

1 Article

2 Pendulum-type hetero-core fiber optic accelerometer 3 for low-frequency vibration monitoring

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10

11 **Abstract:** In this paper, a novel pendulum-type accelerometer based on hetero-core fiber optics has
12 been proposed for structural health monitoring targeting large-scale civil infrastructures. Vibration
13 measurement is a non-destructive method for diagnosing the failure of structures by assessing
14 natural frequencies and other vibration patterns. The hetero-core fiber optic sensor utilized in the
15 proposed accelerometer can serve as a displacement sensor with robustness to temperature changes
16 in addition to immunity to electromagnetic interference and chemical corrosions. Thus the hetero-
17 core sensor inside the accelerometer measures applied acceleration by detecting the rotation of an
18 internal pendulum. A series of experiments showed that the hetero-core fiber sensor linearly
19 responded to the rotation angle of the pendulum ranging within $\pm 5^\circ$, and furthermore the proposed
20 accelerometer could reproduce the waveform of input vibration in a frequency band of several Hz
21 order.

22 **Keywords:** structural health monitoring; fiber optic sensor; accelerometer; hetero-core; low-
23 frequency vibration measurement

24

25 1. Introduction

26 For preventing serious disasters caused by civil infrastructures collapsed, the existence of
27 structural damage and any other faults in infrastructures have to be early and accurately detected, in
28 spite of uncountable infrastructures and the inadequate number of inspectors. The process of
29 implementing fault diagnosis strategy is referred to as structural health monitoring (SHM), in which
30 the failure in a structure is automatically observed by monitoring, data processing, and health
31 evaluation systems [1].

32 As one of SHM techniques, fault diagnosis based on vibration monitoring was widely known as
33 a non-destructive sensing and analysis scheme for global fault diagnosis and has been studied so far
34 in the literature [2-5]. Typically, a decrease in stiffness of the structure is observed in a natural
35 frequency getting lower than expected value. In contrast, the frequency higher than expected implies
36 that the structure was supported stiffer than expected.

37 There have been a lot of studies researched for analyzing the faults in infrastructures from
38 vibration information. For instance, Soman et al. employed several stiffness indices based on modal
39 frequency, displacement and strain for quantifying the degree of structural health by use of a multi-
40 metric measurement system [6]. Moreover, a practical non-destructive bridge condition assessment
41 was also examined on an actual bridge using accelerometers in combination with cable tension
42 sensors, anemometers and thermistors [7]. However, when considering practical usages in actual
43 infrastructures, the SHM system should be robust to ambient noise, temperature changes and
44 corrosions in harsh measurement environments [8].

45 With the development of vibration monitoring methodology, a number of sensing devices have
46 also been proposed aiming at SHM based on electric-based sensors based on piezoelectric, capacitive,
47 and [9-13]. However, it is necessary for the sensor devices to be highly robust to electromagnetic
48 interference and severe temperature changes in order to monitor civil infrastructures for long term
49 in harsh environments, which is always exposed to changes in weather, or sometimes water leakage
50 and lightning damages. In comparison, fiber optic sensors were known as an alternative sensing
51 technology to electric sensors for SHM because of their remarkable merits. Fiber optic sensors are
52 non-electrical and passive devices, immune to electromagnetic interference and chemical corrosions,
53 and have the ability of remote sensing, thereby those of which makes them suited to outdoor long-
54 term smart sensing [14].

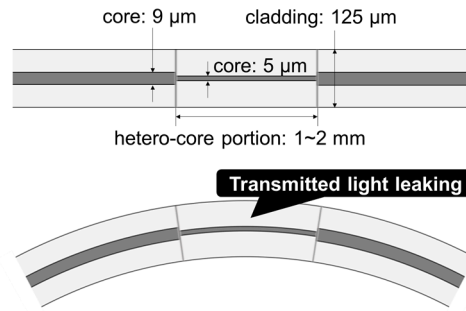
55 There have been a variety of fiber optic sensors developed for smart structures by use of Brillouin
56 scattering [15, 16], fiber Bragg grating (FBG) [17-24], and optical interferometer [25, 26]. Minardo et
57 al. introduced distributed fiber optic temperature and strain sensors using Brillouin optical time-
58 domain analysis (BOTDA) into railway infrastructures and succeeded to dynamically monitor strain
59 distribution due to train passage [15]. Moreover, a considerable number of fiber Bragg grating (FBG)
60 sensors have also been proposed for detecting strain in structure by embedding sensors themselves
61 [17-19], and they can perform as accelerometers by means of sensing the deformation of on an
62 oscillated cantilever beam [20-22]. A fiber optic accelerometer based on Fabry-Perot interference were
63 also investigated [25]. This accelerometer detects vibration in a range up to some hundreds Hz in
64 such a way to measure the flexure of an optical fiber cantilever beam by Fabry-Perot interference.

