis of alternative aviation fuels (LH₂ and biofuels) is conducted, compared to above studies. The metrics used for this comparative WTWa analysis are climate impacts and air quality, water consumption in fuel manufacturing stage, and in non-use phase it estimates acidification, eutrophication, respiratory effects, and smog. This study encompasses aircraft and airport emissions (construction and use-phase) and considers a simplistic estimation of effect of contrails toward climate impacts. The methodology of the said study for the WTWa emissions analysis is eclectic as the authors use published models for different segments of the life cycle. For example, the author uses GREET model for the fuel manufacturing phase emissions. The author uses all available LH₂ production pathways in GREET, and for biofuel only FT, HEFA, and ATJ are used. The study by Miller, though is slightly more detailed compared to the above studies as it consider tens of several feedstocks and/or pathways for LH2 and biofuels along with the effect of contrails cirrus in WTWa analysis, it misses out on PtL or electro-fuels and STJ fuel, and the WTWa results are limited only to short or medium range aircraft. The author provides analysis in the use-phase for different aircraft size at 930 km range (500 nmi) and further examines a Boeing 787-800 type for a 6,500 km range (3,500 nmi), for its WTWa performance. The author observes that the WTWa performance of combustion based LH2 aircraft is strongly dependent on the fuel production pathway. LH2 produced from renewable energy sources (wind, hydroelectric, and geothermal) are strong contenders but scalability is a concern. Additionally, the author points out the need for further evaluation of contrail cirrus effect of LH2 aircraft to refine the WTWa estimates and help decision making towards LH₂ powered aviation investments.

A study by Mukhopadhaya et al. [61] evaluates the performance of regional and small range LH₂ aircraft along with cost analysis and embodied carbon. The authors model the LH₂ aircraft while considering its poor volumetric energy density characteristic along with the effect of cryogenic tank gravimetric index, which penalises the energy performance. The authors compare the embodied carbon and cost per revenue passenger km of LH2 (blue and green) and e-kerosene (PtL) with Jet-A for 2035 and 2050 timeframe in EU and USA. However, the authors consider fewer ways of LH2 production and do not consider the effect of contrails cirrus in their WTWa analysis for the fuels considered. The authors find that the cost of powering aircraft with green hydrogen is expected to be more than Jet-A but lesser than e-kerosene or blue hydrogen. Also, if carbon pricing is included then green LH2 could be cost competitive with Jet-A or cost lesser. Additionally, to maximise the climate impact reduction potential for LH₂ aircraft and enable successful penetration of green hydrogen (low embodied carbon) in aviation, there is a need to account the life cycle effects appropriately along with policies. Moreover, for LH2 aircraft to be successful, the above efforts need to be complemented with supportive government policies such as low-carbon fuel standards, alternative fuel mandates to bridge the cost gap with Jet-A, carbon pricing, and/or bolstering fuel efficiency policies, to enable required investments for fostering R&D, design, testing, and establishing infrastructure for LH2 production, distribution, and storage.

The FlyZero report [31] studies different fuels and propulsion type (with technology improvement in 2040 timeframe) and compares them on WTWa basis, for regional short-range to midsize medium range aircraft. In terms of fuels and propulsion type, the study examines LH₂ (fuel cell and combustion based, considering performance penalty due to installation of cryogenic tank), PtL, and biofuel. The study reports the climate impact due to contrail cirrus along with the climate impacts due to oxides of nitrogen (NO_x), H₂O, and CO₂ in the use-phase, and emissions in the fuel manufacturing phase. The study uses a combination of models for the life cycle assessment which includes use of SimaPro software (hydrogen) and CORSIA values (for biofuels). The FlyZero report accounts the performance penalty due to cryogenic tank installation and the impact of contrail cirrus

in their WTWa analysis of LH₂, PtL, and biofuel for small to mid-size aircraft. However, their analysis is limited to a few selected feedstocks and/or pathways of manufacturing LH₂, PtL, and biofuel. LH₂ is assumed to be produced only from renewable electricity and biofuels manufactured using a mix of feedstocks from CORSIA eligible fuels (limited feedstocks only). Overall, the study finds that both combustion-based and fuel-cell powered LH₂ aircraft are preferred candidates as their WTWa CO₂ equivalent emissions are lesser than other alternatives and Jet-A fuel, except for very-short range where electric aircraft would be preferred vehicles.