65 In this paper, we have proposed a novel accelerometer based on a hetero-core fiber optic sensor
66 for natural vibration monitoring on large-scale infrastructures. Hetero-core fiber optic sensors consist
67 of two single-mode fibers fusion-spliced with different core diameters to obtain high sensitivity to
68 macro-bending on the processed fiber line [27]. Compared to conventional fiber optic sensors, the
69 hetero-core fiber sensors have some attractive features such as a cost-effective measurement scheme
70 by use of a light emitting diode (LED) and a photo diode (PD), and temperature independency on
71 sensor responses [28].

72 The scope of this study is to realize a pendulum-type accelerometer targeting low-frequency
73 natural vibration on large-scale infrastructures, in which an embedded hetero-core sensor detects the
74 rotation position of an internal pendulum. The movement of the pendulum was subjected to a
75 combination of a weight and springs equipped with the pendulum so that they dominantly
76 determined the resonance characteristics of the proposed accelerometer. On the other hand, it was so
77 far demonstrated that the hetero-core sensor was able to measure displacement with high linearity
78 by a conversion mechanism to transform displacement to macro-bending [29], thereby the rotation
79 of pendulum can be detected in such a way that the hetero-core optical fiber measured the
80 displacement on a given point on the pendulum. A series of experiments were conducted to confirm
81 that the hetero-core optical fiber inside the accelerometer measures the rotation of internal pendulum
82 with a good linearity. In addition, it was revealed that the accelerometer performed well in low-
83 frequency vibration measurement by tuning the sensitivity and frequency response characteristics
84 depending on dynamic parameters of the pendulum.

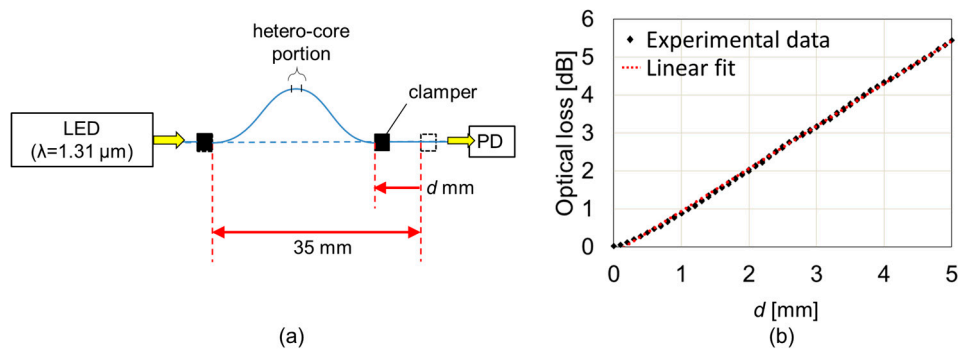
85 2. Sensor principle

86 As shown in Figure 1, a hetero-core fiber optic sensor proposed in this paper is composed of a
87 short single mode (SM) fiber segment called a hetero-core portion, inserted by fusion splicing into an
88 SM fiber transmission line. The core diameters of the hetero-core portion and the transmission line
89 are 5 μm and 9 μm , respectively, and the length of the hetero-core portion is about 1-2 mm. It was
90 previously confirmed that light transmitted through the core partially leaks into a cladding layer at
91 a boundary between the transmission fiber and the hetero-core portion, and the degree of light
92 leakage increases with a bending radius of the hetero-core portion [27].

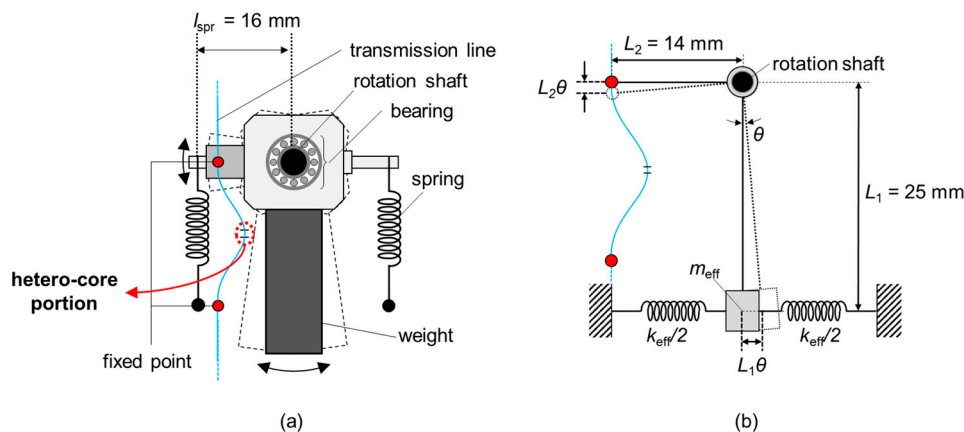


93 **Figure 1.** Schematics of a hetero-core fiber optic sensor.

94 A displacement sensor based on hetero-core fiber optics has already been reported [29] by
 95 employing the conversion mechanism from displacement to bending, as illustrated in Figure 2(a).
 96 The hetero-core optical fiber is clamped across the hetero-core portion by a pair of fiber clampers, one
 97 of which is set to move and the other is fixed. When the displacement of the clampers d increases, the
 98 optical loss of the sensor was linearly increased as the increment of bending radius on the hetero-core
 99 portion, as shown in Figure 2(b). The accuracy of the sensor to the displacement was less than 0.1
 100 %FS so that the hetero-core fiber optic sensor can be employed as a highly-accurate displacement
 101 sensor.
 102



103 **Figure 2.** Sensing principle of a hetero-core fiber optic displacement sensor: (a) experimental setup
 104 and (b) optical loss response to applied displacement.



105 **Figure 3.** (a) Appearance and (b) schematics of a pendulum-type accelerometer with a hetero-core
 106 fiber optic sensor.

107 Figure 3 illustrates a schematic drawing of the proposed pendulum-type accelerometer based
 108 on hetero-core fiber optics. As shown in Figure 3(a), an internal pendulum built in the accelerometer
 109 equips a weight and two springs and rotates on a shaft fixed on a chassis of the sensor. A hetero-core

110 optical fiber is clamped at two points, one of which is on the pendulum and the other is fixed on the
 111 chassis, thereby the sensor detects the rotation angle of the pendulum as a displacement of the
 112 clamped point on the pendulum. When a whole system of the accelerometer was accelerated, the
 113 internal pendulum rotates due to an inertial force and makes the position of the clamped point of the
 114 hetero-core fiber displaced. Figure 3(b) shows the simplified vibration model of the proposed
 115 accelerometer, in which it is supposed that the weight of pendulum exists at a centroid of weight (far
 116 from the rotation shaft by $L_1 = 25$ mm) and is physically subjected to two springs. The effective mass
 117 of the weight m_{eff} is determined from the mass of the weight m , those of other components m_i , and
 118 the distance of components from the rotation shaft l_i by moment equation as follows:

$$m_{\text{eff}} = m + \frac{1}{L_1} \sum_i m_i l_i, \quad (1)$$

119 In addition, the effective elastic coefficient k_{eff} is also expressed from k , the elastic coefficient of
 120 two employed springs as follows:

$$k_{\text{eff}} = 2k \frac{l_{\text{spr}}}{L_1}, \quad (2)$$

121 in which l_{spr} denotes the distance of the springs from the rotation shaft. Therefore, when the
 122 pendulum is rotated by input acceleration $\alpha(t)$, the equation of motion in the pendulum is described
 123 as follows:

$$m_{\text{eff}} L_1 \ddot{\theta} + \lambda(\dot{\theta}) + k_{\text{eff}} L_1 \theta = m_{\text{eff}} \alpha(t), \quad (3)$$

124 where θ denotes a rotation angle of the pendulum and λ represents a damping factor of the system
 125 including friction around the rotation shaft. From this equation, the resonant frequency of this
 126 vibration system f_0 can be derived as follows:

$$f_0 = \frac{1}{2\pi} \sqrt{\frac{k_{\text{eff}}}{m_{\text{eff}}}}. \quad (4)$$

127 Furthermore, when the frequency of input vibration $\alpha(t)$ is sufficiently lower than the resonant
 128 frequency, the following relational expression holds for α and θ because of the first and second terms
 129 in the right side of Eq. (3) can be neglected.