2. Fuel Manufacturing Process

2.1. Jet-A and 100% SPK

The fuel manufacturing process can be broadly broken down into four steps – raw material extraction, transportation of raw materials to the fuel refining unit, fuel refining, and fuel transportation to the storage facility.

Kerosene or jet fuel is manufactured from crude oil (fossil fuel). The extracted crude oil is transported to the refining plant where the crude oil undergoes refining, distillation, cracking, reforming, and separation [77]. During the production of kerosene several other co-products like petrol, gas oil, diesel oil, and residual oil are formed. After kerosene is manufactured, it is transported to the storage facility and/or airport.

In this work, four SPK production pathways are considered for evaluation – ATJ, STJ, HEFA, and FT. The production process of these four pathways is discussed below:

- ATJ fuel is produced from alcohols that are sourced from sugar containing feedstocks and lignocellulosic biomass by fermentation [77]. In the ATJ production, alcohols are dehydrated, and water is removed to form hydrocarbon chains via oligomerisation. Later these are distilled for fractioning the hydrocarbons in terms of chain lengths or separate products.
- STJ fuel is manufactured from biomass feedstocks that contain sugars [77]. When such feedstocks are fermented, hydrocarbons are formed which are further refined and catalytically improved to form SPK.
- 3. FT fuel is primarily manufactured from syngas which can be sourced from any carbon containing matter [77]. Coal and natural gas are presently used as common feedstocks for producing FT fuel. Syngas is commonly manufactured via gasification of lignocellulosic biomass feedstocks such as energy crops, municipal solid waste, and agricultural residues. The biomass undergoes chemical process at very high pressure and temperature in the presence of air and steam. The need for external electric power for the FT process is low. This is because during the FT conversion process, syngas is partly converted to wax and partly to produce electricity that is used for FT conversion process. The co-products of this process are utilised to make hydrogen that is required for hydrocracking of the wax formed during the process. This is followed by separation of products.
- 4. HEFA SPK is produced from feedstocks that contain renewable oil i.e., oil extracted from plants, waste greases, and animal fats [77]. Renewable oil has properties similar to jet fuel and conversion to SPK is an easy process. Such oils undergo hydrotreatment followed by hydrocracking to produce the required hydrocarbon chain. During this process the oxygen from oil is removed. The products are separated into naphtha, diesel, and jet fuel.

The present fuel readiness level of ATJ, STJ, FT, and HEFA are 7, 6 – 7, 7, and 9 respectively.

2.2. Hydrogen

Natural gas

Hydrogen is produced from natural gas using process known as steam methane reformation (SMR), which is an endothermic process. At present, hydrogen is primarily produced using SMR

process. In SMR, methane reacts with steam at high pressure (3–25 bar pressure) using a catalyst to produce carbon monoxide, hydrogen, and a small amount of carbon dioxide [78].

Coal

Organic matter, for example coal, is converted to hydrogen, carbon dioxide, and CO at temperatures above 700°C without combustion, and this process is called as gasification [79]. Further CO reacts with water to form more hydrogen and carbon dioxide via water-gas shift reaction. Hydrogen is separated from the products using special membranes and/or adsorbers.

Coke oven gas (COG)

COG is formed when coal is heated to 1100° C without the presence of air. COG typically contains ethylene (5%), carbon monoxide (CO) (10%), methane (CH₄) (34%), and H₂ (51%) [80]. COG undergoes steam reforming followed with water-gas shift reaction to produce H₂ and CO₂ [81].

• By-product of natural gas liquid (NGL) steam cracking (SC)

NGLs are hydrocarbons such as ethane, propane, butane, isobutane, and pentane [82]. Steam crackers convert hydrocarbon/NGLs to light olefins using thermal cracking process (converting long-chain hydrocarbon to short-chain hydrocarbons) to produce hydrogen as a by-product [83].

Solar PV

The electricity produced from solar energy using PV system, is used for the electrolysis of water to produce hydrogen [84].

Nuclear thermochemical cracking of water

Sulphur – Iodine process is a common thermo-chemical process used for splitting water to produce hydrogen [85]. The first step is called as Bunsen reaction where iodine and sulphur dioxide react with water at 120 °C to produce hydriodic acid and sulphuric acid. In the second step, sulphuric acid is decomposed to sulphur dioxide at high temperature (around 1,000 °C). In the final step, hydriodic acid is decomposed to produce iodine and hydrogen at above 300 °C.

Nuclear high temperature gas reactor (HTGR)

A nuclear high temperature gas reactor is a graphite moderated and helium cooled reactor [85]. A heat exchange system is used to cool off the helium and heat up water to produce steam. Electricity is produced from the steam using turbines and hot water from the turbine outlet is electrolysed to produce hydrogen, using the electricity produced from turbine.

Biomass

Hydrogen can be produced from biomass using gasification process which is described above. Since growing biomass reduces atmospheric carbon dioxide, the net carbon emissions of this process could be low, especially if carbon capture and storage technology is employed [79].

Integrated fermentation

Hydrogen can be produced from biological routes that use sunlight (photosynthesis and photolysis), fermentative bacteria (dark fermentation i.e. no light), and electrolysis of organic matter via microbial metabolism with small external energy input (microbial electrolysis cell) [86]. The effluent from the dark fermentation route is rich in organic matter and this could be utilised by microbial electrolysis cell for producing additional hydrogen. This process is an integrated pathway that combines microbial electrolysis cell and dark fermentation for hydrogen production.

High temperature electrolysis (HTE) with solid oxide electrolysis cell (SOEC)

HTE uses both electricity and thermal energy for splitting water employing SOEC and hydrogen is produced [87]. In this process, water is heated so that it evaporates which requires a high amount thermal energy. This thermal energy is provided either directly in the form of steam to the electrolyser or via an external heat input. The heat input could come from renewable energy source like solar, wind, nuclear, etc. and the embodied carbon for hydrogen production could be reduced. In this work, two electricity source types are studied – Nuclear HTGR and natural gas combined cycle.

By-product of chlorine manufacturing plant

The electrolysis of a sodium chloride solution (brine) is called as the Chlor-alkali process [88]. Hydrogen gas is one of the co-products of this process along with chlorine and sodium hydroxide.

2. Methodology

Table 6. Engine parameters required to estimate EI_{NOx} during cruise for different fuels (source: [12])

BWB fuel cases	P ₃ (MPa)	<i>T</i> ₃ (K)	$f/_a$	ф	τ (ms)	MW of combustion products
Jet-A	2.11	838	0.0241	0.351	0.854	28.86
100% SPK	2.11	838	0.0236	0.343	0.854	20.00
LH ₂ (case 3)	2.16	843	0.0061	0.206	0.842	27.45

Table 7. Operational conditions and fuel burn for different fuels at various flight segments

	Altitude	20	т				Fuel burn (kg)***		
	(m)*	p _{amb} (Pa)**	<i>T_{amb}</i> (K)**	Speed (m/s)	Mach**	Jet-A	100% SPK	LH2	
Engine start and warm-up	0	101,325	288.15	0	0	847	824	241	
Taxi-out	0	101,325	288.15	10.29 [89]	0.03	844	821	241	
Take-off	0	101,325	288.15	77.17 [90], [91]	0.23	420	409	120	
Climb	5,334	51,652	253.48	149.19*	0.47	584	568	162	
Cruise	10,668	23,842	218.81	248.10*	0.84	59,035	57,727	19,003	
Descent and approach	5,334	51,652	253.48	128.61 [90]	0.44	626	612	216	
Loiter	1,500	84,556	278.40	200.85*	0.60	2,678	2,623	894	
Land, taxi-in and shutdown	0	101,325	288.15	79.74 [90], [92]	0.23	491	481	172	

^{*}known from [12]

^{**}calculated using source [93] for given altitude known from [12]

^{***}known from [13]

Table 8. Emission index of different species and contrails for the three fuels

Cracica	Emiss	Reference		
Species	Jet-A	LH_2	100% SPK	
CO ₂	3,158 [8]	0	3,092 [8]	[8]
H ₂ O	1,240 [94]	8937 [95]	1,370 [96]	[94]–[96]
SO ₂	0.8 [94]	0	0 [8]	[8], [94]
VOC	0.17	0	0.17	[8]
CO	1.49	0	1.51	[8]
PM_{10}	0.20	0	0.02	[8]
PM _{2.5}	0.20	0	0.02	[8]
SO _x (total)	1.40	0	0.00	[8]
ВС	0.06	0	0.00	[8]
OC	0.06	0	0.00	[8]
CH ₄	1.18E-03	0	1.21E-03	[8]
N ₂ O	2.31E-03	0	2.37E-03	[8]
Contrails	65	40	50	[31]