$$\frac{d\theta}{d\alpha} = \frac{m_{\text{eff}}}{L_1 k_{\text{eff}}} = \frac{1}{L_1 (2\pi f_0)^2}. \quad (5)$$

130 On the other hand, the hetero-core fiber optic sensor detects the rotation as the displacement
 131 $L_2 \theta$, when θ is sufficiently small. The relation between the rotation angle θ and an optical loss of the
 132 hetero-core sensor is described as follows, on the condition that the sensor linearly responds to the
 133 displacement:

$$\text{Loss} = \gamma' L_2 \theta = \gamma \theta, \quad (6)$$

134 in which the coefficient γ means the ratio of optical loss response to the rotation angle. Therefore,
 135 considering Eq. (5) and (6), the sensitivity of the accelerometer can be obtained by the following
 136 formula:

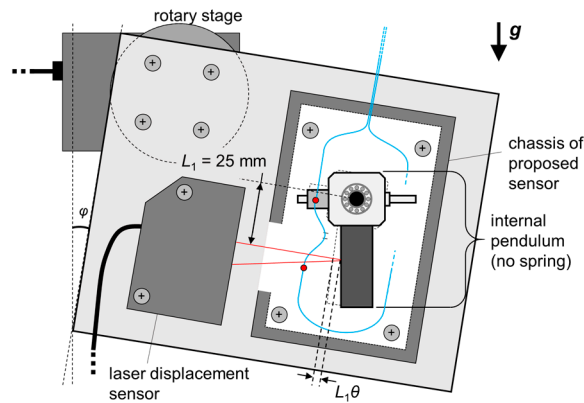
$$\frac{d\text{Loss}}{d\alpha} = \frac{\gamma}{L_1 (2\pi f_0)^2}. \quad (7)$$

137 It can be seen from Eq. (7) that the sensitivity has trade-off relation to the resonant frequency
 138 which determines the width of a measurable frequency band, a combination of a weight and springs
 139 should be modulated in balance for measuring low-frequency natural vibration in infrastructures
 140 with high sensitivity.

141 3. Static rotation response

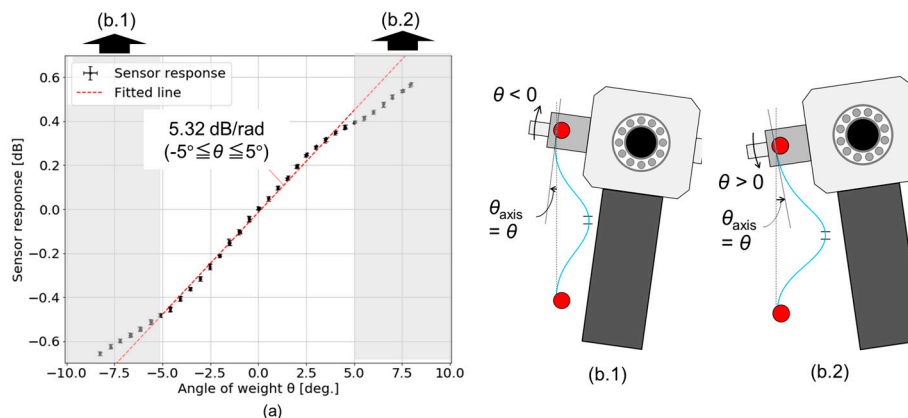
142 For evaluating the sensitivity and linearity of a hetero-core fiber optic sensor to the rotation of
 143 pendulum, a static rotation test was conducted, in which the relation between a rotation angle of the
 144 pendulum and optical loss response was monitored. As shown in Figure 4, the proposed

145 accelerometer and a laser displacement sensor measuring the centroid position of the pendulum were
 146 fixed on a firm plate rotated by a rotary stage. The rotary stage got the plate slowly inclined by φ
 147 in the range of $\pm 8^\circ$ with a stepwise of 0.5° , and the optical loss and the displacement of weight were
 148 simultaneously recorded. In this experiment, two springs inside the accelerometer were removed in
 149 advance in order to reduce the difference between the inclination angle φ and the internal rotation
 150 angle of pendulum θ . Furthermore, the displacement of weight measured by the laser displacement
 151 sensor was regarded as $L_1\theta$ because the rotation angle θ would be sufficiently small.
 152



153 **Figure 4.** Experimental setup for hetero-core fiber sensitivity to the rotation of a pendulum in the
 154 inclined accelerometer.