Table 9. GWP of NOx and H2O at different mission segments

	A166. Ja ()	GWP	
	Altitude (m) —	NOx	H ₂ O
Engine start and warm-up	0	-11 [8], [97]	0 [97]
Taxi-out	0	-11 [8], [97]	0 [97]
Take-off	0	-11 [8], [97]	0 [97]
Climb	5,334	9.17 [97]	0 [97]
Cruise	10,668	114 [2]	0.06 [2]
Descent and approach	5,334	9.17 [97]	0 [97]
Loiter	1,500	-11 [8], [97]	0 [97]
Land, taxi-in and shutdown	0	-11 [8], [97]	0 [97]

Table 11 (SI §3) lists the production capacities of different fuels and production feedstocks/pathways per plant. It is to be noted that only shortlisted pathways of LH₂ and 100% SPK from §3.1.3 are listed in this table. The IF manufacturing capacities are not known. In Table 11 (SI §3) the global manufacturing capacity for Jet-A is purposely listed for 2019 (normal operations prior to the pandemic). It can be observed from Table 11 (SI §3) that for both LH₂ and 100% SPK the manufacturing units and capacities and liquefication capacity must be ramped up significantly (multiplication factor of 400 - 3,500 for LH₂ and 600 - 6,300 for 100% SPK, depending on the pathway) if these fuels have to replace Jet-A fuel.

Table 10. GWP of all emissions in the WTP and PTWa phase

	WTP	PTWa
NOx	-11 [8]	Variable (see Table 9)
H ₂ O	0 [97]	Variable (see Table 9)
ВС	900 [8]	1166 [2]
SO ₂	-	-226 [2]
CO ₂	1 [8]	1 [8]
VOC	5 [8]	5 [8]
CO	3 [8]	3 [8]
OC	-69 [8]	-69 [8]
CH ₄	30 [8]	30 [8]
N ₂ O	265 [8]	265 [8]

Table 11. Production capacities of different fuels and production feedstocks/pathways (per plant)

	F				
Fuel	Feedstock/pathway	tonne/day	Source		
H ₂ (gas)	SMR with CS (H-vision initiative)	360 – 480	[98]		
	Solar PV (hypothetical case estimation)	1,233	[98]		
	Renewable source (solar, wind), plant for 2025	650	[99]		
	Nuclear HTGR	230	[100]		
	Biomass	2,000	[101]		
Global hydrogen liquefaction capacity 290 tonnes/day [100]					
100% SPK	Neste/HEFA	933	[25]		
	EU sub-group (HEFA)	1,348	[25]		
	FT	456	[25]		
		187	[25]		
		249	[25]		
	ATJ	477	[25]		
	STJ	127	[25]		
Jet-A (2019,	Petroleum	7,90,165	[102]		
Global)					

3. Results

In §3.1.3 LH₂ is identified as the fuel that has the potential to make long-range flight emit near-zero and/or negative WTWa CO₂ equivalent emissions, where LH₂ is manufactured from Biomass w CS LUSME, Biomass w CS LBIGCC and IF w CS feedstock/pathways. Table 12 provides a summary of other unintended life cycle environmental and social impacts for these three LH₂ production routes in comparison with Jet-A. These three routes are compared with Jet-A for aspects like indirect use of fossil fuels, water consumption, and WTWa VOC, CO, NO_x, PM₁₀, PM_{2.5}, SO_x, BC, and OC emissions. A colour coded ranking system is used for simplicity and this is indicated in Table 12. Similar data

for remaining LH₂ production routes and 100% SPK cases are discussed below for completeness (Figures 3 to 12).

Figure 3 provides the comparison of fossil fuel use for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios. It can be observed that for the three identified LH₂ cases (Biomass w CS LUSME, Biomass w CS LBIGCC, and/or IF w CS), the use of fossil fuels is significantly greater (indicative of poor process efficiency) compared to Jet-A production (difference of order of magnitude).