155 Figure 5(a) shows the variation of optical loss as a function of the rotation angle of weight θ
 156 measured by the laser displacement sensor. It can be seen that the optical loss linearly changed with
 157 a sensitivity of $0.0928 \text{ dB}/^\circ$ in the range $-5^\circ < \theta < 5^\circ$. In spite of the linear response of a hetero-core
 158 fiber optic sensor to the displacement as mentioned in section 2, nonlinear responses in ranges $\theta < -$
 159 5° and $\theta > 5^\circ$ would result from the change of an axial angle of the optical fiber at a fixed point on
 160 the pendulum. In particular, when the pendulum rotated by $\theta < 0^\circ$ as depicted in Fig. 5(b.1), θ_{axis} , the
 161 axial angle of the optical fiber at a fixed point on the pendulum, was also rotated as much as θ , which
 162 made the curvature on the hetero-core portion higher than the case that the only displacement $L_2\theta$
 163 was applied with $\theta_{\text{axis}} = 0^\circ$. Similarly in the case $\theta > 0^\circ$, as shown in Fig. 5(b.2), the change of θ_{axis} got
 164 lower the increment of curvature on the hetero-core portion due to the applied displacement.
 165 Although this phenomenon would always appear in this sensing mechanism, the effect on sensor
 166 response appeared to be negligible in the small rotation angle so that the sensitivity of the hetero-
 167 core optical fiber to rotation angle γ is $0.0928 \text{ dB}/^\circ$ with a linearity in the range of $-5^\circ < \theta < 5^\circ$.
 168



169 **Figure 5.** (a) Optical loss variation of a hetero-core optical fiber inside a proposed accelerometer as
 170 functions of angle of weight θ , and (b.1) (b.2) schematics of excess changes in the axial angle of the
 171 optical fiber line θ_{axis} due to the rotation movement of fixed point on the pendulum.

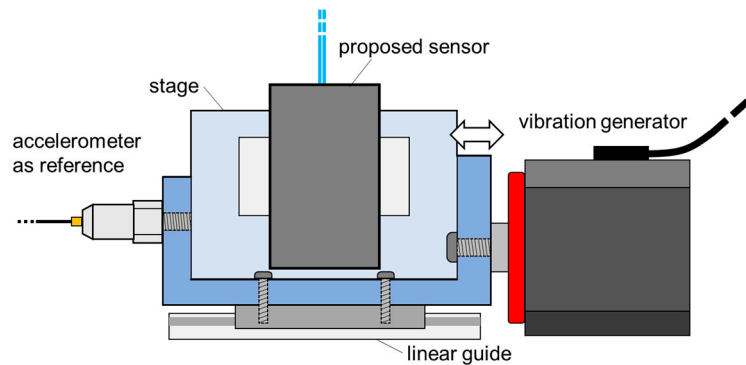
172 4. Frequency response characteristics

173 To understand the optimum specification of the proposed accelerometer for monitoring low-
 174 frequency vibration, the frequency responses were monitored with arranging combinations of the
 175 weight and springs as listed in table 1. The values of f'_0 and $dLoss/d\alpha'$ were calculated as approximate
 176 resonant frequencies and sensitivities without considering the effective masses of pendulum
 177 components except for a weight. In this experiment, as illustrated in Figure 6, the proposed
 178 accelerometer was attached together with an electric accelerometer used as a reference on a stage
 179 which was horizontally vibrated by a vibration generator (WaveMaker01, asahi seisakusyo). The
 180 waveform applied by the vibration generator was set to be sinusoidal with a frequency ranging from
 181 1 Hz to 50 Hz, and the duration of each test was for 10 seconds in each frequency. During the test,
 182 both input acceleration and output data by the reference and the proposed accelerometer were
 183 simultaneously measured by a sampling rate of 4 kHz.

184 **Table 1.** The combinations m and k of the proposed accelerometer, and approximate values of
 185 resonant frequency f'_0 and sensitivity $dLoss/d\alpha'$.