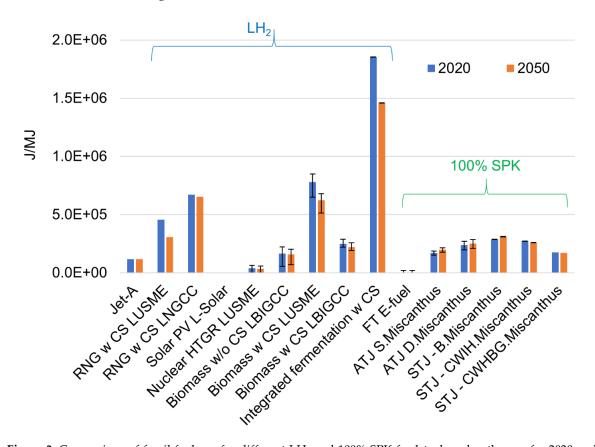


Figure 3. Comparison of fossil fuel use for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

The net WTWa CO₂ equivalent emissions are lower for these three cases due to the employment of carbon sequestration. The greater use of fossil fuels (indirectly) for producing LH₂ from the above three feedstocks/pathways compared to Jet-A, is an unintended natural resource (environmental) impact which none of the studies in literature have accounted. This greater (indirect) use of fossil fuels for the three feedstock/pathways is due to the poor manufacturing (supply-chain) process energy efficiency (from the point of cultivating crops to the point of producing hydrogen from biomass). This needs to be reduced for US based LH₂ production from the above three routes or could be lesser in regions that employ greater renewable energy resources (like European countries) for power production. This effect of greater renewable energy resources in the energy mix can be observed from Figure 3 between 2020 and 2050 US energy mix scenarios. Additionally, the indirect use of fossil fuels can be reduced for the said three routes by improving the energy efficiency of fuel manufacturing and its supply chain. For 100% SPK cases, the fossil fuel use is greater than Jet-A but of the similar order.

Figure 4 shows the comparison of water consumption for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios. It can be observed that for the three identified LH₂ cases (Biomass w CS LUSME, Biomass w CS LBIGCC, and/or IF w CS), the water consumption is significantly greater compared to Jet-A production due to the water required for cultivation of biomass feedstocks/crops. The greater use of water for producing LH₂ from the above

three feedstocks/pathways compared to Jet-A, is an unintended resource (environmental and social) impact which none of the studies in literature have accounted. For biomass based 100% SPK cases, water consumption is greater than Jet-A due to the water required for crop cultivation.

Table 12. Summary of other unintended life cycle environmental and social impacts

Ranking colour scheme						
1	2	3		4		
				LH ₂		
Aspects	Jet-A	Biom	ass w CS	Biomass v	w CS	IF w CS
		LU	JSME	LBIGC	C.C	IF W C3
Fossil fuel use						
Water consumption						
WTWa VOC						
WTWa CO						
WTWa NOx						
WTWa PM ₁₀						
WTWa PM2.5						
WTWa SO _x						
WTWa BC						
WTWa OC						

It can be observed from Table 12 that the three LH₂ production routes perform poorly in terms of (indirect) use of fossil fuels and water consumption, compared to Jet-A. For WTWa CO, NO_x, PM₁₀, PM_{2.5}, BC, and OC emissions, Biomass w CS LBIGCC route performs the best. Jet-A performs best for VOC and SO_x emission (effect of lesser use of fossil fuels). The IF w CS route performs poorly on most of the aspects studied in Table 12, though this route provides a net negative WTWa CO₂ equivalent emissions compared to Jet-A (due to carbon sequestration). Therefore, it is important to examine the unintended environmental and social impacts of a selected route i.e., widening the boundary for analysis than just focusing on climate impacts. This is supported by a review study by Pinheiro Melo et al. [20], where the authors observe that though advanced technologies and alternative fuels can provide solutions for mitigating aviation emissions, there might be new socio-economic and environmental challenges associated with these. Thus, they establish the need for diversifying environmental indicators beyond GHG emissions and the need to consider social and economic aspects.

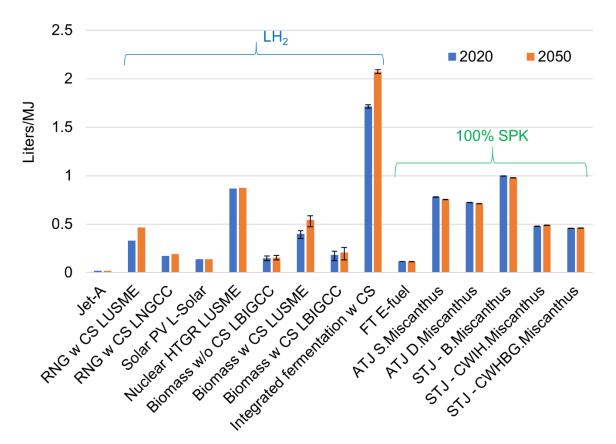


Figure 4. Comparison of water consumption for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

The above is the significance of this section and these aspects are not considered in the studies reviewed in §1. It is important to minimise the (indirect) use of fossil fuels (resource impact) and water. The aspect of higher use of fossil fuels can be addressed by increasing the share of renewable energy sources in the energy mix and increasing the efficiency of fuel manufacturing processes/plants and fuel supply chain. The water consumption aspect can be addressed by using ocean or wastewater and treating it to be used for fuel production. This entails accounting for water treating energy and emissions in WTWa analysis.

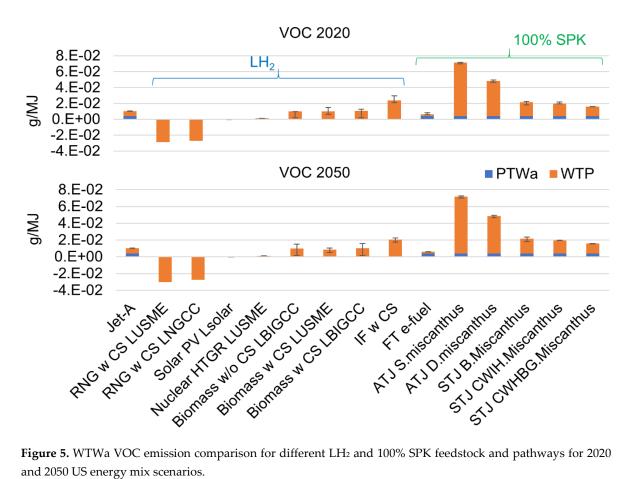
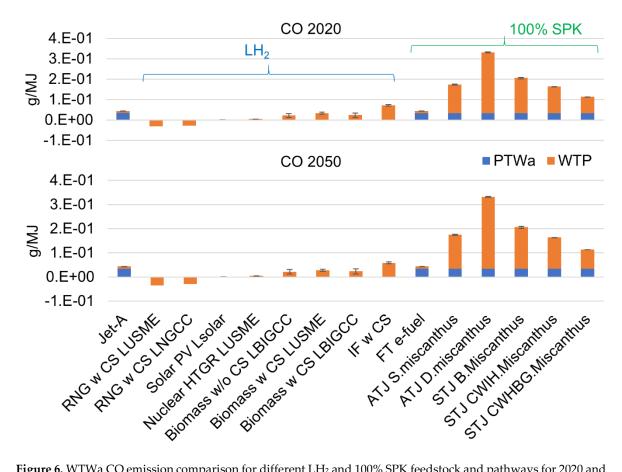


Figure 5. WTWa VOC emission comparison for different LH2 and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.



 $\textbf{Figure 6.} \ \ WTWa\ CO\ emission\ comparison\ for\ different\ LH{\scriptstyle 2}\ and\ 100\%\ SPK\ feedstock\ and\ pathways\ for\ 2020\ and\ solve and\ pathways\ for\ 2020\ and\ pat$ 2050 US energy mix scenarios.

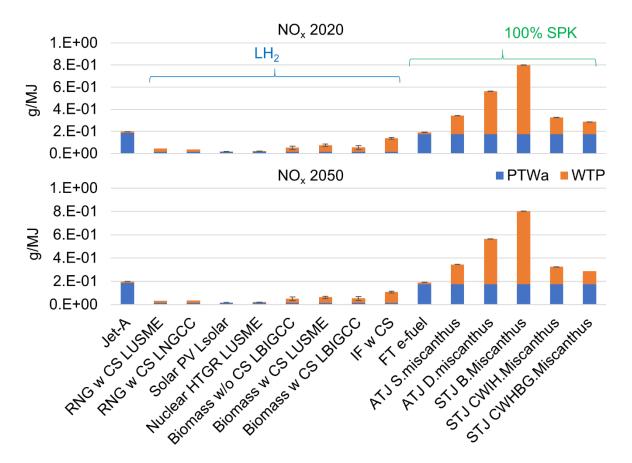


Figure 7. WTWa NO $_x$ emission comparison for different LH $_2$ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