Accelerometer	m [kg]	k [N/m]	$f'_0 = \frac{1}{2\pi} \sqrt{\frac{k_{eff}}{m}}$ [Hz]	$\frac{dLoss'}{d\alpha} = \frac{\gamma}{L_1(2\pi f'_0)^2}$ [dB/ms ⁻²]
Pattern 1	0.042	20	3.93	0.349
Pattern 2	0.00644	20	10.0	0.0536
Pattern 3	0.00644	230	34.0	0.00465

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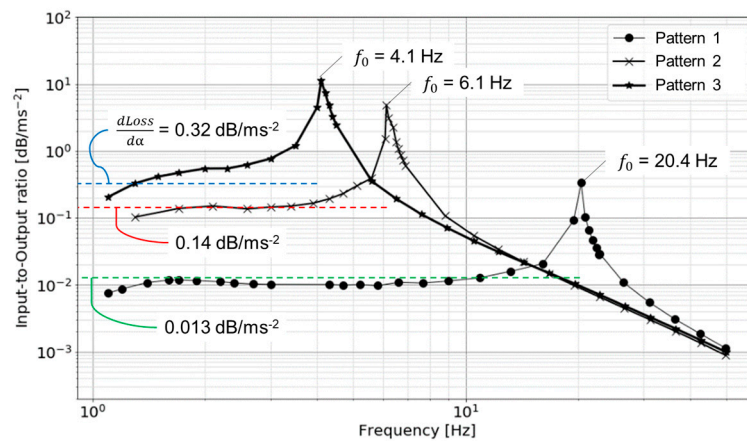


187 **Figure 6.** Experimental setup for frequency response characteristics of the proposed accelerometer.

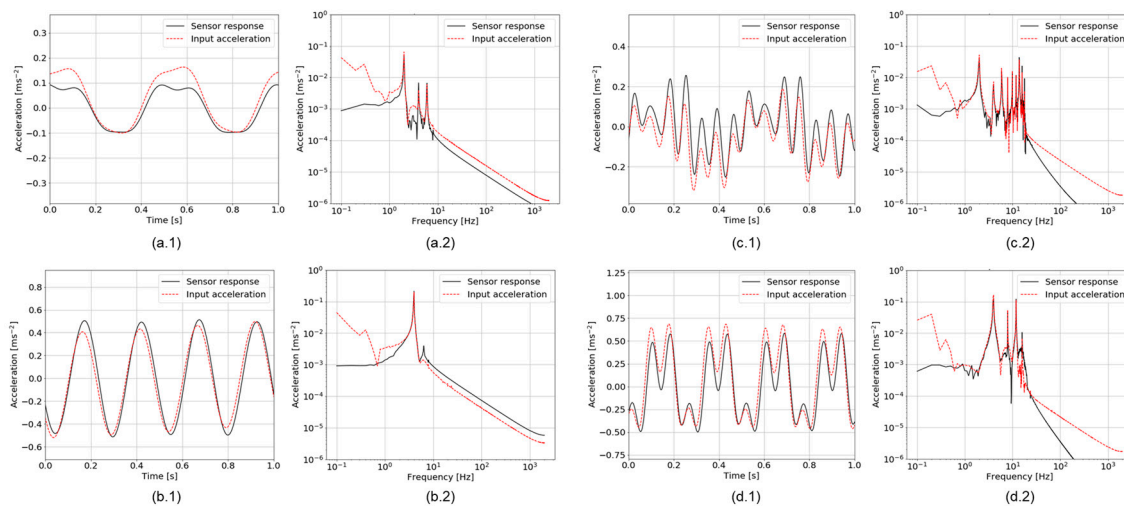
188 Figure 7 shows the frequency response of the proposed accelerometer, which was obtained as
 189 an input-to-output ratio by comparing the amplitudes of the fast Fourier transform (FFT) spectra for
 190 each frequency. It was observed that the resonant frequency f_0 shifted with a tendency like f'_0
 191 described in table 1, and the sensitivity decreased inversely to the resonant frequency. When
 192 calculating $dLoss/d\alpha$ by obtained f_0 according to Eq. (7), the values plotted as dashed lines in Fig. 7
 193 were found to overlap with the measured in frequency bands lower than f_0 . In a case of Pattern 1,
 194 the input-to-output ratio seems to gradually increase as the increment of frequency. In the other cases of
 195 Patterns 2 and 3, however, the stable input-to-output ratios were observed in low-frequency bands
 196 up to 4 Hz and up to 15 Hz, respectively.