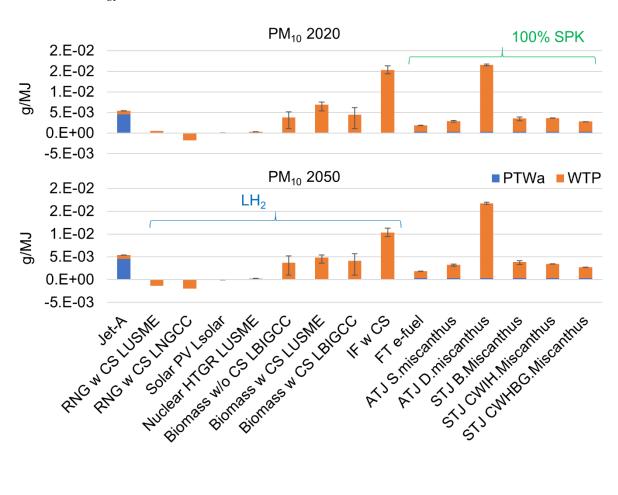


Figure 8. WTWa PM_{10} emission comparison for different LH_2 and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

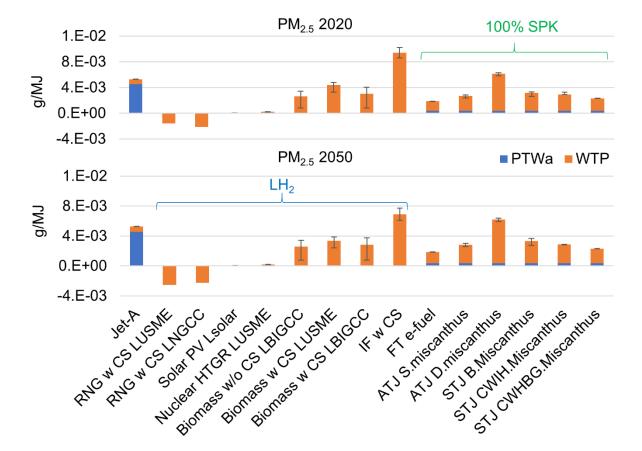


Figure 9. WTWa $PM_{2.5}$ emission comparison for different LH_2 and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

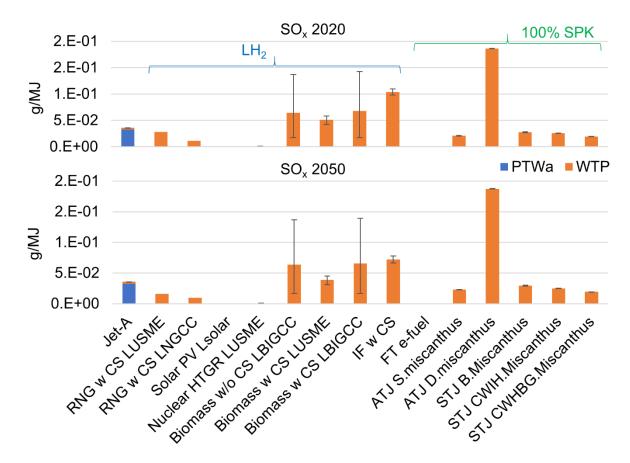


Figure 10. WTWa SO_x emission comparison for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

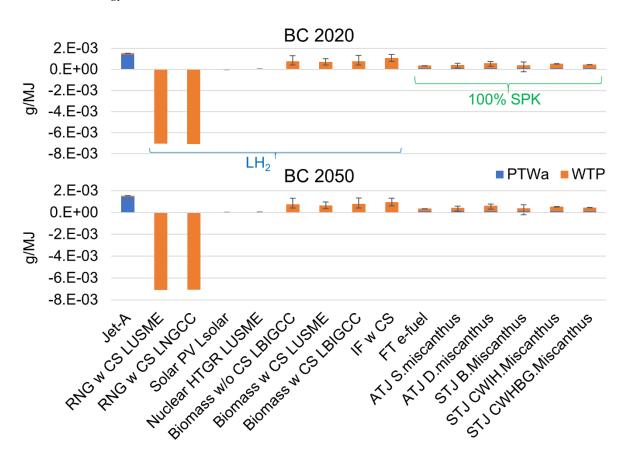


Figure 11. WTWa BC emission comparison for different LH₂ and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

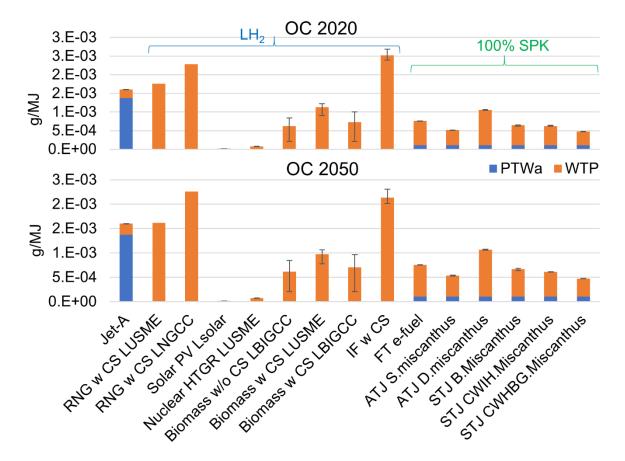


Figure 12. WTWa OC emission comparison for different LH_2 and 100% SPK feedstock and pathways for 2020 and 2050 US energy mix scenarios.

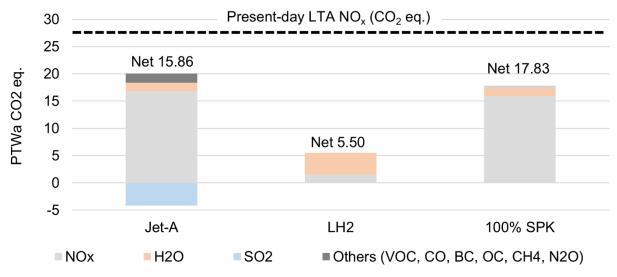


Figure 13. Comparative analysis of PTWa NO_x, H₂O, SO₂ and other emissions (CO₂ equivalent) for Jet-A, LH₂ and 100% SPK (present-day LTA NO_x emission is known from GREET and cruise GWP is known from Table 9).

More information:

First author's other research work can be found in [6,11–112,121–125].

Nomenclature

ASTM	American Society for Testing and Materials	LUSME	Liquefication using US mix electricity	
ATJ	Alcohol-to-jet	L-	Liquefication using electricity	
		Nuclear	from nuclear energy	
B.	Biological plant type			
ВС	Black carbon	L-Solar	Liquefication using electricity from solar energy	
BWB	Blended wing body	MSW	Municipal solid waste	
CH ₄	Methane	MW	Molecular weight	
CO	Carbon monoxide	NBC	Non-biogenic carbon	
CO ₂	Carbon dioxide	NGL	Natural gas liquid	
COG	Coke oven gas	NO_x	Oxides of nitrogen	
CORSIA	Carbon offsetting and	N ₂ O	Nitrous oxide	
	reduction scheme for international aviation	- 1-0		
CS	Carbon sequestration	OC	Organic carbon	
CWHBG	Catalytic with H ₂ from	p_{amb}	Ambient pressure (in Pa)	
	biomass gasification plant		•	
	type			
CWIH	Catalytic with in-situ H ₂	PFAD	Palm fatty acid distillate	
	plant type		•	
CWM	Corn wet mill	PM	Particulate matter	
D.	Distributed	PtL	Power-to-liquid	
$f/_a$ or	Fuel air ratio	PTWa	Pump-to-wake	
FAR				
FT	Fischer–Tropsch	PV	Photovoltaics	
GEMIS	Global emission model for	P_3	Combustor inlet pressure (in	
	integrated systems		MPa)	
GHG	Greenhouse gas	RNG	Renewable natural gas	
GREET	Greenhouse Gases,	SMR	Steam methane reformation	
	Regulated Emissions, and			
	Energy Use in Technologies			
GWP	Global warming potential	SOEC	Solid oxide electrolysis cell	
HEFA	Hydro-processed esters and	SO_x	Oxides of sulphur	
	fatty acids			
HRJ	Hydro-processed renewable	SO ₂	Sulphur Dioxide	
	jet fuel			
HTE	High temperature	SPK	Synthetic paraffinic kerosene	
	electrolysis			

HTGR	High temperature gas	STJ	Sugar-to-jet fuel
	reactor		
HTL	Hydrothermal liquefaction	S.	Standalone
ICAO	International Civil Aviation	T_{amb}	Ambient temperature (in K)
	Organization		
IF	Integrated fermentation	T_3	Combustor inlet temperature
			(in K)
ILUC	Indirect land-use change	UCO	Used cooking oil
LBIGCC	Liquefied using electricity	VOC	Volatile organic compound
	from biomass integrated		
	gasification combined cycle		
LCA	Life cycle assessment	w CS	With carbon sequestration
LH_2	Liquid hydrogen	WTP	Well-to-pump
LNG	Liquid natural gas	WTWa	Well-to-wake
LNH3	Liquid ammonia	w/o CS	Without carbon sequestration
LNGCC	Liquification using power	Φ	Equivalence ratio
	from natural gas combined		
	cycle		

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