197 Additionally, Figure 8 shows profiles of sensor responses in cases of Patterns 2 and 3 when 2-
 198 and 4-Hz vibrations were applied. In order not to take into consideration vibrations excited due to
 199 resonance, high-frequency components were numerically removed by the finite impulse response
 200 (FIR) low-pass filtering with a cut-off frequency of $0.8 \times f_0$. In a case of Pattern 2, it can be seen in Figs.
 201 8(a.1) and (b.1) that the proposed accelerometer responded to vibration with a negligible phase shift
 202 comparing to input acceleration data measured by the reference sensor. Moreover, the frequency
 203 spectra depicted in Figs. 8(a.2) and (b.2) indicated that the proposed accelerometer well reproduced
 204 the frequency spectrum of input acceleration in the range from 1 Hz to the cut-off frequency. There
 205 were multiple peaks observed when 2-Hz vibration was applied as shown in Fig. 8(a.2), which were

206 derived from unwanted harmonic components of input vibration generated by the vibration
 207 generator. On the other hand, spectral components under 1 Hz tended to be lower in the sensor
 208 response than in input vibration. It would be caused by that the proposed accelerometer had a
 209 reduced sensitivity at around 1 Hz, which can be supposed in Fig. 7. These features in terms of a
 210 phase shift and a spectral form in a frequency band ranging from 1 Hz to $0.8 \times f_0$ Hz were similarly
 211 observed in the case of Pattern 3 as shown in Figs. 8(c) and (d). As a result, it was confirmed that the
 212 proposed accelerometer well performed with a stable sensitivity in a frequency band more than 1 Hz
 213 and less than f_0 , with keeping the theoretical trade-off relation between the sensitivity and resonant
 214 frequency. However, the lower resonant frequency of accelerometer was, the higher but the less
 215 stable sensitivity was obtained. For such a reason, the accelerometer should be utilized with moderate
 216 adjustments on the internal weight and springs such as in Patterns 2 and 3.



217 **Figure 7.** Frequency response characteristics of the proposed accelerometer which was modulated in
 218 three patterns shown in table 1.



219 **Figure 8.** Profiles of responses of the proposed accelerometer and input vibration in (x.1) time domain
 220 and (x.2) frequency domain: (a) 2-Hz vibration applied to the proposed accelerometer in pattern 2, (b)
 221 4-Hz vibration to the accelerometer in pattern 2, (c) 2-Hz vibration to the accelerometer in pattern 3,
 222 and (d) 4-Hz vibration to the accelerometer in pattern 3.

223 5. Conclusion

224 This paper described a novel pendulum-type accelerometer based on hetero-core fiber optics for
 225 monitoring low-frequency natural vibrations in large-scale infrastructures such as bridges, building,
 226 and tunnels. This accelerometer contained a pendulum with a weight and springs, whose rotation
 227 angle was measured by an embedded hetero-core fiber optic sensor. Considering an equation of

228 motion in the system of pendulum, the rotation angle was proportional to applied acceleration in a
229 range under a resonant frequency.

230 Through a static rotation test, it was found that the hetero-core fiber sensor linearly responded
231 to the rotation angle in a range within $\pm 5^\circ$, although it nonlinearly responded out of this range because
232 the rotation of a fixed point of a fiber line on the pendulum broke the linearity of the sensor response.
233 Moreover, the ratio of rotation angle to applied acceleration varied inversely with respect to the value
234 of resonant frequency f_0 , the sensitivity of the accelerometer was able to be calculated from f_0
235 experimentally observed. Additionally, it was also confirmed that the accelerometer showed a stable
236 sensitivity and reproduced a waveform of input acceleration in a frequency band from 1 Hz to $0.8 \times$
237 f_0 Hz when the accelerometer was modulated to have f_0 of 6.1 Hz and 20.4 Hz. Consequently, the
238 findings from these experiments have suggested that the proposed accelerometer was suitable for
239 monitoring vibration in a frequency band of several Hz order, and therefore performed well for
240 monitoring low-frequency natural vibrations in large-scale infrastructures.

241 **Author Contributions:** Conceptualization, K.W.; Methodology, K.W. and H.Y.; Validation, H.Y. and I.K.; Formal
242 Analysis, H.Y. and I.K.; Investigation, H.Y.; Data Curation, H.Y.; Writing-Original Draft Preparation, H.Y.;
243 Writing-Review & Editing, M.N. and K.W.; Visualization, H.Y.; Supervision, M.N. and K.W.; Project
244 Administration, K.W.

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248 **Conflicts of Interest:** The authors declare no conflict of interest.

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